

# **ANALYSIS OF OPEN CANOPY STRUCTURES WITH PARAPETS UNDER WIND**

By

**Augusto Poitevin Vera**

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Approved by:

---

Luis A. Godoy, Ph. D.  
Chairman, Graduate Committee

---

Date

---

Ricardo R. López, Ph. D.  
Member, Graduate Committee

---

Date

---

Luis E. Suárez, Ph. D.  
Member, Graduate Committee

---

Date

---

Raúl E. Zapata López, Ph. D.  
Member, Graduate Committee

---

Date

---

Sonia Bartolomei, Ph. D.  
Representative, Graduate Studies

---

Date

---

Ismael Pagán Trinidad, MSCE.  
Chairman, Civil Engineering & Surveying Department

---

Date

## **ABSTRACT**

A literature review shows that there is a lack of information on the wind load pressures acting on canopy structures with a parapet on the roof perimeter. On the other hand, a large number of those structures suffered catastrophic damages due to hurricanes, for example during Hurricanes Katrina and Rita in 2005. The latest version of the predominant standard and commentary for wind design in the eastern part of the US, the ASCE 7-05, introduced for the first time the issue of open structures. However, the code recommendations do not address open structures with parapets. In this research, the author addresses the problem using two different (but complementary) approaches: a numerical simulation using Computational Fluid Dynamics (CFD) and wind tunnel testing. Values of pressure coefficients,  $C_p$  and  $C_n$ , were obtained with the use of both methods, on top and bottom surfaces of the open canopy, and windward and leeward surfaces for parapets, showing that the obtained values had good agreement between both methodologies.

With the use of CFD, parametric studies further explore different plan geometries and incremental parapet heights in order to obtain extreme  $C_n$  values. The obtained extreme values were implemented on four case studies of collapsed open canopy structures, due to Hurricanes Rita and Katrina in 2005. With the use of structural analysis software, the structural members suffering extreme stresses, were identified and compared with the actual collapsed structures on the case studies selected. Structural design procedure is suggested for the analysis of open canopy structures with parapets and is implemented on each of the cases studied.

## RESUMEN

La revisión de la literatura demuestra la falta de información disponible sobre las presiones de viento actuando sobre estructuras abiertas con parapetos en la periferia del techo de las mismas. Por otra parte, un gran número de este tipo de estructuras sufren daños catastróficos, por ejemplo durante los Huracanes Katrina y Rita que ocurrieron en el año 2005. La más reciente versión del estándar y comentario en la parte Este de los Estados Unidos, el ASCE 7-05, introdujo por vez primera el tema de las estructuras abiertas. Es de notar, que las recomendaciones de este código no incluyen las estructuras abiertas con parapetos. En esta investigación, el autor investiga el problema usando dos enfoques distintos, pero complementarios: Usando Dinámica Computacional de Fluidos (CFD) y pruebas hechas en un túnel de viento, valores de presión,  $C_p$  y  $C_n$  fueron obtenidos con el uso de ambos métodos, en las superficies superiores e inferiores de la estructura abierta, y sotavento y barlovento en las superficies de los parapetos, siendo los valores obtenidos muy parecidos entre ambas metodologías.

Mediante el uso de CFD, estudios paramétricos exploran más profundamente diferentes geometrías en planta y diferentes alturas para poder obtener valores  $C_n$  extremos. Estos valores son entonces implementados en estudio de casos particulares, de estructuras abiertas que colapsaron debido a los efectos de los Huracanes Katrina y Rita en el año 2005. Con el uso de un programa de análisis de estructuras, miembros que sufrieron esfuerzos máximos, pudieron ser identificados y comparados con las estructuras colapsadas en los casos seleccionados. Un

procedimiento de diseño estructural para estructuras abiertas con parapetos es sugerido e implementado para cada uno de los casos estudiados.

## **DEDICATION**

I would like to dedicate this work, to my wife Yvonne and to my daughter Amanda. All the time and effort during all this years, those two beautiful ladies were near me, encouraging me and supporting me. I will be forever grateful for all the help, love and support from them during all this time.

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# **CHAPTER 1. PROBLEM STATEMENT**

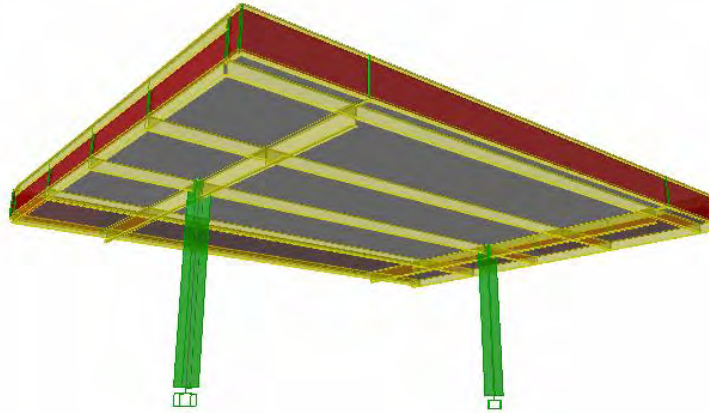
## **1.1 General Information**

Open canopies are frequently used in the construction of civil engineering facilities, either as components of larger structures or as self supported structures. An example of the second type may be found in most gas stations throughout the nation, in which the roof covers the gas pumps and incoming vehicles. There are many other applications, including parts of industrial buildings and processing plants, and sport courts. A large number of such structures suffered catastrophic damages during hurricanes, including Hurricanes Katrina and Rita in 2005, (NIST 2006).

In their typical configurations, open canopies are commonly supported by interior columns in different patterns, without having any perimeter walls. The roof is formed by a system of beams in two directions to support the roof panels. The supporting columns may be aligned in one or more rows, depending on the size of the roof and on the functionality of the facility. In small gas stations, canopies are frequently supported by a single row of columns as illustrated in Figure 1.1. Typical dimensions vary from 6 m (20 ft.) to 15 m (50 ft.) in each horizontal direction and from 3.7 m (12 ft.) to 6 m (20 ft.) in height.

From the structural point of view, it would be desirable to have a frame structure that integrates columns and beams into a single resisting structure. However, the inspection of many structures of this kind in the United States clearly shows that the beam-column connections are not rigid connections, with the consequence that the majority of the connections used between the elements are simple shear and tension connections. The majority of the canopy columns inspected are designed as cantilever elements taking all the lateral forces due to wind pressure

and wind uplift. In addition, maintenance to open canopy structures appears to be a very important factor that may reduce their factor of safety, but maintenance work is often neglected with the consequence that corrosion is present in many canopies.



**Figure 1.1: Schematic view of a canopy used in gas stations.**

In hurricane-prone areas, such as the Caribbean islands and the coastal areas in the US, the most critical structural conditions occur during high winds due to hurricanes. As it is known, several hurricanes in recent years have been of category 3 and 4. This generates sustained wind velocities of above 64.82 m/s (145 mph) in some regions, and those levels of wind velocities and pressures have been incorporated in design codes such as ASCE 7-05, Figure 6-1. With such high wind velocities, a surprise comes associated to the absence of information on the wind load pressures acting on open structures with parapets. This void in our current state of knowledge is also reflected in the design recommendations available to engineers and the catastrophic effects on such structures due to the lack of design data and testing.

The majority of canopy structures have a parapet on the roof perimeter (as schematically illustrated in Figure 1.1). The wind pressures that these types of structures are exposed to are

very complex, because pressures on the top and bottom surfaces of the roof are not uniform and their values are different on each surface. Open canopy structures which include parapets have not been studied in detail in the research literature or in the current codes used for design in the United States.

## 1.2 Motivation

The predominant building code for wind design in the eastern part of the United States, is the ASCE -7. It did not properly address the issue of open structures for a number of years. The most recent version ASCE-7 05, was the first of various versions of this code that includes open structures. However, it does not address the issue of open structures with parapets, which is perhaps the most common configuration found in real constructions (see, for example Figure 1.2).



**Figure 1.2: Open canopy of a Gas Station, Quebradillas, PR. (Photograph by the Author).**

The pressure coefficients currently employed for the design of open canopy structures do not include parapets on the structure perimeter. The Uniform Building Code, 1997 Edition, is the

predominant building code in the western part of the United States. This code addresses the issue of open structures but does not provide any recommendations regarding the effects of parapets in open structures.

In the event of a hurricane, the results on such structures have proven to be devastating. Illustrations from some of 2005 hurricanes, Katrina and Rita, are shown in Figure 1.3 and Figure 1.4. The structure shown in Figure 1.3 was a new construction, completed during 2005. However improper knowledge of the pressures acting due to wind effects may have been a major factor contributing to the collapse. It was surprising during the field inspection in Texas and New Orleans after hurricanes Katrina and Rita by Godoy (2006), the close relation of the poor maintenance of open canopies and failure cases. There are several mechanisms leading to a rapid deterioration of the structure: first, environmental action; second poor roof drainage (the drains are located inside the hollow columns); and third, lack of preventive inspection.

Therefore, motivation to carry out this research is the need to establish basic recommendations for the safe design of the main wind force resisting system (MWFRS) for such structures. These recommendations will help in the proper maintenance and even possible retrofit of existing structures and to improve their safety level.



**Figure 1.3: Failed canopy in Chalmette, New Orleans, during Hurricane Katrina, (Photograph by L. Godoy).**

The case shown in Figure 1.4 is catastrophic in the sense that a total collapse occurred. However, even in this case it is possible to reconstruct the main mechanisms leading to the collapse, provided that rational estimates of wind pressures are taken into consideration. Thus, one motivation for this research is the need to reconstruct failure mechanisms and understand what design considerations need to be improved.



**Figure 1.4: Canopy structure collapsed at Meraux, New Orleans, Hurricane Katrina, (Photograph by L. Godoy).**

The results of a preliminary study of the causes of open canopy structures during high winds (Godoy and Poitevin, 2006) revealed that the current design code at that time, ASCE-7 98, did not address open structures. For the preliminary study, wind pressures and coefficients of enclosed structures were used. The pressure values for this study were generated using a wind design program named Wind Loads on Structures 2005 by Standard Design Group (SDGS 2005). This computer program generates wind pressures for enclosed and partially enclosed structures, including the MWFRS and the cladding on any structure.

Using the ASCE 7-98, values of the wind pressures were obtained from the basic design wind conditions specified for the location of the stations. Those wind pressures were applied to various three dimensional structural models. The structural analysis software used was ETABS developed by the firm called Computers and Structures, (CSI 2009). One of the computer models used in the study is shown in Figure 1.5. Figure 1.6 shows the results corresponding to the same model. A linear elastic analysis was performed on various model configurations taken from field measurements made during the reconnaissance of the hurricane effects of Rita and Katrina in 2005, to evaluate the stresses throughout the structure. The stress ratio refers to the ratio between the actual stresses and the allowable stresses using the Allowable Stress Design Code, AISC 1989 edition, as illustrated in Figure 1.6.



## **1.3 Importance**

Failures of open canopy structures are so common that during the event of a hurricane or high winds, one does not need to specifically search for the collapse of those types of structures, because they are easily found in most towns, as shown by the site reconnaissance made by Godoy (2006). The collapse of an open canopy structure, in a gas station, interrupts the supply of gasoline to the public and government agencies that need continued gas supply specially during an emergency period. Therefore, this type of structures should be designed as a critical and of high importance (essential facilities) in terms of human risks and security in the time of an emergency. Unfortunately, this type of structure has been not been investigated in depth until the current research, as shown by the literature review reported in Chapter 2. Research of open structures without parapets has been conducted in several countries, including Australia, Canada, Japan and in the United States. However, open canopy structures like those of gas stations have a parapet on the perimeter of the roof. As mentioned before, the effects of that parapet and the correlation between the height of the parapet and the geometry of the building are of crucial importance to estimate wind pressures and have not been investigated in detail.

## **1.4 Objectives**

The main objectives of this research may be divided in three major areas:

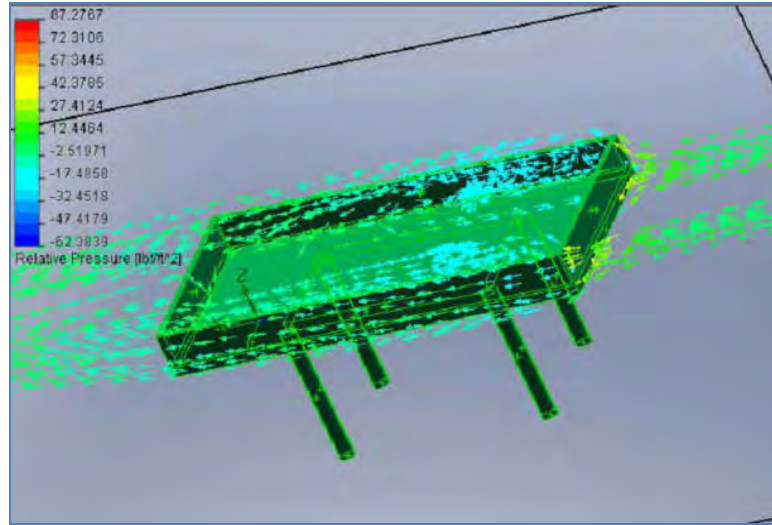
1. To investigate the characteristics of wind flow through open canopy structures with parapets, in order to evaluate wind coefficients for the main wind force resisting system (MWFRS) due to such conditions.

2. To explain the structural behavior of open canopy structures under wind, leading to the identification of most severely stressed components and possibly of collapse mechanisms and design problems.
3. To propose recommendations for design and future research based on rational basis for open canopy structures with parapets.

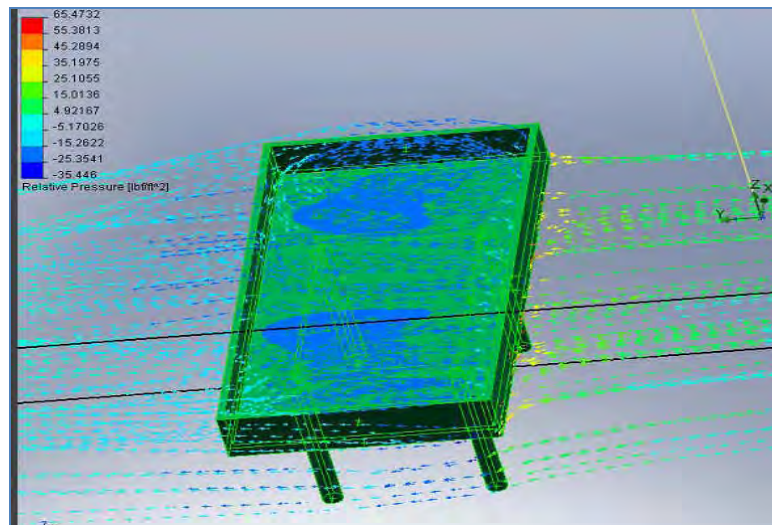
## **1.5 Proposed Methodology**

The proposed methodology in this work includes two stages. In a first stage, pressure coefficients will be evaluated under the assumption of a rigid structure. In a second stage, the pressure coefficients will be used as the loads acting on an elastically deformable structure to estimate the response of the structure.

The first stage will be tackled by two different (but complementary) approaches: first, a wind tunnel testing simulation will be carried out. Second, a computational simulation will be performed using Computational Fluid Dynamics (CFD). A commercial CFD software named EFD.Lab (Engineering Fluid Dynamics), developed by a firm called Flomerics (2009), will be used for the computational fluid analysis. Such CFD simulation is able to calculate wind pressures on the top and bottom surfaces of the structure with a wind velocity similar as those specified by design codes such as ASCE 7 (2005). Factors such as turbulence, roughness, humidity, and temperature can also be included in the CFD analysis to emulate real conditions. As an introduction to the type of results expected from a CFD simulation, Figure 1.7 and 1.8 illustrate velocity field vectors as computed using EFD.Lab.



**Figure 1.7: Flow and wind pressure distribution in the longitudinal direction.**



**Figure 1.8: Flow and wind pressure distribution in the transversal direction.**

With the use of wind tunnel results, a calibrated CFD model can be used to explore various models with different parapet heights. The wind tunnel facility used for this research is located at the Universidad Nacional del Nordeste (UNNE) in Resistencia, Argentina. The UNNE wind tunnel facility is a low velocity atmospheric boundary layer wind tunnel, built with the aim of performing aerodynamic studies of structural models.

The facility is an open circuit tunnel with a length of 22.8 m (74.8 ft), the testing chamber being a square section of 2.4 m (7.87 ft) width and 1.8 m (5.9 ft) in height and uses a 2.25 m (7.38 ft) diameter fan with a 92 kW motor. The maximum wind velocity that may be obtained in this tunnel is 25 m/s (55.9 mph) when the testing section is empty. Further details are given in Chapter 4.



**Figure 1.9: Wind tunnel at UNNE, (Photograph by B. Natalini).**

The results obtained from the models tested in the wind tunnel will be used to calibrate the computational simulation. This would help to assess if the method of using CFD for analysis can be used with confidence for the parametric analysis instead of only using the wind tunnel method. Pressure coefficients and wind pressures through the selected models will be determined experimentally. Once those results are determined, structural analysis of selected configurations, using commercially available structural software will be investigated. Actual shapes and

geometries of open canopy structures will be analyzed to assess their safety levels to withstand design wind velocities.

## **1.6 Original Contributions**

The proposed research will produce contributions to advance both, academic research and engineering practice. On the academic front, pressure coefficients for the design of open canopy structures with parapets do not exist at present. Open canopy structures without parapets have been recently investigated and wind pressure distributions and design coefficients have been proposed on a small number of previous investigations. However, no previous testing has been reported with open canopy structures that consider the effects of parapets on wind flow. Because this may be a controversial topic (due to its engineering significance), it is desirable to have methodological redundancy to make sure that adequate pressure coefficients are reliable. Correlation between the possible modeling and prediction of open canopy structures with the use of CFD software has not been previously investigated. The possibility of the confirmation of the use of CFD in the analysis of open canopy structures with parapets is another original contribution of this research. Finally, a design procedure is needed to be used for the safe and secure structural design of such structures.

## **CHAPTER 2. LITERATURE REVIEW**

### **2.1 Introduction**

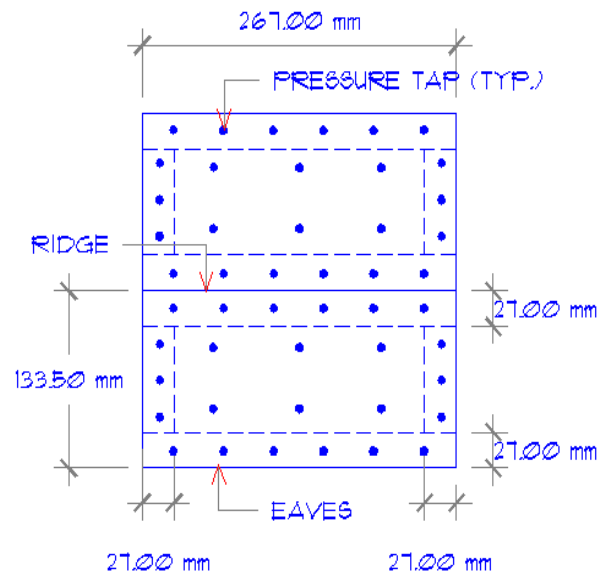
The focus of this investigation is the effect of wind on open canopy structures with parapets. This chapter contains a review of previous work presented by researchers on open canopies and in the area of CFD, which is relevant to the current investigation. The literature review covers the area of wind tunnel testing and parapet pressures.

### **2.2 Open Canopies**

The past half century, has witnessed interesting developments in the understanding of wind loading on structures. During this time, the description of wind load has moved from simple static drag forces to sophisticated models (Davenport, 2002). A review of recent literature on wind pressures in similar structures shows a wide variety of previous investigations. The need for more detailed information on the wind flow and perhaps on open canopies is a consequence of the collapse of a large number of open canopies in gas stations in areas affected by hurricanes Katrina and Rita in 2005 (Godoy 2006).

Several researchers had the opportunity to study wind loads on open canopy structures. Gumley (1984) made a parametric study investigating the effects of the roof shape, roof pitch, roof aspect ratio, eave height, and wind direction and internal stacking arrangements. He measured pressures averaged on roof areas using wind tunnel procedures. The effects of stacking patterns under the roof were also investigated, but only the envelope results were presented. Figure 2.1 shows the location of pressure gages used on Gumley's investigation. The drawing is interesting because it shows the number and location of pressure gauges employed by other

authors in wind tunnel tests. The results were used for updating the Eurocode (2002) and Australian (1989) wind loading codes.



**Figure 2.1: Pressure gages used on Gumley's investigation**

Full scale measures of agricultural canopy structures were reported by Robertson et al. (1985). These structures had an aspect ratio  $l/b$  (length/width) of approximately 2. Based on those results they proposed a set of wind force coefficients for designing such structures. Various wind tunnel studies were subsequently carried to validate the conditions obtained from the full-scale measurements by Robertson et al. (1985).

Another set of experiments were performed by Letchford and Ginger (1992) and Ginger and Letchford (1994), who measured the mean and peak point pressures over several roof areas. They compared the obtained results with the Australian wind loading code at the time, to conclude that the code provisions underestimated the wind loads. Altman (2001) made extensive measurements of the forces and moments on mono-sloped and gable roofs at Clemson University, USA. The roof models used in the study were made of high-density foam of 6 mm

thick. Figure 2.2 shows the model used on Altman's study. He compared the obtained results against various codes. In some cases, the code provisions underestimate and in others they overestimated. From the measurements taken by Altman, based on the obtained experimental results, he proposed wind force coefficients to be used for the design of main wind force resisting systems.



**Figure 2.2: Model used on Altman study, at Clemson University, USA. (Photograph by the Author).**

Letchford et al. (2000) measured mean wind forces on solid and porous canopy models. The mean drag and lift forces on various open canopy roof geometries were investigated. In conclusion, from the obtained result, the lift forces decrease as the pitch decreases and drag forces increases as porosity increases. Lam and Zhao (2002) performed wind tunnel tests on large cantilevered roofs, which are used mostly as grandstand roofs. The objective of the investigation was to identify the generation mechanism of wind pressure and peak lifting action on a large cantilevered roof. It was found that a horizontal roof is under a mean lifting action at

most wind incidence angles. However, at the wind incidence from the front of the roof, very high suction was found on the front edge of the roof.

Natalini et al. (2002) investigated the pressure distribution on curved canopy roofs with the use of wind tunnel testing. Curved canopy roofs are a very common type of structure in South America. Mean pressure coefficient from the wind tunnel tests were presented on the investigation. Paluch et al. (2003) investigated arch roof industrial buildings, adding the effect of attached canopies on the sides. Six scale models with five different canopies were investigated. The results showed that the aerodynamic coefficients for the roof are not affected by the canopies, in the case of  $0^\circ$  from the main axis. However, the influence on the pressure distribution is noticeable for wind incidence perpendicular to the main axis of the arch roofs and for other incidences as well.

Uematsu et al. (2007) tested three types of roof geometries, (i.e. gable, troughed and mono sloped roofs). Wind pressures were measured at many points both on the top and bottom surfaces of the roof model at various wind directions. The conclusions at which Uematsu and co-workers arrived based on their investigation and those of other authors may be summarized as:

- a) The roof pitch affects the wind forces significantly.
- b) There are significant differences in the results when the roof pitch is smaller than  $15^\circ$ .
- c) The influence of roof aspect ratio (length/width from 1 to 4) on the wind force coefficients is small.
- d) The experimental data for mono sloped and gable roofs is limited.
- e) Roof thickness and supporting systems significantly affects the results.

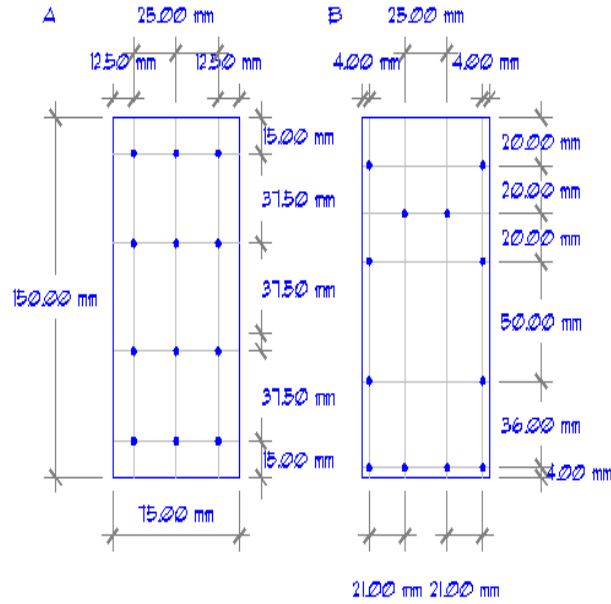
Roof is supported by slender columns and no walls, so that wind action is directly exerted on the top and bottom surfaces. These roofs seem to be more vulnerable to wind actions than

those of enclosed buildings. Local wind pressures and overall wind forces and moments acting on free standing canopy roofs have been investigated experimentally. Based on the results for the distribution of the most critical positive and negative peak pressure difference coefficients irrespective of wind direction, the peak wind force coefficients for the design of cladding and its immediately supporting structures were proposed in Uematsu et al. (2007).



**Figure 2.3: Wind tunnel model (Uematsu et al. 2008)**

Figure 2.3 shows a canopy roof model on the turntable of the wind tunnel of Concordia University. Special care was taken in decreasing the roof thickness and column width to avoid the distortion of the flow around the roof. The roof model is made of two galvanized steel sheets 0.3 mm thick and consists of a sandwich structure.



**Figure 2.4: Pressure taps arrangement (Uematsu, et al. 2008)**

Twelve pressure taps of 0.4 mm diameter were drilled on each side of the basic model. Taping locations were identical on both the upper and lower surfaces so that the net pressure difference could be obtained. Two types of tap arrangements were used; one is for overall wind measurements and the other is for the local pressure measurements.

Very large negative peak values are induced in the leeward ridge corner for the gable roof and in the windward eave corner for the troughed and mono-sloped roofs. The most critical values, both positive and negative, generally increase in magnitude with an increase in  $\beta$ , the roof pitch. When the roof pitch is the same, larger peak values are induced in mono-sloped roofs than in gable and troughed roofs (Uematsu et al. 2008).

## 2.3 Computational Fluid Dynamics (CFD)

Computational fluid dynamics (CFD) in Wind Engineering was initiated and has progressed over the past two decades. The rapid growth of computer power, which makes possible power acquisition and analysis of large amounts of experimental data, has led to the increasing use of CFD techniques (Baker 2007). The ultimate goal of CFD is to represent the physical events that occur in the fluids flow around and within designated objects. These events are related to the action and interaction of dissipation, diffusion, convection, shock waves, slip surfaces, boundary layers and turbulence. In the field of aerodynamics all these phenomena are governed by the Navier-Stokes equations (Lomax and Pulliam, 1999).

Computational fluid dynamics constitutes a new approach in the study and development of fluid dynamics, which was previously dominated by wind tunnel testing. At present, most researchers in the field of wind engineering agree that there is a need to have better theory and experiments in order to gain understanding of wind acting on structures. The recent success obtained through CFD simulations are indicative that both, physical and numerical approaches can be used as complementary techniques, rather than on eliminating the other. Computational fluid dynamics results are analogous to wind tunnel results obtained in a laboratory, in the sense that they both provide data for given flow configurations. However, unlike a wind tunnel which is generally heavier, a computer program can be carried on and accessed remotely by computers (Anderson, 1995).

Computational fluid dynamics (CFD) has recently made enormous strides. However, techniques for obtaining time dependent pressures induced by turbulent flows do not allow the routine and confident use of CFD as a substitute for wind tunnel testing of structures, although

can be a compliment for such testing. The role of CFD in structural engineering applications may be expected to become more important in the future (Simiu and Miyata 2006).

Use is often made of commercially available CFD codes because of their ready availability, well developed interfaces and broad verification and validation. The atmospheric boundary layer (ABL) extends for a considerable distance above the earth's surface relative to the average building height. CFD can only represent a smaller finite distance because of hardware limitation and the complexity of including a meteorological model. Currently, smaller features such as vegetation and small buildings cannot be included in the computational grid using personal computers. The  $k$ - $\epsilon$  model, that includes energy dissipation, is generally incorporated through a wall function approach that is based on boundary layer theory (Hargreaves and Wright 2007).

Accurate simulation of ABL flow in the computer domain is imperative to obtain accurate and reliable predictions of the related atmospheric process. In a CFD simulation, the flow profiles of mean wind speed and turbulence quantities that are applied at the inlet plane of the computational domain are generally fully developed profiles. These profiles should be representative of the roughness characteristics of that part of the upstream terrain that is not included in the computational domain. This is expressed by the presence of either the appropriate aerodynamic roughness length or the appropriate power law exponent of the terrain (Blocken et al. 2007).

The most common CFD techniques are capable of predicting the mean pressures on buildings with reasonable accuracy, but are not sufficiently accurate at evaluating the fluctuating and peak pressures. The poor representation of the pressure fluctuations is primarily because it is necessary to incorporate over simplified representations of the turbulence in the fluid flow

equations. However, CFD techniques are capable of providing useful insights into wind flow around building for environmental considerations (Holmes, 2001).

The lack of validation with the full scale, as was done in the early years on wind tunneling, could easily mislead a well intentioned structural engineer into thinking that the CFD package is generating real design loads. Engineers need to take the lead to ensure that non-validated data are not taken as gospel (Cochran, 2006).

## **2.4 Wind Tunnel Testing**

Wind tunnel testing was mostly carried out in aeronautical wind tunnels in smooth uniform flow. There were some isolated exceptions where the variation of wind speed with height was simulated. Although there were significant differences in the pressures between uniform and boundary layer flows, it appeared to be of academic interest only (Davenport, 2002).

A significant development during the 1950s was due to Jensen, who undertook a comparison of the mean pressures on small buildings in full scale and in wind tunnel model experiments. They were carried out in a variety of boundary layers, and he stated his model law, “The correct model test for phenomena in the wind must be carried out in a turbulent boundary layer and the model law requires the boundary layer to be scaled as regards the velocity profile” (Davenport, 2002).

Wind tunnel test on wind loading on structures require the simulation of the atmospheric boundary layer (ABL). Several methods have been proposed since 1960's to reproduce the atmospheric flow. It is accepted that the best atmospheric boundary layer simulation is obtained with rough floor surface, although simulation scales reached by this method are too small for

usual applications in structural aerodynamics. It has been shown that when comparing with atmospheric data, it is preferable to use comparative procedures, which do use the boundary layer thickness as a scaling factor (De Bortoli et al. 2002).

Wind tunnels have evolved as an indispensable aid to the practice of civil engineering. Boundary layer wind tunnels and currently data acquisition systems reveals that such tests continue to provide even more comprehensive wind load information for structural design (Cermak 2003).

With the basics of the aerodynamics of bluff bodies and the detailed characteristics of the atmospheric surface layer discussed in the previous sections, one can now approach the wind tunnel simulation process with confidence. The important element in the section on bluff body aerodynamics is the role played by the turbulence (small scale and large scale) in the formation of vortices under separated shear layers (Tieleman, 2003).

Independent tests conducted at six prominent wind tunnel laboratories on models of two industrial buildings showed that test results can vary significantly from laboratory to laboratory. Because of some variations in results, some structural engineering firms have engaged in the design of important structures, commission wind tunnel tests to more than one laboratory (Simiu and Miyata 2006).

## **2.5 Parapet Pressures**

The influence of parapets has been investigated for closed structures, perhaps in relation to wind effects on closed buildings. It is clear that low parapets may significantly increased the roof corner suctions for oblique wind directions. Data indicates that, in general, the higher the

parapet, the lower are the pressure coefficients. This is due to the fact that the parapet tends to lift the vortices away from the roof surface. A wind study reflecting a comparison with existing wind code provisions, NBCC (National Building Code of Canada) and ANSI 1985, against obtained experimental results on wind pressures and suctions on flat roofs with parapets (Stathopoulos and Baskaran, 1987).

Pressure studies have shown that parapet height can influence pressure coefficients at the roof edge. Low parapets, of the order of 0.5m in height, have been found to increase peak suction on the roof in comparison to a roof without a parapet. On the other hand, higher parapets, in the order of 1 m or higher, cause a significant reduction in peak suction (Stathopoulos et al. 2002).

The parapets tend to raise the corner vortex above the roof surface. For lower parapets, this acts to increase the local suctions; for higher parapets, the loads are decreased below those of no parapets. In all cases, the extents of the vortices on the roofs are expanded. The results of a systematic study on the effects of parapets on structural loads for low buildings indicate that parapets are no benign. The distance from the flow separation at the eave's edge to the first reattachment point on the roof for normal wind increases significantly. This leads to an increased load of about 10% in interior frames. Bay uplift is increased on end bays by similar amounts (Kopp et al. 2005).

## **CHAPTER 3. SIMULATION OF WIND FLOW USING COMPUTATIONAL FLUID DYNAMICS (CFD)**

There are several ways to represent wind flow on structural components, either through a physical representation of the structure (as would be done in a wind tunnel) or else using computer modeling. This chapter deals with the computational simulation of wind flow on structures using what is known as Computational Fluid Dynamics (CFD), whereas wind tunnel representations are discussed in the next chapter.

### **3.1 Theoretical background**

Computational fluid dynamics (CFD) techniques have been under development for a number of years (Holmes 2001) as a branch of fluid dynamics. At present, CFD constitutes a new approach in the study of wind effects on structures. The introduction of fast digital computers, together with the development of accurate numerical algorithms for solving numerical problems, has revolutionized the way researchers study and practice fluid dynamics today. This complements the approaches of the pure theory and pure experiments but, as Anderson (1995) states, it will never fully replace either of these approaches.

CFD results are directly analogous to wind tunnel results obtained in a laboratory; they both represent sets of data for given flow configurations at different Mach numbers, Reynolds numbers, etc. A CFD computer program is, therefore, a readily transportable tool, and could be thought as analogous to a transportable wind tunnel (Anderson 1995).

The physical aspects of fluid flows are governed by three fundamental postulates:

1. Mass is conserved

2. Newton's second law is satisfied, force = mass  $\times$  acceleration
3. Energy is conserved

The first step in a CFD problem is to specify the problem to be solved; second, an appropriate set of governing equations must be selected for the domain and boundary conditions; third, appropriate numerical approximations should be made. To obtain basic equations of fluid motion, one need to identify the relevant physical features, apply these features to a model of the flow and, from this application, extract the mathematical equations. The resulting equations are the well known Navier-Stokes equation, which are discussed in the following for the sake of completeness.

For convenience of the presentation, a Cartesian coordinate system is adopted here, where velocity and density ( $\rho$ ) are functions of space ( $x, y, z$ ) and time  $t$ , which considers the flow through an infinitesimal element of sides  $dx, dy$ , and  $dz$ . The scalar density is given by

$$\rho = \rho(x, y, z, t) \quad 3.1$$

Using the chain rule from differential calculus, it is possible to obtain  $d\rho$ :

$$d\rho = \frac{\partial \rho}{\partial x} dx + \frac{\partial \rho}{\partial y} dy + \frac{\partial \rho}{\partial z} dz + \frac{\partial \rho}{\partial t} dt \quad 3.2$$

From equation (3.2), the rate of change of density with time is represented by:

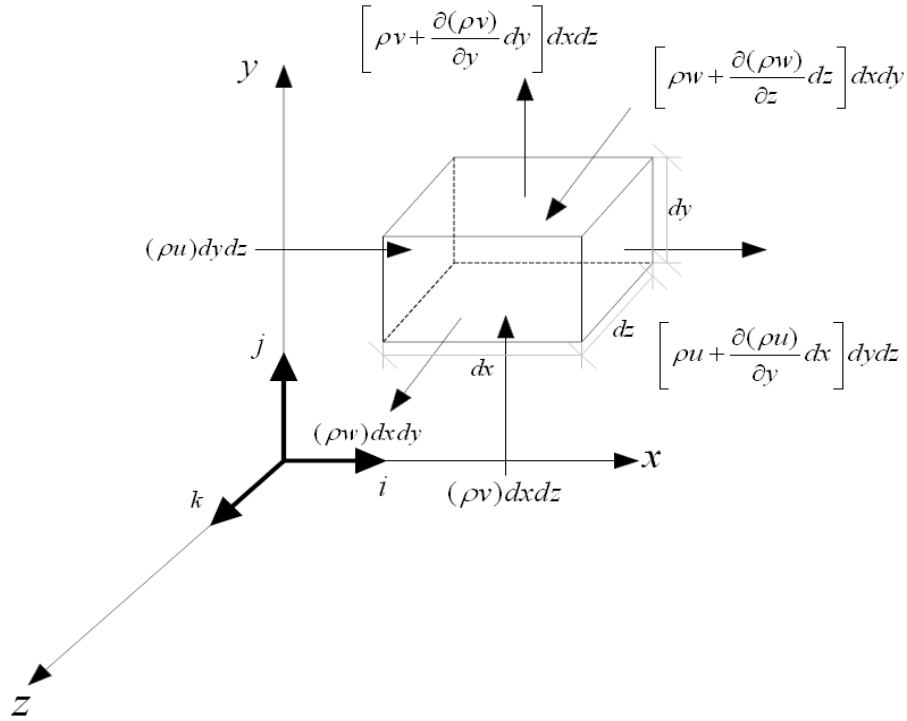
$$\frac{d\rho}{dt} = \frac{\partial \rho}{\partial t} + \frac{\partial \rho}{\partial x} \frac{dx}{dt} + \frac{\partial \rho}{\partial y} \frac{dy}{dt} + \frac{\partial \rho}{\partial z} \frac{dz}{dt} \quad 3.3$$

Since  $\frac{dx}{dt} = u$ ,  $\frac{dy}{dt} = v$ ,  $\frac{dz}{dt} = w$ , where  $u$ ,  $v$ ,  $w$  are velocity components, equation (3.3)

becomes

$$\frac{d\rho}{dt} = \frac{\partial\rho}{\partial t} + u\frac{\partial\rho}{\partial x} + v\frac{\partial\rho}{\partial y} + w\frac{\partial\rho}{\partial z} \quad 3.4$$

In CFD this equation is called the substantial derivative and represents a total derivative with respect to time. The governing flow equations are frequently expressed in terms of the substantial derivative.



**Figure 3.1: Model of an infinitesimal small element assumed fixed in space.**

Consider a flow model like the one shown in Figure 3.1, namely, an infinitesimal element fixed in space, with the fluid moving through it, for which a mass flow occurs through this element. From Figure 3.1 we have a net outflow in the x direction given by (Anderson 1995),

$$\left[ \rho u + \frac{\partial(\rho u)}{\partial x} dx \right] dydz - (\rho u) dydz = \frac{\partial(\rho u)}{\partial x} dx dydz \quad 3.5$$

The net outflow in the y direction is given by

$$\left[ \rho v + \frac{\partial(\rho v)}{\partial y} dy \right] dx dz - (\rho v) dx dz = \frac{\partial(\rho v)}{\partial y} dx dy dz \quad 3.6$$

Finally, the net outflow in the z direction is

$$\left[ \rho w + \frac{\partial(\rho w)}{\partial z} dz \right] dx dy - (\rho w) dx dy = \frac{\partial(\rho w)}{\partial z} dx dy dz \quad 3.7$$

Hence, the net mass flow out of the element is given by

$$\text{Net mass flow} = \left[ \frac{\partial(\rho u)}{\partial x} + \frac{\partial(\rho v)}{\partial y} + \frac{\partial(\rho w)}{\partial z} \right] dx dy dz \quad 3.8$$

The physical principle that mass is conserved, when applied to a fixed element, must equal the time rate of decrease of mass inside the element. Denoting the mass decrease by a negative quantity, this statement can be expressed as,

$$\left[ \frac{\partial(\rho u)}{\partial x} + \frac{\partial(\rho v)}{\partial y} + \frac{\partial(\rho w)}{\partial z} \right] dx dy dz = - \frac{\partial \rho}{\partial t} (dx dy dz) \quad 3.9$$

Or

$$\frac{\partial \rho}{\partial t} + \left[ \frac{\partial(\rho u)}{\partial x} + \frac{\partial(\rho v)}{\partial y} + \frac{\partial(\rho w)}{\partial z} \right] = 0 \quad 3.10$$

In Equation (3.10) the terms in brackets are  $\nabla \cdot (\rho V)$  where the operator  $\nabla$  is the divergence operator. Then, equation 3.10 becomes

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho V) = 0 \quad 3.11$$

$V$  is the flow velocity. Equation (3.11) is the partial differential equation form of the continuity equation, which was derived on the assumption of an infinitesimal small element fixed in space, as shown in Figure 3.1.

The second postulate is Newton's Law,  $F = ma$ , the momentum equation. Newton's second law, when applied to the moving fluid element states that the net force on the fluid element equals its mass times the acceleration of the element. This is a vector relation and can be split into three scalars relations in the x, y and z directions. Let us consider the x component of Newton's second law; the total force in the x direction is given by (Anderson 1995),

$$F_x = \left[ -\frac{\partial p}{\partial x} + \frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{xy}}{\partial y} + \frac{\partial \tau_{zx}}{\partial z} \right] dx dy dz + \rho f_x dx dy dz \quad 3.12$$

where  $f_x$  is the body force per unit mass acting on the element,  $p$  = surface pressure and  $\tau$  are the components of the stress tensor:  $\tau_{xy}$  and  $\tau_{zx}$  are the shear stress related to the time of change of shearing deformation of the fluid element in the xy and zx, respectively and  $\tau_{xx}$  the normal stress is related to the time rate of change of volume of the fluid element. The mass  $m$  of the fluid element is given by,

$$m = \rho dx dy dz \quad 3.13$$

Since we are following a moving fluid element, the rate of change with respect to time in the  $x$  direction is given by,

$$a_x = \frac{Du}{Dt} \quad 3.14$$

Combining equations (3.12), (3.13) and (3.14), we can obtain the  $x$  component of the momentum equation for a viscous flow. The  $x$ ,  $y$  and  $z$  components are the components of the momentum equations (Anderson 1995).

$$\rho \frac{Du}{Dt} = -\frac{\partial p}{\partial x} + \frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{yx}}{\partial y} + \frac{\partial \tau_{zx}}{\partial z} + \rho f_x \quad 3.15$$

$$\rho \frac{Dv}{Dt} = -\frac{\partial p}{\partial y} + \frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{zy}}{\partial z} + \rho f_y \quad 3.16$$

$$\rho \frac{Dw}{Dt} = -\frac{\partial p}{\partial z} + \frac{\partial \tau_{xz}}{\partial x} + \frac{\partial \tau_{yz}}{\partial y} + \frac{\partial \tau_{zz}}{\partial z} + \rho f_z \quad 3.17$$

The partial differential equations were obtained from an application of the fundamental physical principle to an infinitesimal fluid element. These are scalar equations and are called the Navier-Stokes equations. These equations are (Anderson 1995),

$$\frac{\partial(\rho u)}{\partial t} + \nabla(\rho u V) = -\frac{\partial p}{\partial x} + \frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{yx}}{\partial y} + \frac{\partial \tau_{zx}}{\partial z} + \rho f_x \quad 3.18a$$

$$\frac{\partial(\rho v)}{\partial t} + \nabla(\rho v V) = -\frac{\partial p}{\partial y} + \frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{zy}}{\partial z} + \rho f_y \quad 3.18b$$

$$\frac{\partial(\rho w)}{\partial t} + \nabla(\rho w V) = -\frac{\partial p}{\partial z} + \frac{\partial \tau_{xz}}{\partial x} + \frac{\partial \tau_{yz}}{\partial y} + \frac{\partial \tau_{zz}}{\partial z} + \rho f_z \quad 3.18c$$

For virtually all practical aerodynamics problems, Stokes in 1845 obtained the constitutive conditions in the form (Anderson 1995)

$$\tau_{xx} = \lambda(\nabla \cdot V) + 2\mu \frac{\partial u}{\partial x} \quad 3.19a$$

$$\tau_{yy} = \lambda(\nabla \cdot V) + 2\mu \frac{\partial v}{\partial y} \quad 3.19b$$

$$\tau_{zz} = \lambda(\nabla \cdot V) + 2\mu \frac{\partial w}{\partial z} \quad 3.19c$$

$$\tau_{xy} = \tau_{yx} = \mu \left[ \frac{\partial v}{\partial x} + \frac{\partial u}{\partial y} \right] \quad 3.19d$$

$$\tau_{xz} = \tau_{zx} = \mu \left[ \frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \right] \quad 3.19e$$

$$\tau_{yz} = \tau_{zy} = \mu \left[ \frac{\partial w}{\partial y} + \frac{\partial v}{\partial z} \right] \quad 3.19f$$

where  $\mu$  is the molecular viscosity coefficient and  $\lambda$  is the second viscosity coefficient. Stokes made the hypothesis that  $\lambda = \frac{2}{3}\mu$ . Substituting (3.19) into (3.18) we obtain the complete Navier-Stokes equations in conservation form. In the CFD literature, a “Navier-Stokes solution” means a solution of a viscous flow problem using the full governing equations. This includes the solution for the momentum, energy and mass equations. They are a system of nonlinear partial

differential equations, and hence are difficult to solve analytically, with the consequence that no general closed form solution exist to these equations at present.

The third postulate states that the energy is conserved. The first law of thermodynamics states that when applied to the flow model of a fluid element moving with the flow, the rate of change of energy inside the fluid element is equal to the heat into the element, plus the rate of work done on the element due to body and surface forces. The rate of doing work by a force exerted on a moving body is equal to the product of the force and the component of velocity in the direction of the force.

Finally, a numerical method and a strategy for dividing the flow domain into cells or elements must be selected. Many different gridding strategies exist, including structured, unstructured, hybrid, composite, and overlapping grids. The most common choices of a numerical method are finite difference, finite volume, finite element, and spectral methods (Zingg 1999).

The most common finite difference representations of derivatives are based on Taylor's series expansions. The majority of the partial differential equations on the governing equations can be replaced by a system of algebraic difference equations for the dependent variables of a difference equation. Finite volume methods have become very popular in CFD as a result of two major advantages: First, they ensure that the discretization of the governing equations is conservative and second, the volume methods do not require transformations in order to be applied to irregular meshes. The basic idea of the finite-volume method is to satisfy the integral form of the conservation law to some degree of approximation for many continuous control volumes, which cover the domain of interest. For this research, the finite volume method is used

(as implemented by the software developer), to satisfy the conservation law to some degree of approximation for contiguous control volumes, which cover the domain of interest.

### **3.2 EFD.Lab program description**

The wind flow on an open canopy structure with parapets can be represented and modeled with the use of CFD. The specific CFD software used in this research is named EFD.Lab (Engineering Fluid Dynamics), developed by Flomerics Inc. (Flomerics 2009). EFD.Lab solves the Navier-Stokes equations, which include the formulations of mass, momentum and energy conservation laws for fluid flows. The equations are supplemented by fluid state equations defining the nature of the fluid and by empirical dependencies of fluid density, viscosity and thermal conductivity on temperature (Flomerics 2009).

EFD.Lab solves the governing equations using the finite volume method (FVM) on a prismatic rectangular computational mesh drawn in a 3D Cartesian coordinate system with the planes orthogonal to its axes. To improve the results, the mesh is refined locally at the solid/fluid interface. Turbulence is incorporated by means of turbulent intensity and turbulent length parameters, which were obtained from the wind tunnel and used in the CFD computations in order to emulate the wind tunnel conditions.

### **3.3 CFD model**

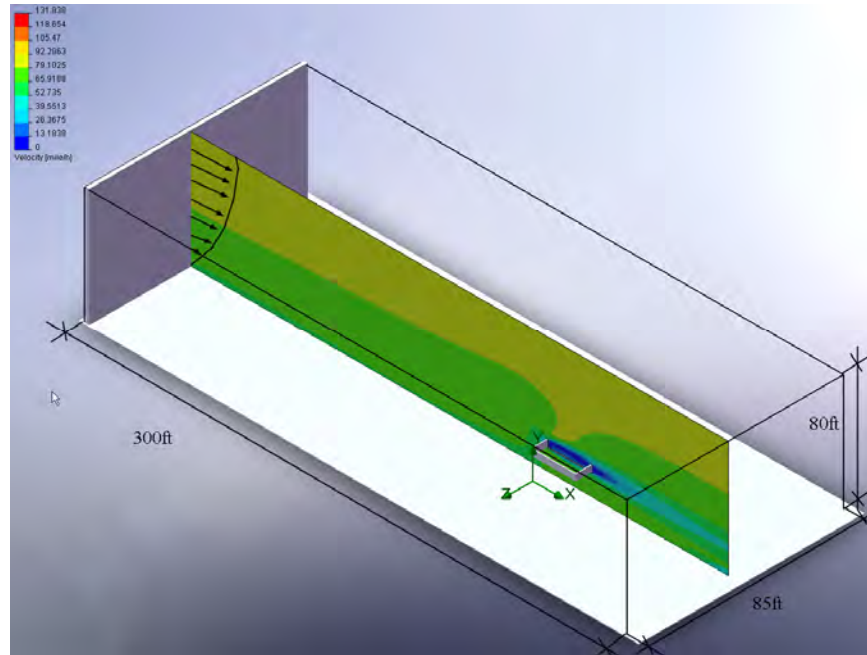
The initial computational mesh used for the simulation is created in a prismatic domain, as shown in Figure 3.2. The domain dimensions assumed in this initial example are 300 ft (91.4 m) in the x direction, 85 ft (25.9 m) in the z direction, and 80 ft (24.4 m) in the y direction. The surface area on the left is defined as the inlet boundary, in which it is possible to introduce the

non-uniform values of a wind profile. The specific wind profile adopted in the first model investigated in this research is a power law (as defined by ASCE 7 05),

$$V(z) = V(z_{ref}) \left( \frac{z}{z_{ref}} \right)^{1/\alpha}$$

This velocity profile has been investigated for the specific wind tunnel used in the research at Universidad Nacional del Nordeste (UNNE) in Resistencia, Argentina, and a value of  $1/\alpha = 0.24$  is recommended by the researchers of that wind tunnel (De Bortoli et al. 2002). The exponent  $1/\alpha$  depends on the surface roughness and on the wind speed averaging time.  $z_{ref}$  is the reference height, namely 10 m (32.8 ft).  $V(z_{ref})$  is the velocity of reference at the height of reference.

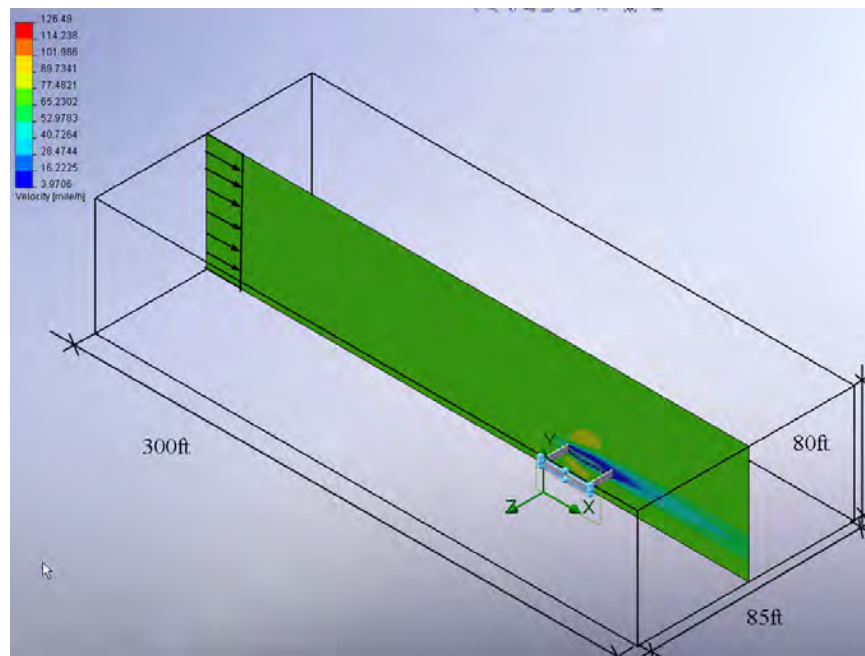
The simulation of a wind flow on a canopy requires the definition of a flow model and an element mesh; each one has been considered in order to identify a mesh providing acceptable results. Our first model considered is shown in Figure 3.2, in which the incoming flow has a velocity profile which follows the power law. Different velocities on the left inlet entrance of Figure 3.2 indicate that a wind profile using the power law was used as a wind velocity input. The canopy induces significant changes to the flow, and the program calculates wind pressures and  $C_p$  values on the model surfaces.



**Figure 3.2: Computational domain for Model #1, with boundary definition and laminar wind profile. The inlet condition is defined in the plane on the left.**

EFD.Lab determines for external flows, the boundary conditions for all the boundaries of the computational domain. The boundary conditions on any CFD problem define the solutions to be obtained from the governing equations. In this particular model, the inlet surface has been defined as mentioned before, a power law wind profile, with different wind velocities specified through the height. The bottom boundary condition is automatically defined based on surface roughness. The outlet boundary is defined as away from the surface boundary as possible, simulating the real physical boundary condition. The rest of the boundaries assume a zero relative velocity between the surface and the fluid immediately at the surface. This is called the no-slip condition at the surface for a viscous flow.

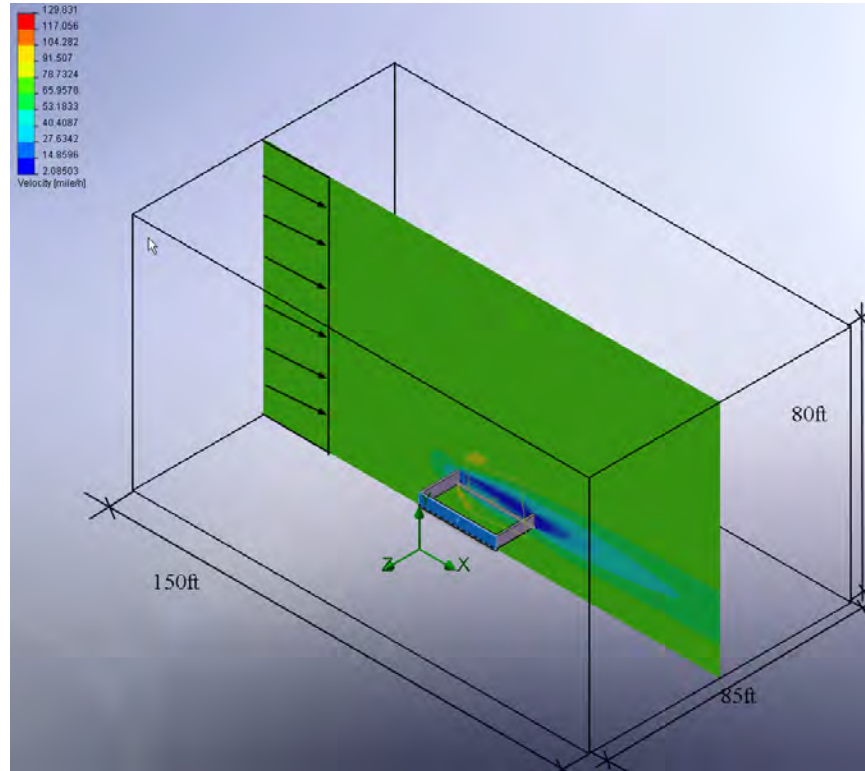
A second computational model, named Model #2, with the same mesh configuration as Model #1, was also built, in which a uniform flow was assumed, i.e. a uniform velocity profile in height was assumed at the inlet boundary (see Figure 3.3). Although the inlet conditions between a power law and a uniform velocity profile are very different, in the present case in which the canopy is a thin body located at a given distance from the terrain, the results in terms of pressures acting on the canopy are remarkably similar. The reasons for this similarity are that the canopy is located away from the turbulence generated at the ground level, and the pressures are dominated by the pressures and velocities that occur at the canopy height. Such similarity is only specific of this case and does not occur in enclosed structures.



**Figure 3.3: Computational domain for Model #2, without boundary definition and uniform wind profile.**

To investigate the sensitivity of the solution to changes in the number of elements adopted, the laminar model was considered. A third computational model (see Figure 3.4), named Model #3 was also created. This model has a smaller computational domain in which

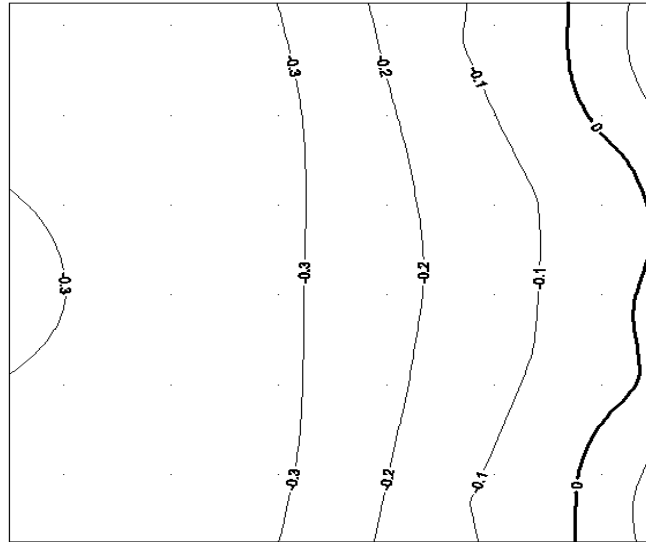
laminar flow was assumed with a constant velocity profile in height at the inlet boundary. The domain dimensions assumed in this initial example are 150 ft (45.7 m) in the x direction, 85 ft (25.9 m) in the z direction, and 80 ft (24.4 m) in the y direction. This computational model was created to verify if similar results can be obtained with a smaller computational domain and number of fluid cells compared to the computational domain on Model #2 and Model #1.



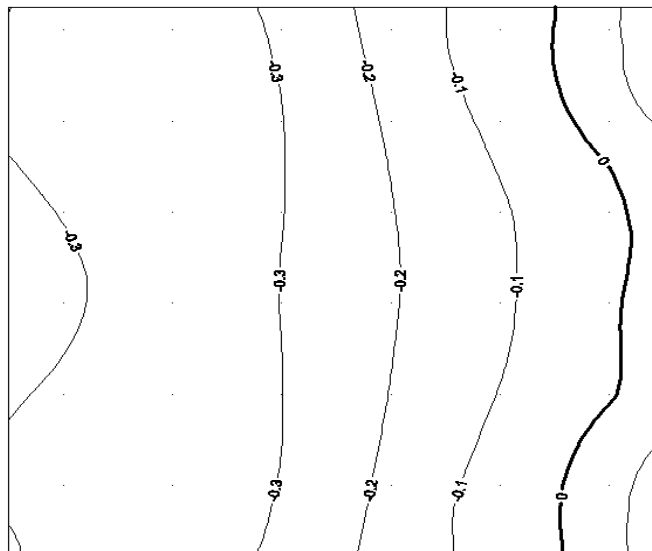
**Figure 3.4: Computational domain for Model #3, without boundary definition and uniform wind profile.**

The results from the initial computational domain, Model #1, with the use of the inlet boundary layer calculated with the power law, were compared against a similar domain in terms of geometry and number of fluid cells. The number of fluid cells generated for Model #1 and Model #2 was 303,123. The domain for Model #3 generated a number of cells of 54,954. The computational time on a personal computer was drastically reduced from 3 hours and 27 minutes

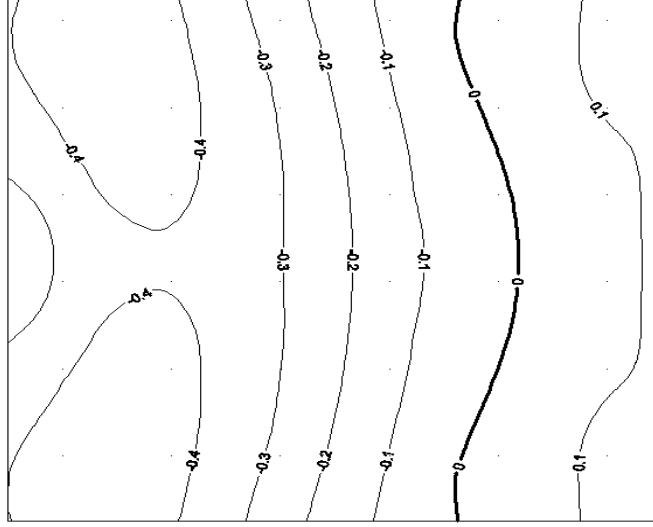
for Model #1 and Model #2, to 23 minutes for Model #3. In addition,  $C_p$  values on top surfaces on each canopy model were compared (See Figure 3.5 to 3.7) and the results obtained were similar. Based on these results, it was decided that further analysis would be performed with the smaller domain of Model #3 using a uniform velocity profile at the inlet surface, as shown in Figure 3.4.



**Figure 3.5: Model #1 top surface  $C_p$  values.**



**Figure 3.6: Model #2, top surface  $C_p$  values.**



**Figure 3.7: Model #3, top surface Cp values.**

The significance of velocity in the model and in real situations is not self-evident and needs some clarification. The sustained wind velocity used for analysis on Model #2 and Model #3 was obtained from the wind velocity in Model #1, with the use of the power law formula. The wind speed velocity calculation with an hourly wind speed of 96 mph (145 mph 3 sec gust wind speed, same as ASCE 7-05 basic wind speed for Puerto Rico) was obtained using Figure C6-2, in the ASCE 7-05 Commentary (ASCE 2005).

$$V(z) = (V_{z_{ref}}) \left( \frac{z}{z_{ref}} \right)^{1/\alpha} = 96 \text{ mph} \left( \frac{12 \text{ ft}}{32.8 \text{ ft}} \right)^{1/4.16} = 75 \text{ mph} \quad 3.21$$

$V_{z_{ref}}$  = hourly wind speed wind speed (mph),  $z$  = mean roof height (ft),  $z_{ref}$  = reference height = 32.8 ft (10 m),  $1/\alpha$  = power law exponent =  $1/4.16 = 0.24$  for open terrain on the UNNE wind tunnel.

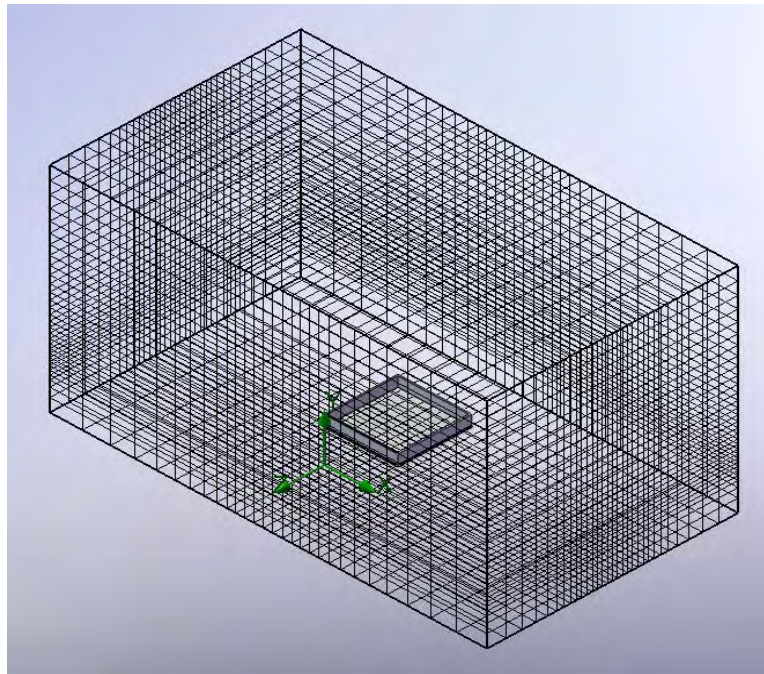
This result is based on the power law and the result is the hourly wind speed at 12 ft averaged over one hour. The averaged one hour wind speed is commonly used as a reference wind speed in wind tunnel simulations. The hourly wind speed obtained will be used in the CFD simulations for Model #2 and Model #3.

The wind speed in Model #1 at the canopy roof height was approximately 75 mph. The same wind velocity of 75 mph, but sustained, instead of a wind profile calculated by the power law, was applied to Model #2 and Model #3. This means that the same wind velocity was applied on all investigated models at open canopy height. The results for laminar (Figure 3.5) and for uniform velocity flow (Figure 3.6) are virtually identical in terms of values and distributions. Based on the agreement and on the significant differences in computer time required, it was decided that the detailed investigations and comparisons with wind tunnel would be done using uniform velocity flow. Regarding the use of a different number of cell elements in the uniform velocity flow (Figures 3.6 and 3.7), some small differences were obtained in the results. The area of exposure of both models against the wind at the same canopy heights is similar. Further analysis was performed with the smaller domain of Model #3, using a uniform velocity profile at the inlet surface. The CFD Reynolds number calculated for the simulation is  $8.52 \times 10^7$ .

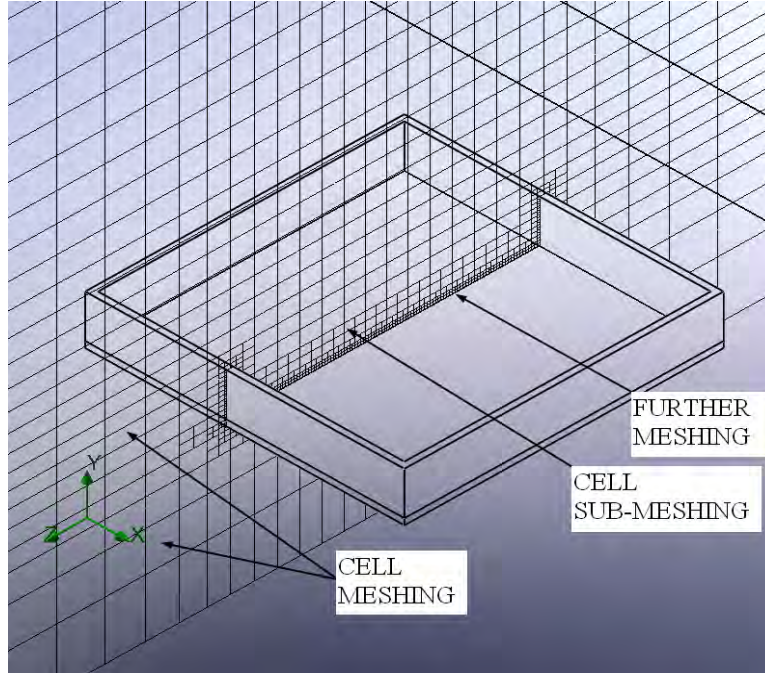
The CFD software yields results of wind pressures on the external (top and bottom) surfaces of the structure. In this case, a wind velocity similar to those specified by design codes has been adopted, with an hourly wind speed of 33.5 m/s (75 mph) for all simulations. This sustained wind speed is equivalent to 64.8 m/s (145 mph) 3 second gust wind speed, as required by ASCE 7-05 for Puerto Rico. Within the context of this research, CFD simulations were extremely useful to identify and define the geometric characteristics of the canopies to be investigated prior to the construction of physical models that were tested in a wind tunnel facility. Preliminary investigations allowed the identification of the high-pressure areas, which were in turn used to specify the location of the wind pressure taps that were used in the small scale models.

The initial geometry of the canopy analyzed consists of a 7.6 m (25 ft) x 7.6m (25ft) with a 1.2m (4ft) parapet. The roof height is located at 3.6m (12ft). The proposed geometry was decided after comparing photographs and visits through many existing gas stations in Puerto Rico. The proposed geometry is very common and used in many gas stations in Puerto Rico.

The mesh cells sides used on the program are orthogonal to the specified axes on the coordinate system. The mesh can be automatically created using a mesh generator, but the mesh in this simulation was specified so that it could be refined on the exposed surfaces. Further sub-meshing was assigned for refinement on all exposed surfaces. Refer to Figure 3.8 and Figure 3.9 for meshing and sub-meshing images.



**Figure 3.8: Computational domain and meshing used in Model #3.**



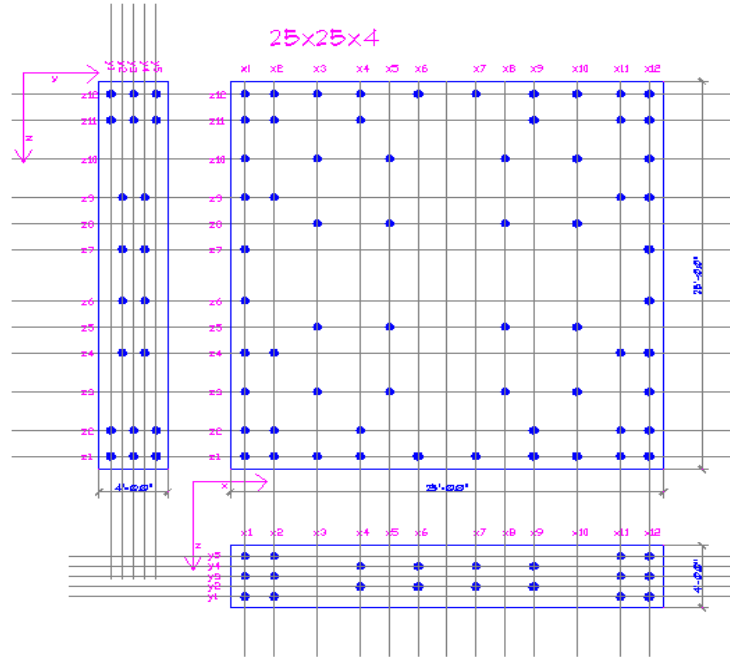
**Figure 3.9: Open canopy showing fluid cell mesh, submeshing and further meshing at Model #3, domain intercept.**

Wind pressures coefficients  $C_p$ 's were obtained from the CFD simulations. To compare the results from the wind tunnel, data was obtained at the same geometric location where the pressure taps were located in the scale model. Specifically, pressure taps were located on the top surface, bottom surface and on all parapets covering inside and outside surfaces, and their locations as indicated in Figure 3.10. The same tap locations were used in the CFD models to specify mesh points. Results of the  $C_p$  (pressure coefficient factor) were obtained from the CFD simulation, and have been represented in Figure 3.11 in terms of contour lines. The expression used in EFD.Lab to evaluate the pressure coefficients is:

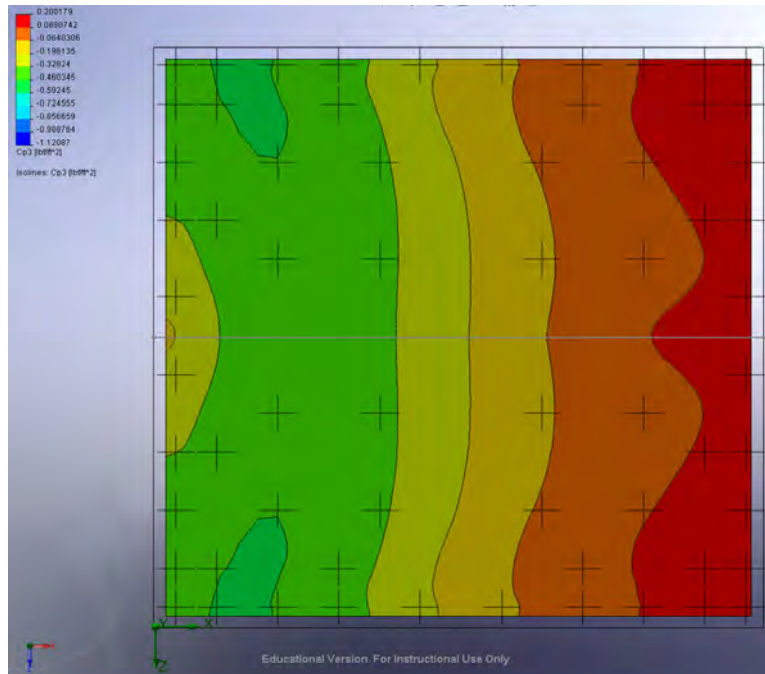
$$C_p = \frac{p - p_0}{\frac{1}{2} \rho V^2} \quad 3.22$$

where  $V$ =reference wind speed (fps),  $\rho$  = air density (slugs),  $p$ = pressure measured at point of interest (psf),  $p_0$  = reference pressure (psf).

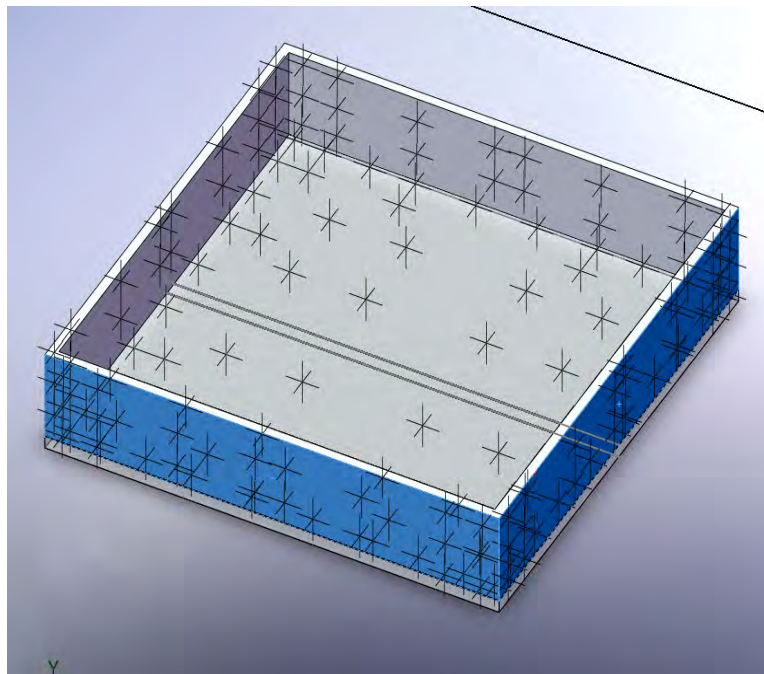
The CFD model with the location of the pressure taps has is shown in Figure 3.12. As mentioned before, the pressure taps are at the same locations as in the model tested at the wind tunnel. Results from those simulations will be reported in Chapter 6 of this dissertation.



**Figure 3.10: Pressure tap locations on top, bottom and parapet surfaces used for CFD and wind tunnel scale model.**

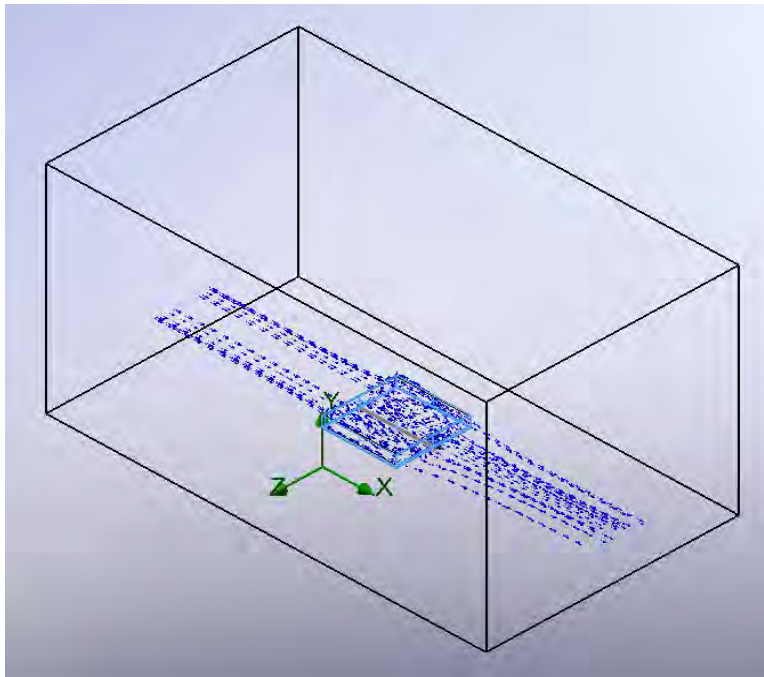


**Figure 3.11: Location of pressure taps and measured wind pressures on the top surface.  
Wind direction acting from the right (0 degrees).**

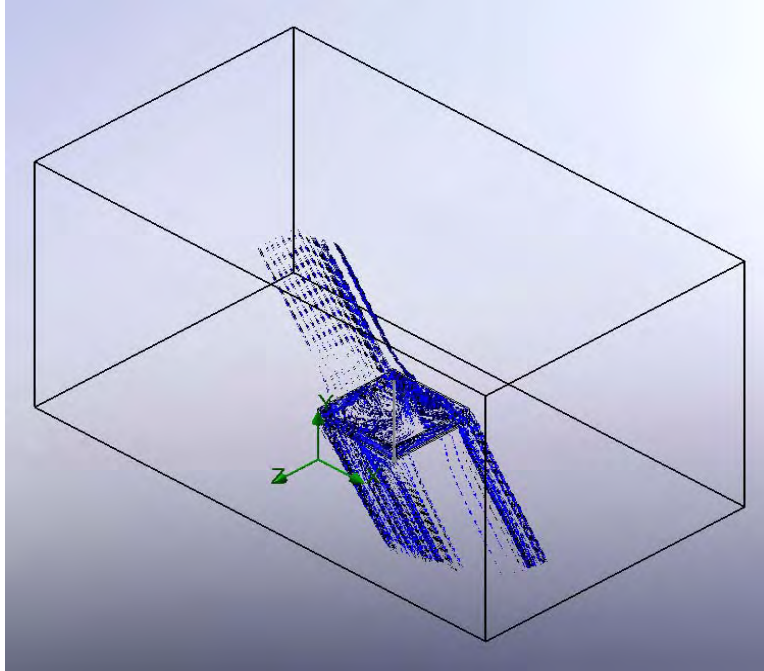


**Figure 3.12: Model pressure taps on the canopy model.**

Two wind directions were used for the computational model, which were the same wind directions used in the wind tunnel. Following Gumley (1981, 1982), and Letchford (1992), one direction was assumed at 0 degrees (see Figure 3.13) and the second one at 30 degrees of the long axis of the canopy (see Figure 3.14). Verification was done with the use of CFD and running one model with the wind applied at different directions, and we confirmed that the choice of 0 degrees and 30 degrees leads to extreme values in terms of  $C_p$ . Considerations regarding the use of CFD models will be made again in Chapter 6, where numerical and experimental results are compared.



**Figure 3.13: Wind direction in CFD model at 0 degrees.**



**Figure 3.14: Wind direction in CFD model at 30 degrees.**

## **CHAPTER 4. WIND TUNNEL TESTING**

### **4.1 General features of wind tunnel studies**

The use of wind tunnels to aid in civil engineering, structural design and planning has been increasing in recent years. The purpose of wind tunnel tests is to provide designers with information on local wind parameters, and wind loads having an accuracy far exceeding that which can be obtained from predictions based on other less expensive means such as theory, numerical analysis, and consulting (Liu 1991).

Some structures warrant a wind tunnel test, like skyscrapers, large structures, constructions having unusual shapes or major structures located in special locations. Examples of special locations are those affected by topographical features, such as hills, cliffs or valleys. In these situations, the use of wind tunnel tests is used to improve the final design. Failure to conduct tests or investigate using the wind tunnel may result in an unsafe design.

A wind tunnel model test conducted in the United States may approximately cost in the order of forty thousand dollars, so that most companies cannot afford a wind tunnel test for every structure to be analyzed. A wind tunnel test cannot be financially justified, unless the expected savings from such a test are far greater than the cost of conducting it. Safety in most of the cases is not the motivation for wind tunnel tests. Savings in cladding are the primary motivation in terms of analyzing structures with complex geometries. In several cases, wind tunnel testing can produce thousands of dollars in savings when actual wind pressures can be obtained from it.

Numerous criteria exist to categorize wind tunnels. According to the flow circuit, a tunnel may be classified as either an open or closed circuit tunnel. An open circuit tunnel is normally a

straight structure. Air is drawn into the tunnel from a funnel-shape intake at one end of the tunnel, and the air exits the tunnel through a funnel shape outlet. The enlarged cross-sectional areas at the two ends prevent undesirable strong winds from being generated outside the tunnel near the inlet or outlet. The closed circuit type is a recirculation loop. It may occupy a small space if the loop is vertical. Advantages of closed circuit tunnels are, (1) they do not cause undesirable winds in laboratories housing the wind tunnels; (2) they generate less noise. Both open and closed circuit tunnels are often used for testing structural models. The tunnel used in this investigation, was an open circuit tunnel at UNNE.

A boundary layer wind tunnel must have a test section that is sufficiently long to generate a thick vertical boundary layer, sufficiently high so that the boundary layer generated will not touch the tunnel ceiling and sufficiently wide so that neighboring structures and topographical features can be incorporated into the model. Furthermore, the blockage ratio, which is the ratio of the cross sectional area of the model blocking the flow and the cross sectional area of the tunnel test section, must be less than one tenth. These requirements necessitate rather large tunnels. The boundary layer tunnels used in commercial testing of structural models normally have a minimum width of 2.4 m (8ft), minimum height of 1.5 m (5 ft), and a minimum length of 10 m (33 ft), all measures referring to test sections.

To facilitate the rapid growth of a vertical boundary layer along the tunnel test section, roughness elements must be placed on the tunnel floor, and additional devices, such as spires, must be installed upstream. The roughness and the spires must be designed to produce the type of velocity profile and the type of turbulence similar to that encountered by the prototype structure. Typically, the model tested in a wind tunnel is placed on a turntable so that it can be

studied for winds from different directions. To simulate the wind field correctly, the model should include the terrain and structure features.

Major components of a wind tunnel include fan, test section, nozzle, diffuser, honeycomb, flow strengtheners, guide vanes, screens, turntable, spires and roughness elements on the floor. The fan is needed for all types of wind tunnels. The test section is where the model is tested and where the atmospheric boundary layer is simulated. The model is always placed near the downstream end of the test section where the boundary layer thickness is maximum. The turntable is round and has a diameter slightly smaller than the width of the test section. The turntable is covered with the structure to be tested. Further upstream the floor is covered with roughness elements to generate and maintain a turbulent boundary layer. The roughness elements are normally cubic elements attached to the tunnel floor (Liu 1991).

## **4.2 Wind tunnel description**

Wind tunnels are equipment designed to obtain air flow conditions, so that similarity studies can be performed, with the confidence that actual operational conditions can be reproduced. The wind tunnel testing was performed during 2008/2009 academic year at Universidad Nacional del Nordeste (UNNE) in Resistencia, Argentina. The UNNE wind tunnel facility is a low velocity atmospheric boundary layer wind tunnel, built with the aim to perform aerodynamic studies of structural models. The distribution of the flow on the structural model must be such that the atmospheric boundary layer at the actual location is reproduced. This is obtained with the help of turbulence promoters and vortex generators, so that wind simulations could be performed (Wittwer and Moller 2000).



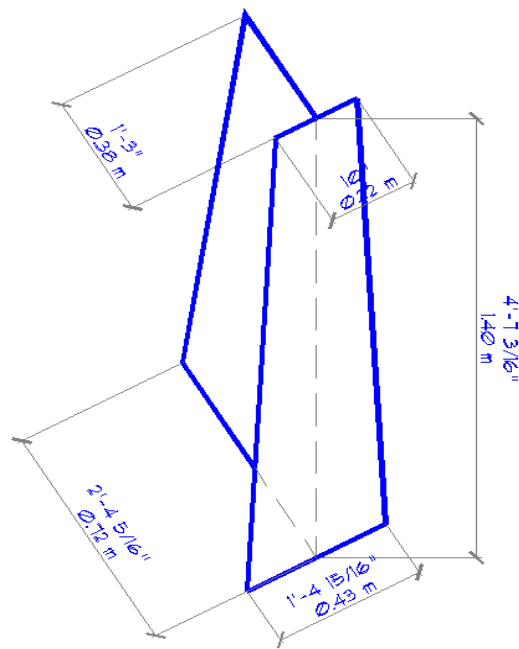
**Figure 4.1: View the UNNE wind tunnel facility. (Photograph by the Author)**

The facility is an open circuit tunnel with a length of 22.8 m (74.8 ft), the testing chamber being a rectangular section of 2.4 m (7.87 ft) width and 1.8 m (5.9 ft) in height and uses a 2.25 m diameter (7.38 ft) fan with a 92 kW motor. The maximum wind velocity that may be obtained in this tunnel is 25 m/s (55.9 mph) when the testing section is empty (the section is empty when there is no model in the testing chamber). Further details regarding this wind facility are given by Wittwer and Möller (2000).

All models were tested under a wind simulation corresponding to a suburban area. The simulation of the natural wind on the atmospheric boundary layer was performed by means of the Counihan and Standen methods with velocity distributions defined as ground covered by several closely spaced obstacles in forest or urban territory (Wittwer and Moller 2000). The Counihan and Standen methods are techniques developed to thicken boundary layers on wind

tunnels. These techniques are generally known as roughness, barrier and mixing-device methods (Natalini et al. 1998). Surface roughness was obtained on 17 m (55.7 ft) test floor section with prismatic 30x30x22 mm (1.18x1.18x.89 in) elements, 80 mm (3.14 in) apart. The choice of element size and packing density was based on the works of Counihan, Fang and Sill, Gartshore and Cross (Natalini et al. 1998).

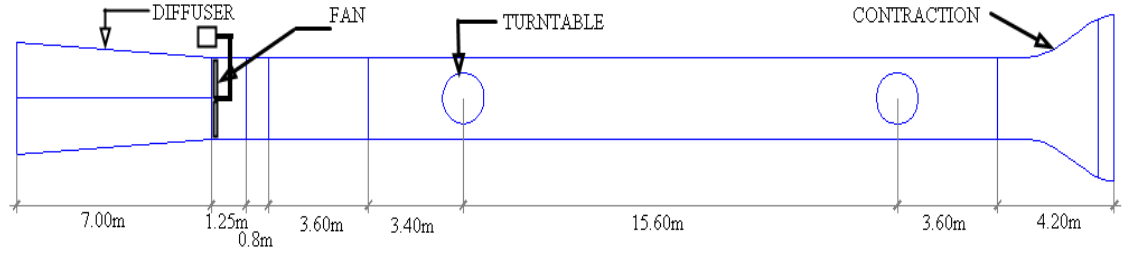
Representative values of power law exponent  $1/\alpha = 0.24$  and the roughness length parameter  $z_0 = 0.7$  m were adopted in order to obtain a scale of 1:500, a usual scale in simulations. Consequently, the design value in the wind tunnel for the specified scale is  $z_0 = 1.4$  mm. The same suburban terrain exponent was used on the CFD simulation. The suburban terrain condition corresponds to open terrain with scattered obstructions having heights of less than 9.1 m (30 ft). According to the ASCE 7-05, it corresponds to Exposure B. In order to obtain a boundary layer depth of  $\delta = 1.3$  m artificial simulation methods were applied. This technique called natural simulation produces simulation scales higher than 1:500, and it requires the use of long wind tunnels. The simulation hardware consisted of two modified Irwin's spires, as shown in Figure 4.2 and 4.3. Figure 4.4 shows the tunnel plan at UNNE.



**Figure 4.2: Typical Irwin spires used on the UNNE wind tunnel.**

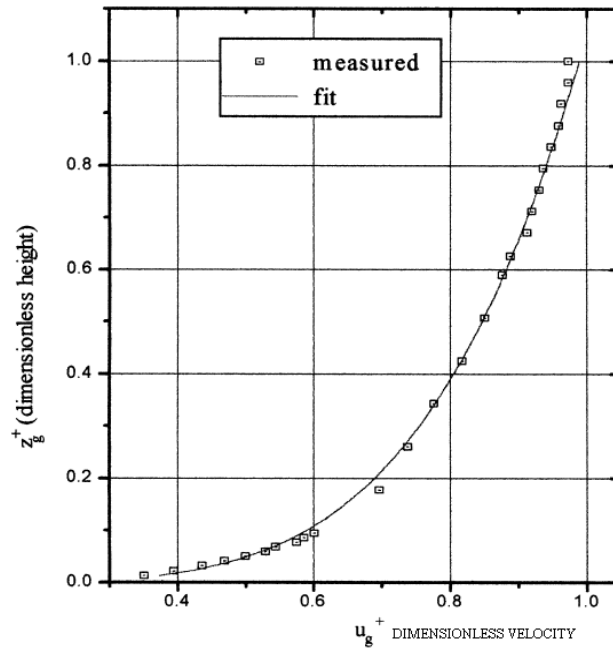


**Figure 4.3: View the UNNE surface roughness and Irwin spires. (Photograph by the Author)**



**Figure 4.4: Wind tunnel plan at UNNE (reproduced from Wittwer and Moller 2000).**

In this way, a part-depth boundary layer simulation of neutrally stable atmosphere was obtained. Mean velocities, when fitted to a potential law, with an exponent of 0.24 obtained at UNNE are shown on Figure 4.5.



**Figure 4.5: Mean velocity profiles at the UNNE wind tunnel (Wittwer and Moller 2000).**

### 4.3 Construction of models

The similitude theory in fluid mechanics requires that all model tests must be conducted under geometric, kinematic and dynamic similarities. Geometrical similarity requires that the shape of the model must be the same as that of the prototype. Kinematic similarity means the velocity field must be similar. Dynamic similarity means the pressure distribution and the generated forces must be similar too (Liu 1991). These three similarities are not independent of each other. For example, in this research a scale ratio of 1:50 is proposed. Making the Reynolds number identical for the model as for the CFD model yields

$$R_m = R_p \Rightarrow \frac{\rho_m V_m L_m}{\mu_m} = \frac{\rho_p V_p L_p}{\mu_p}$$

where  $R_m$ =Reynolds number for the model,  $R_p$ =Reynolds number for CFD,  $\rho$ =air density,  $\mu$ =dynamic viscosity,  $V$ =velocity and  $L$ =length. Air density and viscosity are similar on both cases, consequently

$$\frac{V_m}{V_p} = \frac{L_p}{L_m} = 50$$

This simple calculation shows that to maintain the same Reynolds number the wind speed for the wind tunnel needs to be 50 times of the prototype velocity. This is very hard to obtain on a wind tunnel. The Reynolds number on the wind tunnel is  $3.67 \times 10^6$ . The calculated Reynolds number on the CFD simulation is  $8.52 \times 10^7$ . As long as the Reynolds number is not smaller than  $10^4$ , it will allow the flow around the model to remain turbulent, and kinematic and dynamic similarities will prevail (Liu 1991).

Experiments were conducted on one 1:50 scale model of a 7.5 m (25 ft)  $\times$  7.5 m (25 ft) square roof with parapets of 1.2 m (4 ft) high, having an eave height of 3.6 m.(12 ft). The usual

scale for wind tunnel testing of open structures varies from 1:50 to 1:100. In this particular research, a scale of 1:50 was chosen in order to install all the PVC tubes to be connected to the testing instruments without any interference.

First, experiments were carried out without parapets in order to compare with other wind tunnel results available in the open literature. Wind load coefficients were measured under wind blowing at angles of  $0^\circ$  and  $30^\circ$  relative to one of the symmetry axis, since, as demonstrated by Gumley (1981, 1982) and Letchford et al. (1992) among others, these directions produce the most severe loads on planar canopy roofs with no parapets. In the absence of additional information, it was expected that such angles would still be relevant for canopies with parapets.

The roof of the model and the parapets were made with a 2 mm thick aluminum plate, and the columns with a 2.5 mm diameter steel rod. As the models have two axes of symmetry, only a quarter of the roof model was instrumented with pressure taps in place, thus reducing the number of tubes needed. In addition, all the tubes were led towards the farthest corner, where they formed a bundle that went into a horizontal pipe through which they went away from the model to finally go under the floor. In this way, the scale distortion in both columns and roof thickness and the possible interference of the tubes upon the measurements were minimized. The model is shown in Figures 4.6 to 4.8.



**Figure 4.6: Model with parapets with PVC tubes and columns. (Photograph by the Author)**

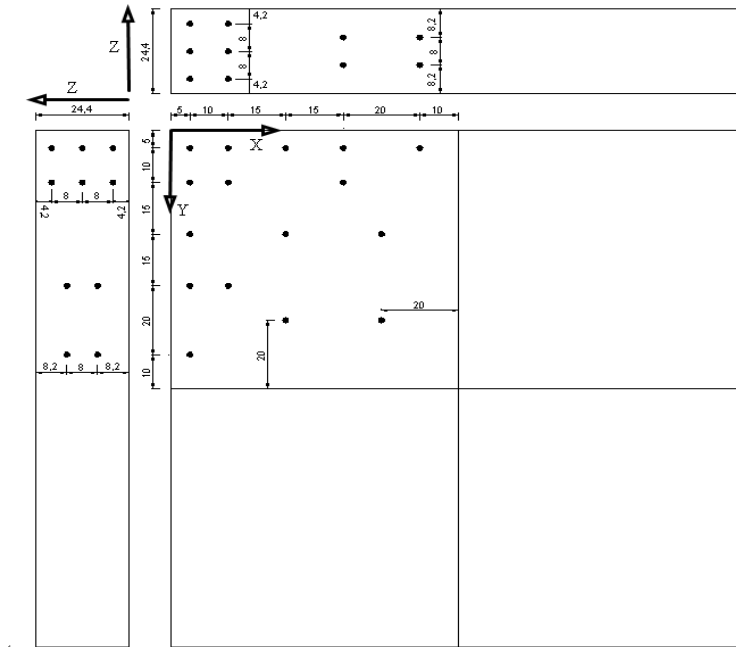


**Figure 4.7: Top view of the model, showing the distribution of 16 pressure taps on the top surface of the canopy and parapets. (Photograph by the Author)**

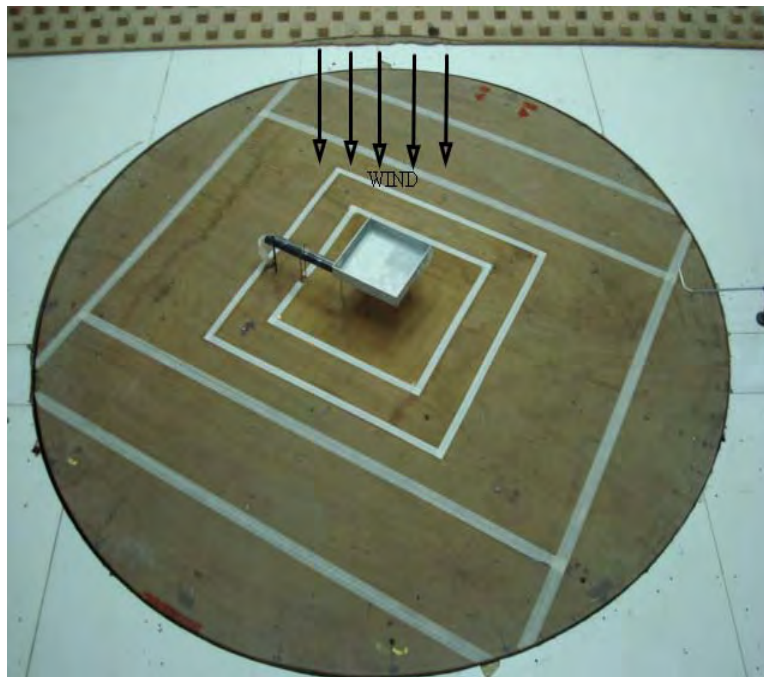


**Figure 4.8: Distribution of pressure taps on the model with bundle of tubes and columns.  
(Photograph by the Author)**

Sixteen pressure taps were spread on the roof and twenty on the parapets. The specific locations of the taps were influenced by the previous computational results which indicated, and suggested places of special interest in terms of pressure values. Figure 4.9 shows the position of the pressure taps on the top surface of the model and Figure 4.10 shows the model placed in the turntable. The turntable is a round table of 1.5 m (4.9 ft) of width, which is rotated during the wind test in order to obtain the desired wind pressures.



**Figure 4.9: Distribution of pressure taps on the model.**



**Figure 4.10: View of the model with parapets on the turntable (Photograph by the Author).**

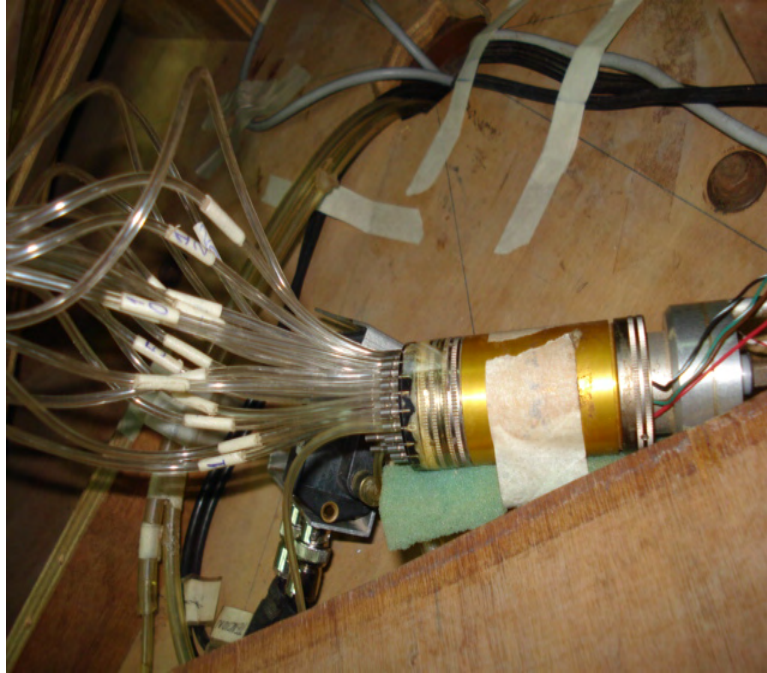
## 4.4 Instrumentation

Pressures were measured using a differential pressure electronic transducer. The brand used in this tunnel is Micro Switch Honeywell 163 PC (Figure 4.11). The transducer converts the pressure differences from the model and the reference static pressures, to differences in voltage.



**Figure 4.11: View of the pressure electronic transducer Honeywell 163PC. (Photograph by the Author)**

A sequential switch Scanivale 48 D9-1/2, which was driven by means of a CTRL2 / S2-S6 solenoid controller, connected the pressure taps to the transducer through PVC tubes of 1.5 mm internal diameter and 650 mm in length (Figure 4.12). The Scanivale system is the only working system available in this tunnel. It has some disadvantages in the sense that measurements cannot be taken simultaneously but should be obtained following a sequence. Each PVC tube connected to the Scanivale is connected to a pressure tap on the scale model. For this experiment, each pressure tap data was acquired once at a time following a numerical order.



**Figure 4.12: View of the Scanivale 48 D9-1/2 w/PVC tubes (Photograph by the Author).**

No resonance problems were detected for tubes of that length. Therefore, restrictors for the tubes were not used for filtering. When a fluctuating signal flows through the tubes in order to reach the Scanivale, verification that the signal is not amplified or with distortion is done through comparison with previous results. Those signals go to the DC transducer output and were read with the aid of a Keithley 2000 digital multimeter. The integration time operation rate of the A/D converter was set to produce mean values over 15 seconds of time integration (Figure 4.13). The multimeter was configured to read and store the values every 0.6 seconds, and every 25 readings and average are recorded every 15 seconds. This procedure can obtain variances in the order of  $\pm 0.02$  of the pressure coefficient.

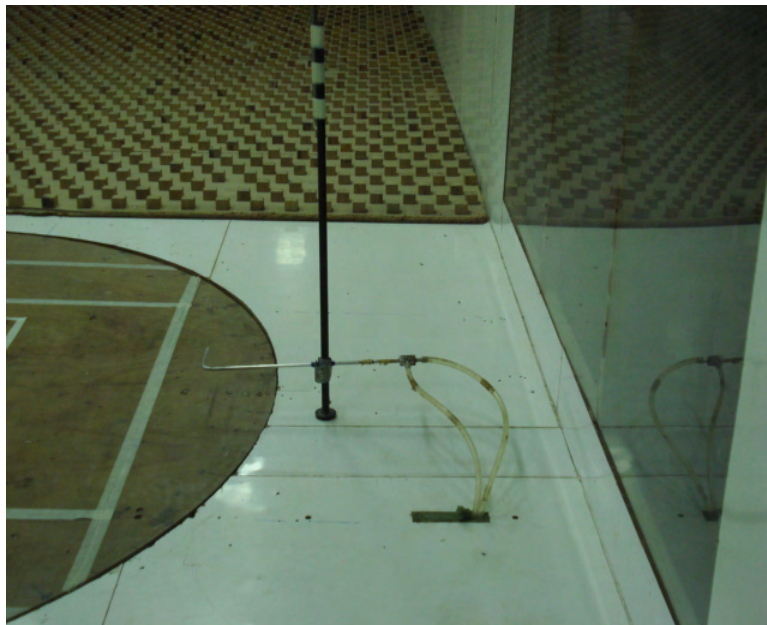


**Figure 4.13: View of the UNNE Keithley 2000 digital multimeter (Photograph by the Author).**

Simultaneously to the pressure measurements being taken on any of the roof surfaces, the reference dynamic pressure,  $q_{\text{ref}}$ , was measured at the eave's height with a Pitot-Prandtl tube connected to a Van Essen 2500 Betz differential micro-manometer of 1 Pa resolution (Figure 4.14). The probe stayed beside the model at a distance of about 0.70 m to avoid mutual interference. The reference static pressure was obtained from the static pressure taps of the same Pitot-Prandtl tube (Figure 4.15).

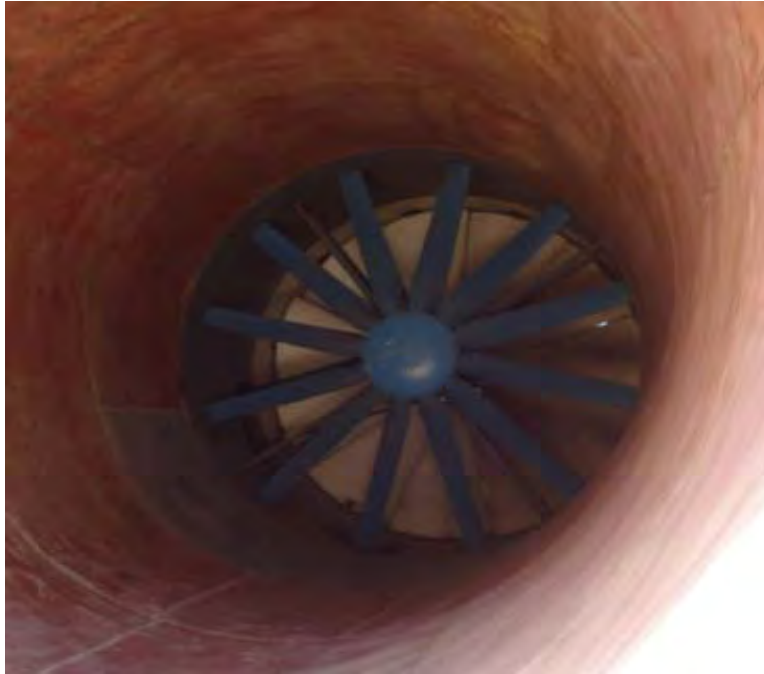


**Figure 4.14: View of the Van Essen 2500 Betz differential micro manometer. (Photograph by the Author)**



**Figure 4.15: View of the Pitot-Prandtl tube. (Photograph by the Author)**

The wind tunnel fan is an axial type, Aerofol model 88-J with a velocity of rotation of 720 rpm. The motor has a power of 100 kW generating a maximum wind velocity of 25 m/s (55.9 mph). Refer to Figure 4.16 for a photograph of the tunnel fan.



**Figure 4.16: View of the UNNE tunnel fan. (Photograph by the Author).**

## **4.5 Data processing**

In order to obtain pressure coefficients from the wind tunnel at the UNNE, the following procedure has been used. The first step is to calibrate the electronic transducer. Inducing pressure and suction on the differential micromanometer, readings from the multimeter on each of the induced pressures are obtained. Figure 4.17 shows the data obtained from the multimeter on the

model without parapets. The procedure is done at the beginning of the test and at the end to verify that there were no sudden pressure changes that can affect the measured data.

| CALIBRACION DE TRANSDUCTOR |                    |       |
|----------------------------|--------------------|-------|
| PRESION<br>[Pa]            | Lecturas [voltios] |       |
|                            | Inicio             | Fin   |
| 0                          | 3.509              | 3.510 |
| 20                         | 3.548              | 3.550 |
| 40                         | 3.588              | 3.590 |
| 60                         | 3.628              | 3.630 |
| -10                        | 3.488              | 3.490 |
| -20                        | 3.468              | 3.470 |
| -30                        | 3.449              | 3.451 |
| -40                        | 3.429              | 3.431 |
| -50                        | 3.409              | 3.410 |
|                            | 16:44              | 18:57 |

**Figure 4.17: Calibration data on model without parapets.**



**Figure 4.18: Photograph showing the author taking data from the wind tunnel test.**

With the data obtained for the calibration, and using a spreadsheet we can obtain the linear relation between the pressure differential and the voltage obtained. The manufacturer of the electronic transducer guarantees that there is a lineal relationship in order to obtain the slope and intercept in the form of:

$$P = a + b * Voltage$$

This processed calibration data is then used for the calculation of the pressure coefficients,  $C_p$  factors.  $C_n$  values are computed from the difference of  $C_p$  values. Figure 4.19 shows a sample of the processed transducer calibration data.

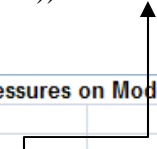
|                              |         |                         |                    |
|------------------------------|---------|-------------------------|--------------------|
| <b>Test February 25 2009</b> |         |                         |                    |
| Exterior $\alpha = 30$       |         | Exterior $\alpha = 210$ |                    |
| Exterior $\alpha = 330$      |         | Exterior $\alpha = 150$ |                    |
| Pressure[Pa]                 | Voltage | P=a+b.Volt.             |                    |
| 0                            | 3.509   | b                       | 501.8434           |
| 20                           | 3.548   | a                       | -1761.1235         |
| 40                           | 3.588   |                         |                    |
| 60                           | 3.628   | Test                    | -50.339333         |
| -10                          | 3.488   |                         |                    |
| -20                          | 3.468   |                         |                    |
| -30                          | 3.449   | Error Analysis          |                    |
| -40                          | 3.429   | a=                      | -1762.7048 1.5813  |
| -50                          | 3.409   | a=                      | -1760.1208 -1.0027 |
| 0                            | 3.51    | b=                      | 502.4224 -0.5789   |
| 20                           | 3.55    | b=                      | 501.4299 0.4135    |
| 40                           | 3.59    |                         |                    |
| 60                           | 3.63    | Least square method     |                    |
| -10                          | 3.49    | to fit data             |                    |
| -20                          | 3.47    |                         |                    |
| -30                          | 3.451   |                         |                    |
| -40                          | 3.431   |                         |                    |
| -50                          | 3.41    |                         |                    |

**Figure 4.19: Processed data for the calibration of the transducer.**

The pressure electronic transducer receives from the PVC tubes a pressure from the model and a pressure from the static pressure Pitot-Prandlt, which is used to measure  $q_{ref}$  or the

dynamic pressure of reference. This potential difference is measured with the digital multimeter. This signal is fluctuating due to the fluctuation of the wind. As mentioned before, the multimeter was configured to read and store the values every 0.6 seconds, and after 25 readings, an average is recorded (every 15 seconds). Figure 4.20 shows the processed data after the transducer was calibrated. The pressure difference, for example, is obtained using the previous equation:

$$P = a + b * Voltage = +(-1761.12 + 501.84 * (3.504)) = 2.664 Pa$$



| Exterior, interior and total pressures on Model 1 |                 |               |                |        |
|---|-----------------|---------------|----------------|--------|
| Without parapets                                  |                 |               |                |        |
| $\alpha = 30$                                     |                 |               |                |        |
| Exterior pressures                                |                 |               |                |        |
| Tap number  | Potential [ V ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $C_e$  |
| 1   | 3.504           | -2.664        | 108.7          | -0.025 |
| 2   | 3.476           | -16.716       | 108.6          | -0.154 |
| 3   | 3.477           | -16.214       | 108.5          | -0.149 |
| 4   | 3.463           | -23.240       | 108.4          | -0.214 |
| 5   | 3.458           | -25.749       | 108.3          | -0.238 |
| 6   | 3.505           | -2.162        | 108.2          | -0.020 |
| 7   | 3.512           | 1.351         | 108.1          | 0.012  |
| 8   | 3.507           | -1.159        | 108.0          | -0.011 |
| 9   | 3.494           | -7.683        | 107.9          | -0.071 |
| 10  | 3.516           | 3.358         | 107.8          | 0.031  |
| 11  | 3.514           | 2.354         | 107.7          | 0.022  |
| 12  | 3.487           | -11.196       | 107.6          | -0.104 |
| 13  | 3.516           | 3.358         | 107.5          | 0.031  |
| 14  | 3.492           | -8.686        | 107.4          | -0.081 |
| 15  | 3.517           | 3.860         | 107.3          | 0.036  |
| 16  | 3.515           | 2.856         | 107.2          | 0.027  |

**Figure 4.20: Processed data for the pressure coefficient  $C_p$ .**

The dynamic pressure of reference,  $q_{ref}$ , is measured with the differential micromanometer. The final pressure coefficient is the dynamic pressure/ $q_{ref}$ . The net pressure coefficient ( $C_n$ ) is the difference between the external pressure coefficient and the internal pressure coefficient. Appendix B through E shows the complete data sets, and the procedure with the calculated values on spreadsheets. The results from the wind tunnel tests will be presented and discussed in Chapter 6.

## **CHAPTER 5. AVAILABLE DESIGN CODE INFORMATION**

### **5.1 General Code Information**

Open structures have been designed and constructed in many countries for a number of years. In each case, the designer follows national standards based on best practices. However, little help is found with reference to the problem posed in this dissertation. For the sake of completeness, the design of open structures according to current building codes has been investigated during this research. Several design codes and standards are currently in use worldwide, such as the Uniform Building Code (1997 Edition), the ASCE 7-05 standard, the Eurocode, the Australian Code and the Japanese Code. Those were investigated to identify their current design criteria for the design of open structures. First, American codes (the UBC 97 and the ASCE 7-05) will be discussed in order to investigate current design criteria for the analysis of open canopy structures with parapets. Next, we will mention other design codes used worldwide, and whether they have available information in the design of open canopy structures with parapets.

### **5.2 UBC 97**

The Uniform Building Code, (1997 Edition) is a design code still in use in the West of the United States, and it is also the required building code in use in Puerto Rico. The UBC 97 is complemented in Puerto Rico with the use of the ASCE 7-05 for the calculation of wind pressures. The Uniform Building code covers the design of many types of structures. The open canopy structures will be included on what is called by the code, “unenclosed structures” as defined by Section 1616.

The basic wind speed is the fastest mile wind speed associated with an annual probability of occurrence of 0.02 measured at a point 33 feet above the ground for an exposure category C (UBC 1997), as defined by Section 1618. Design wind pressures are defined on Section 1620. Wind pressure for buildings and calculations are obtained in accordance to

$$P = C_e C_q q_s I_w \quad (\text{UBC 20-1})$$

where  $C_e$  = Gust factor coefficient, as given in UBC Table 16-G,  $C_q$  = Pressure coefficient, as given in UBC Table 16-H,  $I_w$  = Importance factor in UBC, Table 16-K,  $P$  = Design wind pressure,  $q_s$  = Wind stagnation pressure, as set forth in UBC Table 16-F.

The primary load resisting system of every structure shall be designed for the pressures using UBC formula 20-1. Two primary methods, Method 1 (normal force method) and Method 2 (projected area method), are the two methods commonly used for the calculation of wind pressures. No specific design procedure is available for the design of open structures with parapets in the Uniform Building Code, 1997 edition.

### **5.3 ASCE 7-05**

The American Society of Civil Engineers provisions, ASCE 7-05, provide three procedures for the calculation of wind loads. Method 1 or the simplified procedure is discussed in Section 6.4 of the ASCE document, Method 2, the analytical procedure is found in Section 6.5 of the document and Method 3, is the wind tunnel procedure which is described in Section 6.6.

The basic wind speed is the 3-second gust speed at 33 ft (10 m) above the ground in open terrain. The basic wind speeds are specified in the ASCE wind maps in their Figure 6-1. In

Puerto Rico the basic 3-second gust wind is 145 mph. The wind speeds are assumed to be the same for all horizontal directions.

Method 2, the analytical procedure on Section 6.5 describes the procedures for determining the wind pressures. Method 1 cannot be used for the design of open structures due to Section 6.4.1.1 item 3, which describes that it can only be used for enclosed structures.

Section 6.5.10 describes the formula for the calculation of the velocity pressure as follows:

$$q_z = .00256 K_z K_{zt} K_d V^2 I \quad (\text{ASCE 6-15})$$

where  $K_z$ =velocity pressure exposure coefficient, defined in Section 6.5.6.6,  $K_{zt}$ =topographic factor, defined in Section 6.5.7.2,  $K_d$ =wind directionality factor, defined in Section 6.5.4.4,  $V$  = velocity from Figure 6-1,  $I$  = Importance factor for the building or other structure which is determined from Table 6-1.

Regarding exposure, for each wind direction being considered, the upwind exposure category shall be based on ground surface roughness, which is determined from the natural topography, vegetation or surrounding constructed facilities (ASCE 7-05). Surface roughness categories are defined in Section 6.5.6.2. Surface roughness C is used for the wind tunnel simulation and CFD model simulation. Gust effect factor is determined on section 6.5.8. For rigid structures, one should use Section 6.5.8.1 and for flexible or dynamically sensitive structures, Section 6.5.8.2.

Enclosure classifications are defined in Section 6.5.9 in such a way that all buildings shall be classified as enclosed, partially enclosed or open. Section 6.5.13 covers the design of wind loads for open buildings with monoslope, pitched or troughed roofs. Parapets shall be designed as explained in Section 6.5.12.2.4. The ASCE 7-05 does not provide coefficients for structures

that combine both, open canopy and parapets in the same structure. No previous studies have been done evaluating the proposed code coefficients.

## **5.4 IBC 2006**

The most recent design code in the United States is the International Building Code, 2006 Edition. This code is used in the entire nation, and almost all the states are adopting it, or are in the process of incorporating the code to their standards. The IBC 2006 makes reference to the ASCE 7-05 for wind design procedures. It uses all the standards, formulas and coefficients of the ASCE 7, which were briefly described in the previous section.

## **5.5 Additional Building Codes**

Building codes in different parts of the world were investigated, to verify if there was available information for the design of open canopy structures. The Eurocode ( CEN TC 250, date: 2002-06, prEN 1991-1-4.6, Eurocode 1: Actions on structures - Part 1-4: General actions – Wind actions), which is used in Europe and many Caribbean islands, does not include any information for the design of open canopy structures with parapets.

The Australian Code, Standards Association of Australia, AS 1170.2-1989, Minimum design loads on structures (known as the SA Loading Code) - Wind loads, 1989, does not include any information for the design of open canopy structures with parapets.

Additional codes include the Japanese Code, which is composed of the Building Standard Law, and the Building Standard Law Enforcement Order. The Building Standard Law, in which objectives of the code, definitions of terms, fundamental concepts are described, was revised in 1998, leading to the performance-based regulation. Following the revision, technical regulations

such as load provisions, and structural calculation provisions, were modified in 2000. They are in the Building Standard Law Enforcement Orders (cabinet orders), the Building Standard Law Enforcement Regulations (Ministry of Land, Infrastructure and Transportation). The former is Ministry of Construction and Notifications of the Ministry of Construction. The revision of the wind load provisions was very drastic, but it was based on the Architectural Institute of Japan (AIJ) recommendations, and it does not include any information for the design of open canopy structures with parapets.

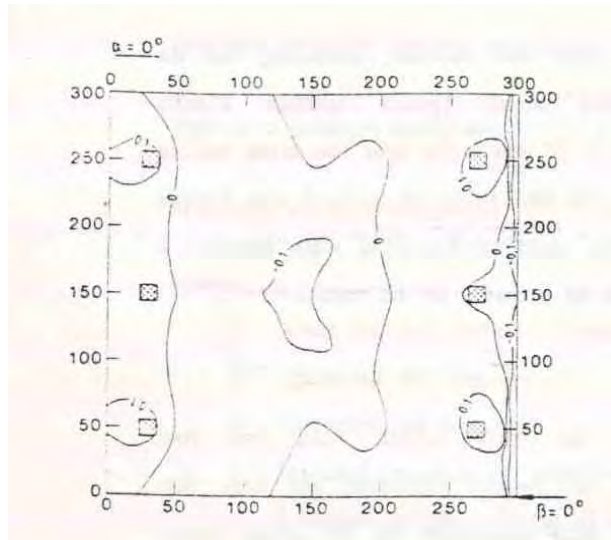
The last two codes explored, the Brazilian Code (Associação Brasileira de Normas Técnicas published “Forças devidas ao vento em edificações NBR 6123”, Jun/1988), and the Canadian Code (NRC-CNRC National Building Code of Canada, Canadian Commission on Building and Fire Codes, 1995), do not provide any information for the design of open canopy structures with parapets.

## **CHAPTER 6. RESULT OF WIND PRESSURES IN CANOPIES**

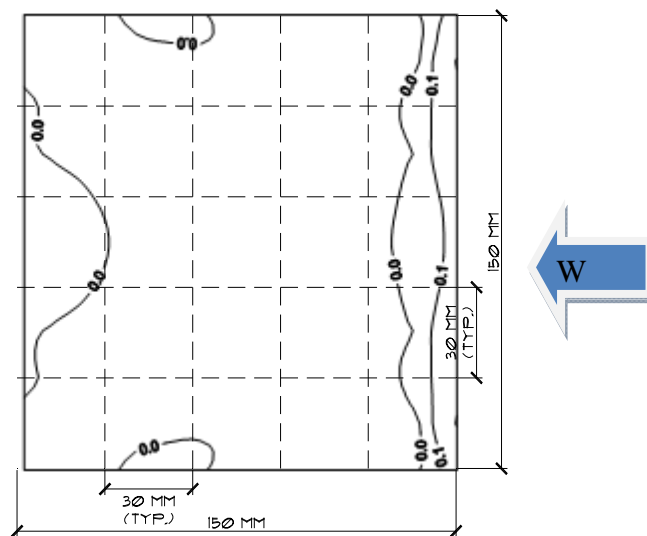
### **6.1 Comparison between wind tunnel results and previous work**

First, wind tunnel experiments were carried out at UNNE on a canopy without parapets, in order to compare the measurement with other wind tunnels results already available in the open literature. This provides the opportunity to check the accuracy of the tunnel boundary layer and the instrumentation available at UNNE. Wind load coefficients were measured under wind blowing at angles of  $0^\circ$  and  $30^\circ$  relative to one of the symmetry axis, since, as demonstrated by Gumley (1981, 1982) and Letchford et al. (1992), among others, these wind directions produce the most severe loads on planar canopy roofs with no parapets.

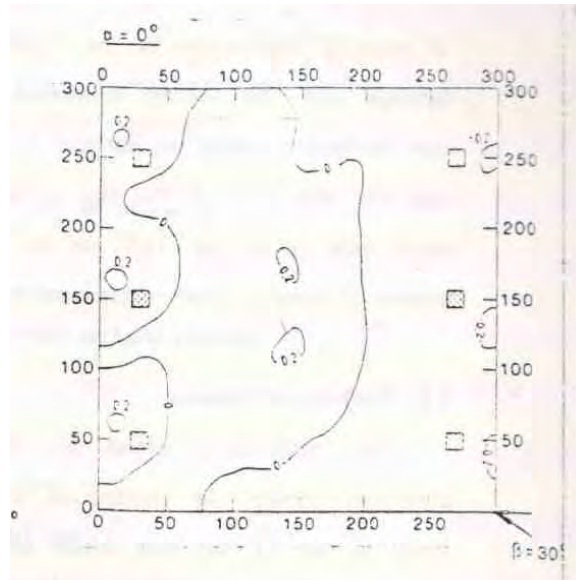
Results are next compared against previous work of Ginger and Letchford (1994). It can be seen from Figure 6.1 to Figure 6.4 that both sets of contour lines are very close in both qualitative and quantitative terms. Note that in the present study (Figure 6.2 and Figure 6.4) the interference of the columns has been eliminated in order to prevent the interference caused by the column sizes.



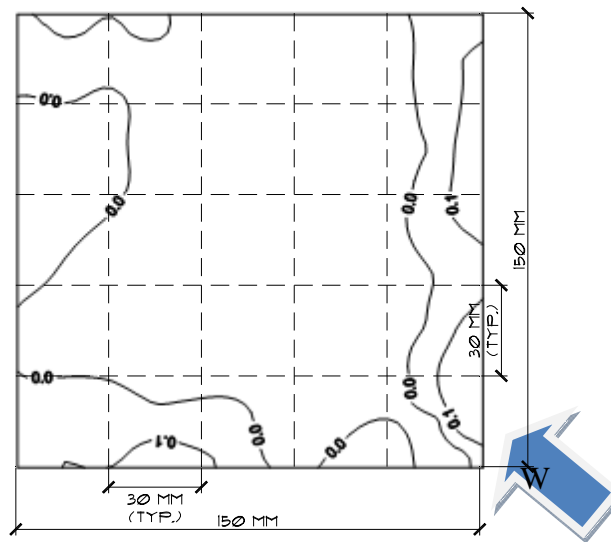
**Figure 6.1: No parapets, net pressures, wind direction of  $0^\circ$  from Ginger et al. (1994)**



**Figure 6.2: No parapets, wind tunnel net pressures, wind direction of  $0^\circ$ , present research.**



**Figure 6.3: No parapets, wind tunnel net pressures, wind direction of 30° (Ginger et al. 1994)**



**Figure 6.4: No parapets, wind tunnel net pressures, wind direction of 30°, present research.**

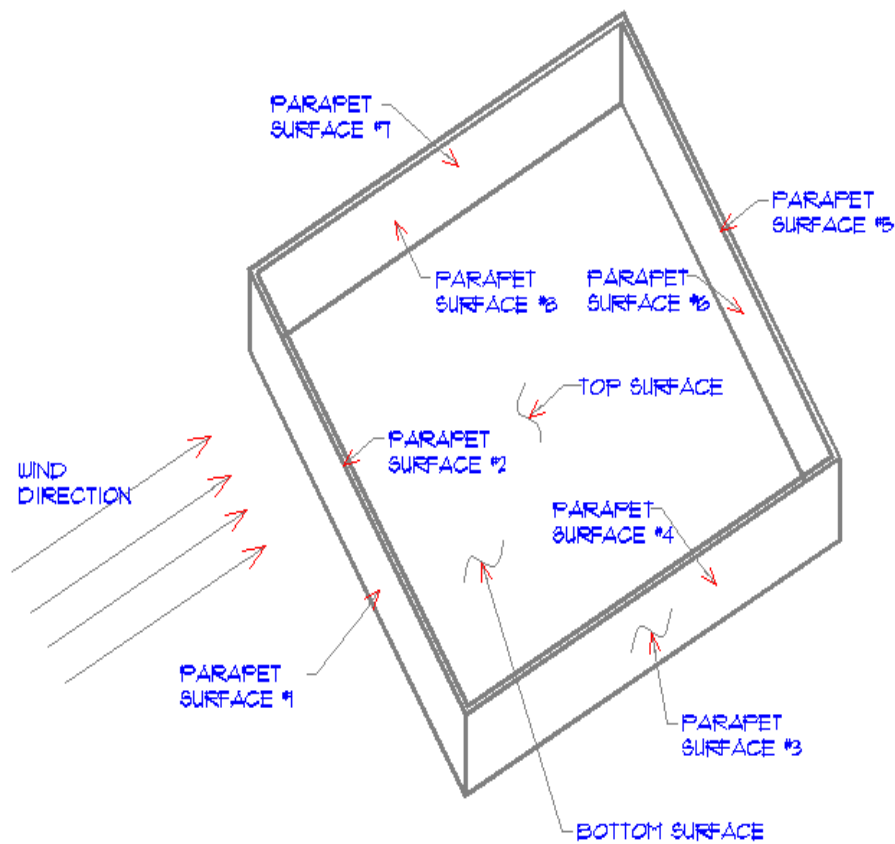
Data from the wind tunnel (and also from computational models) were recorded on the top and bottom surfaces at 64 pressure points, from which  $C_p$  values were obtained for each surface. Two wind conditions were considered in this experimental part of the research: one with

an incident wind oriented along the long axis direction ( $0^\circ$ ), as shown in Figure 6.1 and 6.2 whereas a second case was considered by rotating the turntable to  $30^\circ$  with respect to the structural axis (see Figure 6.3 and 6.4). Blue arrows in the figures indicate the wind direction in the present study.

## **6.2 CFD and wind tunnel results for top surface, wind at 0 degrees**

Data from the wind tunnel and computational models were obtained on the top and bottom surfaces at 64 pressure points leading to  $C_p$  values for each surface. Refer to Figure 4.9 for pressure tap distribution. Two wind conditions were considered: one with incident wind oriented in the long axis direction ( $0^\circ$ ), whereas a second case was considered by rotating the wind to  $30^\circ$  with respect to the structural axis. Refer to Figure 3.13 and Figure 3.14 for wind direction representation on EFD.Lab.

Figure 6.5 shows the designations of the canopy surfaces used in this chapter, including the top, bottom and all the parapet surfaces. All exterior parapet surfaces are identified by an odd number, whereas all interior parapet surfaces are identified by even numbers. The wind direction at 0 degrees is also represented. For the CFD computations and the wind tunnel tests, only one configuration has been tested, the canopy with dimensions 7.6 m (25ft) x 7.6 m (25 ft) and parapet of 1.22 m (4 ft)



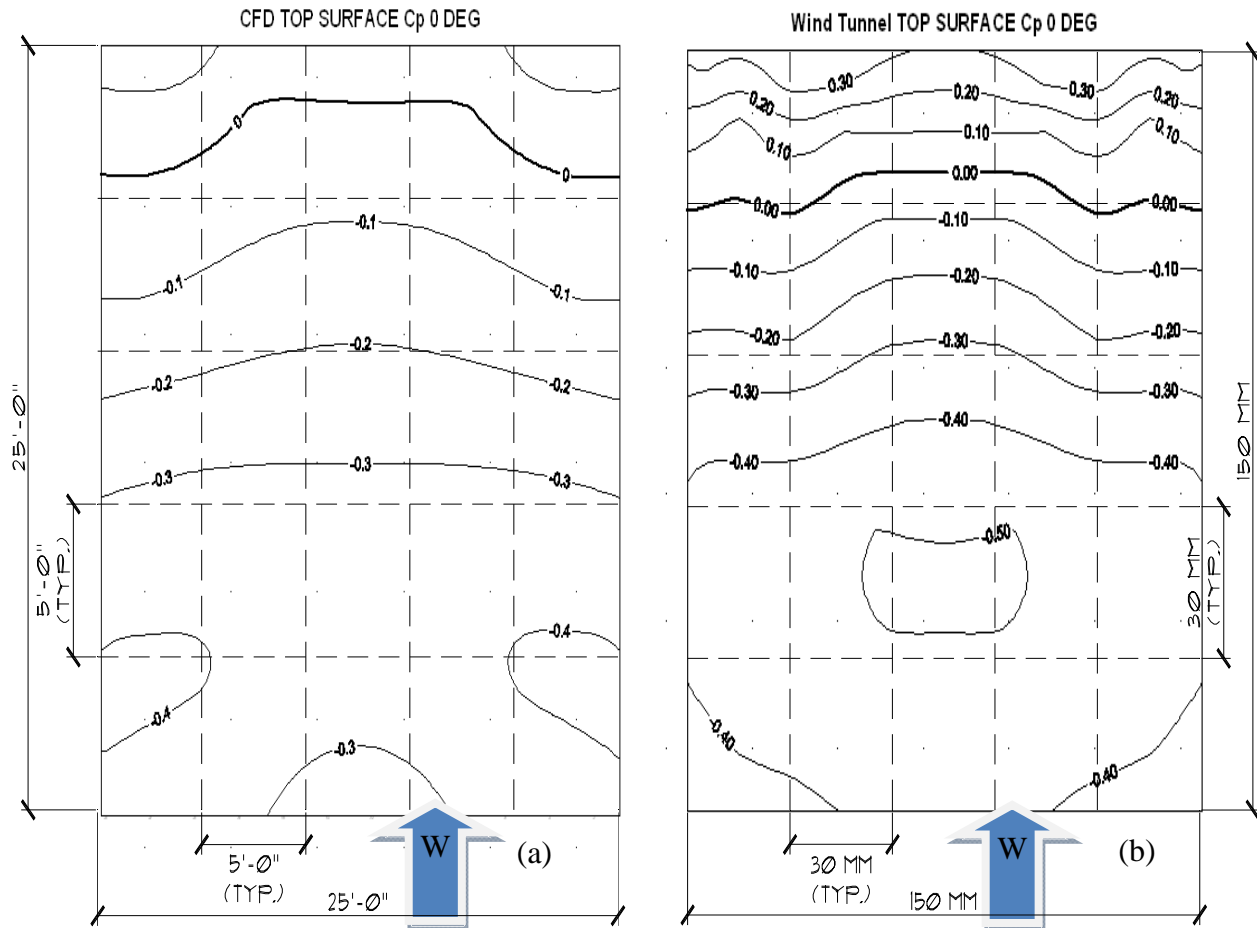
**Figure 6.5: Canopy surface nomenclature.**

Figure 6.6a shows the  $C_p$  values for the top surface of the open canopy, from the CFD simulations  $C_p$  results for wind at  $0^\circ$ . A blue arrow in the figures indicates the wind direction on both models. In the CFD simulation, the top surface shows an uplift almost until the end of the surface where a downward pressure is clearly shown at the end of the current figure.

Figure 6.6b shows  $C_p$  results obtained from the wind tunnel for the top surface of the canopy. Again, the wind tunnel measurements on the top surface show uplift almost until close to the end of the surface where a downward pressure is noticed. Thus, approximately 1/6 of the canopy is under positive pressures and the rest is under suction. Some changes in  $C_p$  values are

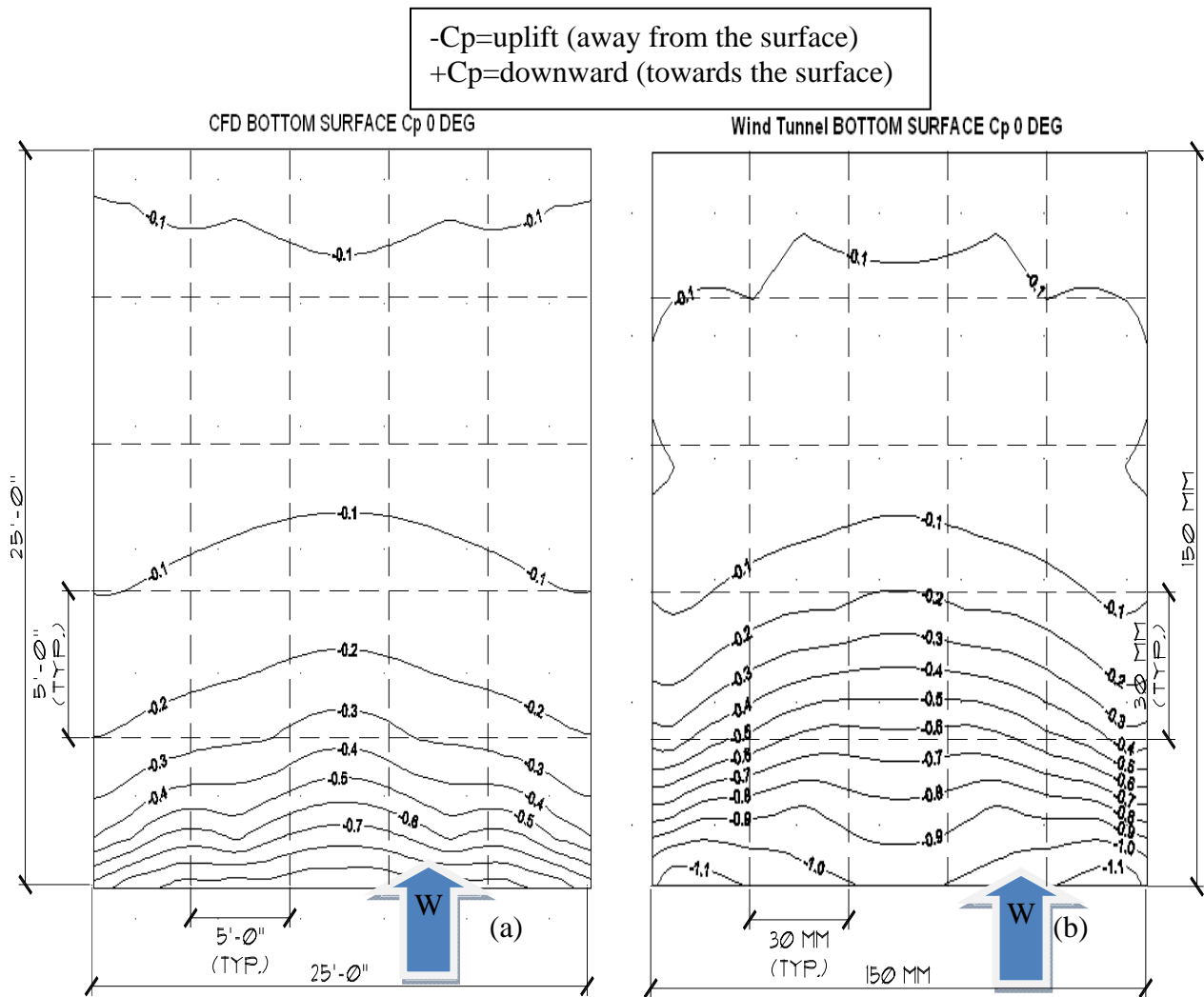
noticed in the transverse direction, appearing from boundary effects on the sides of the canopy, generally, the pattern of contours follows fairly regular lines parallel to each other.

-Cp=uplift (away from the surface)  
+Cp=downward (towards the surface)



**Figure 6.6:  $C_p$  results from top surface from CFD (a) and wind tunnel (b) model for incident wind at 0 degrees from the structural axis.**

### 6.3 CFD and wind tunnel results for bottom surface, wind at 0 degrees



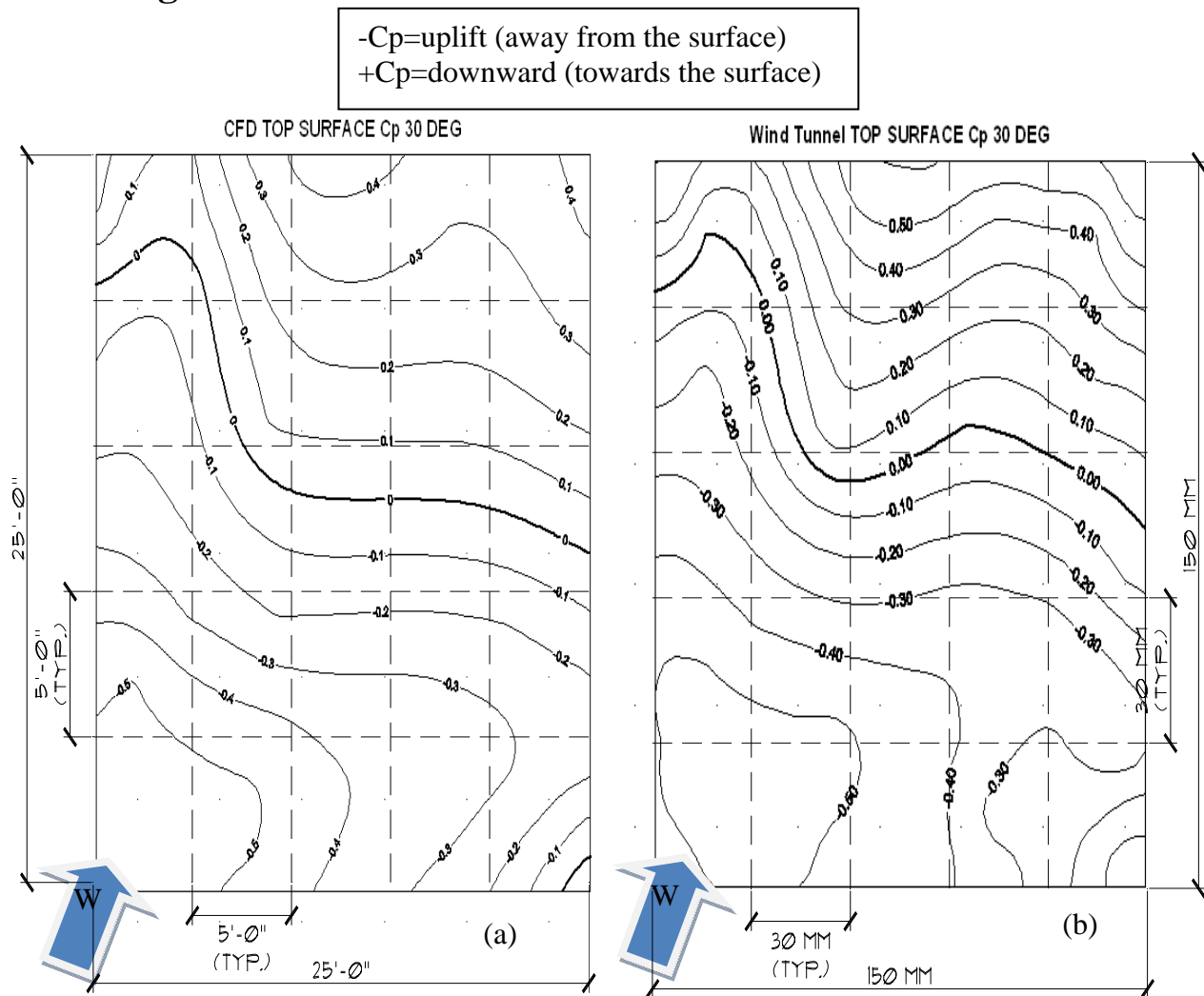
**Figure 6.7: Cp results from bottom surface CFD (a) and wind tunnel (b) model for incident wind at 0 degrees from the structural axis.**

Figure 6.7a show the Cp values for the bottom surface of the canopy using CFD, for the wind at 0°. The bottom surface shows on both, the wind tunnel and the CFD simulation, uniform downward pressure through the entire bottom surface.

The Cp values for the bottom surface of the canopy at the wind tunnel for the wind at 0° are shown in Figure 6.7b. The wind tunnel results have an almost uniform downward pressure gradient affecting the entire bottom surface. The comparisons indicate that Cp values derived

from the CFD and the wind tunnel tests are in good agreement. A t-sample statistical analysis, where the difference of the two set of data is compared, indicate a difference of less than 5 percent through the present research for the  $C_p$  and  $C_n$  values for top and bottom surfaces. Refer to Appendix G for an example of the statistical analysis procedure.

## 6.4 CFD and wind tunnel results for top surface, wind at 30 degrees



**Figure 6.8:  $C_p$  results from top surface CFD (a) and wind tunnel (b) model for incident wind at 30 degrees from the structural axis.**

Figure 6.8a shows the  $C_p$  values for the top surface of the canopy, obtained with CFD, for the wind at  $30^\circ$ . The CFD simulation on the top surface shows an uplift for the first half of the top surface (windward region) and a downward pressure on the other half (leeward).

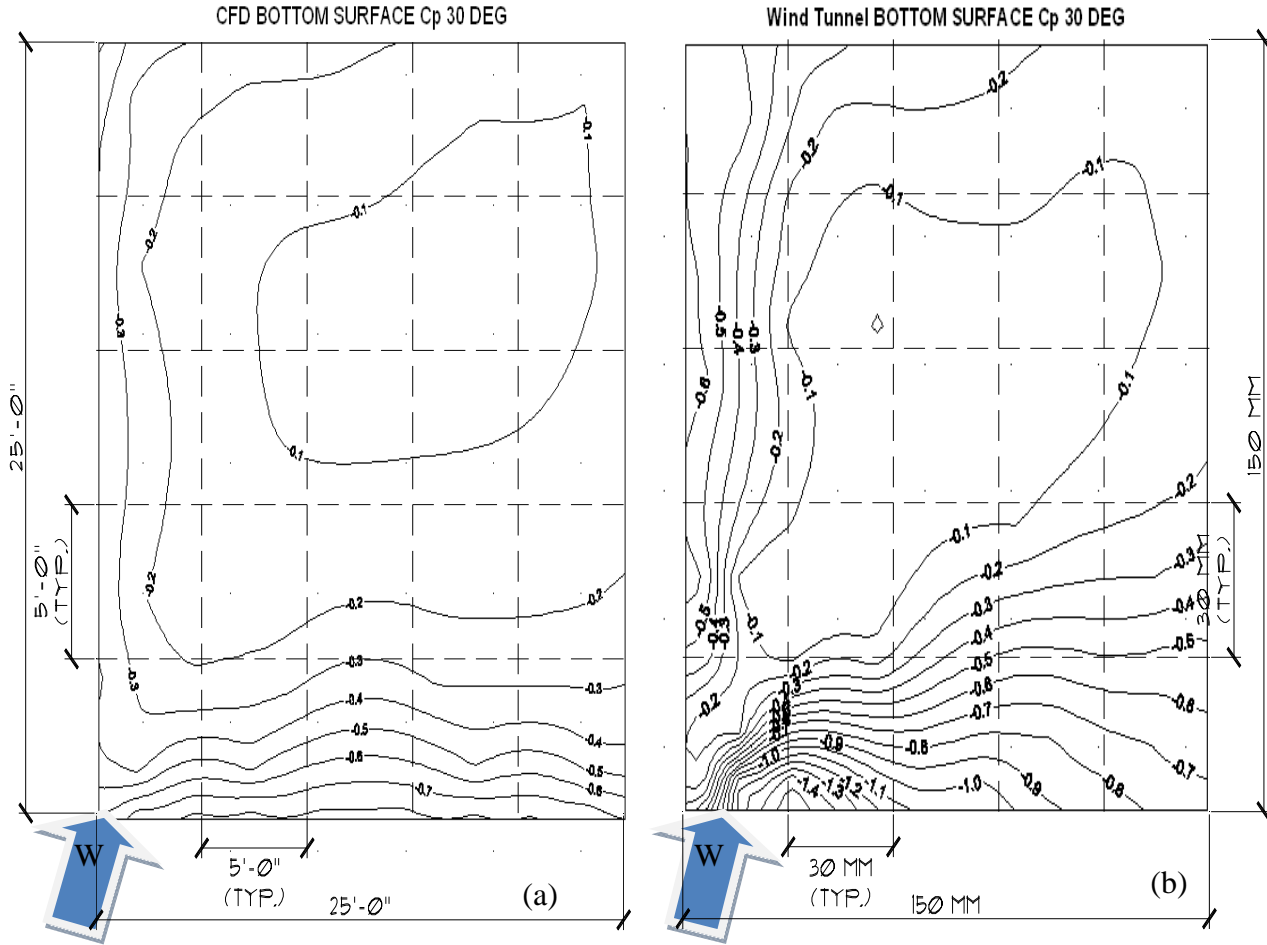
Next, a configuration at  $30^\circ$  was investigated by rotating the turntable in the wind tunnel. Figure 6.8b shows the  $C_p$  values for the top surface of the canopy from the wind tunnel for the wind at  $30^\circ$ , indicating that there is an uplift on the first half (windward region) of the top surface and a downward pressure on the other half (leeward). This effect is clearly shown on both of the previous figures (Figure 6.8a and 6.8b).

## **6.5 CFD and wind tunnel results for bottom surface, wind at 30 degrees**

Figure 6.9a shows the CFD  $C_p$  values for the bottom surface of the canopy for the wind at  $30^\circ$ . The bottom surface shows uniform negative  $C_p$  values, meaning that there is an uplift pressure on the surface, i.e. downward pressure (suction) through the entire bottom surface.

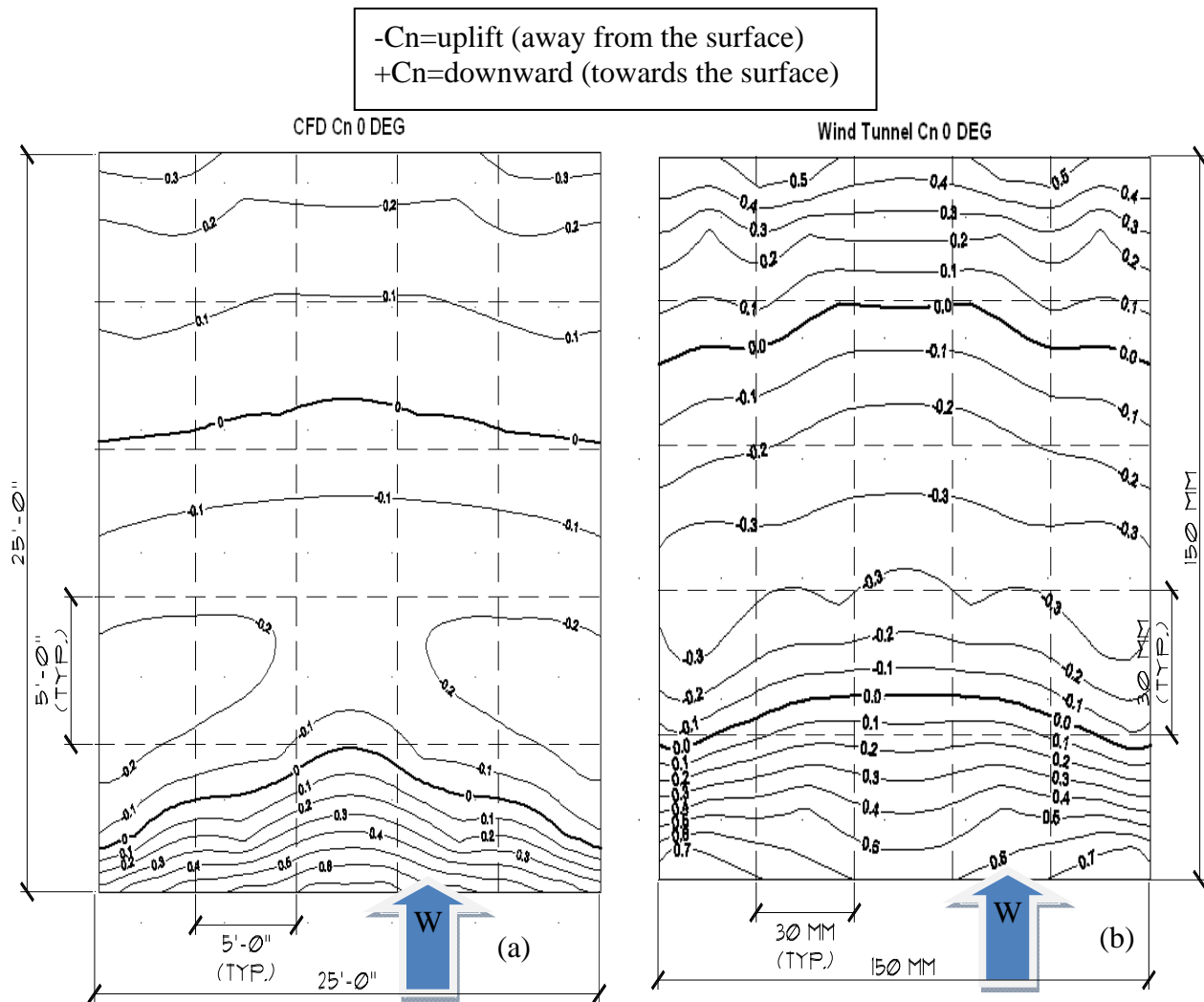
Figure 6.9b shows the  $C_p$  values for the bottom surface of the canopy for the wind tunnel for the wind at  $30^\circ$  incident angle. The bottom surface shows on the wind tunnel and uniform downward pressure through the entire bottom surface. Both figures are in good agreement regarding the values obtained.

-Cp=uplift (away from the surface)  
+Cp=downward (towards the surface)



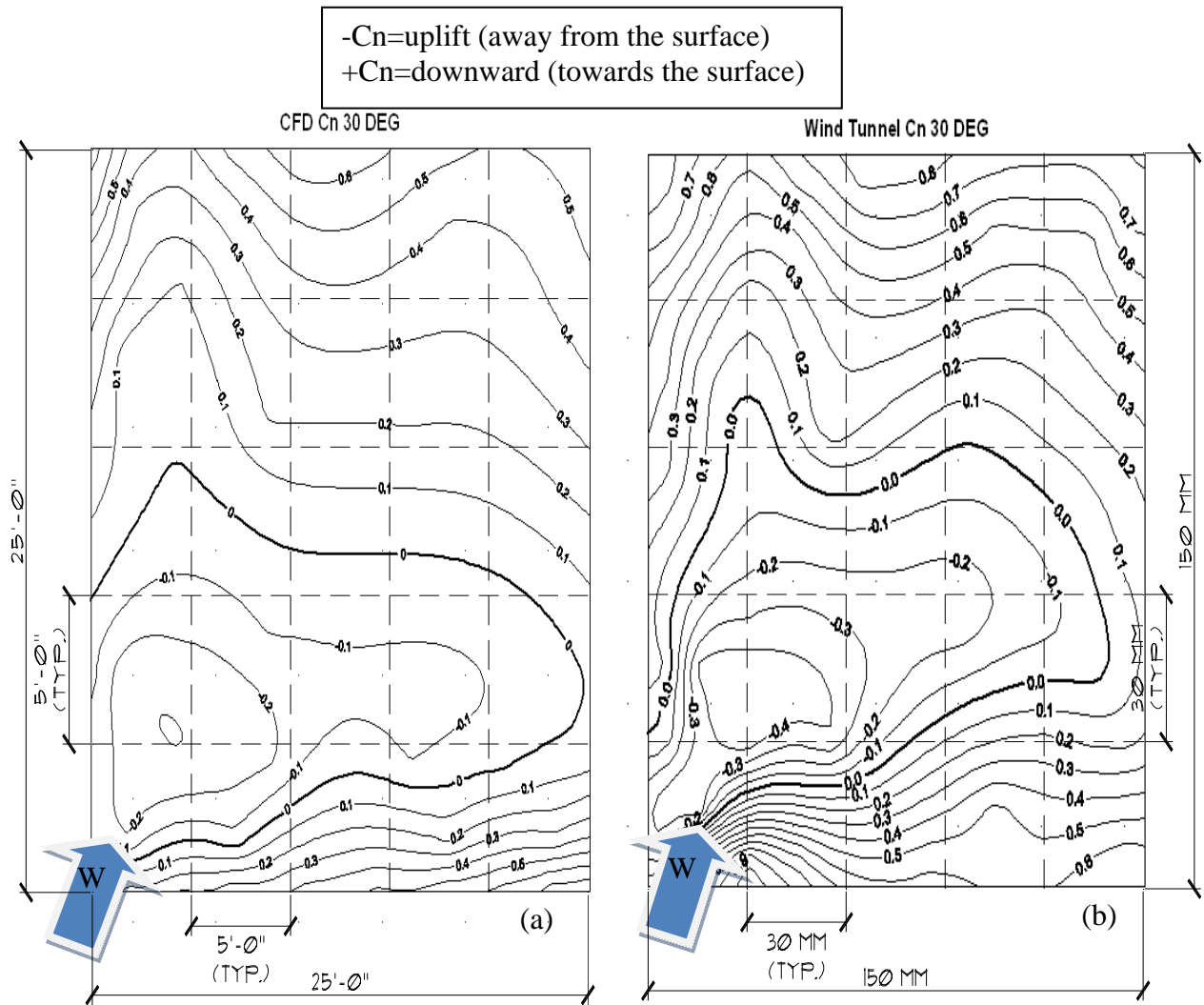
**Figure 6.9: Cp results from bottom surface CFD (a) and wind tunnel (b) model for incident wind at 30 degrees from the structural axis.**

Net values, denoted by  $C_n$ , were calculated as the difference between the  $C_p$  values obtained at the same location and on the top and bottom surfaces. Figure 6.10a shows the  $C_n$  results for the wind at  $0^\circ$  for the CFD simulation, and Figure 6.10b was drawn using the wind tunnel results. Both figures show an initial downward pressure through the first 1/3 of the length, and uplift for the next 1/3 of the length, and a downward pressure for the last 1/3 of the surface.



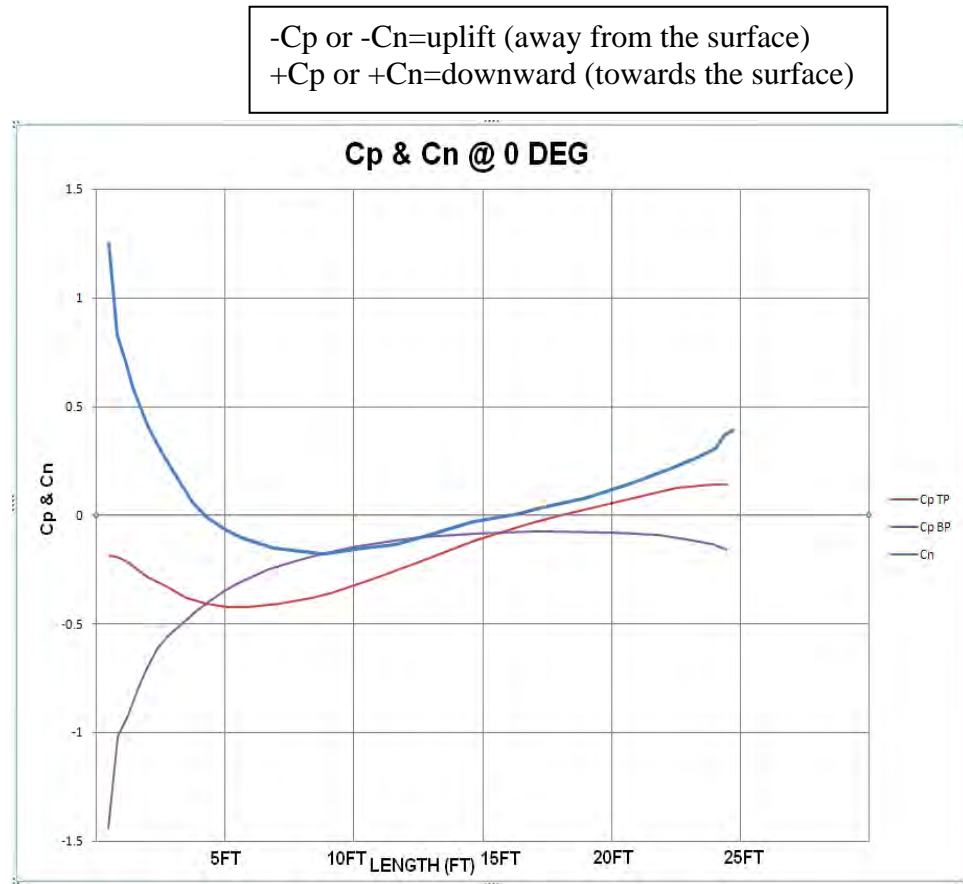
**Figure 6.10: Cn results from CFD (a) and wind tunnel (b) model for incident wind at 0 degrees from the structural axis.**

Cn results for the wind at 30° are shown on Figure 6.11a and Figure 6.11b. The net pressures in this case show an uplift area located in the center of the surface, with a downward pressure zone around that uplift center area. The wind in both studies, the CFD and the wind tunnel, are in good agreement in the location of the areas where uplift and downward pressure are located.



**Figure 6.11: Cn results from CFD (a) and wind tunnel model (b) for 30 degrees from the structural axis.**

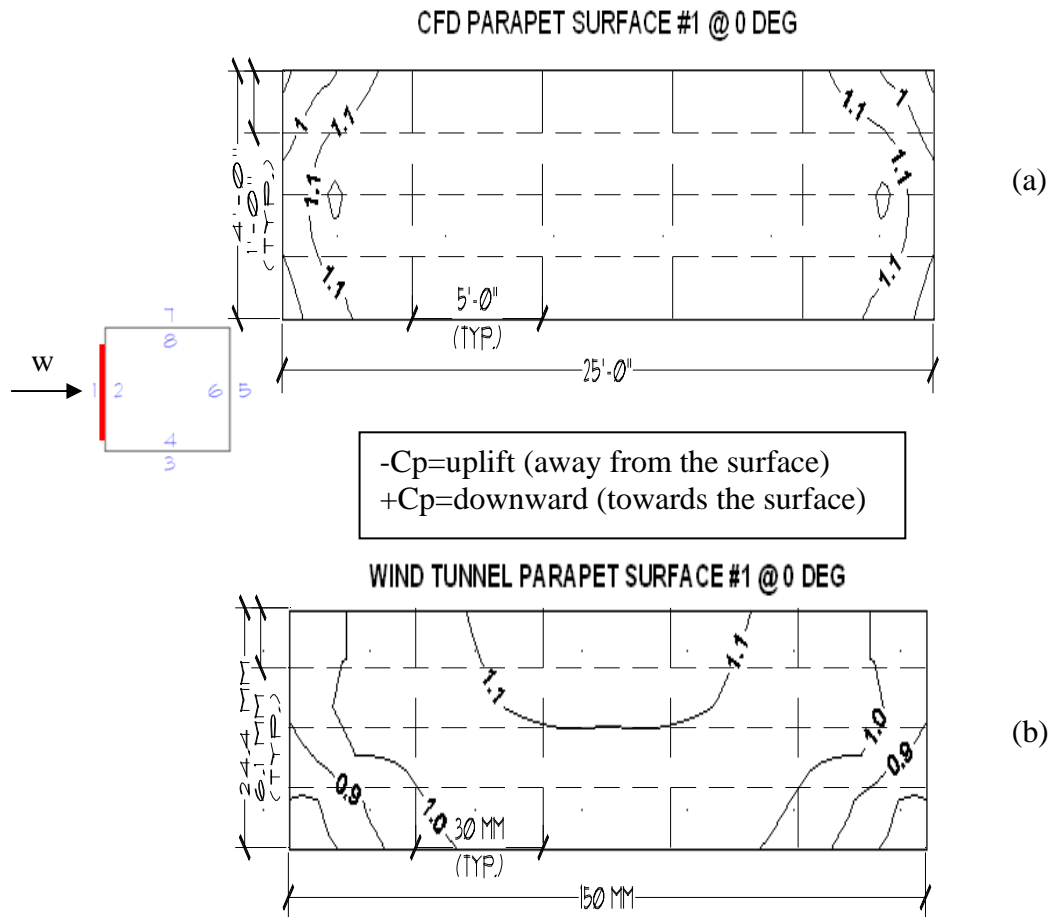
The extreme values for  $C_p$  and  $C_n$  along the wind axis are shown in Figure 6.12. Values of  $C_p$  TP ( $C_p$  at the top surface),  $C_p$  BP ( $C_p$  at the bottom surface), and  $C_n$  for the canopy with a wind at  $0^\circ$  are plotted. The  $C_n$  values show a  $C_p$  of +1.3 maximum behind the parapet. The  $C_n$  values changes along the center of the roof to a  $C_n$  of -0.5, whereas at the end of the surface it changes to a  $C_n$  of +0.5. The  $C_n$  values change from + (positive=downward) to - (negative=uplift) and to + (positive=downward)  $C_n$  values along the roof geometry.



**Figure 6.12: Cp and Cn results for CFD model at 0 degrees for (25x25x4 ft) canopy.**

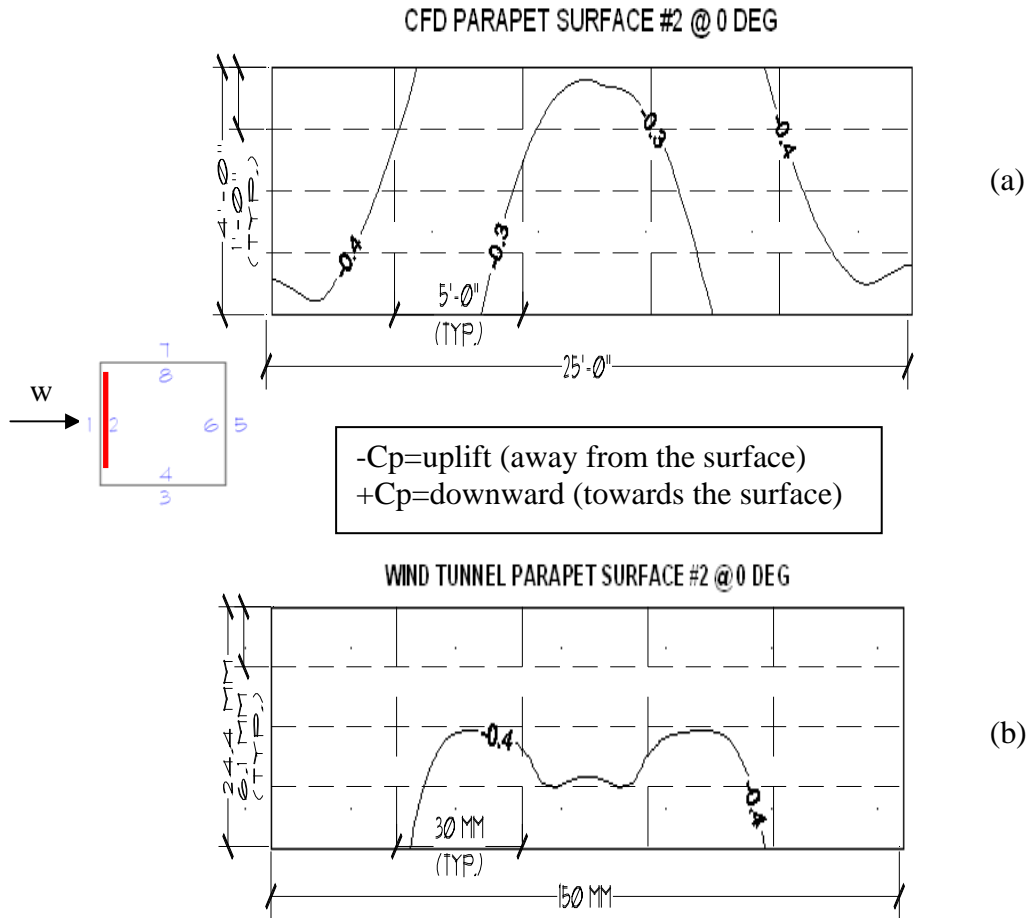
## 6.6 CFD and wind tunnel results for parapets, wind at 0 degrees

Cp values for the parapet surface #1 at 0° for CFD and wind tunnel are shown in Figure 6.13a and Figure 6.13b. Positive Cp values indicating pressure downward on the surface. Values vary from +0.9 to +1.1.



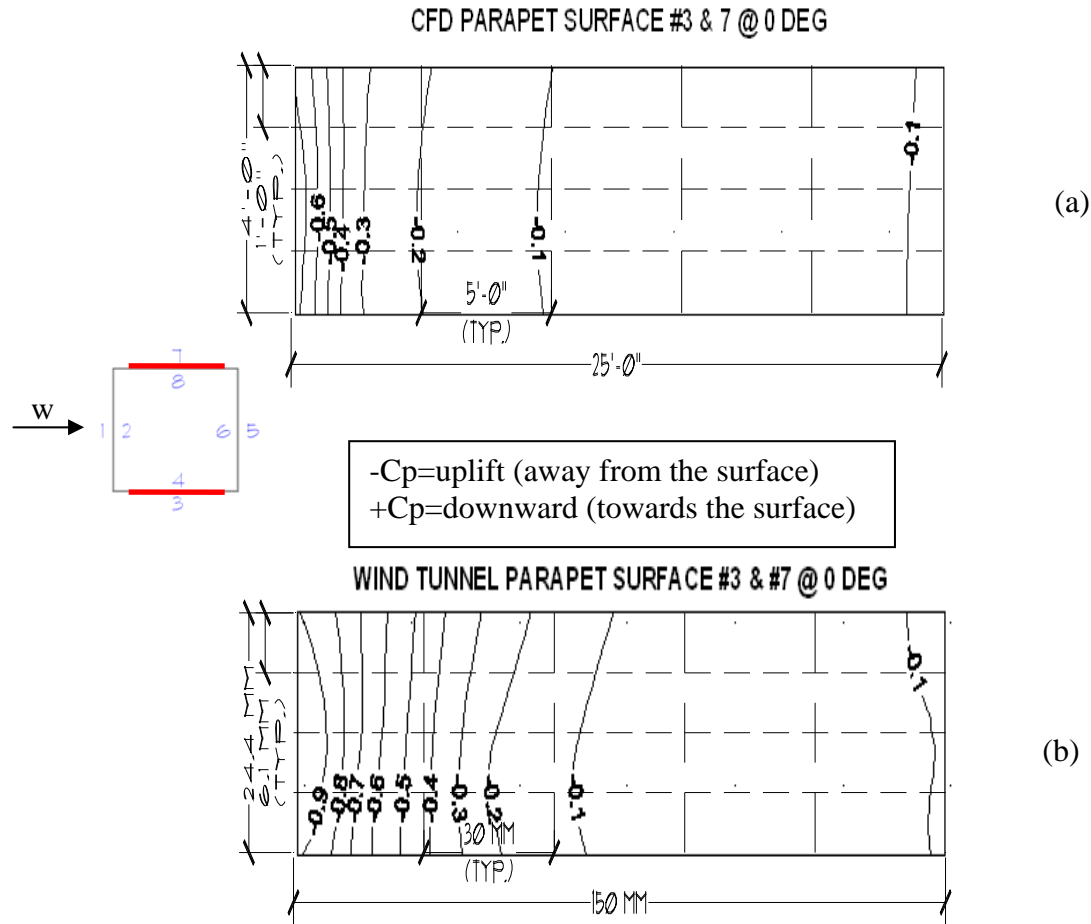
**Figure 6.13:  $C_p$  results at parapet surface #1 at 0 degrees from CFD (a) and wind tunnel (b).**

$C_p$  values for the parapet surface #2 at  $0^\circ$  for CFD and wind tunnel are shown in Figure 6.14a and Figure 6.14b. Negative  $C_p$  values indicating uplift pressure away from the surface. Values vary from -0.3 to -0.4.



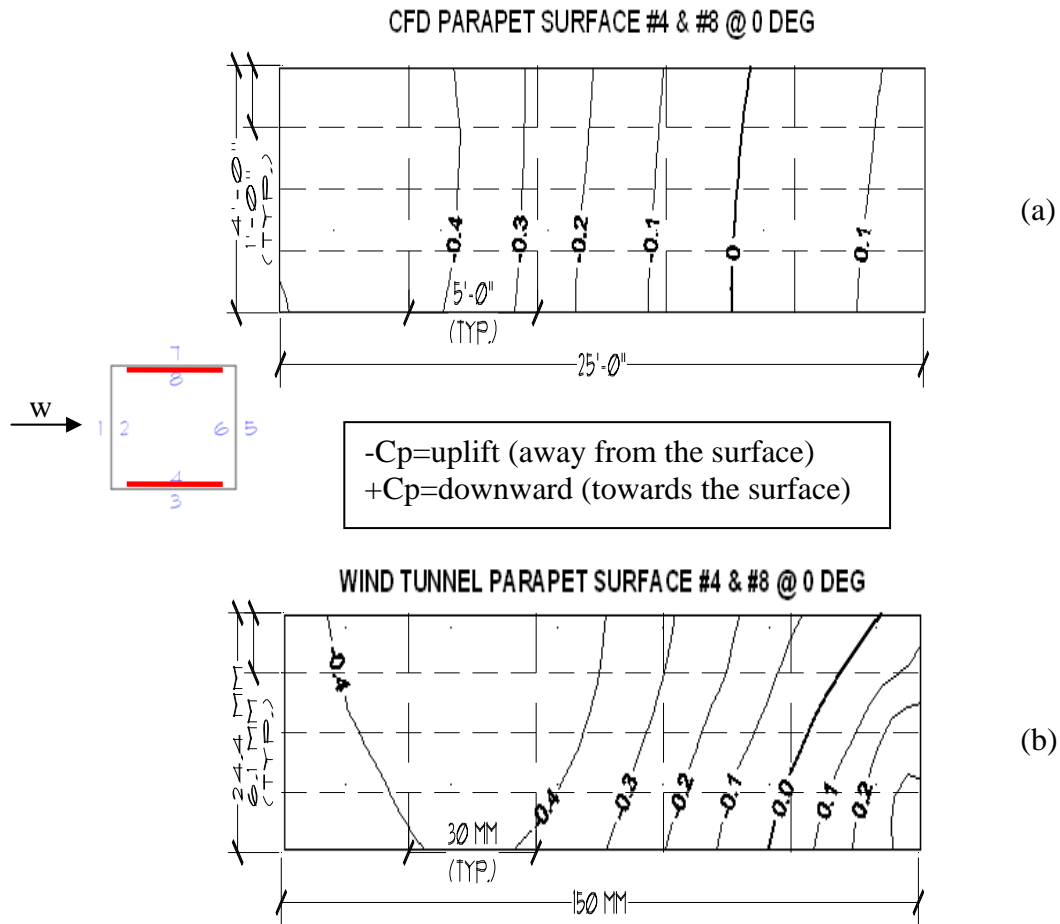
**Figure 6.14: Cp results at parapet surface #2 at 0 degrees for CFD (a) and wind tunnel (b).**

Cp values for the parapet surface #3 and parapet surface #7 at 0° for CFD and wind tunnel are shown in Figure 6.15a and Figure 6.15b. Negative Cp values indicating uplift pressure away from the surface. Values vary from -0.9 to -0.1.



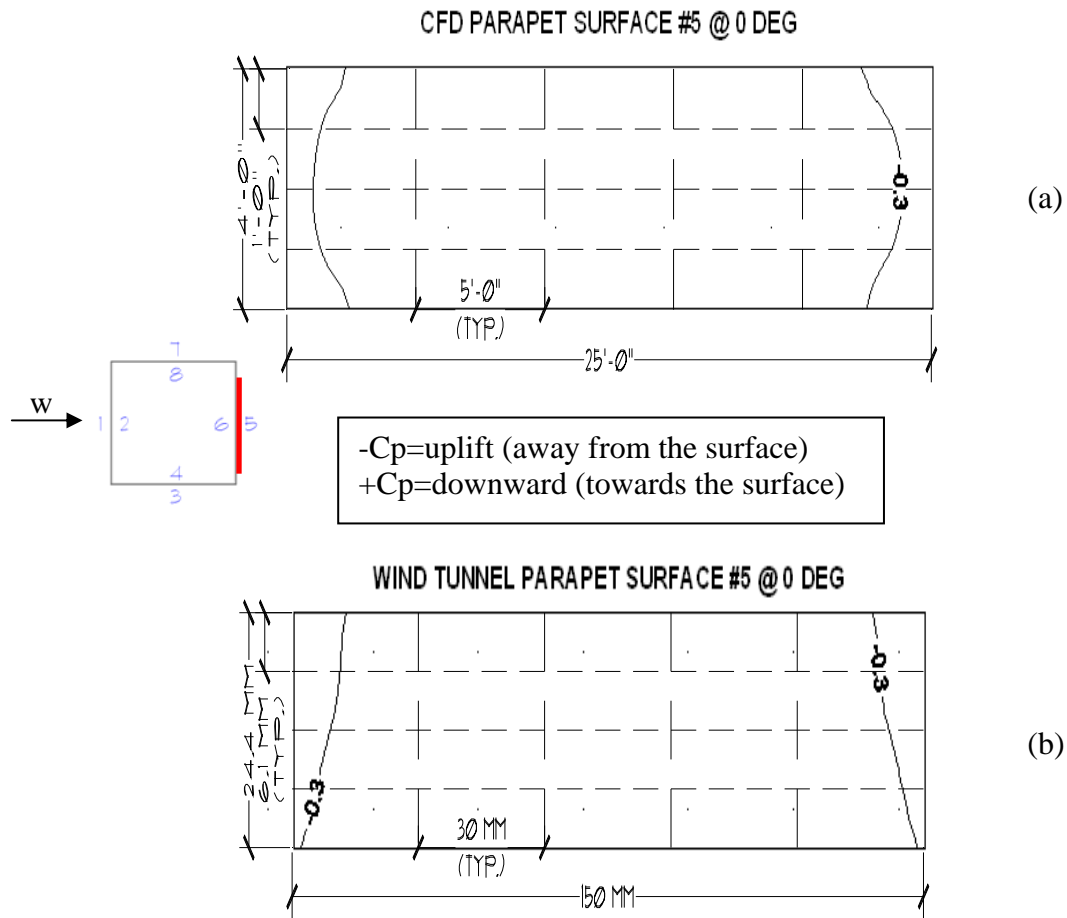
**Figure 6.15:  $C_p$  results at parapet surface #3 and #7 at 0 degrees for CFD (a) and wind tunnel (b).**

$C_p$  values for the parapet surface #4 and parapet surface #8 at  $0^\circ$  for CFD and wind tunnel are shown in Figure 6.16a and Figure 6.16b. Negative  $C_p$  values indicating uplift, or pressure away from the surface. Positive  $C_p$  values indicating pressure towards on the surface. There is a transition from negative to positive values at  $2/3$  of the horizontal surface. Values vary from -0.4 to +0.2.



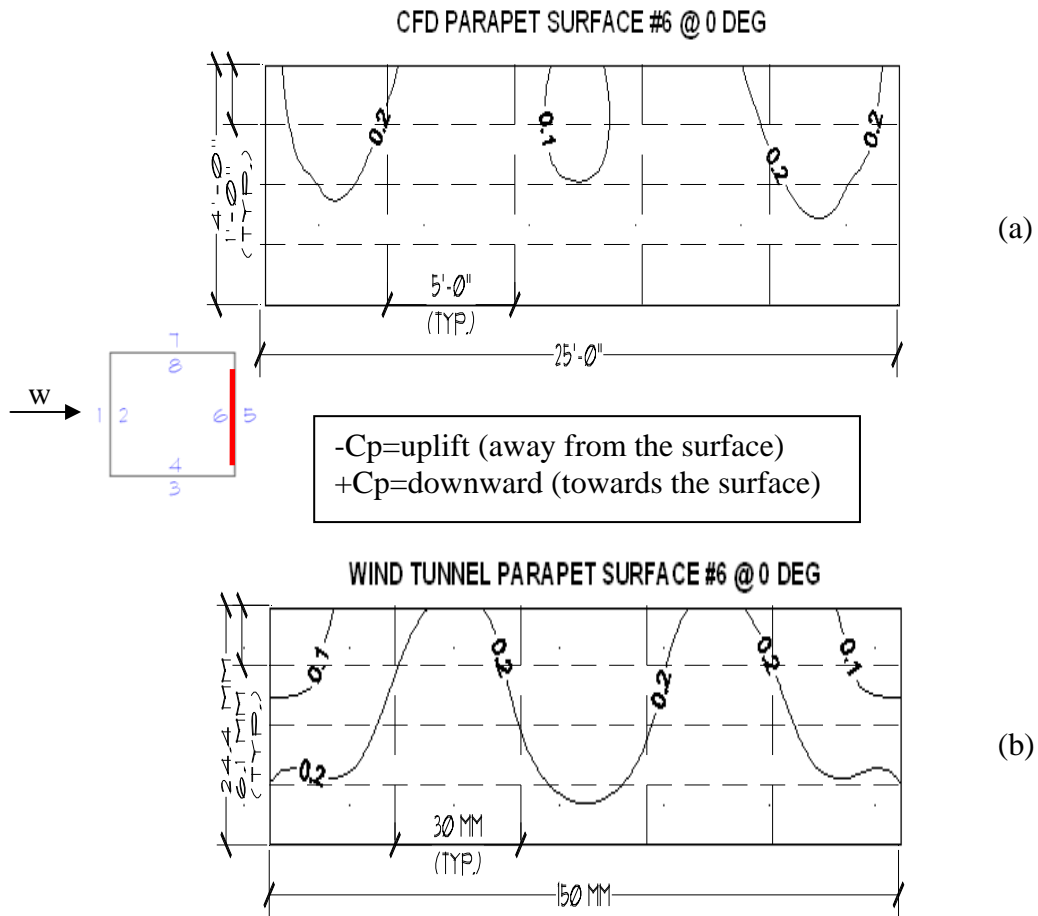
**Figure 6.16:  $C_p$  results at parapet surface #4 and #8 at 0 degrees for CFD (a) and wind tunnel (b).**

$C_p$  values for the parapet surface #5 at  $0^\circ$  for CFD and wind tunnel are shown in Figure 6.17a and Figure 6.17b. Negative  $C_p$  values indicating uplift, pressure away from the surface are shown on both figures. Values are -0.3 on both simulations.



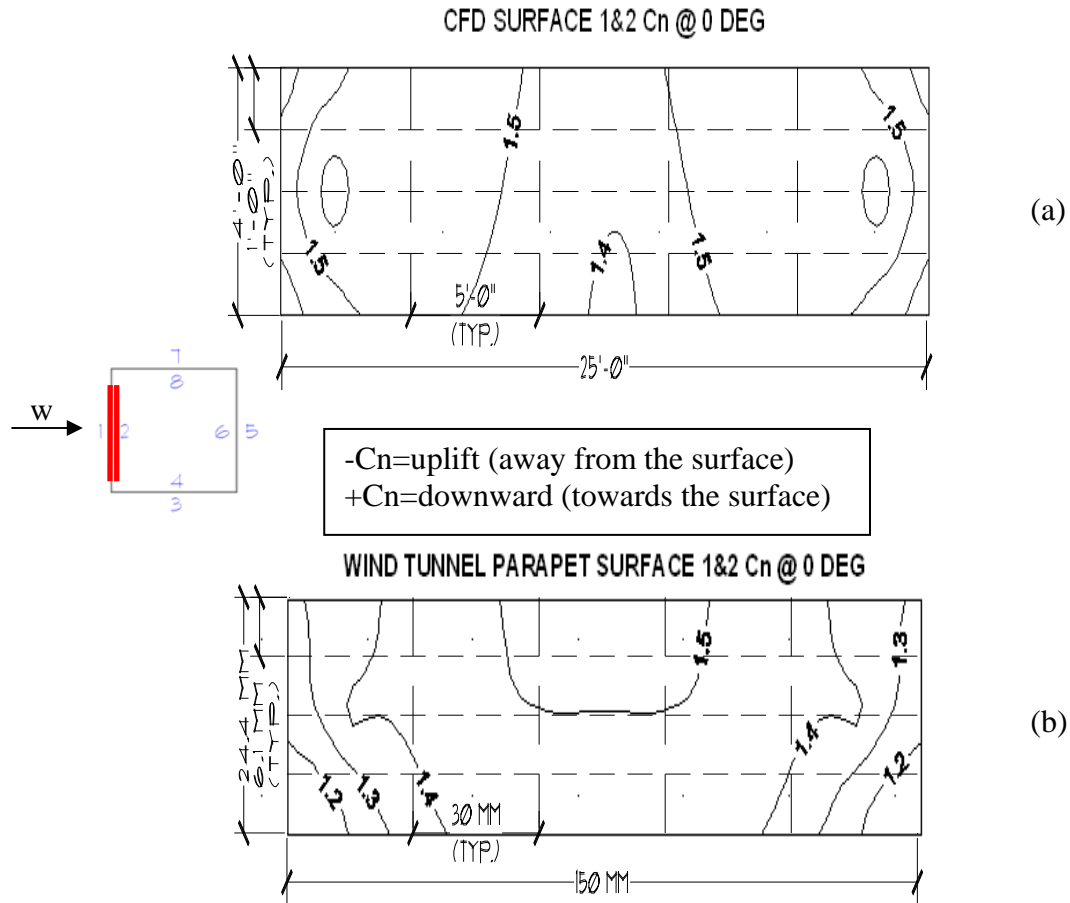
**Figure 6.17:  $C_p$  results at parapet surface #5 at 0 degrees for CFD (a) and wind tunnel (b).**

$C_p$  values for the parapet surface #6 at  $0^\circ$  for CFD and wind tunnel are shown in Figure 6.18a and Figure 6.18b. Positive  $C_p$  values indicating pressure downward on the surface are shown on both figures. Values vary from +0.1 to +0.2.



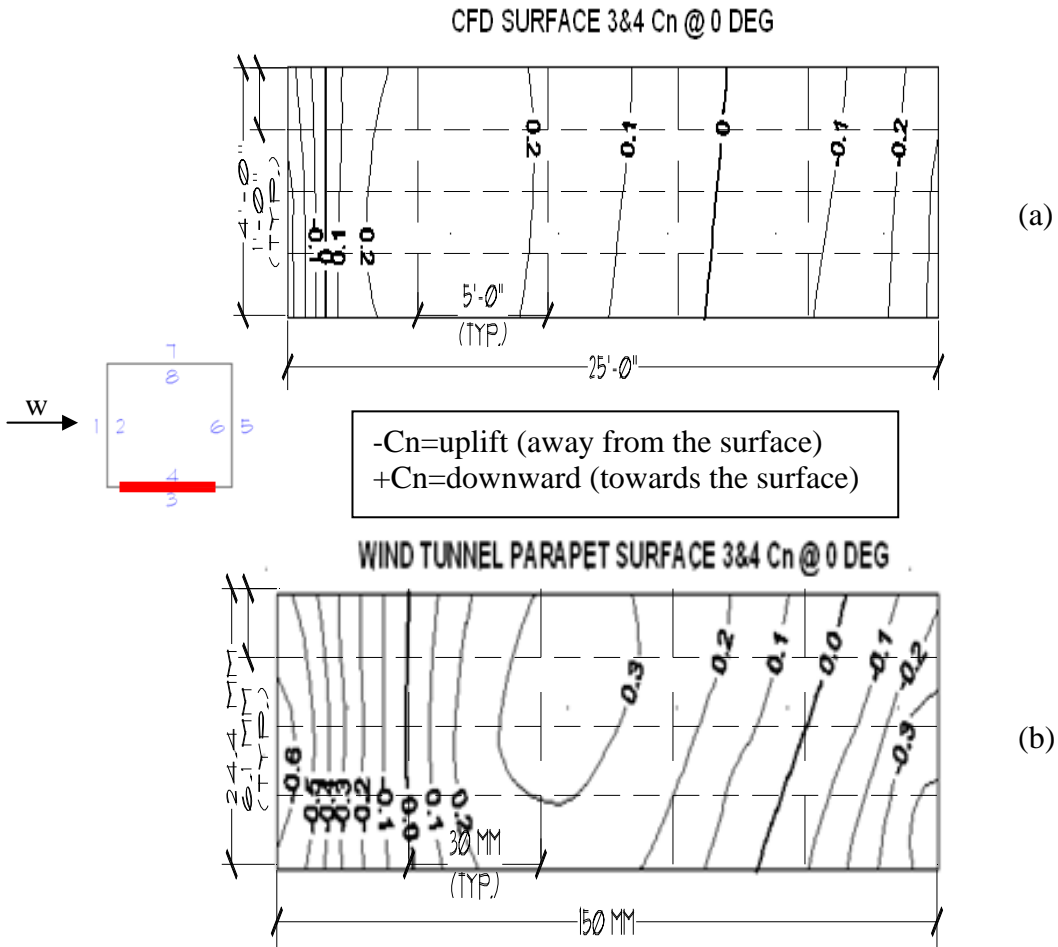
**Figure 6.18:  $C_p$  results at parapet surface #6 at 0 degrees for CFD (a) and wind tunnel (b).**

$C_n$  values for the parapet surface #1 and parapet surface #2 at  $0^\circ$  for CFD and wind tunnel are shown in Figure 6.19a and Figure 6.19b. Net positive values indicating pressure downward on the surface is shown on both figures. Values vary from +1.2 to +1.5.



**Figure 6.19:  $C_n$  results at parapet surface #1 and 2 at 0 degrees for CFD (a) and wind tunnel (b).**

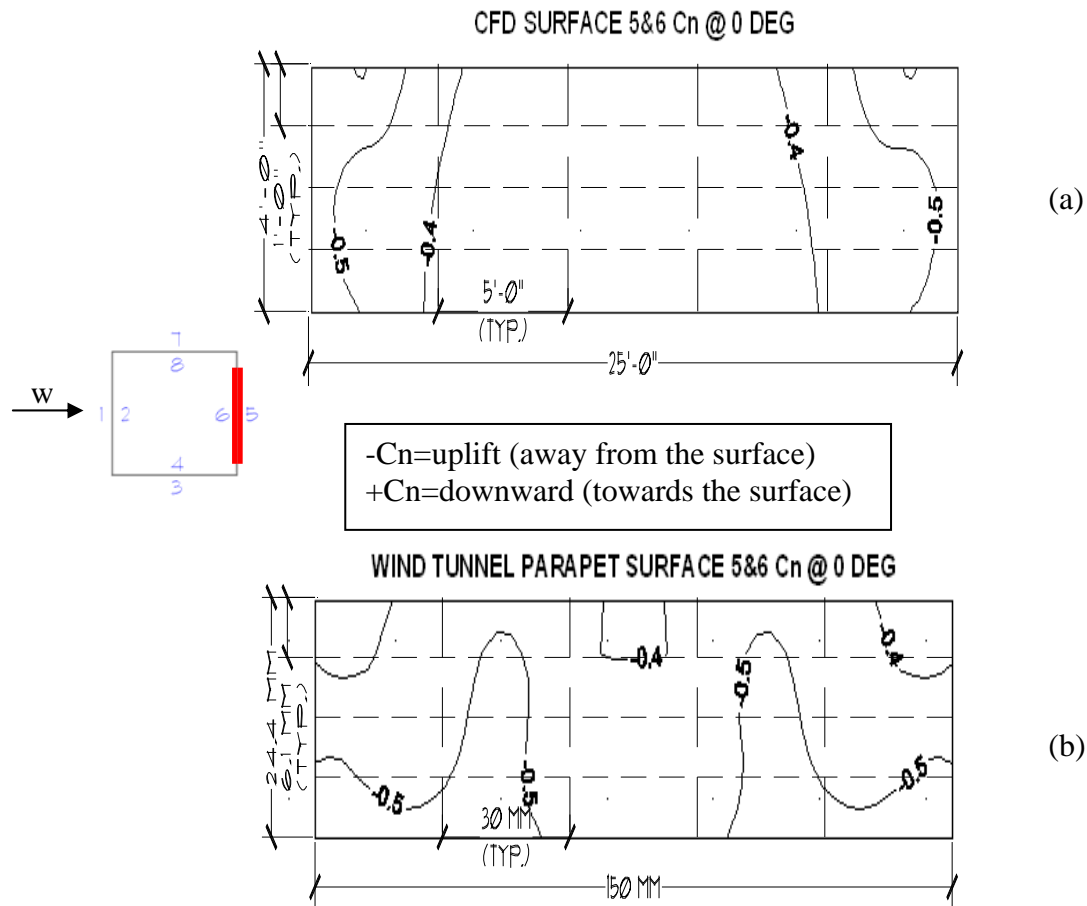
$C_n$  values for the parapet surface #3 and parapet surface #4 at 0° for CFD and wind tunnel are shown in Figure 6.20a and Figure 6.20b. Net negative and positive values indicating changes in pressure downward and uplift on the surface is shown in both figures. Values vary from -0.5 to be 0.0 the first 1/3 of the horizontal surface.  $C_n$  values for the parapet surface #3 and parapet surface #4 at 0° for CFD and wind tunnel are shown in Figure 6.20a and Figure 6.20b.



**Figure 6.20:  $C_n$  results at parapet surface #3 and 4 at 0 degrees for CFD (a) and wind tunnel (b).**

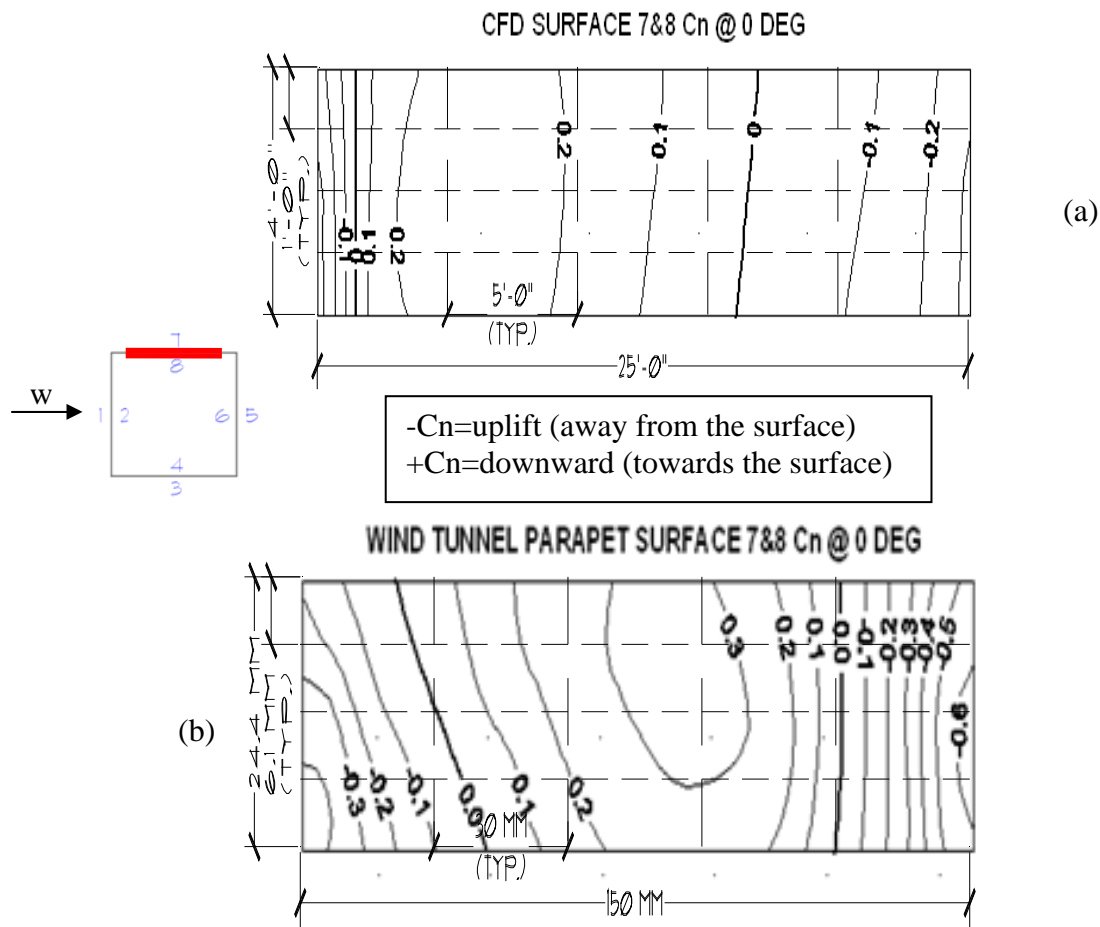
Net negative and positive values indicating changes in pressure downward and uplift on the surface is shown in both figures. Values vary from -0.5 to be 0.0 the first 1/3 of the horizontal surface. The values continue changing from 0.0 to +0.3 and to 0.0 for 2/3 of the horizontal surface. The last 1/3 of the horizontal surface,  $C_n$  surface values changes from 0.0 to -0.3. The CFD simulation is able to capture the transition between the changes in downward pressure and uplift. The values at the beginning of the horizontal surface are higher on the wind tunnel parapet surface.

Cn values for the parapet surface #5 and parapet surface #6 at 0° for CFD and wind tunnel are shown on Figure 6.21a and Figure 6.21b. Net negative values indicating uplift on the surface is shown on both figures. Values vary from -0.5 to -0.4.



**Figure 6.21: Cn results at parapet surface #5 and 6 at 0 degrees for CFD (a) and wind tunnel (b).**

Cn values for the parapet surface #7 and parapet surface #8 at 0° for CFD and wind tunnel are shown on Figure 6.22a and Figure 6.22b. Net negative and positive values indicating changes in pressure towards and away on the surface are shown on both figures.

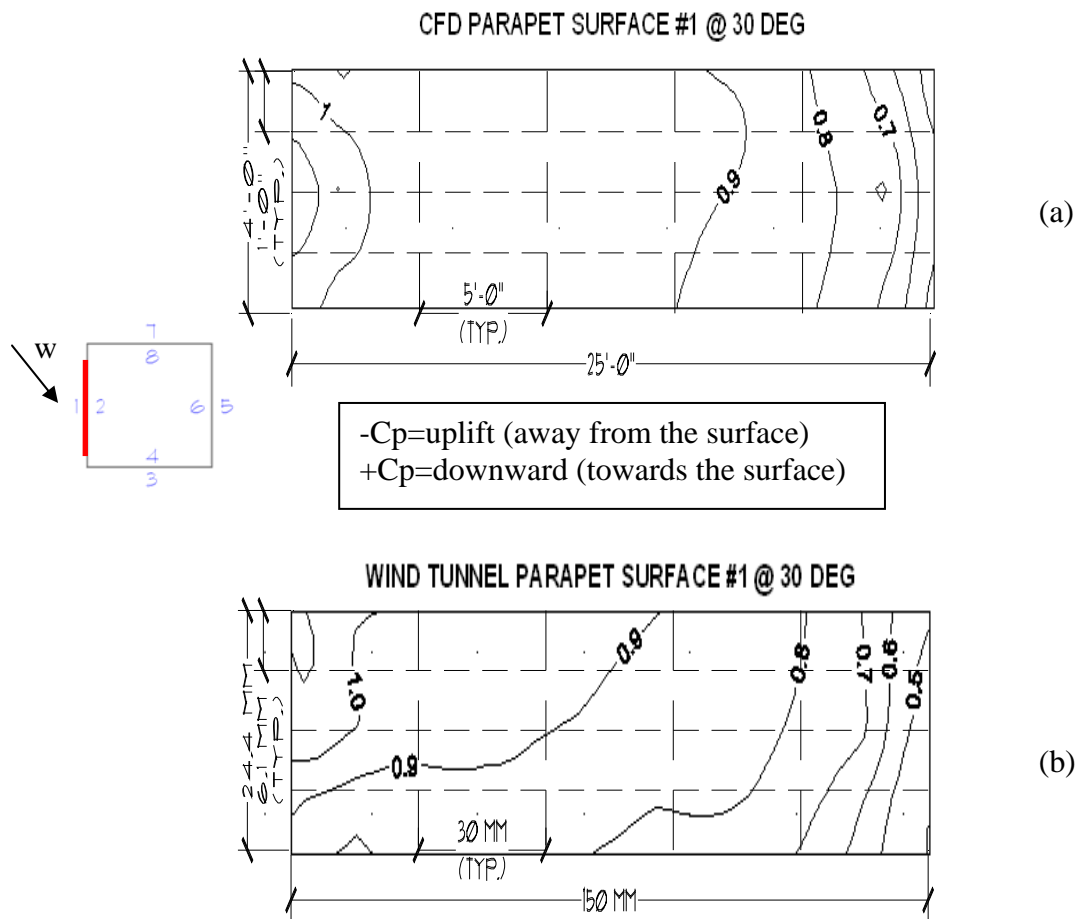


**Figure 6.22:  $C_n$  results at parapet surface #7 and 8 at 0 degrees for CFD (a) and wind tunnel (b).**

Values vary from -0.6 to 0.0 at the first 1/3 of the horizontal surface. The values continue changing from 0.0 to +0.3, going back to 0.0 for 2/3 of the horizontal surface. The last 1/3 of the horizontal surface,  $C_n$  surface values changes from 0.0 to -0.3. The CFD simulation is able to capture the transition between the changes in towards pressure and away from surface. The higher values at the beginning of the horizontal surface are higher on the wind tunnel parapet surface.

## 6.7 CFD and wind tunnel results for parapets, wind at 30 degrees

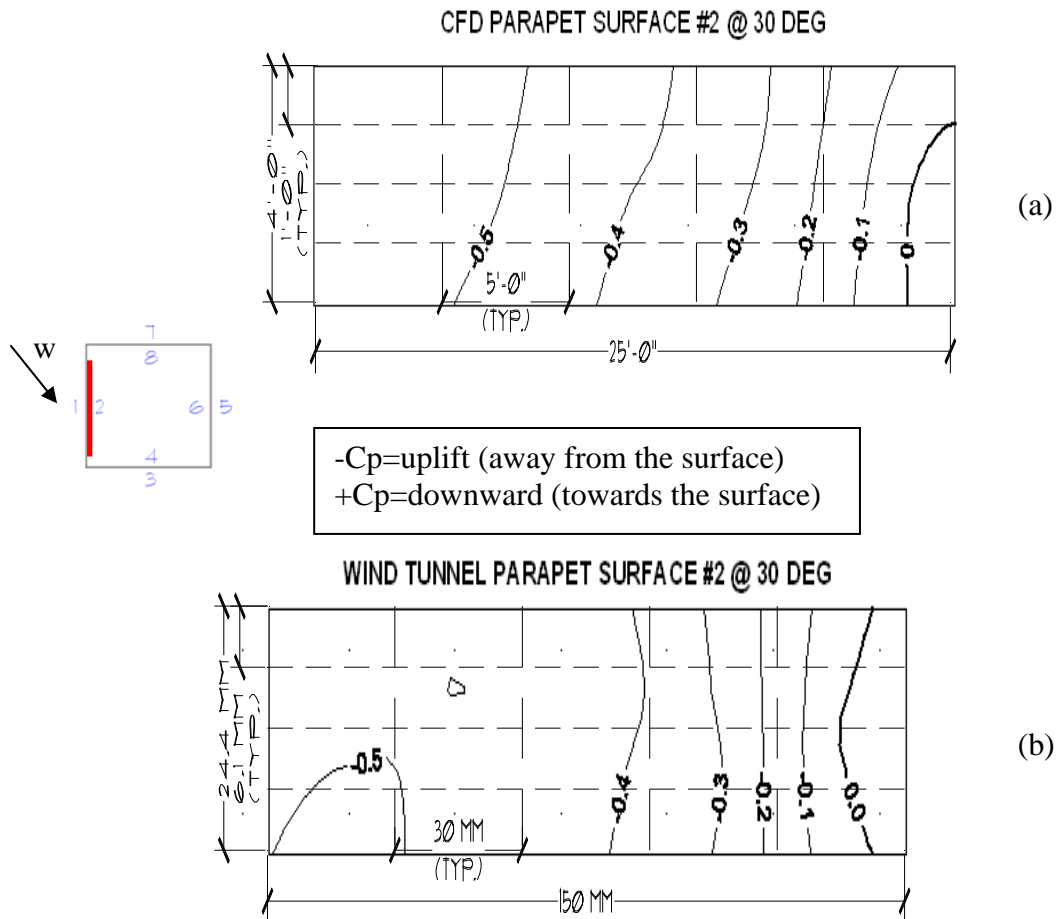
Cp values for the parapet surface #1 at 30° for CFD and wind tunnel are shown in Figure 6.23a and Figure 6.23b. Positive Cp values indicating pressure downward on the surface. Values vary from +0.5 to +1.1.



**Figure 6.23: Cp results at parapet surface #1 at 30 degrees for CFD (a) and wind tunnel (b).**

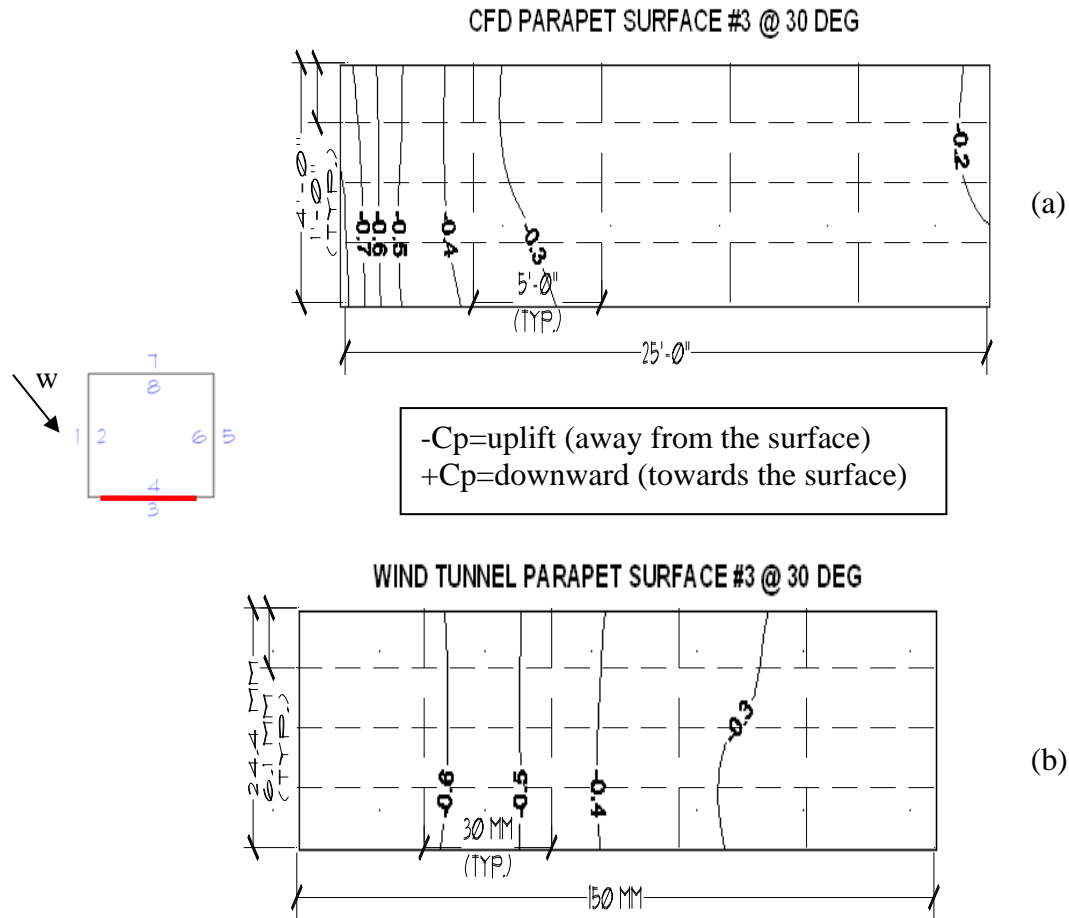
Cp values for the parapet surface #2 at 30° for CFD and wind tunnel are shown in Figure 6.24a and Figure 6.24b. Negative Cp values indicating uplift pressure away from the surface and

a transition zone where the  $C_p$  values changes from negative to positive values at the last 1/6 of the horizontal surface. Values vary from -0.5 to +0.1.



**Figure 6.24:  $C_p$  results at parapet surface #2 at 30 degrees for CFD (a) and wind tunnel (b).**

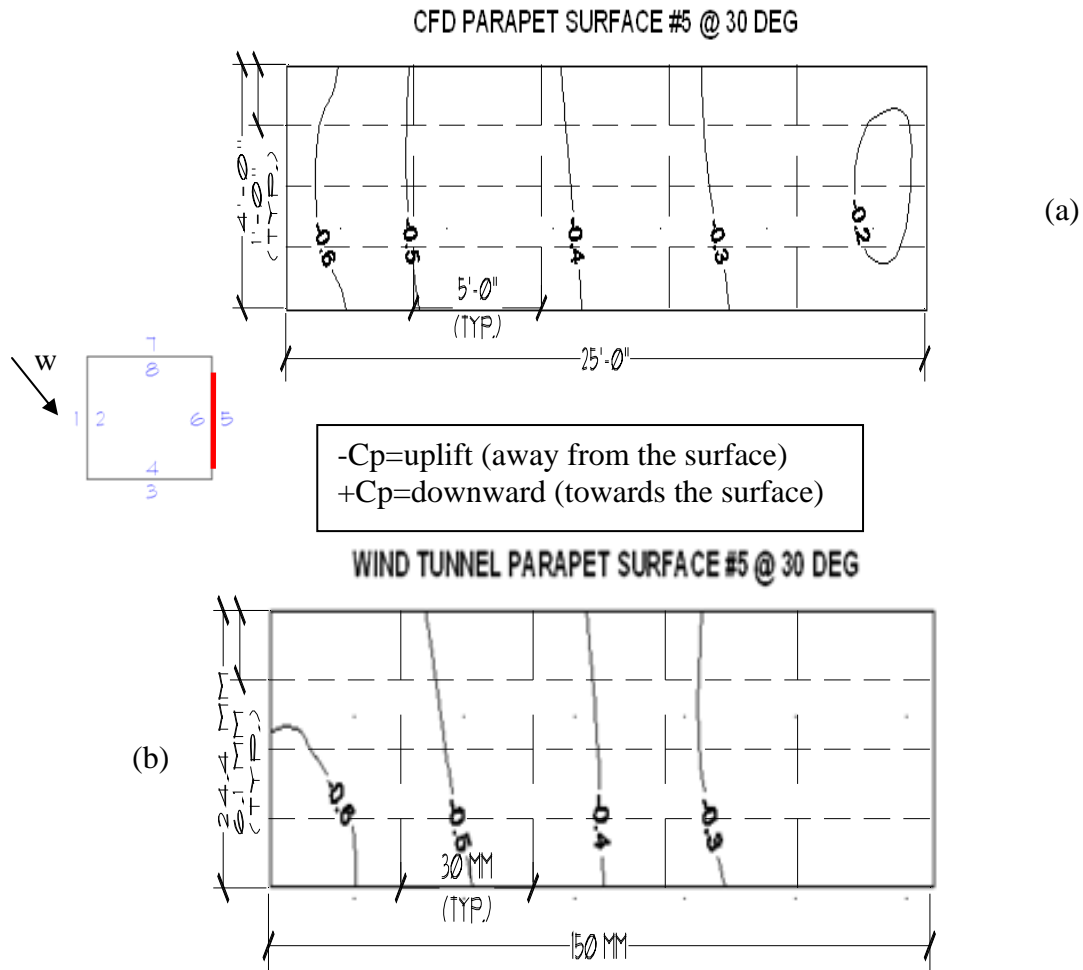
$C_p$  values for the parapet surface #3 and parapet surface #7 at 30° for CFD and wind tunnel are shown in Figure 6.25a and Figure 6.25b. Negative  $C_p$  values indicating uplift pressure away from the surface. Values vary from -0.7 to -0.2.



**Figure 6.25:  $C_p$  results at parapet surface #3 at 30 degrees for CFD (a) and wind tunnel (b).**

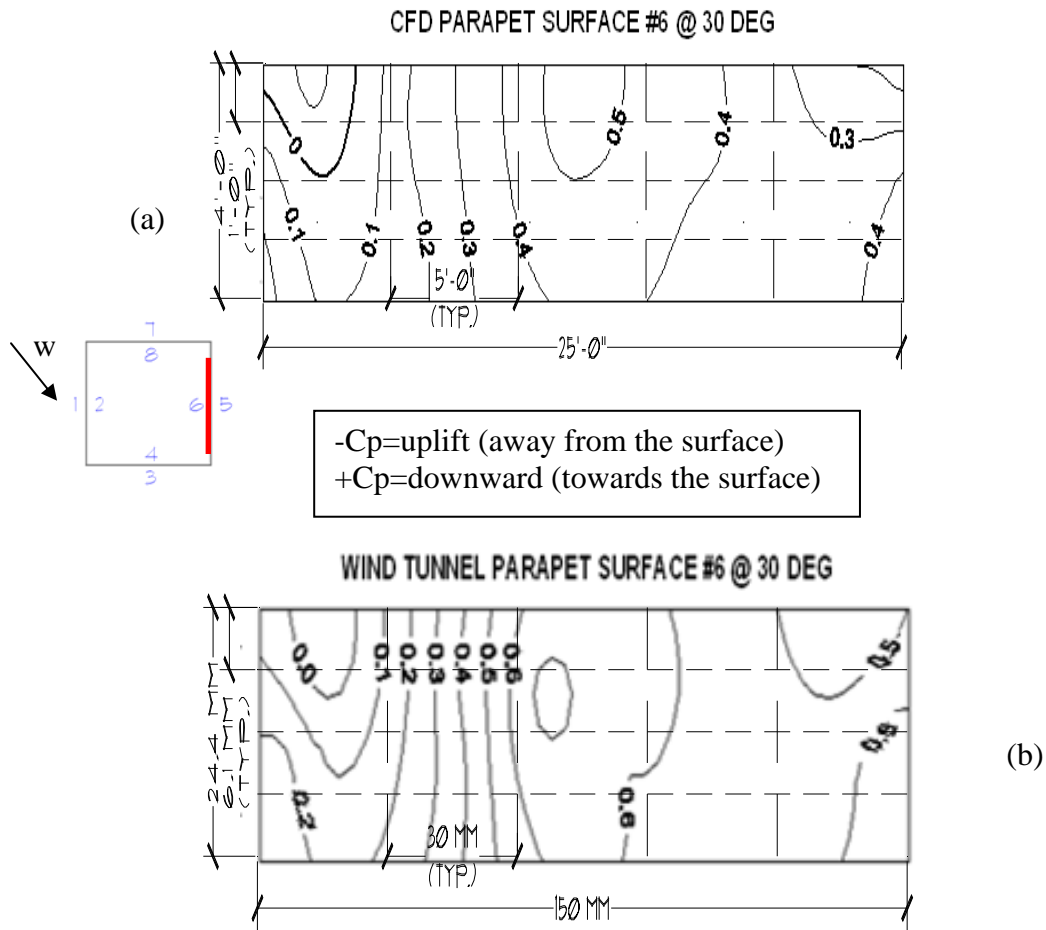
$C_p$  values for the parapet surface #4 at 30° for CFD and wind tunnel are shown in Figure 6.26a and Figure 6.26b. Negative  $C_p$  values indicating uplift, pressure away from the surface. Positive  $C_p$  values indicating pressure downward on the surface. There is a transition from negative to positive values at 1/2 of the horizontal surface. Values vary from -0.2 to 0.0 on the first half, and from 0.0 to 0.7 on the second half.





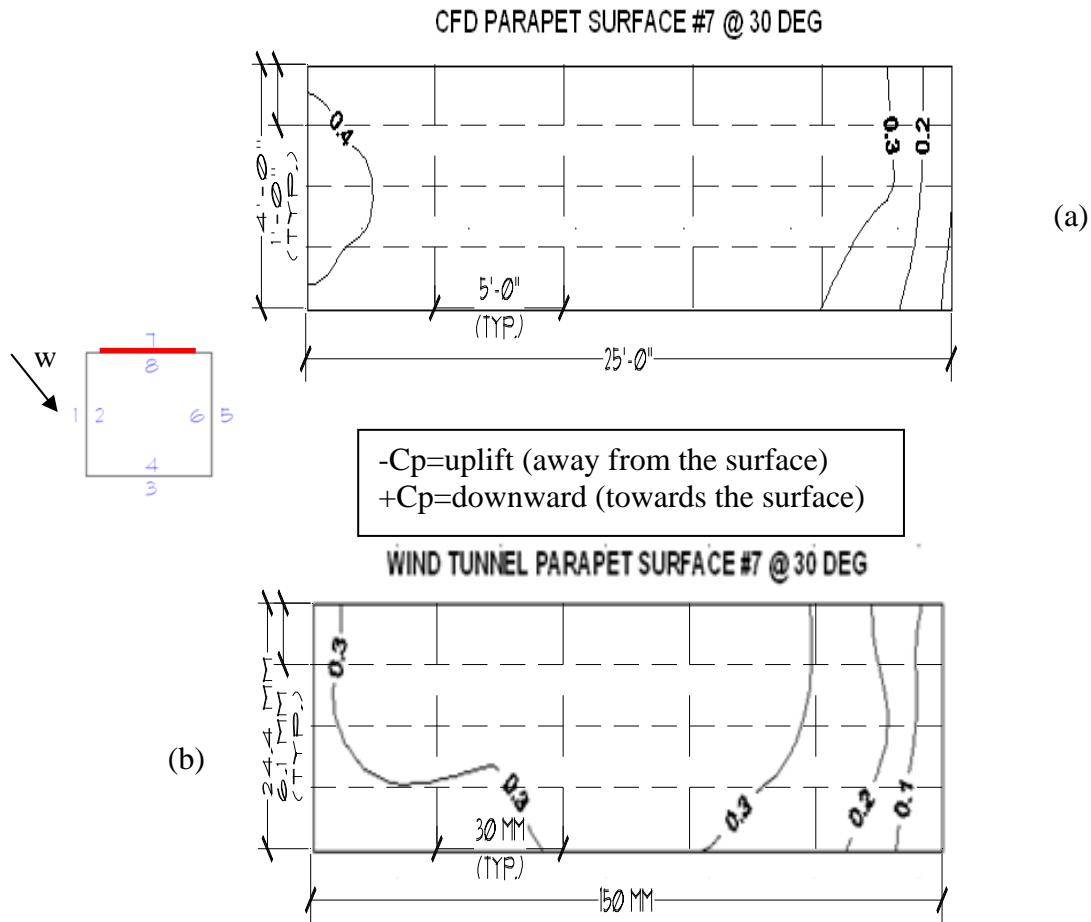
**Figure 6.27:  $C_p$  results at parapet surface #5 at 30 degrees for CFD (a) and wind tunnel (b).**

$C_p$  values for the parapet surface #6 at 30° for CFD and wind tunnel are shown in Figure 6.28a and Figure 6.28b. Positive net values indicating downward pressure on the surface is shown on both figures. There is a small region where the values changes from positive to negative  $C_p$  values at the end of the horizontal surface. Values vary from +0.6 to 0.0.



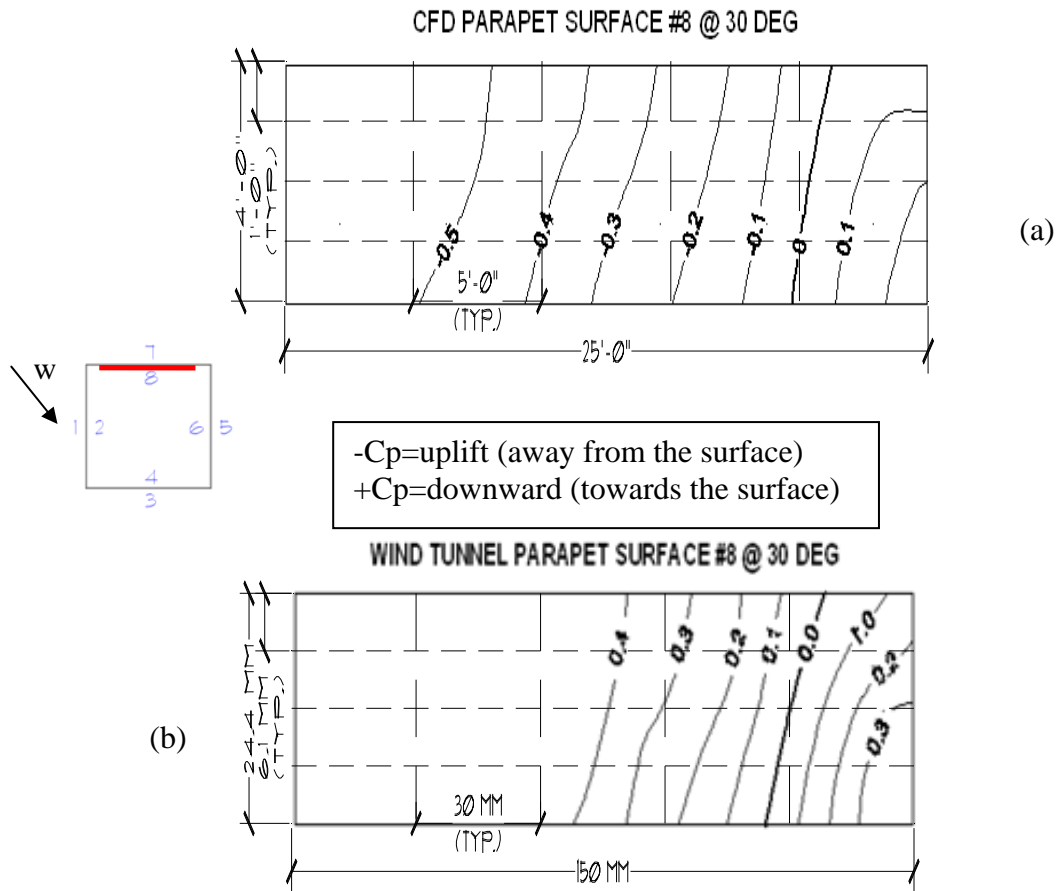
**Figure 6.28: Cp results at parapet surface #6 at 30 degrees for CFD (a) and wind tunnel (b).**

Cp values for the parapet surface #7 at 30° for CFD and wind tunnel are shown in Figure 6.29a and Figure 6.29b. Positive Cp values indicating towards pressure on the surface are shown on both figures. Values vary from +0.4 to +0.1.



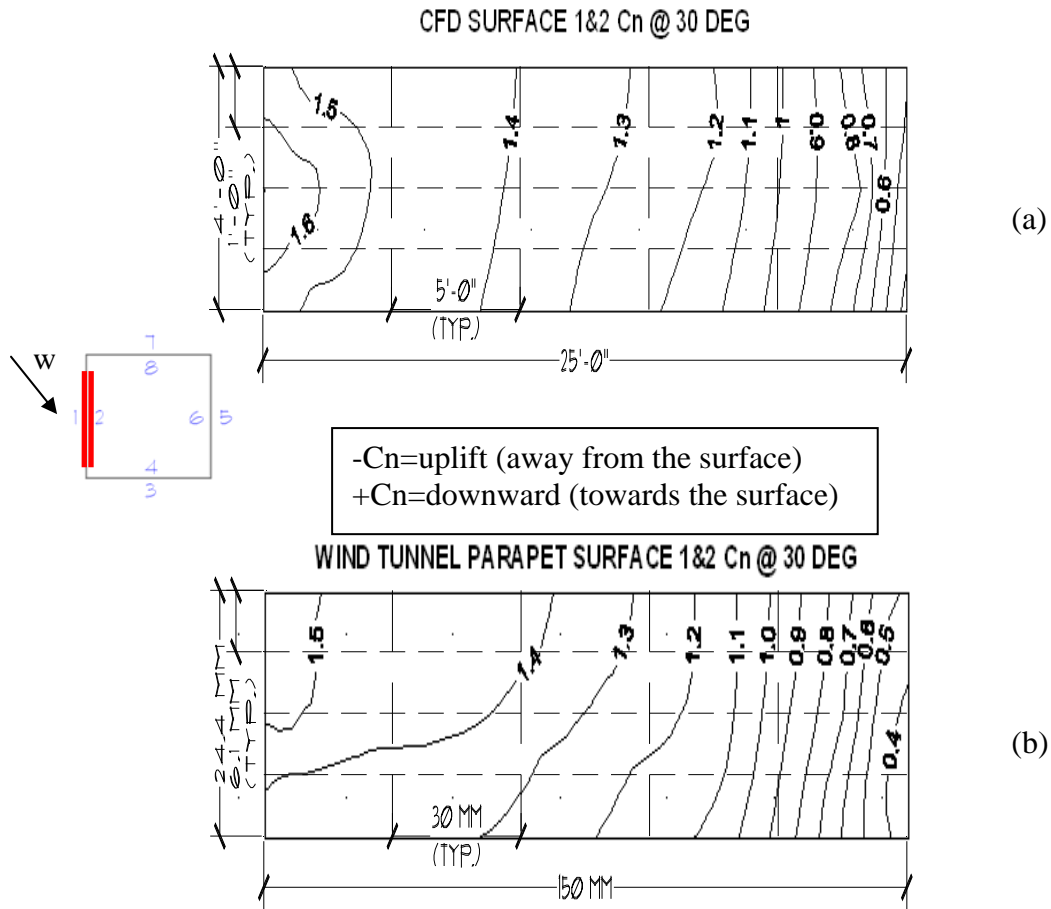
**Figure 6.29:  $C_p$  results at parapet surface #7 at 30 degrees for CFD (a) and wind tunnel (b).**

$C_p$  values for the parapet surface #8 at 30° for CFD and wind tunnel are shown in Figure 6.30a and Figure 6.30b. Negative  $C_p$  values indicating uplift, pressure away from the surface. Positive  $C_p$  values indicating pressure downward on the surface. There is a transition from negative to positive values at 2/3 of the horizontal surface. Values vary from -0.4 to 0.0 and from 0.0 to +0.3.



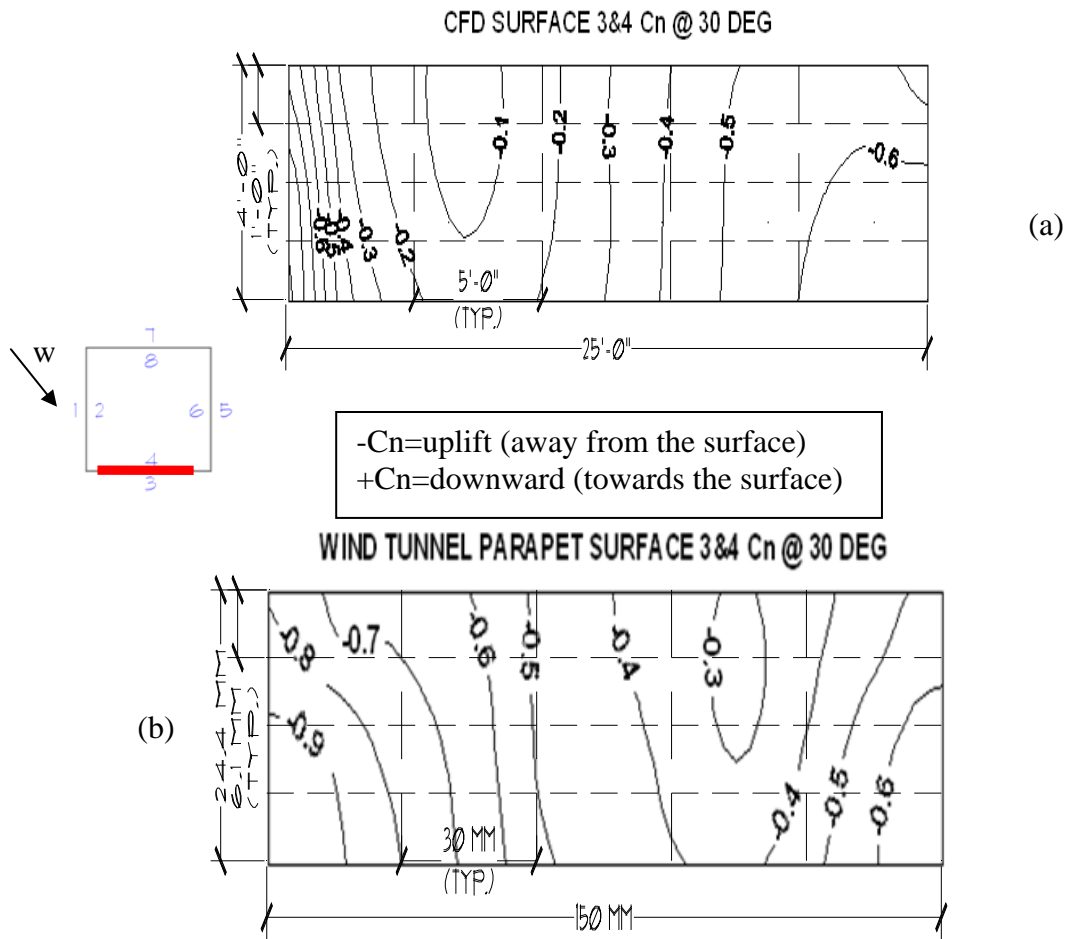
**Figure 6.30:  $C_p$  results at parapet surface #8 at 30 degrees for CFD (a) and wind tunnel (b).**

$C_n$  values for the parapet surface #1 and parapet surface #2 at 30° for CFD and wind tunnel are shown in Figure 6.31a and Figure 6.31b. Net positive values indicating pressure downward on the surface is shown on both figures. Values vary from +1.5 to +0.5.



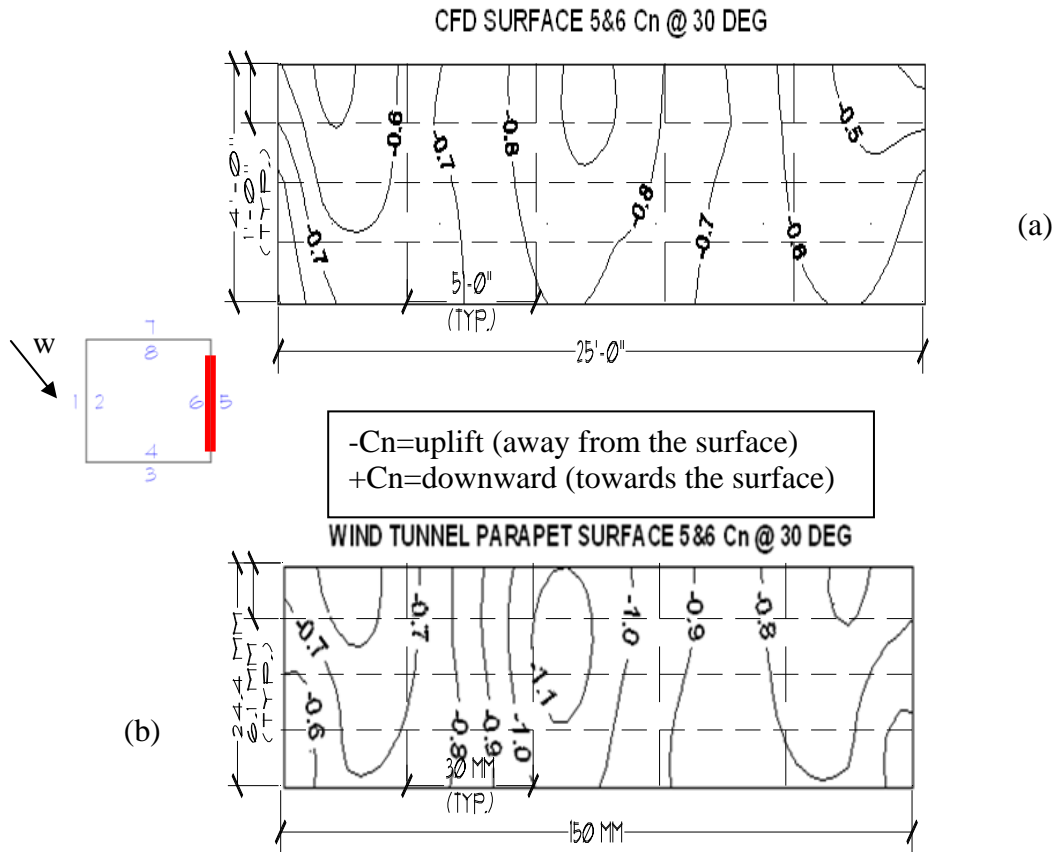
**Figure 6.31:  $C_n$  results at parapet surface #1 and 2 at 30 degrees for CFD (a) and wind tunnel (b).**

$C_n$  values for the parapet surface #3 and parapet surface #4 at 30° for CFD and wind tunnel are shown in Figure 6.32a and Figure 6.32b. Net negative values indicating changes in pressure and uplift on the surface are shown on both figures. Values vary from -0.9 to -0.3 for the first 1/2 of the horizontal surface. The values continue changing from -0.3 to -0.6 at the end of the horizontal surface.



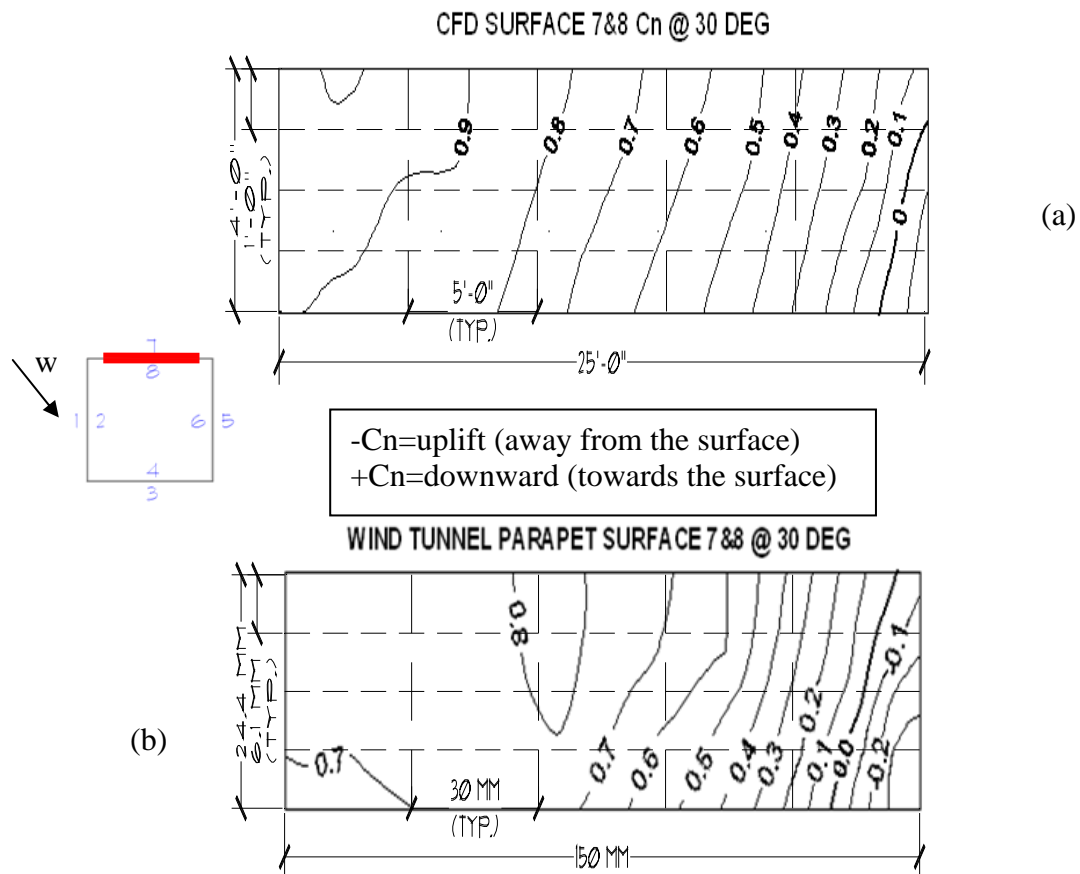
**Figure 6.32: Cn results at parapet surface #3 and 4 at 30 degrees for CFD (a) and wind tunnel (b).**

Cn values for the parapet surface #5 and parapet surface #6 at 30° for CFD and wind tunnel are shown in Figure 6.33a and Figure 6.33b. Net negative values indicating uplift on the surface is shown on both figures. Values vary from -1.1 to -0.6.



**Figure 6.33:  $C_n$  results at parapet surface #5 and 6 at 30 degrees for CFD (a) and wind tunnel (b).**

$C_n$  values for the parapet surface #7 and parapet surface #8 at 30° for CFD and wind tunnel are shown in Figure 6.34a and Figure 6.34b. Net positive and negative values indicating changes in uplift and pressure downward are shown on both figures. Values vary from +0.9 to 0.0 until the last 1/6 of the horizontal surface. The values continue changing from 0.0 to -0.3 on the last 1/6 of the horizontal surface.



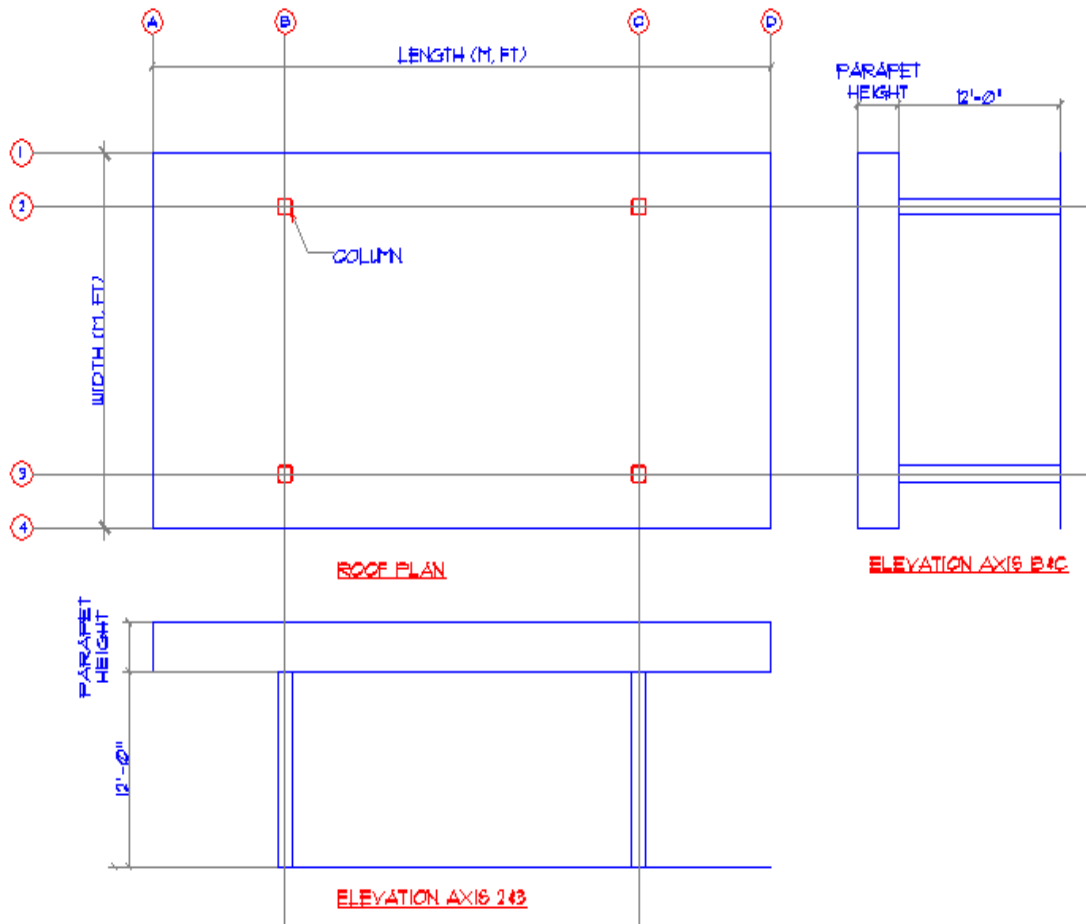
**Figure 6.34: Cn results at parapet surface #7 and 8 at 30 degrees for CFD (a) and wind tunnel (b).**

## **CHAPTER 7. PARAMETRIC STUDIES OF PRESSURE COEFFICIENTS IN CANOPIES USING CFD**

### **7.1 Description**

The results of the previous chapter shows that EFD.Lab, the CFD computer program used in this investigation, can generate similar  $C_p$  values in comparison to the wind tunnel results. For the CFD computations and the wind tunnel tests, only one configuration has been tested, the canopy with dimensions 7.6 m (25ft) x 7.6 m (25 ft) and parapet of 1.22 m (4 ft). However, it is necessary to investigate different geometries in order to determine how the  $C_p$  and the  $C_n$  values vary with the plan geometry and parapet differences, and establish design  $C_n$  values based on those results.

Three different configurations are investigated in this chapter. Each configuration is investigated individually with four different parapet heights. Figure 7.1 indicates the nomenclature used for the canopy geometry. Four different parapet heights are investigated in combination with the different plan models. Parapet heights attempt to represent realistic cases and vary from 0.91 m (3 ft), 1.22 m (4 ft), 1.52 m (5 ft) and 1.83 m (6 ft). Two different wind directions are modeled using the CFD program, 0 and 30 degrees as done on the wind tunnel. The combination of all the plan and parapet geometries generates a total of 24 models to be investigated.



**Figure 7.1: Nomenclature used for description of open canopy investigation.**

Table 7.1 indicates the number of models and geometry description. A total of twenty four (24) models were generated and investigated. Values for  $C_p$  and  $C_n$  were obtained for each model, and results for all models are presented on the following sections.

**Table 7.1: Model geometry description for the CFD parametric study.**

| Model # | Model Description (LxWxPH) | Width |    | Length |    | Parapet height |    | Wind Angle degrees |
|---------|----------------------------|-------|----|--------|----|----------------|----|--------------------|
|         |                            | m     | ft | m      | ft | m              | ft |                    |
| 1       | 25x25x3 0 DEG              | 7.62  | 25 | 7.62   | 25 | 0.91           | 3  | 0                  |
| 2       | 25x25x3 30 DEG             | 7.62  | 25 | 7.62   | 25 | 0.91           | 3  | 30                 |
| 3       | 25x25x4 0 DEG              | 7.62  | 25 | 7.62   | 25 | 1.22           | 4  | 0                  |
| 4       | 25x25x4 30 DEG             | 7.62  | 25 | 7.62   | 25 | 1.22           | 4  | 30                 |
| 5       | 25x25x5 0 DEG              | 7.62  | 25 | 7.62   | 25 | 1.52           | 5  | 0                  |
| 6       | 25x25x5 30 DEG             | 7.62  | 25 | 7.62   | 25 | 1.52           | 5  | 30                 |
| 7       | 25x25x6 0 DEG              | 7.62  | 25 | 7.62   | 25 | 1.83           | 6  | 0                  |
| 8       | 25x25x6 30 DEG             | 7.62  | 25 | 7.62   | 25 | 1.83           | 6  | 30                 |
| 9       | 25x40x3 0 DEG              | 7.62  | 25 | 12.19  | 40 | 0.91           | 3  | 0                  |
| 10      | 25x40x3 30 DEG             | 7.62  | 25 | 12.19  | 40 | 0.91           | 3  | 30                 |
| 11      | 25x40x4 0 DEG              | 7.62  | 25 | 12.19  | 40 | 1.22           | 4  | 0                  |
| 12      | 25x40x4 30 DEG             | 7.62  | 25 | 12.19  | 40 | 1.22           | 4  | 30                 |
| 13      | 25x40x5 0 DEG              | 7.62  | 25 | 12.19  | 40 | 1.52           | 5  | 0                  |
| 14      | 25x40x5 30 DEG             | 7.62  | 25 | 12.19  | 40 | 1.52           | 5  | 30                 |
| 15      | 25x40x6 0 DEG              | 7.62  | 25 | 12.19  | 40 | 1.83           | 6  | 0                  |
| 16      | 25x40x6 30 DEG             | 7.62  | 25 | 12.19  | 40 | 1.83           | 6  | 30                 |
| 17      | 25x50x3 0 DEG              | 7.62  | 25 | 15.24  | 50 | 0.91           | 3  | 0                  |
| 18      | 25x50x3 30 DEG             | 7.62  | 25 | 15.24  | 50 | 0.91           | 3  | 30                 |
| 19      | 25x50x4 0 DEG              | 7.62  | 25 | 15.24  | 50 | 1.22           | 4  | 0                  |
| 20      | 25x50x4 30 DEG             | 7.62  | 25 | 15.24  | 50 | 1.22           | 4  | 30                 |
| 21      | 25x50x5 0 DEG              | 7.62  | 25 | 15.24  | 50 | 1.52           | 5  | 0                  |
| 22      | 25x50x5 30 DEG             | 7.62  | 25 | 15.24  | 50 | 1.52           | 5  | 30                 |
| 23      | 25x50x6 0 DEG              | 7.62  | 25 | 15.24  | 50 | 1.83           | 6  | 0                  |
| 24      | 25x50x6 30 DEG             | 7.62  | 25 | 15.24  | 50 | 1.83           | 6  | 30                 |

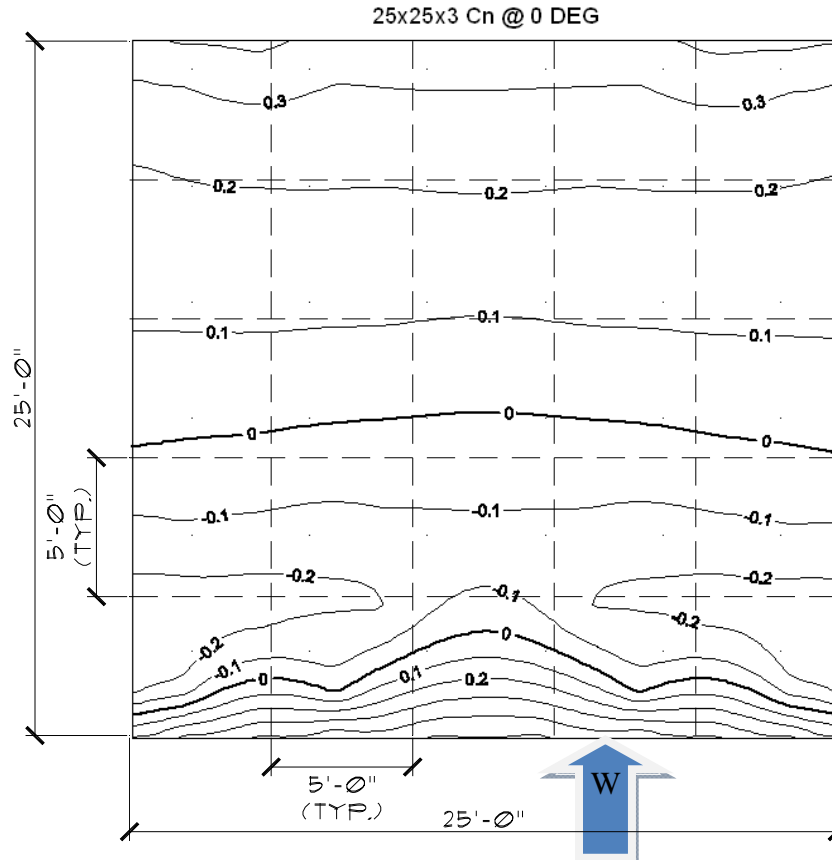
CFD model results are represented in two different graphical formats. First, top view of the contour surface representing the  $C_n$  values is presented for each model. Second, a 3D contour plot showing the  $C_n$  values are also represented. The purpose of that graphical representation is that it helps the visualization of the  $C_n$  values. The maximum and minimum are clearly represented, especially when the wind is applied at 30 degrees from the structural axis to the model.

After each group of four models is discussed, a 2D graph is shown to plot the  $C_n$  values for each parapet height, for each of the geometries. Observations and comments for each graphical result are in the following sections.

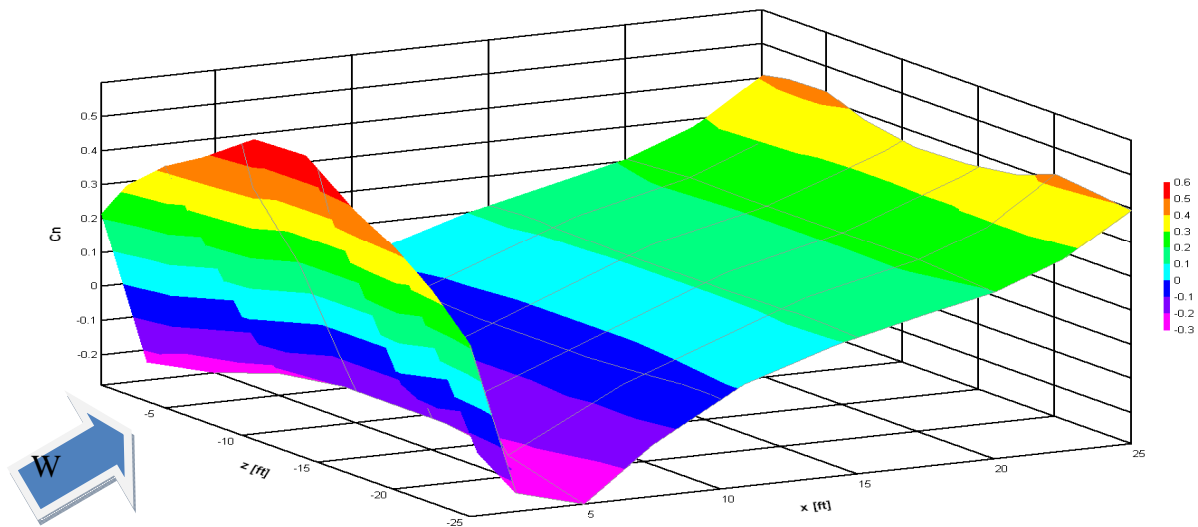
The importance of this parametric study is to investigate the effects of different geometries on the  $C_n$  values, with the aim of using them for structural design.

## **7.2 Open canopy models, 7.6 m (25 ft) x 7.6 m (25 ft) at 0 degrees**

Model #1 is a canopy with dimensions of 7.6 m (25 ft) x 7.6 m (25 ft) x 0.91 m (3 ft) with the wind at 0 degrees, as shown in Figure 7.2. Values of pressure coefficients were computed for this case and the results in terms of contour lines are plotted in Figure 7.2. Figure 7.3 shows a 3D view of the  $C_n$  values of the canopy. Values range from a maximum of +0.6 close the windward (and due to turbulence generated by the front parapet) to a minimum of -0.3. The results plotted show clear bounds with equal pressure values in the transverse direction, and only close to the front parapet does turbulence generate a small transverse variation.

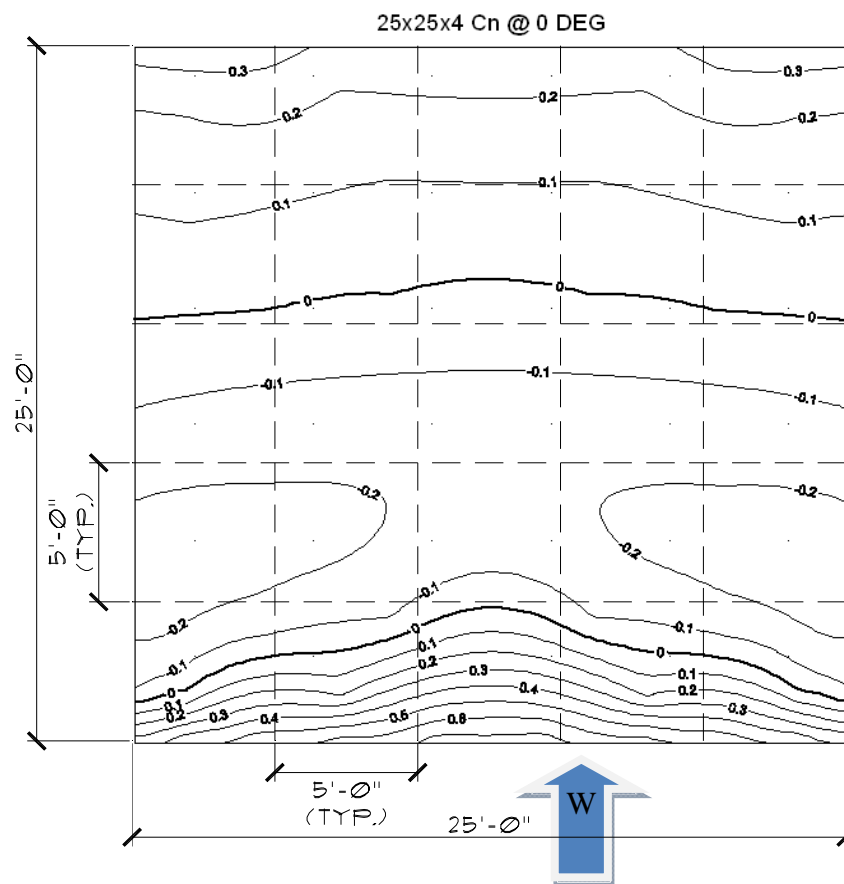


**Figure 7.2: Model #1, contour plan for Cn values for wind at 0 degrees.**  
25x25x3 0 DEG

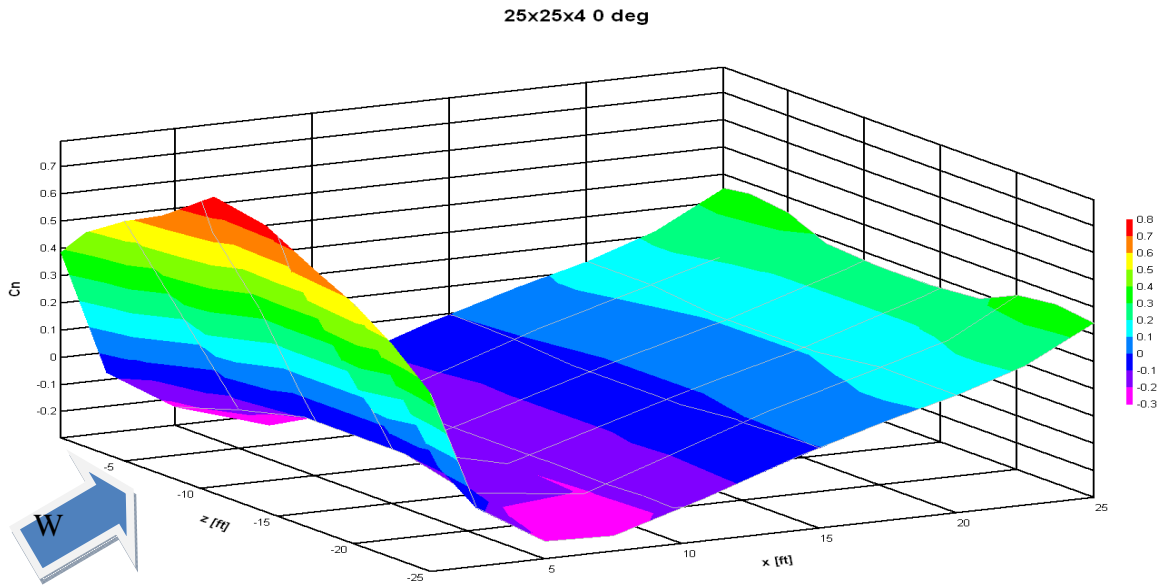


**Figure 7.3: Model #1, 3D contour plot for Cn values for wind at 0 degrees.**

For the same dimensions in plan, the new model has a higher parapet. Cn values for contour plan for Model #3, with dimensions of 7.6 m (25 ft) x 7.6 m (25 ft) x 1.22 m (4 ft) with the wind at 0 degrees are shown in Figure 7.4. Figure 7.5 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.8 to a minimum of -0.3. In comparison with the previous model with a shorter parapet, the maximum positive values have increased from 0.6 to 0.8 for a change of parapet from 3 to 4 ft.



**Figure 7.4: Model #3, contour plan for Cn values for wind at 0 degrees.**



**Figure 7.5: Model #3, 3D contour plot for  $C_n$  values for wind at 0 degrees.**

A higher parapet is investigated in the next configuration.  $C_n$  values for contour plan for Model #5, with dimensions of 7.6 m (25 ft) x 7.6 m (25 ft) x 1.52 m (5 ft) with the wind at 0 degrees are shown on Figure 7.6. Figure 7.7 shows a 3D view of the  $C_n$  values of the canopy. Values range from a maximum of +0.9 to a minimum of -0.3.

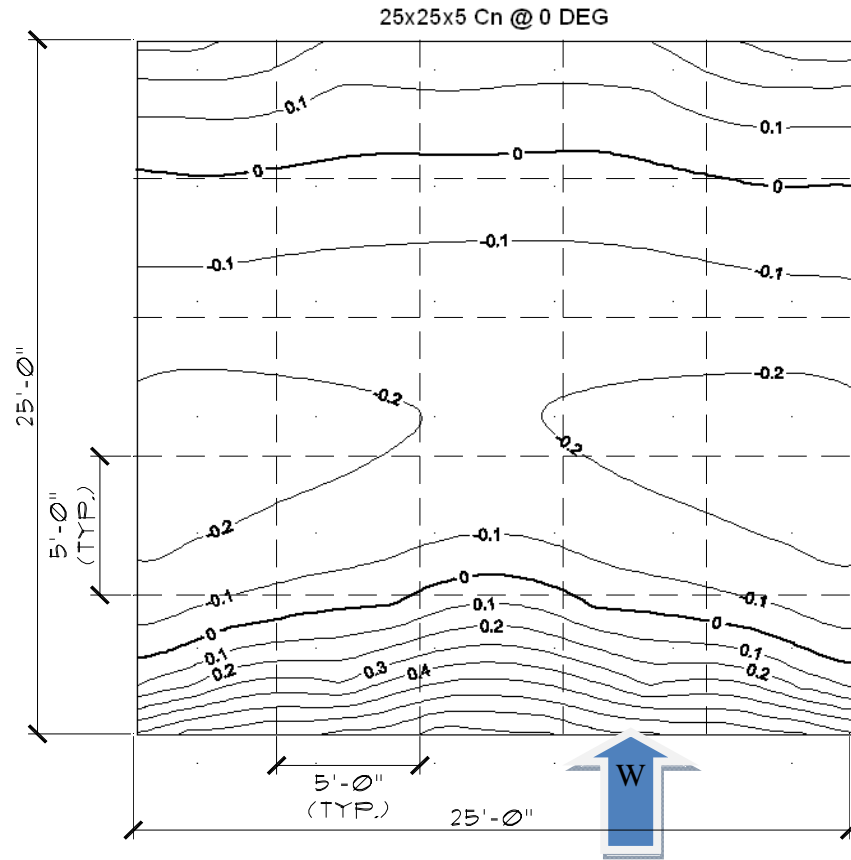


Figure 7.6: Model #5, contour plan for Cn values for wind at 0 degrees.

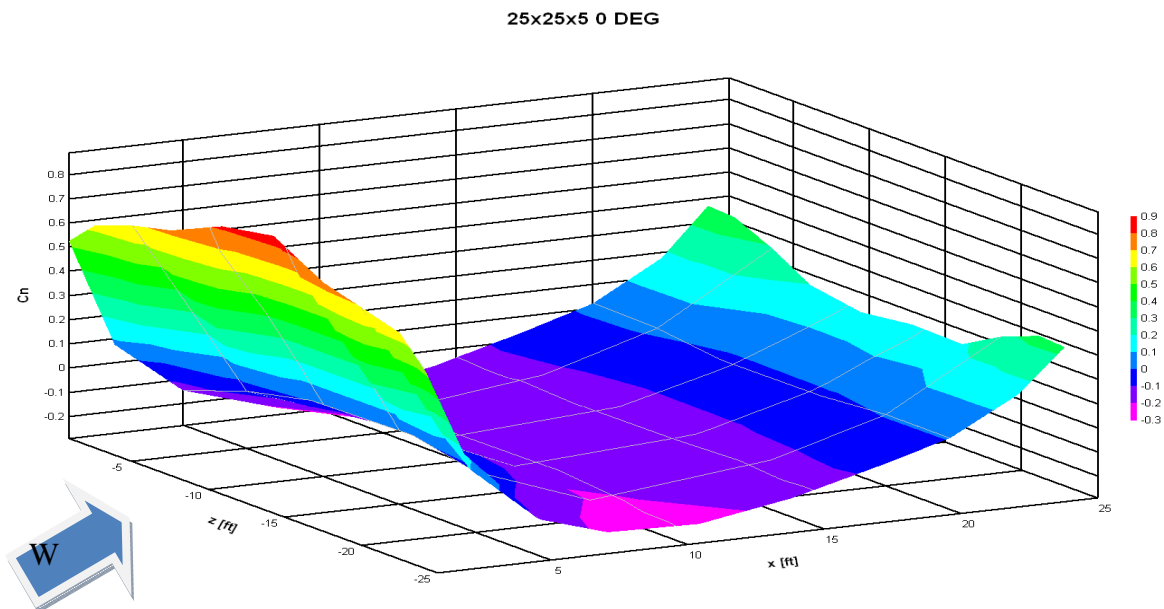
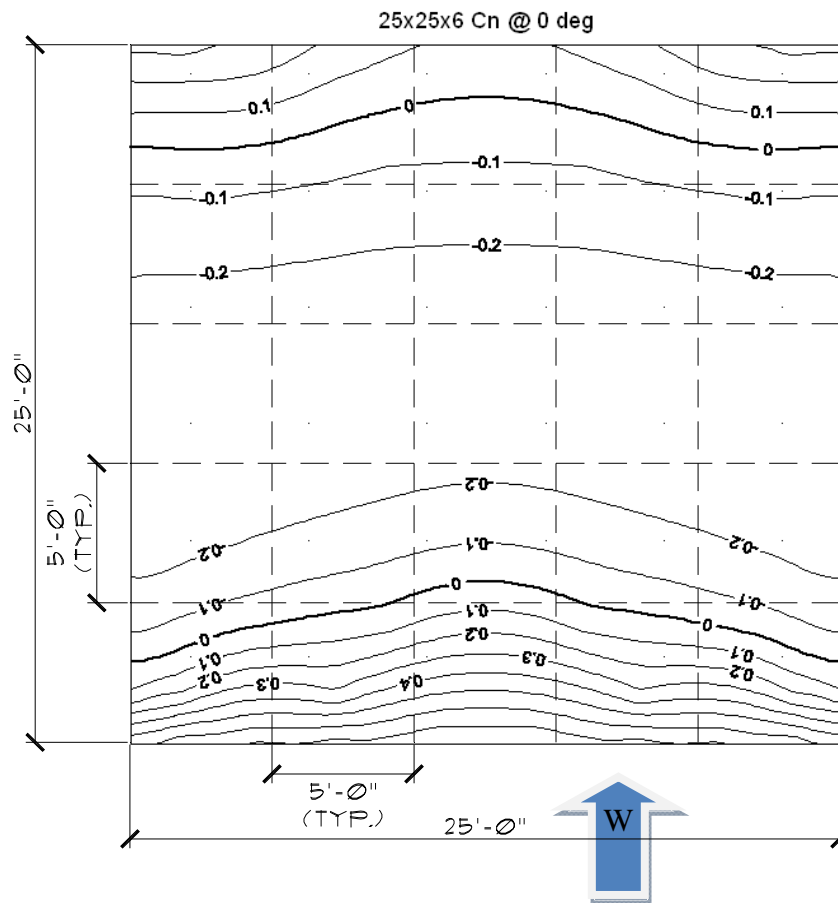
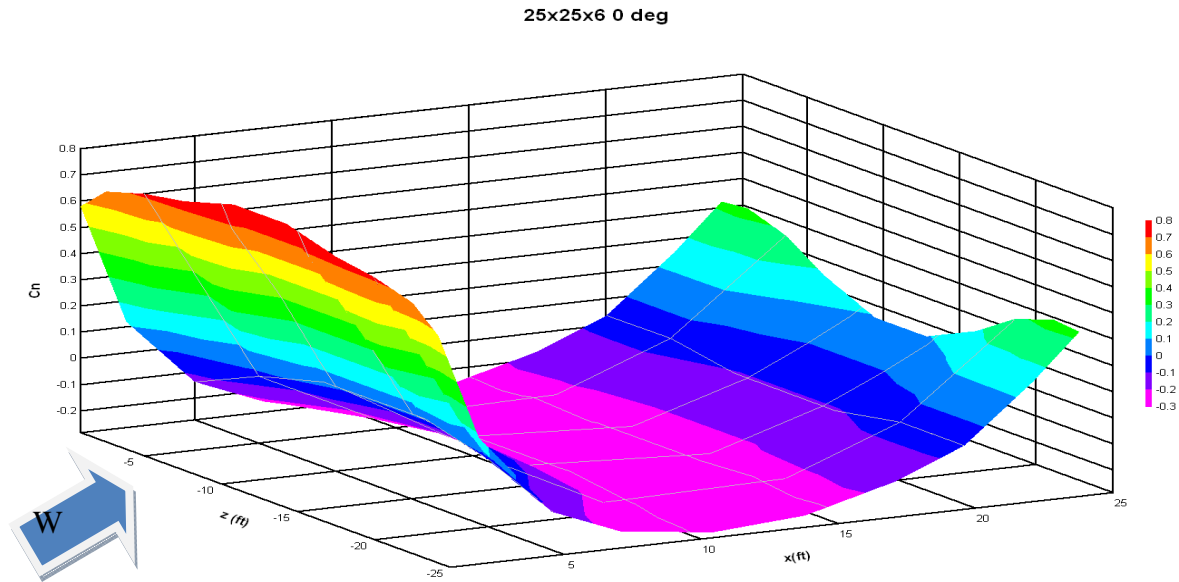


Figure 7.7: Model #5, 3D contour plot for Cn values for wind at 0 degrees.

Finally, the parapet is increased with respect to the previous models. Cn values for contour plan for Model #7, with dimensions of 7.6 m (25 ft) x 7.6 m (25 ft) x 1.82 m (6 ft) with the wind at 0 degrees, are shown on Figure 7.8. Figure 7.9 shows a 3D view of the Cn values of the canopy. Values ranges from a maximum of +0.9 to a minimum of -0.3.

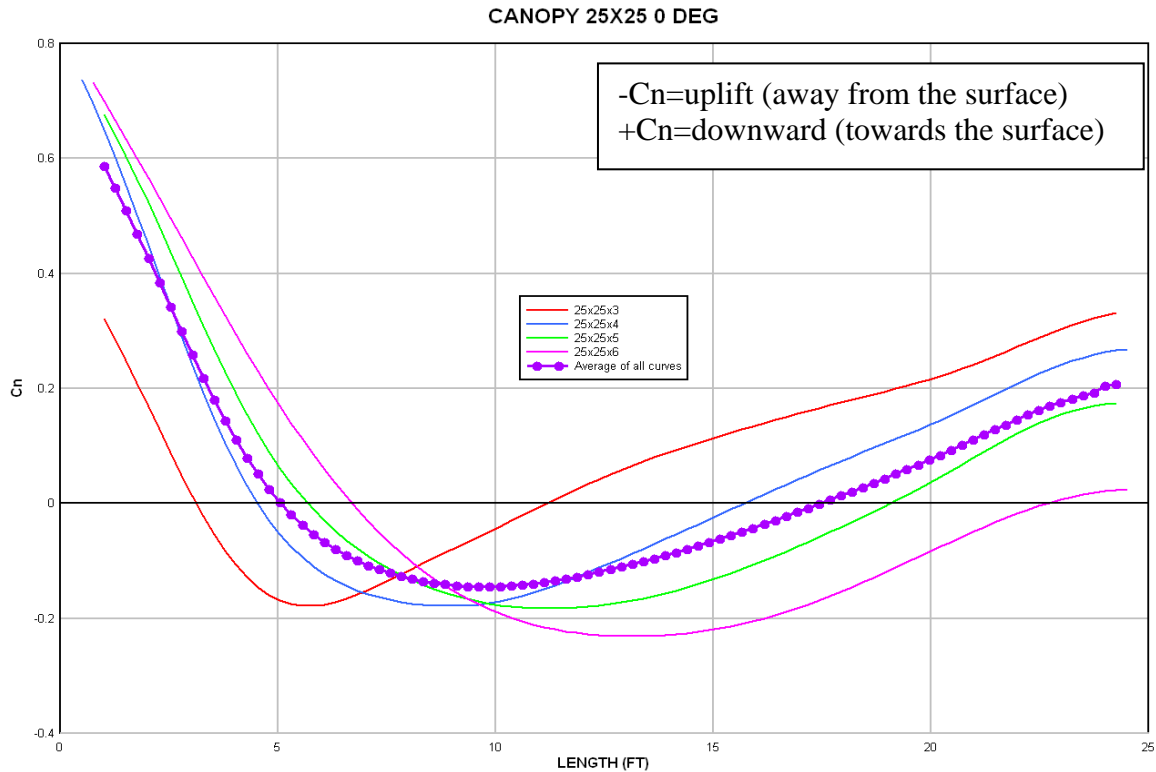


**Figure 7.8: Model #7, contour plan for Cn values for wind at 0 degrees.**



**Figure 7.9: Model #7, 3D contour plot for  $C_n$  values for wind at 0 degrees.**

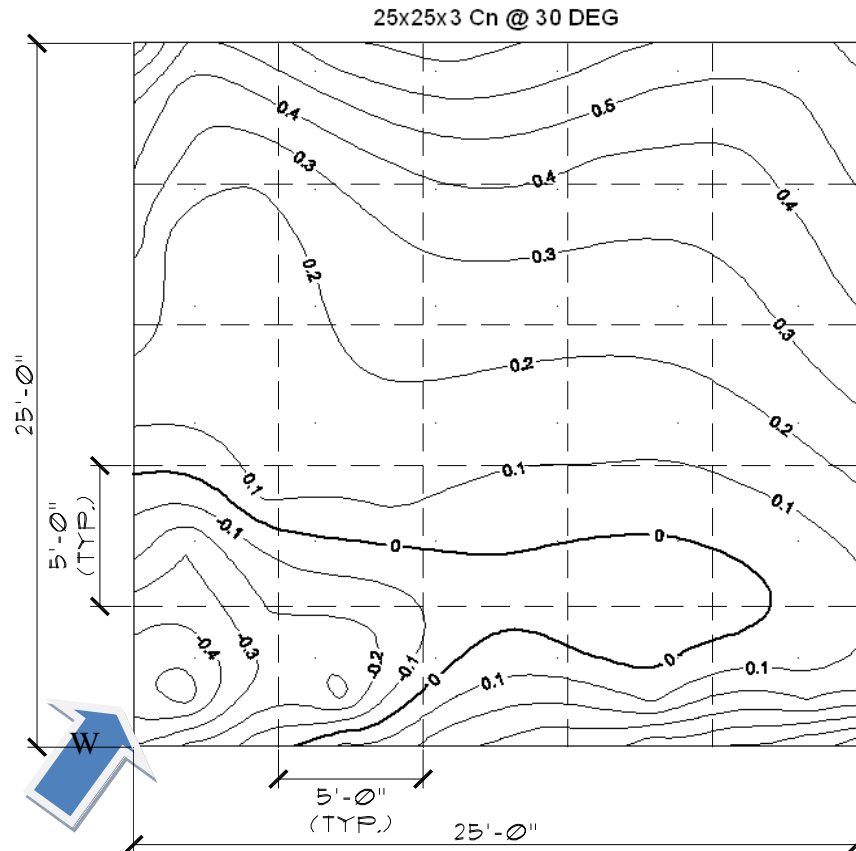
Figure 7.10 shows four  $C_n$  graphs, one for each parapet height on the 7.6 m (25 ft) x 7.6 m (25 ft) open canopy. The 2D graph data was obtained at the center of the canopy width in the principal axis direction. Every parapet height exhibits in the graph a segment of positive  $C_n$  values (acting downward) on the first  $L/5$  of the horizontal length. After the downward segment, uplift pressure acting on a length of  $L/3$  followed, to end with a downward pressure for the rest of the horizontal distance. It may be seen that the downward values on the area close to windward are larger than those computed on the leeward zone.



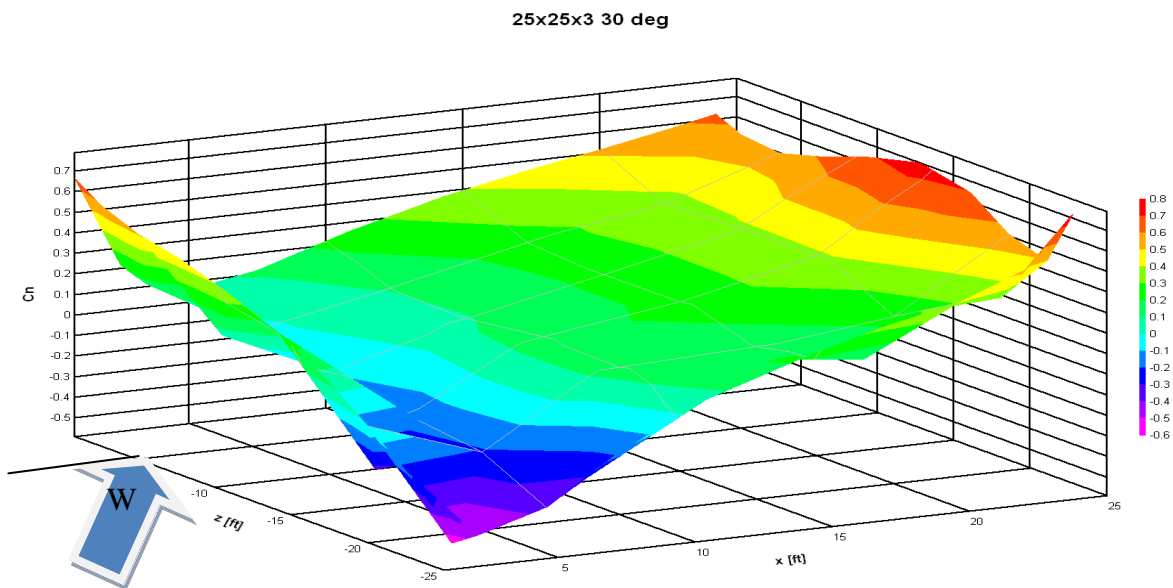
**Figure 7.10: Influence of parapet height. 2D graph of  $C_n$  values for all 7.6 m (25 ft) x 7.6 m (25 ft) @ 0 degrees.**

### 7.3 Open canopy models, 7.6 m (25 ft) x 7.6 m (25 ft) at 30 degrees

The configuration in the previous section was aligned with the wind direction; however, other wind directions may lead to more stringent pressure values.  $C_n$  values for contour plan for Model #2, with dimensions of 7.6 m (25 ft) x 7.6 m (25 ft) x 0.91 m (3 ft) with the wind at 30 degrees are shown in Figure 7.11. Figure 7.12 shows a 3D view of the  $C_n$  values of the canopy. The results indicate an asymmetric pressure pattern in the canopy, in which two parapets seen to play a role. Values range from a maximum of +0.8 to a minimum of -0.6. These values are significantly higher than those computed for Model #1, which were +0.6 and -0.3 respectively.

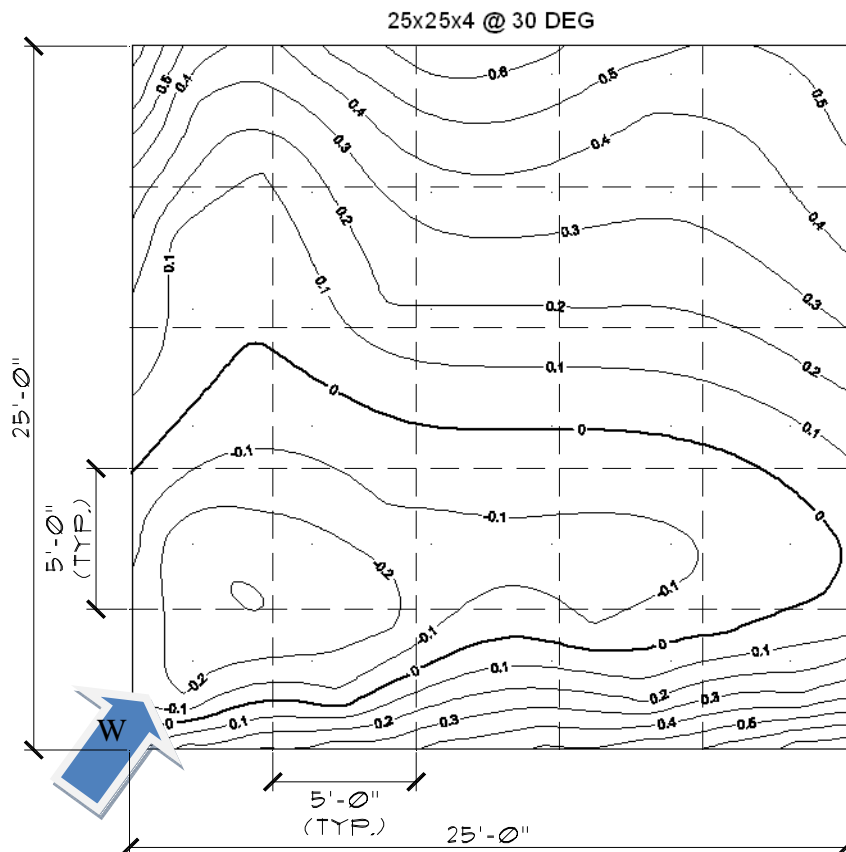


**Figure 7.11: Model #2, contour plan for Cn values for wind at 30 degrees.**

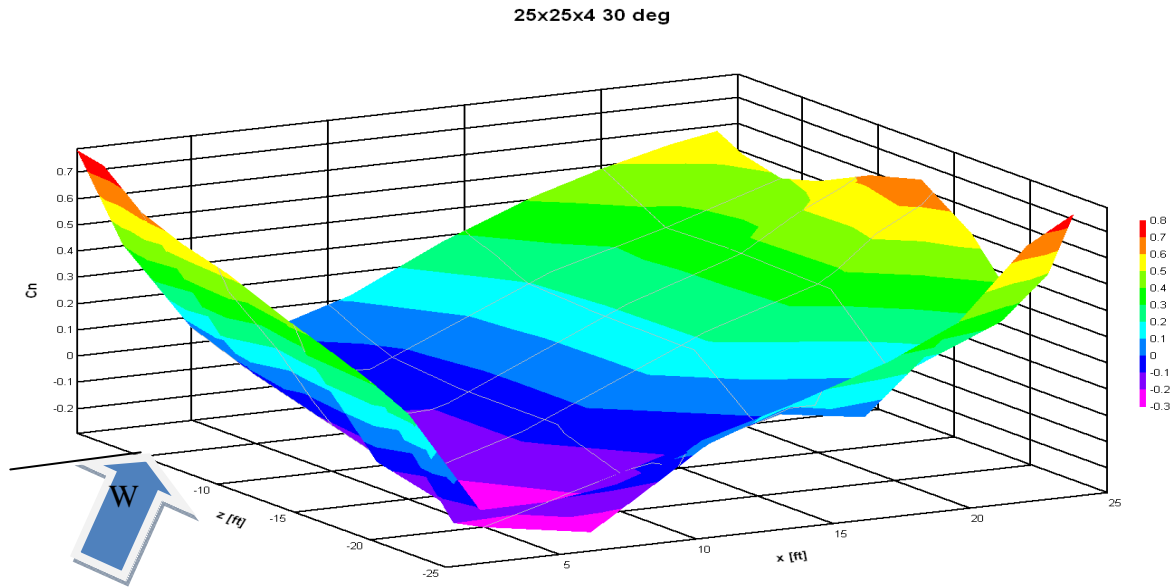


**Figure 7.12: Model #2, 3D contour plot for Cn values for wind at 30 degrees.**

Cn values for contour plan for Model #4, with dimensions of 7.6 m (25 ft) x 7.6 m (25 ft) x 1.22 m (4 ft) with the wind at 30 degrees are shown in Figure 7.13. Figure 7.14 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.8 to a minimum of -0.3. Those are the same extreme values computed for the wind incidence at 0°.

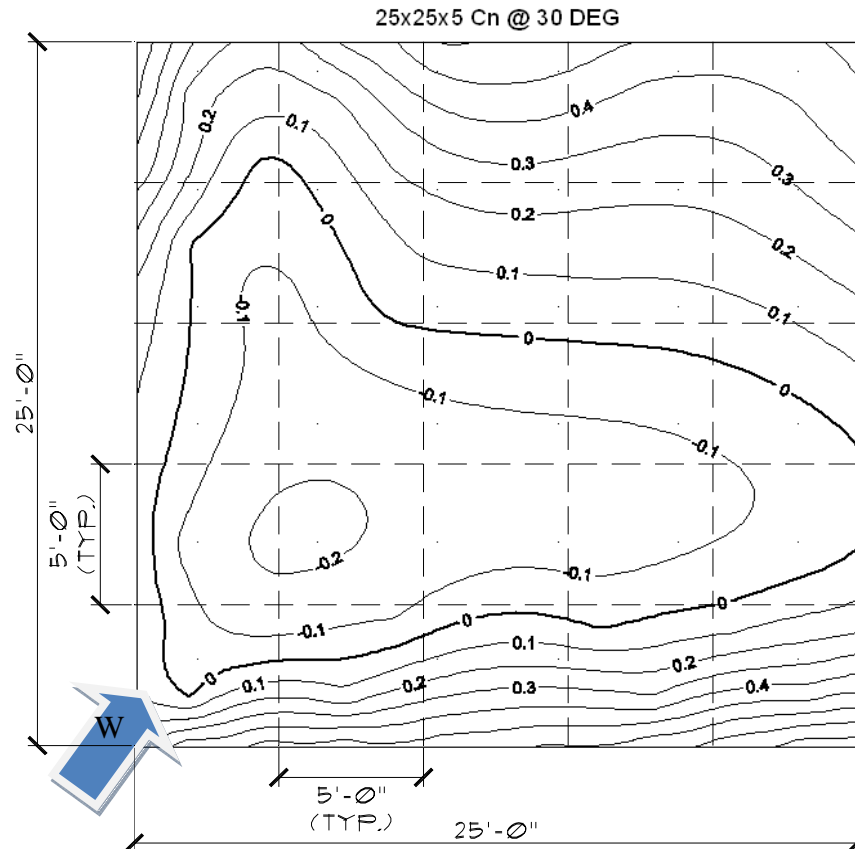


**Figure 7.13: Model #4, contour plan for Cn values for wind at 30 degrees.**

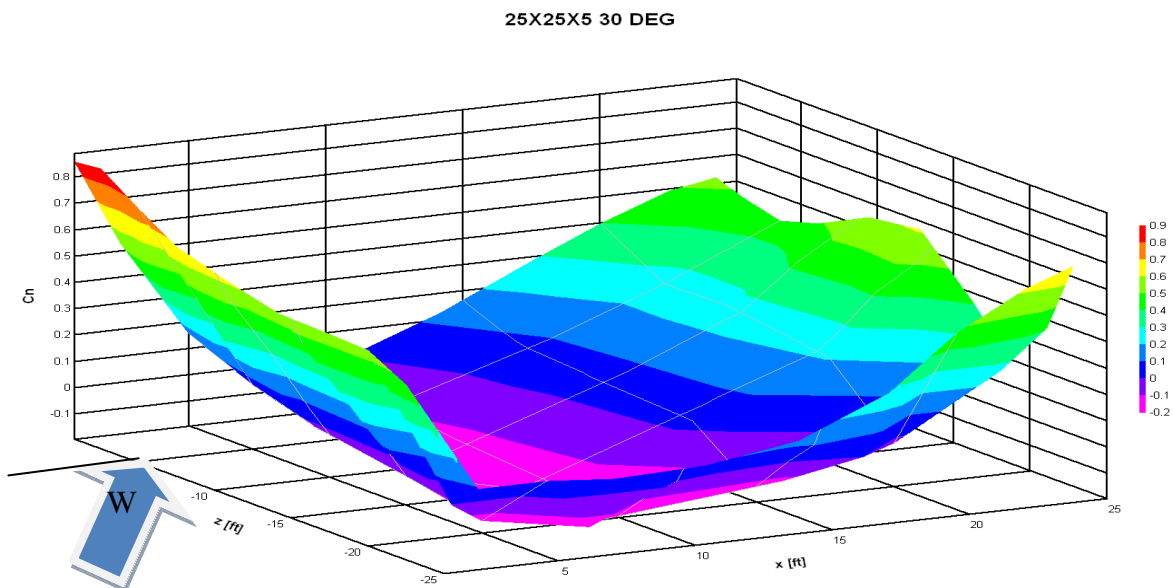


**Figure 7.14: Model #4, 3D contour plot for Cn values for wind at 30 degrees.**

Cn values for contour plan for Model #6, with dimensions of 7.6 m (25 ft) x 7.6 m (25 ft) x 1.52 m (5 ft) with the wind at 30 degrees are shown in Figure 7.15. Figure 7.16 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.9 to a minimum of -0.2, which are similar to the values computed for Model #5 (with 0°).

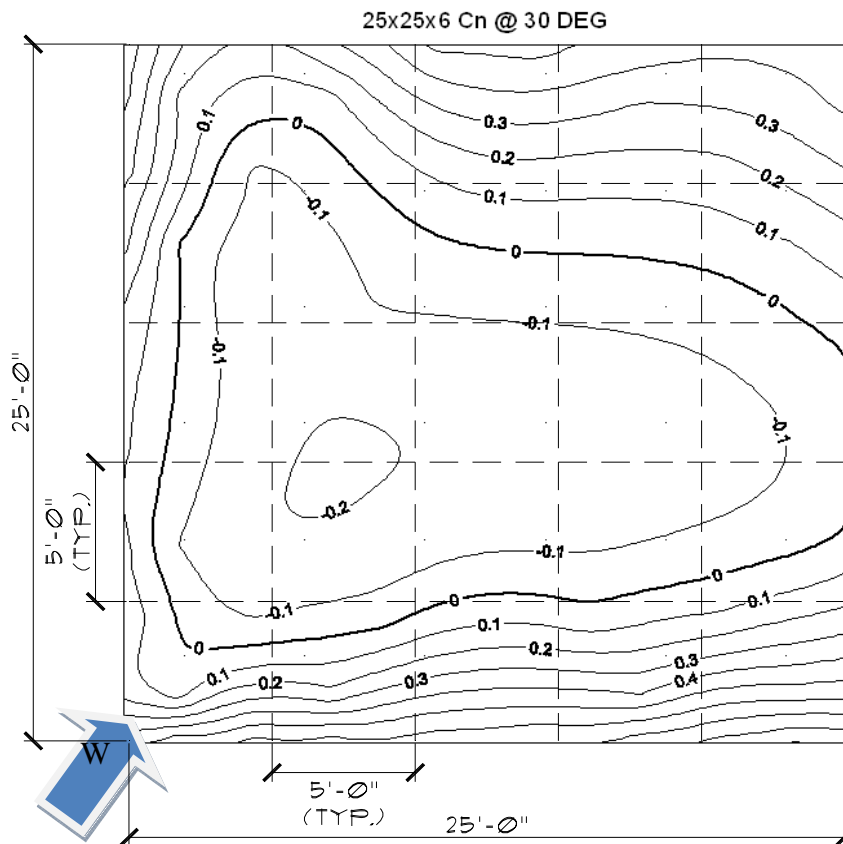


**Figure 7.15: Model #6, contour plan for Cn values for wind at 30 degrees.**

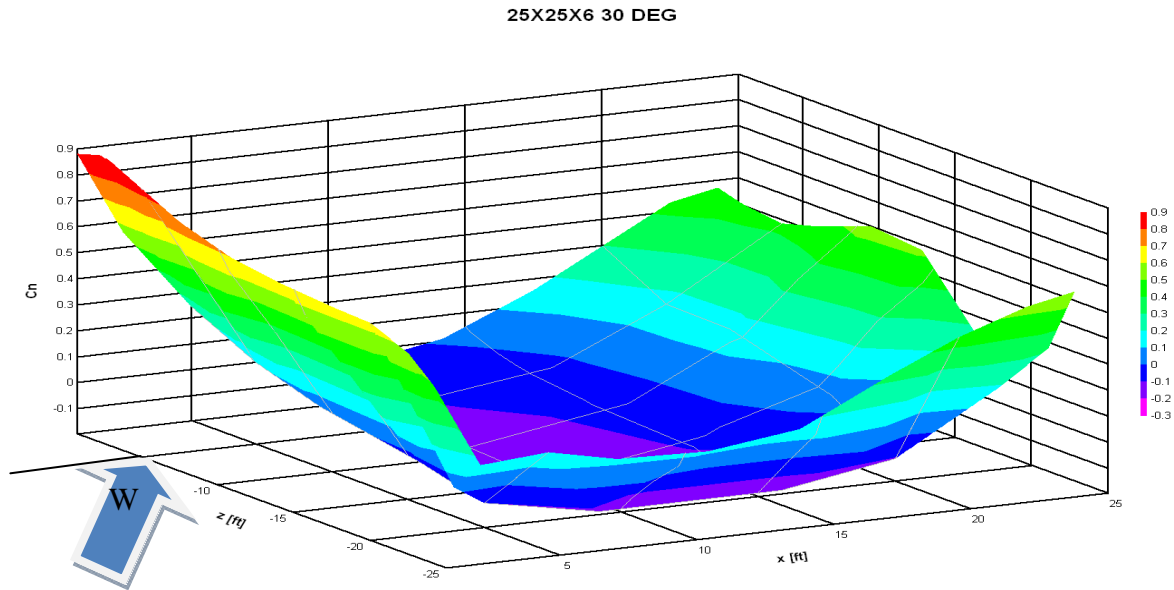


**Figure 7.16: Model #6, 3D contour plot for Cn values for wind at 30 degrees.**

Cn values for contour plan for Model #8, with dimensions of 7.6 m (25 ft) x 7.6 m (25 ft) x 1.82 m (6 ft) with the wind at 30 degrees are shown in Figure 7.17. Figure 7.18 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.9 to a minimum of -0.2, which are smaller than those computed at 0° (+1.0 and -0.2 respectively).

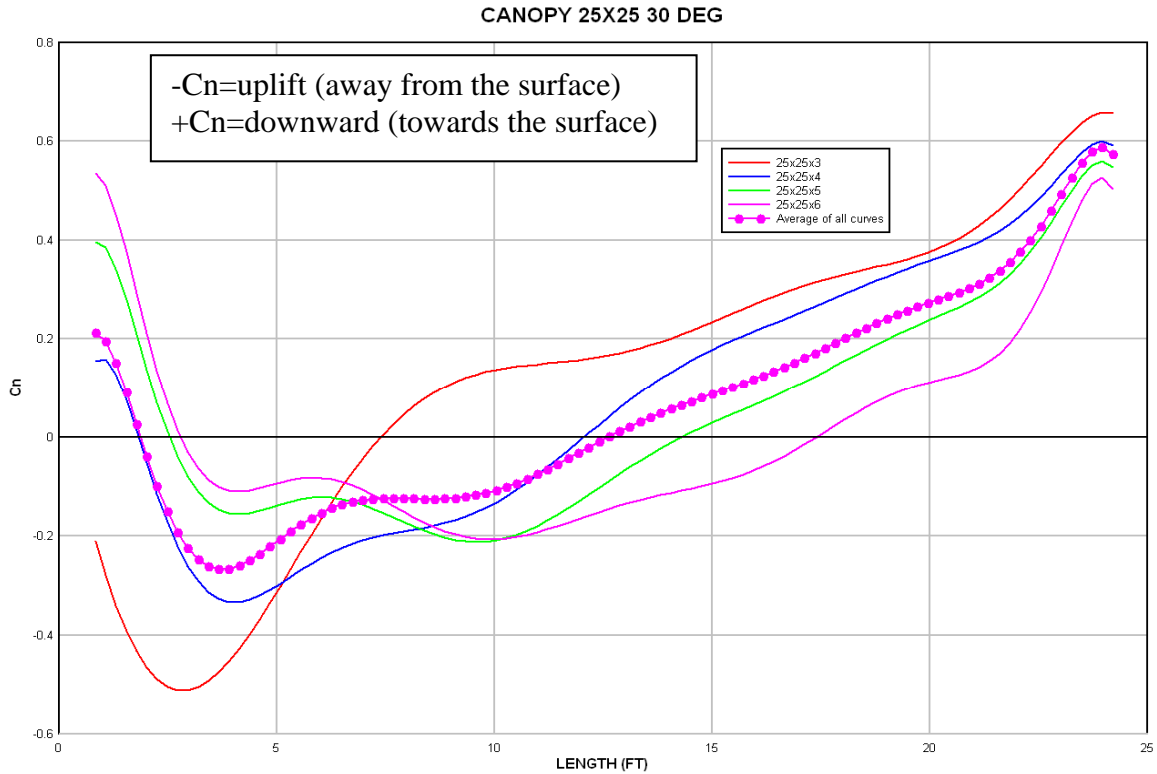


**Figure 7.17: Model #8, contour plan for Cn values for wind at 30 degrees.**



**Figure 7.18: Model #8, 3D contour plot for  $C_n$  values for wind at 30 degrees.**

Figure 7.19 shows four  $C_n$  graphs, one for each parapet height on the 7.6 m (25 ft) x 7.6 m (25 ft) open canopy. The 2D graph data was obtained at the center of the canopy width in the angle of the applied wind, in these case thirty (30) degrees from the principal axis direction. All parapet height exhibit on the graph a segment of positive  $C_n$  values (downward) on the first  $L/5$ , except for the parapet height of 0.91 m (3 ft) and 1.22 m (4 ft), which were all negative  $C_n$  values. After the uplift segment, an uplift pressure length of  $L/3$  followed a downward for the rest of the horizontal distance. As said before, the extreme values shown by  $0^\circ$  and  $30^\circ$  are similar, but the actual distributions are the same, as can be seen from comparisons of Figures 7.10 and 7.19.



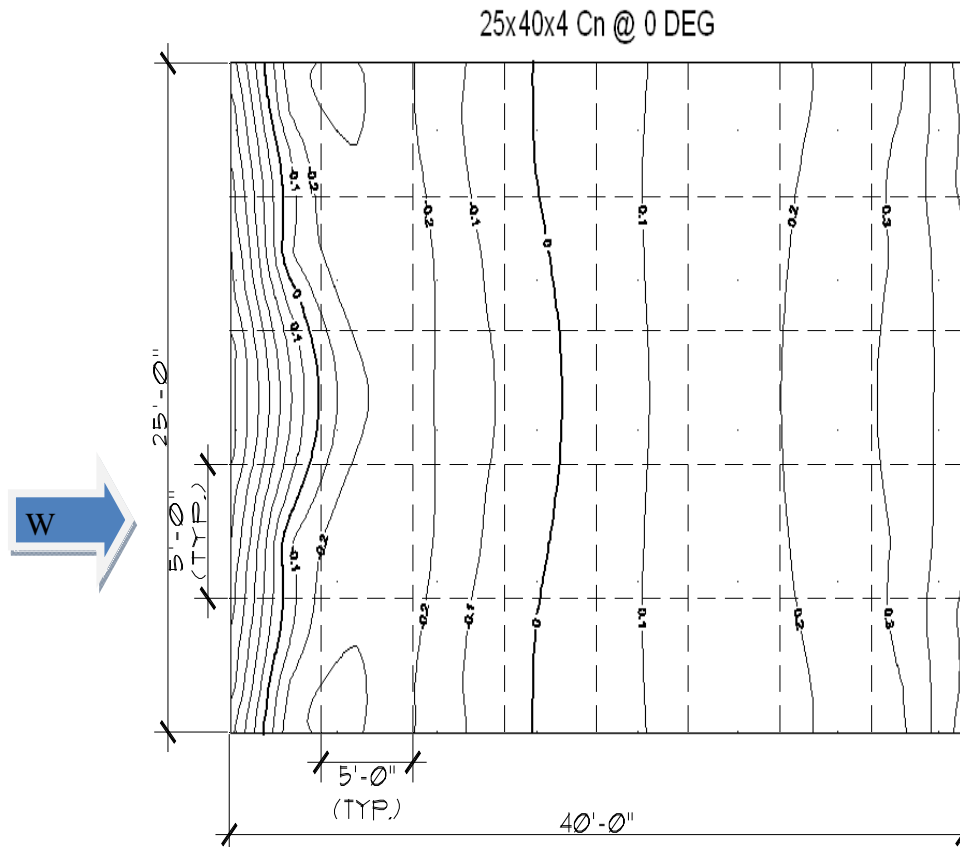
**Figure 7.19: 2D graph of Cn values for all 7.6 m (25 ft) x 7.6 m (25 ft) @ 30 degrees.**

## 7.4 Open canopy models, 7.6 m (25 ft) x 12.2 m (40 ft) at 0 degrees

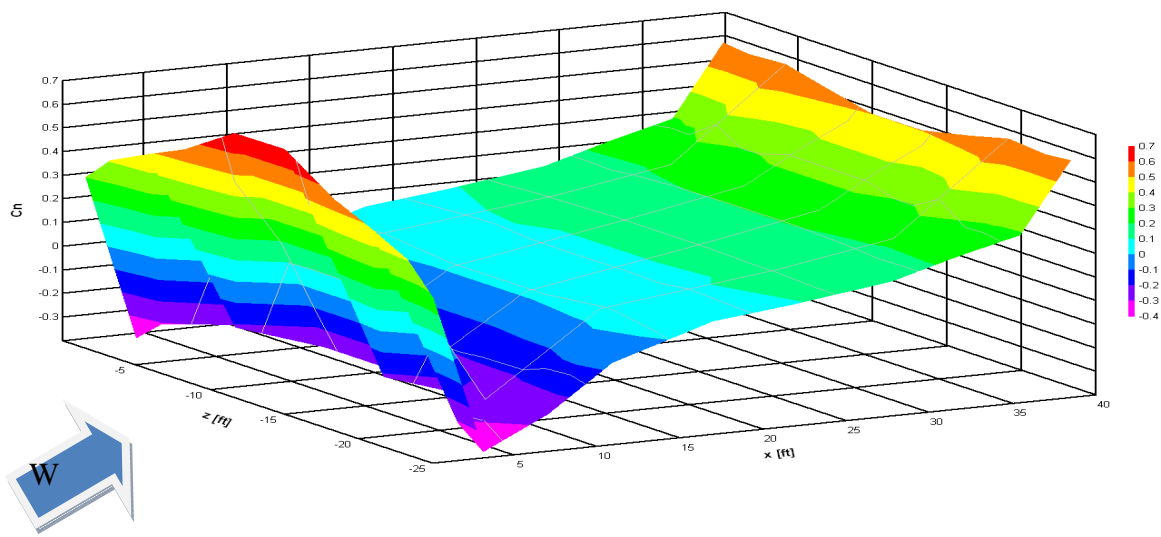
The previous configurations were representative of square canopies, and the influence of having a rectangular configuration is investigated next. Again, the influence of parapet height is taken into account. Cn values for contour plan for Model #9, with dimensions of 7.6 m (25 ft) x 12.2 m (40 ft) x 0.91 m (3 ft) with the wind at 0 degrees are shown in Figure 7.20. Figure 7.21 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.7 to a minimum of -0.4. The values are slightly higher than in the square configuration #1 (+0.6 and -0.3 respectively).



view of the  $C_n$  values of the canopy. Values range from a maximum of +0.7 to a minimum of -0.3. In the square configuration, we reported extreme values of +0.8 and -0.3.

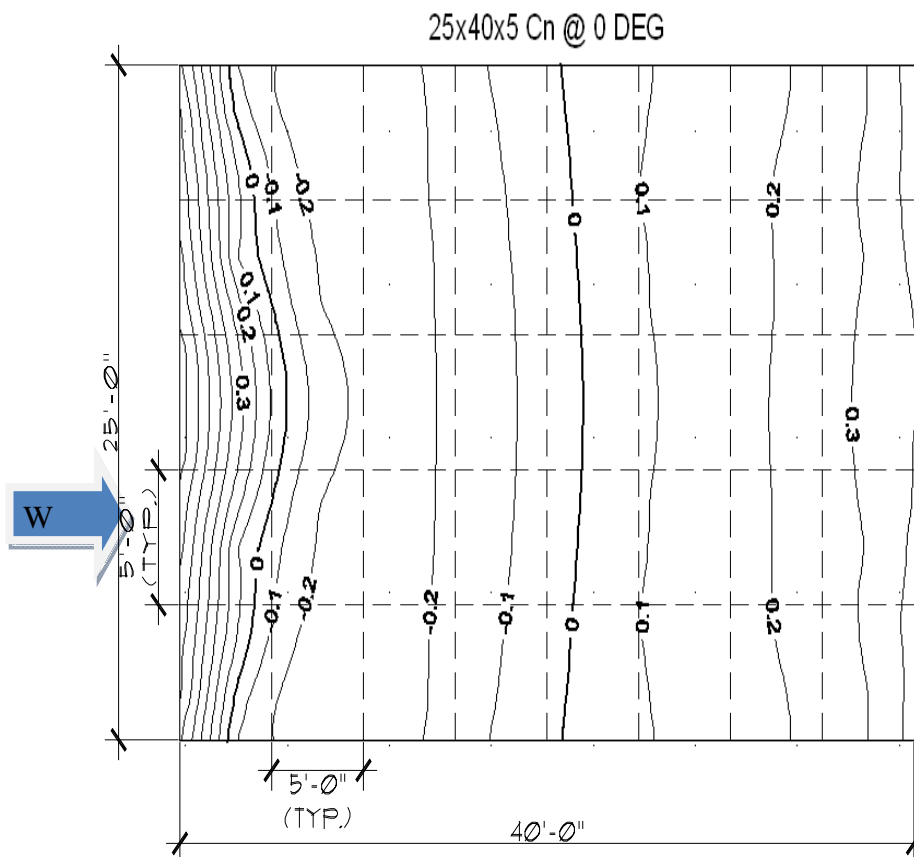


**Figure 7.22: Model #11, contour plan for  $C_n$  values for wind at 0 degrees.**  
25x40x4 0 deg

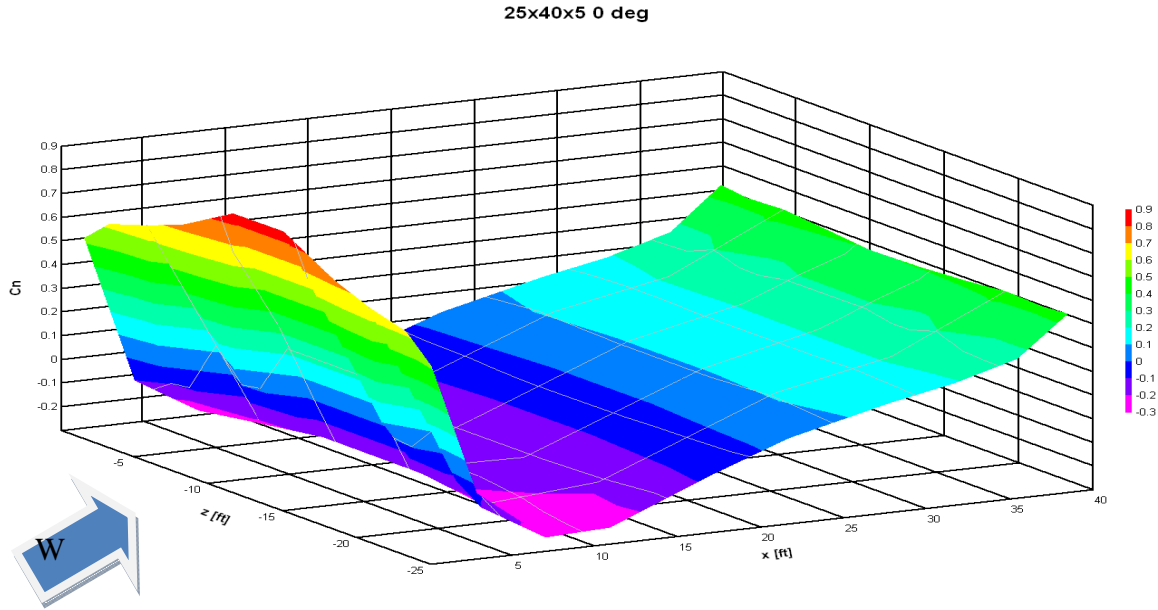


**Figure 7.23: Model #11, 3D contour plot for  $C_n$  values for wind at 0 degrees.**

Cn values for contour plan for Model #13, with dimensions of 7.6 m (25 ft) x 12.2 m (40 ft) x 1.52 m (5 ft) with the wind at 0 degrees are shown in Figure 7.24. Figure 7.25 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.9 to a minimum of -0.3. The same extreme values were obtained in the square canopy.

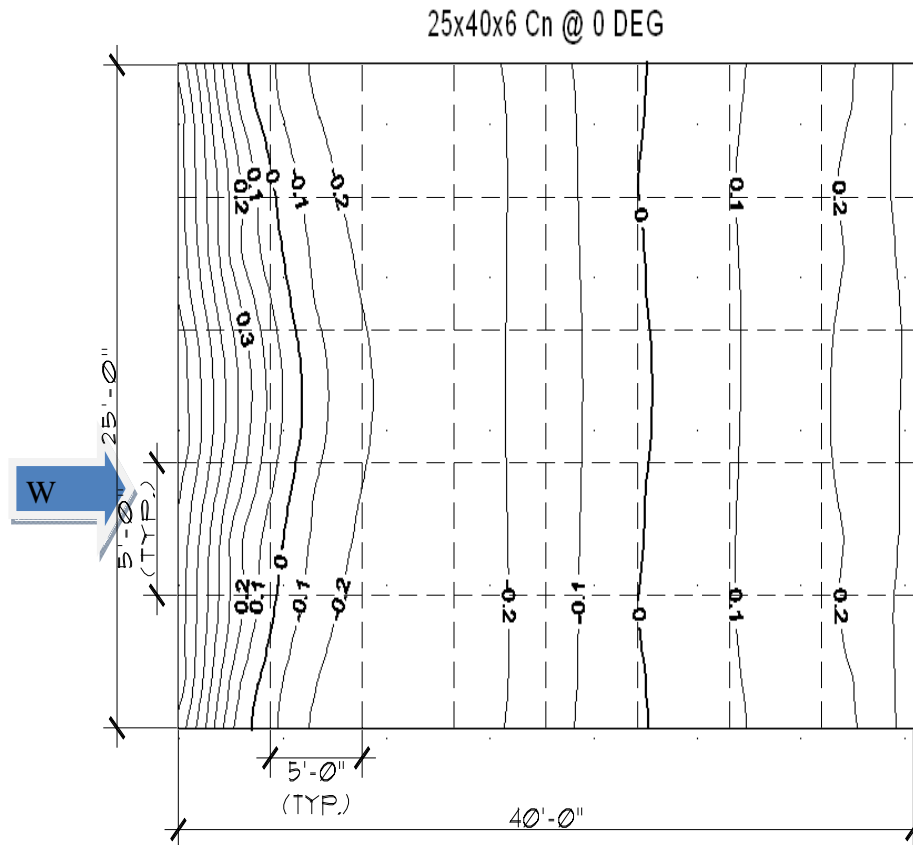


**Figure 7.24: Model #13, contour plan for Cn values for wind at 0 degrees.**

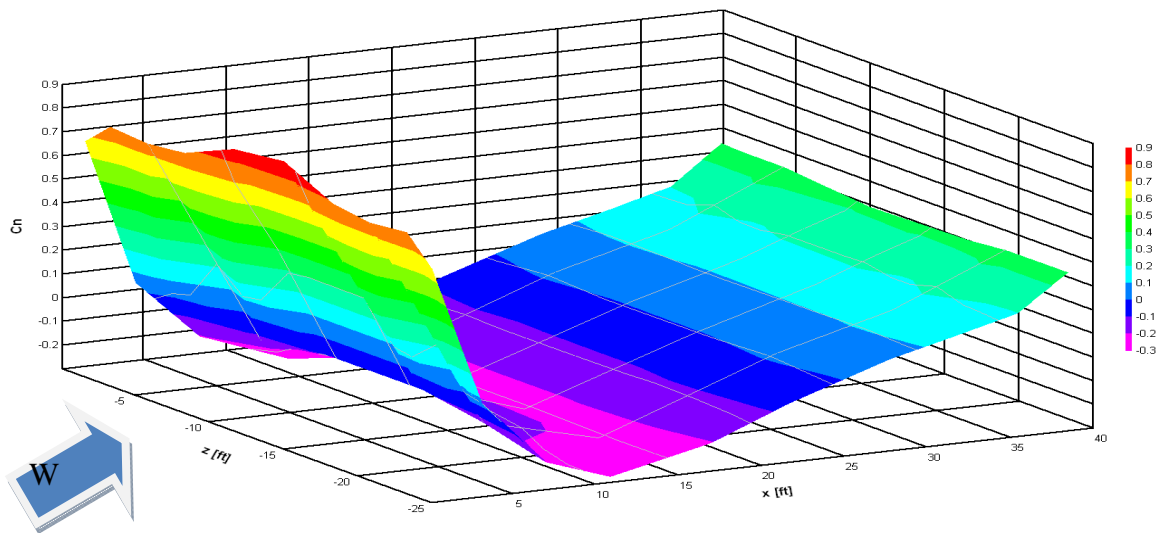


**Figure 7.25: Model #13, 3D contour plot for Cn values for wind at 0 degrees.**

Cn values for contour plan for Model #15, with dimensions of 7.6 m (25 ft) x 12.2 m (40 ft) x 1.82 m (6 ft) with the wind at 0 degrees are shown in Figure 7.26. Figure 7.27 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.9 to a minimum of -0.3. As a reference value, values of +1.0 and -0.2 resulted for the square configuration.

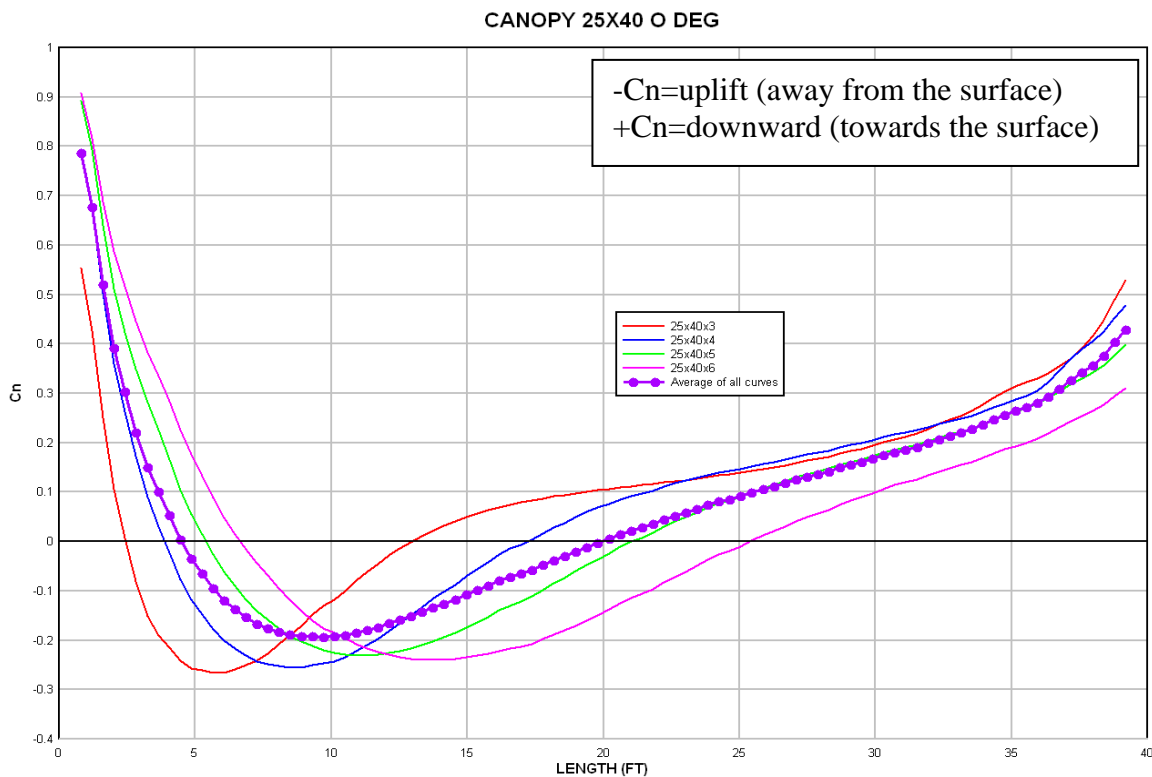


**Figure 7.26: Model #15, contour plan for Cn values for wind at 0 degrees.**  
25x40x6 0 deg



**Figure 7.27: Model #15, 3D contour plot for Cn values for wind at 0 degrees.**

Figure 7.28 shows four Cn graphs, one for each parapet height on the 7.6 m (25 ft) x 12.2 m (40 ft) open canopy. The 2D graph data was obtained from the middle of the canopy width in the principal axis direction. All parapet heights exhibit a segment of positive Cn values (downward) for the first L/5 of the horizontal length. After the downward segment, an uplift pressure length of L/3 occurs, followed by a downward segment for the rest of the horizontal distance.

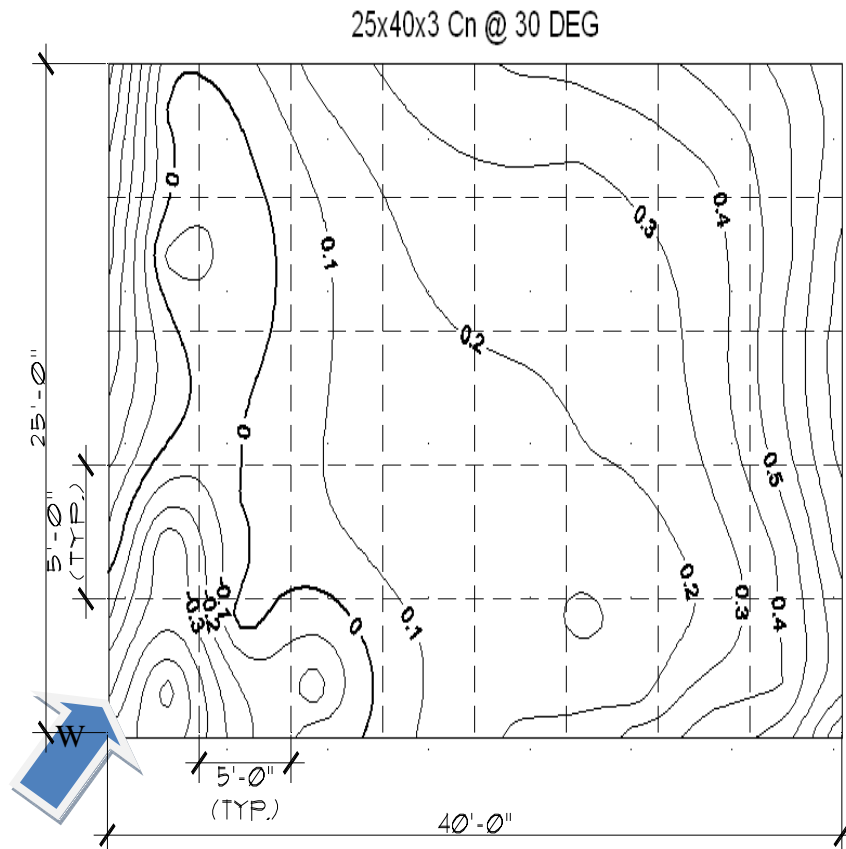


**Figure 7.28: 2D graph of Cn values for all 7.6 m (25 ft) x 12.2 m (40 ft) @ 0 degrees.**

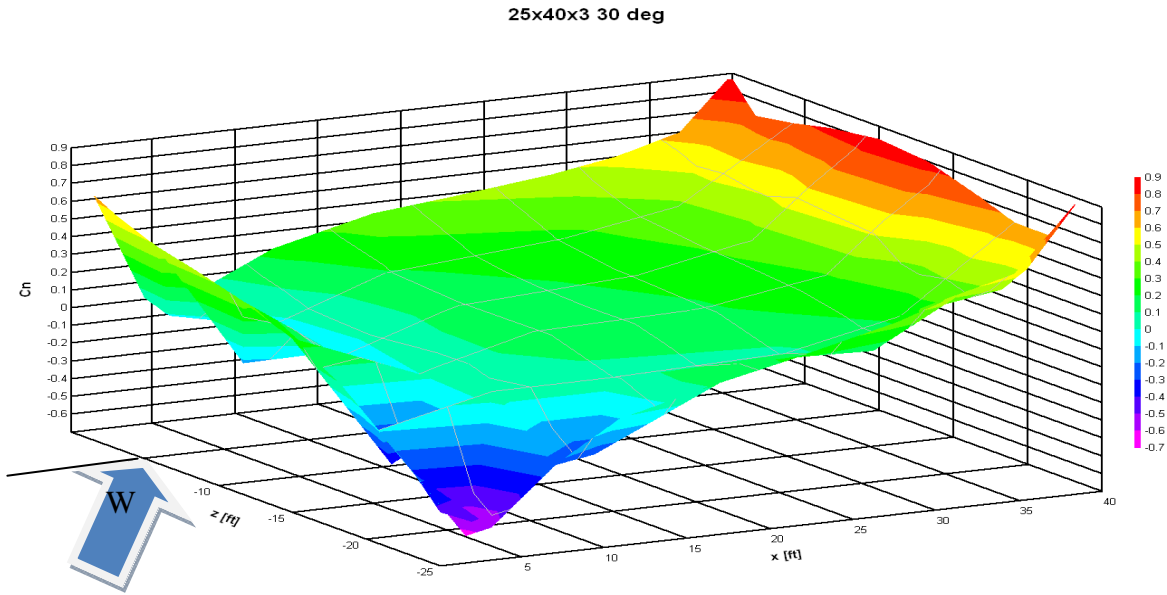
## 7.5 Open canopy models, 7.6 m (25 ft) x 12.2 m (40 ft) at 30 degrees

Cn values for contour plan for Model #10, with dimensions of 7.6 m (25 ft) x 12.2 m (40 ft) x 0.91 m (3 ft) with the wind at 30 degrees are shown in Figure 7.29. Figure 7.30 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.9 to a minimum of -

0.7. The values are slightly higher than in the square configuration #2 (+0.7 and -0.4 respectively).

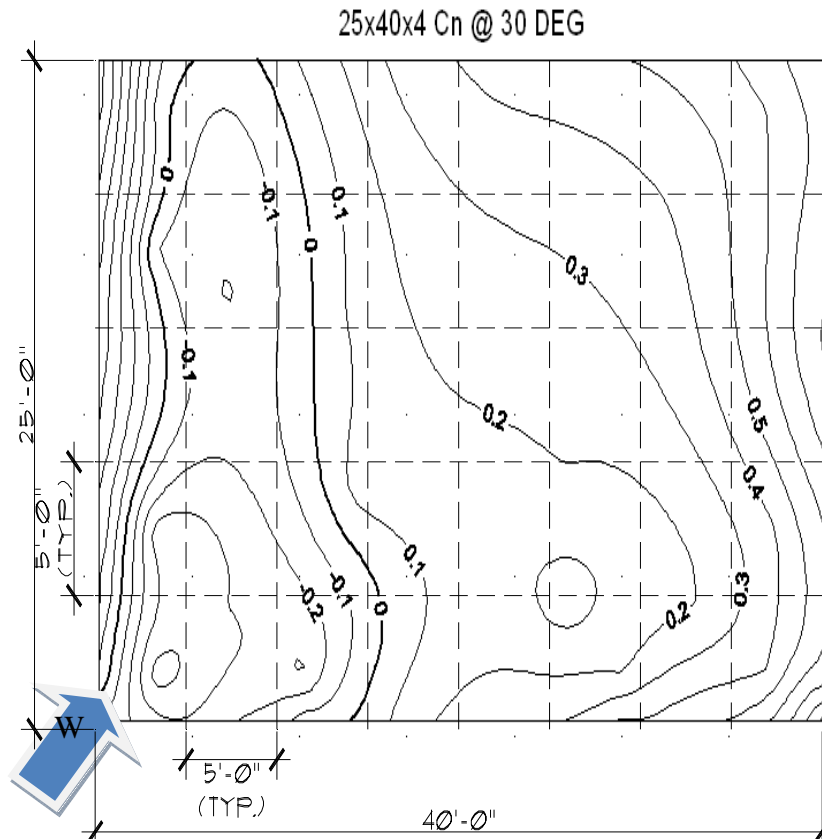


**Figure 7.29: Model #10, contour plan for Cn values for wind at 30 degrees.**

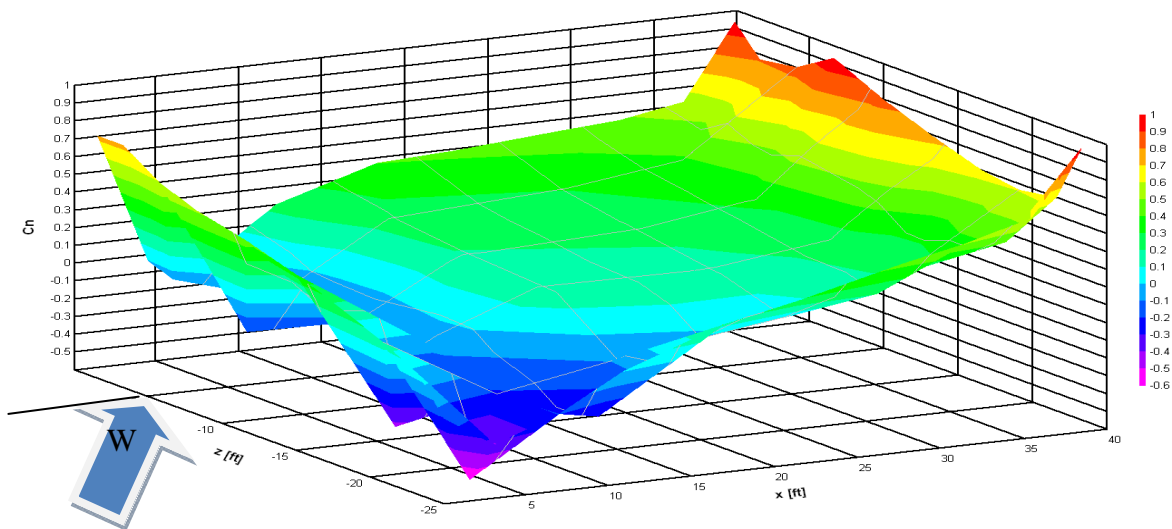


**Figure 7.30: Model #10, 3D contour plot for  $C_n$  values for wind at 30 degrees.**

$C_n$  values for contour plan for Model #12, with dimensions of 7.6 m (25 ft) x 12.2 m (40 ft) x 1.22 m (4 ft) with the wind at 30 degrees are shown in Figure 7.31. Figure 7.32 shows a 3D view of the  $C_n$  values of the canopy. Values range from a maximum of +1.0 to a minimum of -0.6. The values are slightly higher than in the square configuration Model #4 (+0.8 and -0.3 respectively).

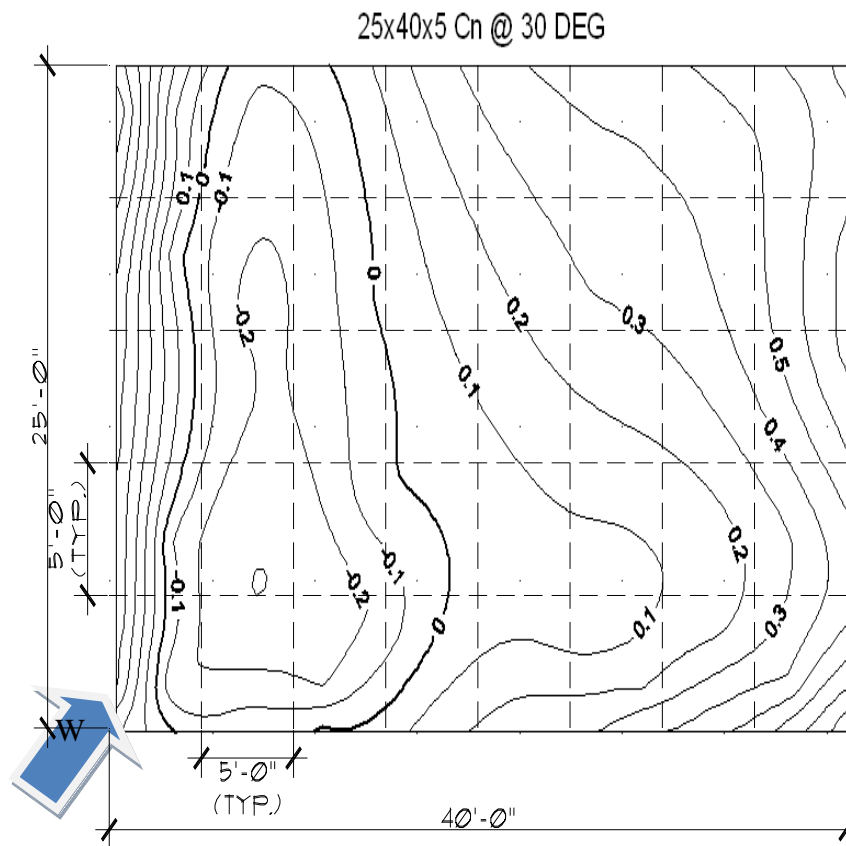


**Figure 7.31: Model #12, contour plan for Cn values for wind at 30 degrees.**  
25x40x4 30 deg

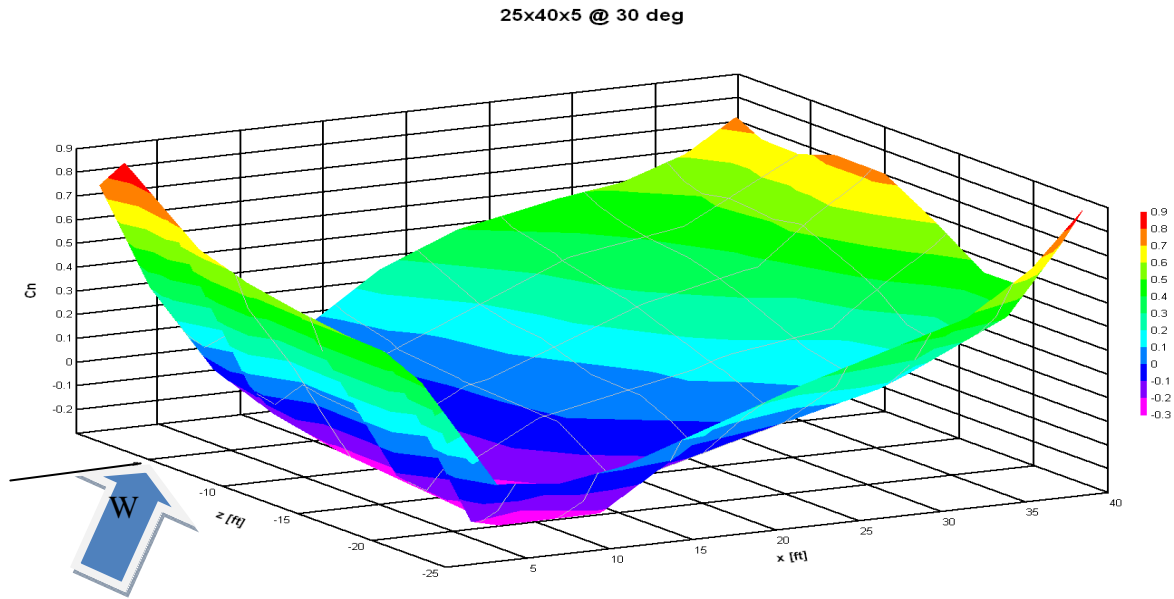


**Figure 7.32: Model #12, 3D contour plot for Cn values for wind at 30 degrees.**

Cn values for contour plan for Model #14, with dimensions of 7.6 m (25 ft) x 12.2 m (40 ft) x 1.52 m (5 ft) with the wind at 30 degrees are shown in Figure 7.33. Figure 7.34 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.9 to a minimum of -0.3. The values are very similar compared to the square configuration Model #6 (+0.9 and -0.2 respectively).

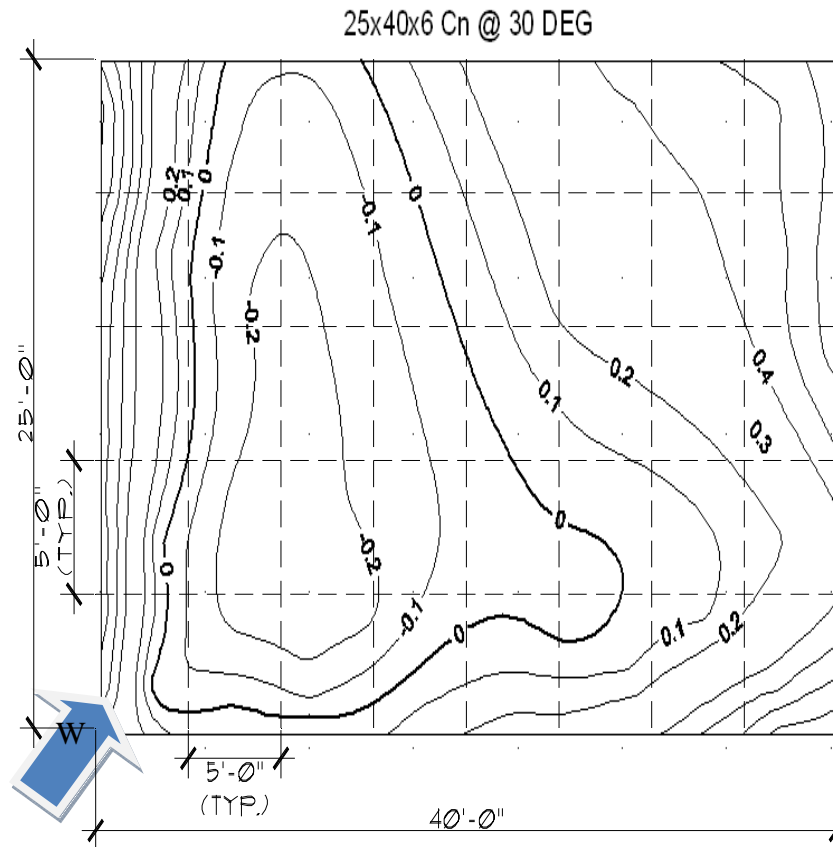


**Figure 7.33: Model #14, contour plan for Cn values for wind at 30 degrees.**

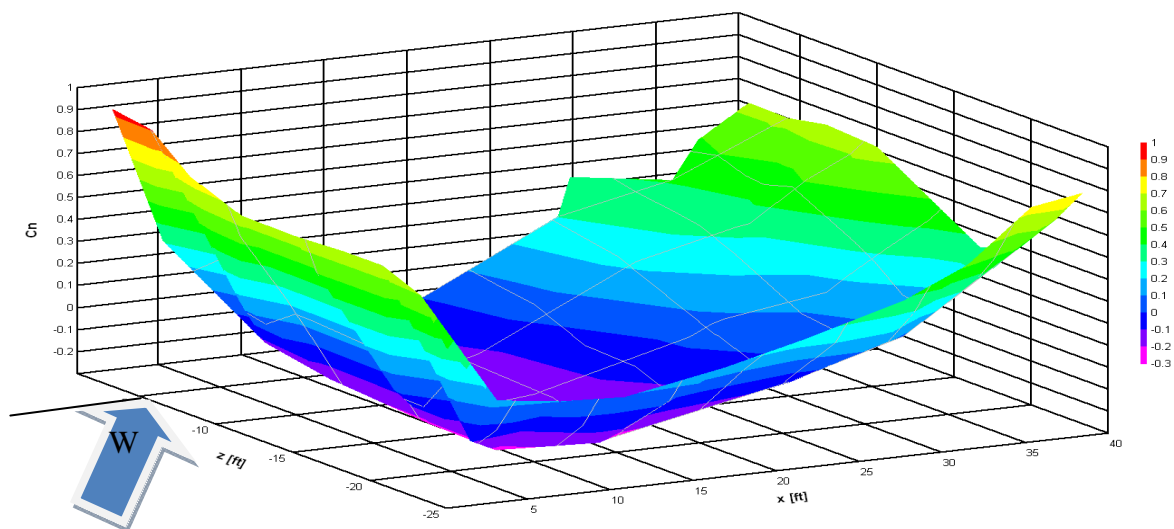


**Figure 7.34: Model #14, 3D contour plot for  $C_n$  values for wind at 30 degrees.**

$C_n$  values for contour plan for Model #16, with dimensions of 7.6 m (25 ft) x 12.2 m (40 ft) x 1.82 m (6 ft) with the wind at 30 degrees are shown in Figure 7.35. Figure 7.36 shows a 3D view of the  $C_n$  values of the canopy. Values range from a maximum of +1.0 to a minimum of -0.3. The values are slightly higher than in the square configuration #8 (+0.9 and -0.2 respectively).

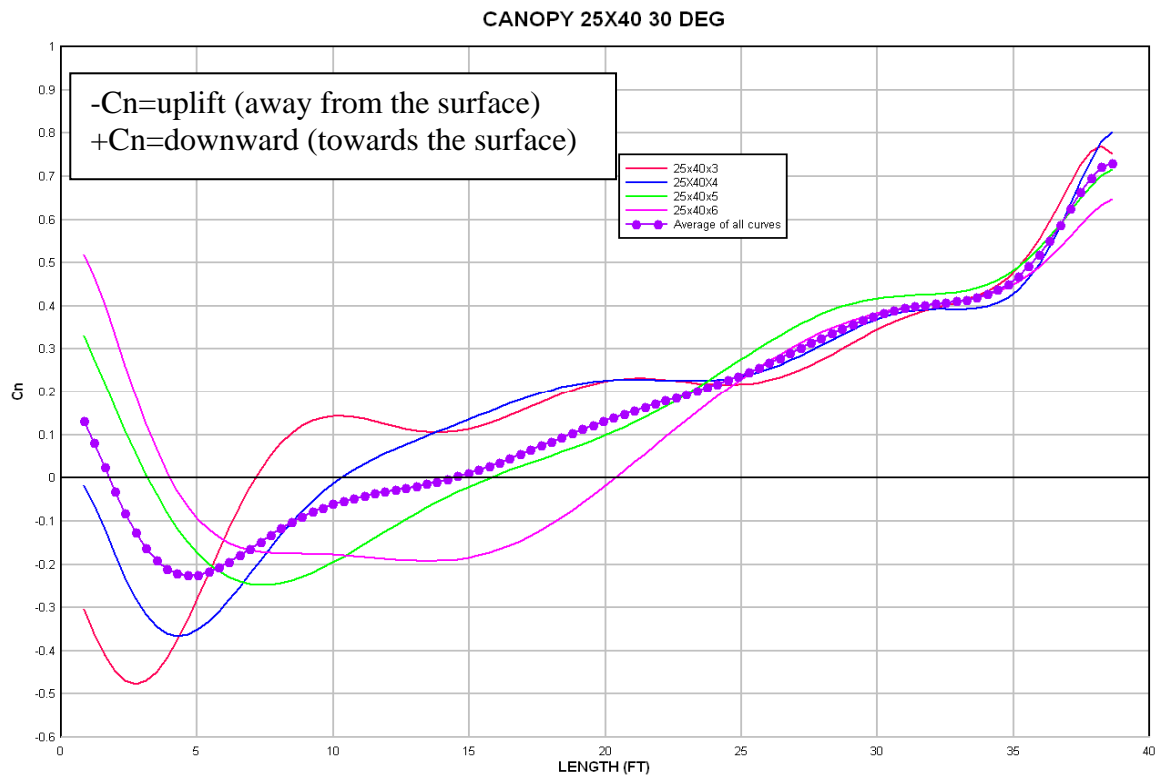


**Figure 7.35: Model #16, contour plan for Cn values for wind at 30 degrees.**  
25x40x6 @ 30 deg



**Figure 7.36: Model #16, 3D contour plot for Cn values for wind at 30 degrees.**

Figure 7.37 shows four Cn graphs, one for each parapet height on the 7.6 m (25 ft) x 12.2 m (40 ft) open canopy. The 2D graph data was obtained from the middle of the canopy width in the angle of the applied wind, in these case thirty (30) degrees from the principal axis direction. All parapet height exhibits on the graph a segment of positive and negative Cn values. Depending on the parapet height, downward and uplift pressure is reflected. This is shown on the first L/5, except the parapet height of 0.91 m (3 ft) and 1.22 m (4 ft), which were all negative Cn values. After the downward and uplift segment, an uplift pressure length of L/3 occurs followed by a downward pressure for the rest of the horizontal distance.



**Figure 7.37: Graph of Cn values for all 7.6 m (25 ft) x 12.2 m (40 ft) @ 30 degrees.**

## 7.6 Open canopy models, 7.6 m (25 ft) x15.2 m (50 ft) at 0 degrees

An even longer canopy in the transverse direction is investigated in this section, in which the length is increased to 50 ft. Cn values for contour plan for Model #17, with dimensions of 7.6 m (25 ft) x 15.2 m (50 ft) x 0.91 m (3 ft) with the wind at 0 degrees are shown in Figure 7.38. Figure 7.39 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.7 to a minimum of -0.4. (Same as in Model #9).

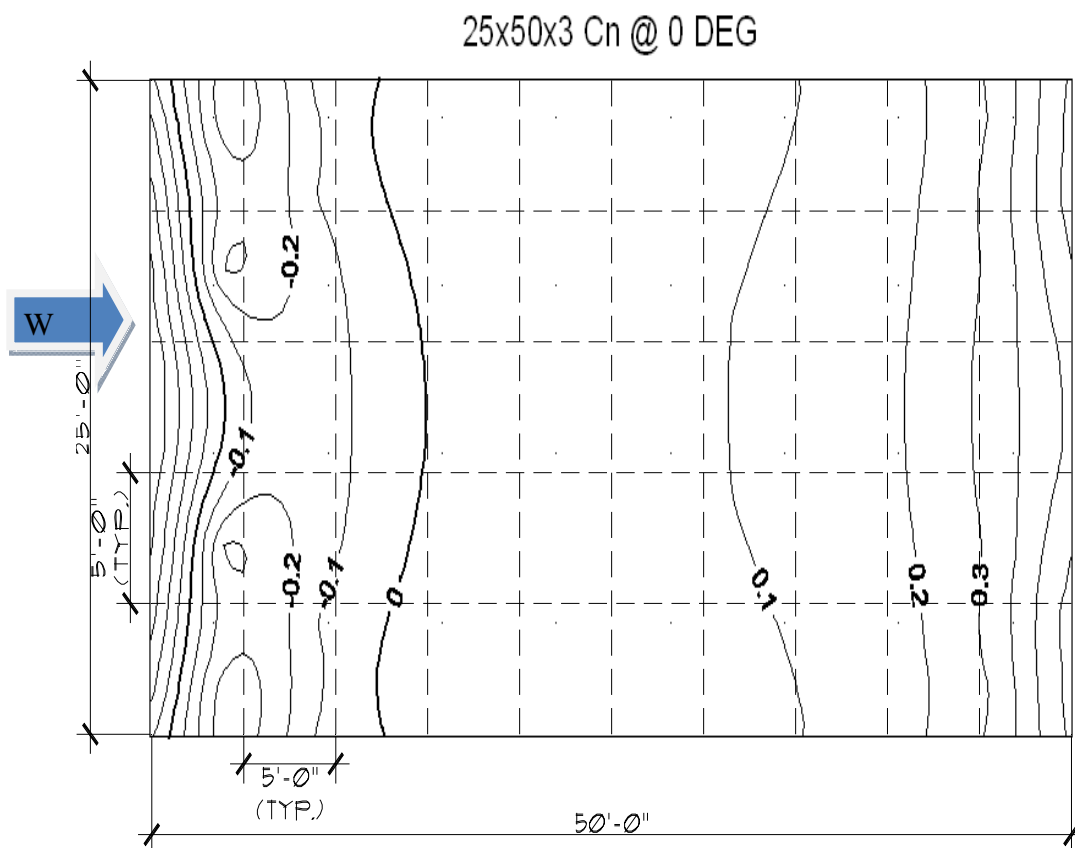
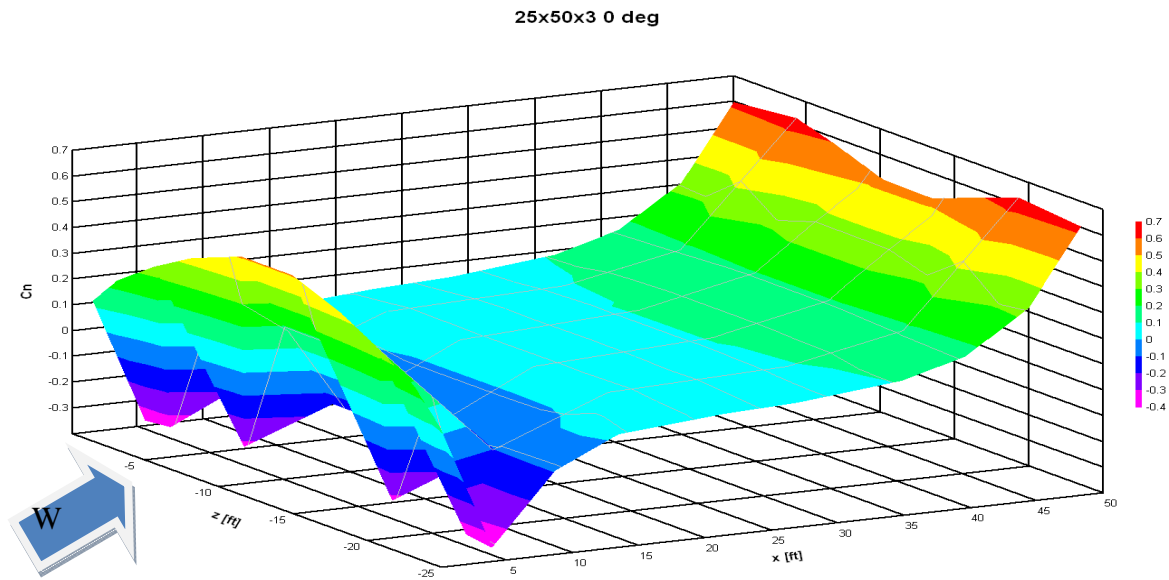
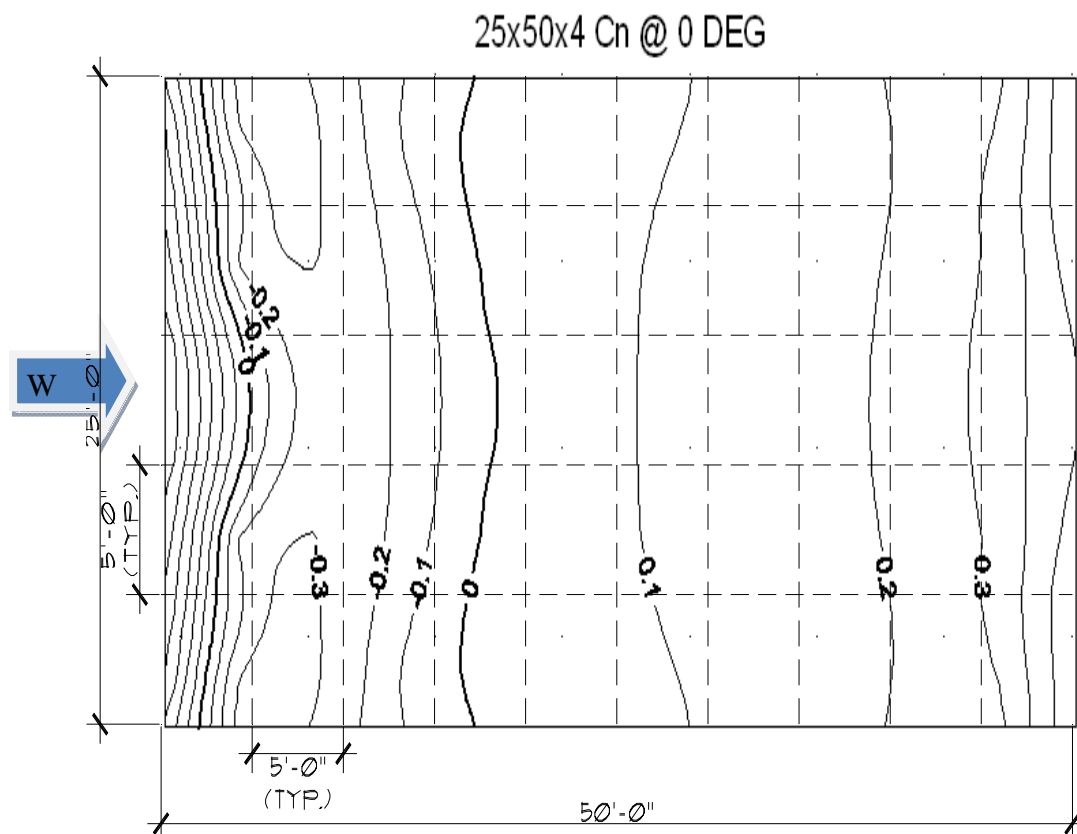


Figure 7.38: Model #17, contour plan for Cn values for wind at 0 degrees.

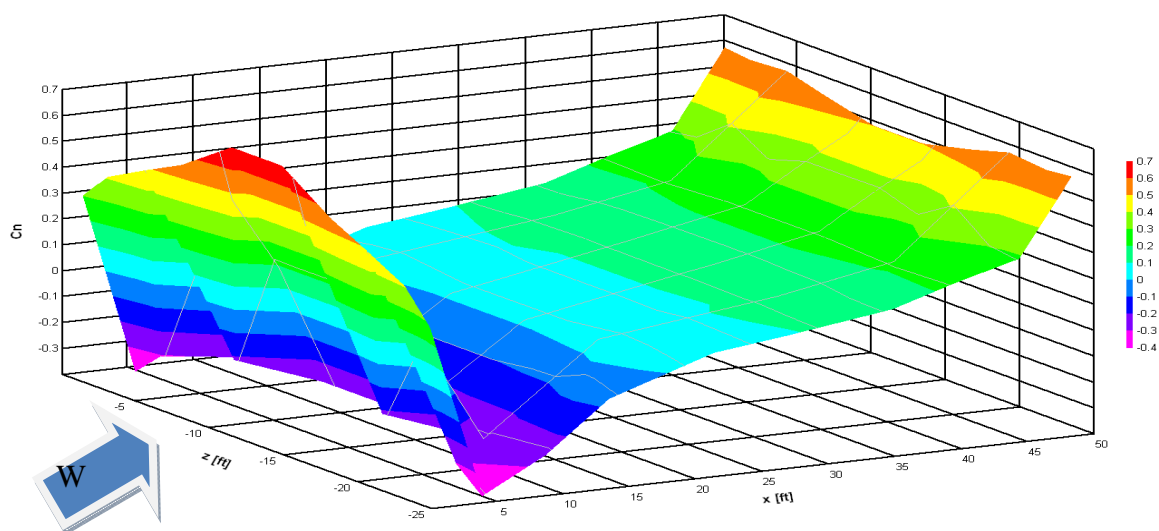


**Figure 7.39: Model #17, 3D contour plot for Cn values for wind at 0 degrees.**

Cn values for contour plan for Model #19, with dimensions of 7.6 m (25 ft) x 15.2 m (50 ft) x 1.22 m (4 ft) with the wind at 0 degrees are shown in Figure 7.40. Figure 7.41 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.7 to a minimum of -0.4. (Similar to Model #11).

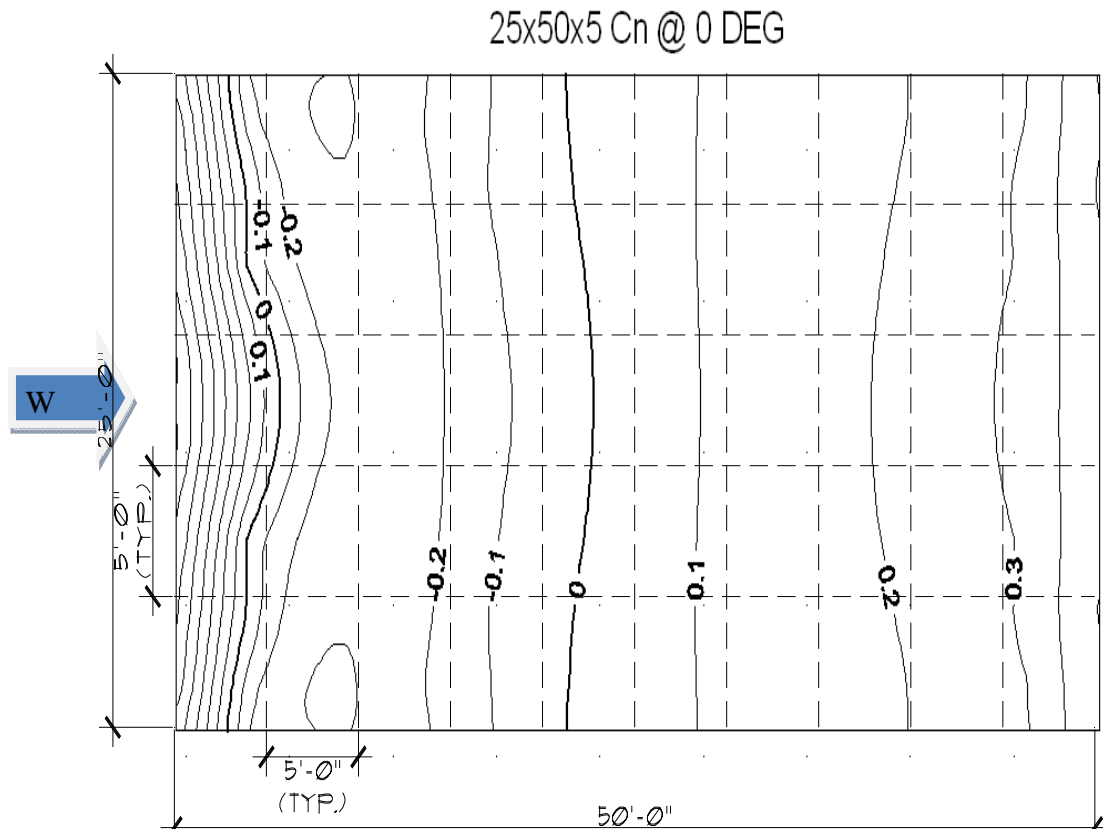


**Figure 7.40: Model #19, contour plan for Cn values for wind at 0 degrees.**  
25x50x4 0 deg

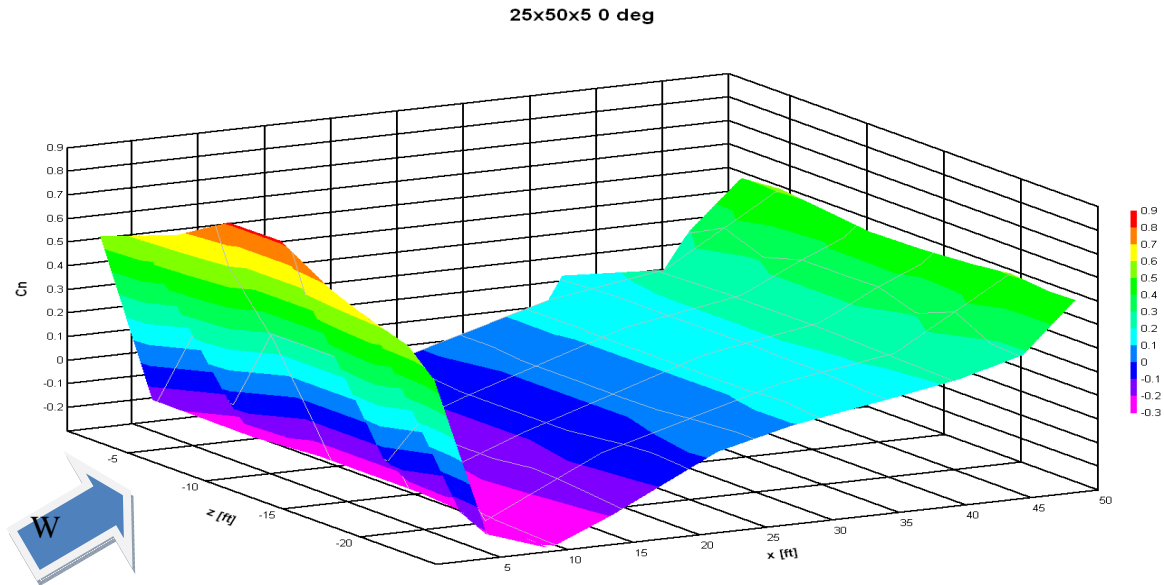


**Figure 7.41: Model #19, 3D contour plot for Cn values for wind at 0 degrees.**

Cn values for contour plan for Model #21, with dimensions of 7.6 m (25 ft) x 15.2 m (50 ft) x 1.52 m (5 ft) with the wind at 0 degrees are shown in Figure 7.42. Figure 7.43 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +0.9 to a minimum of -0.3. (Similar to Model #13).

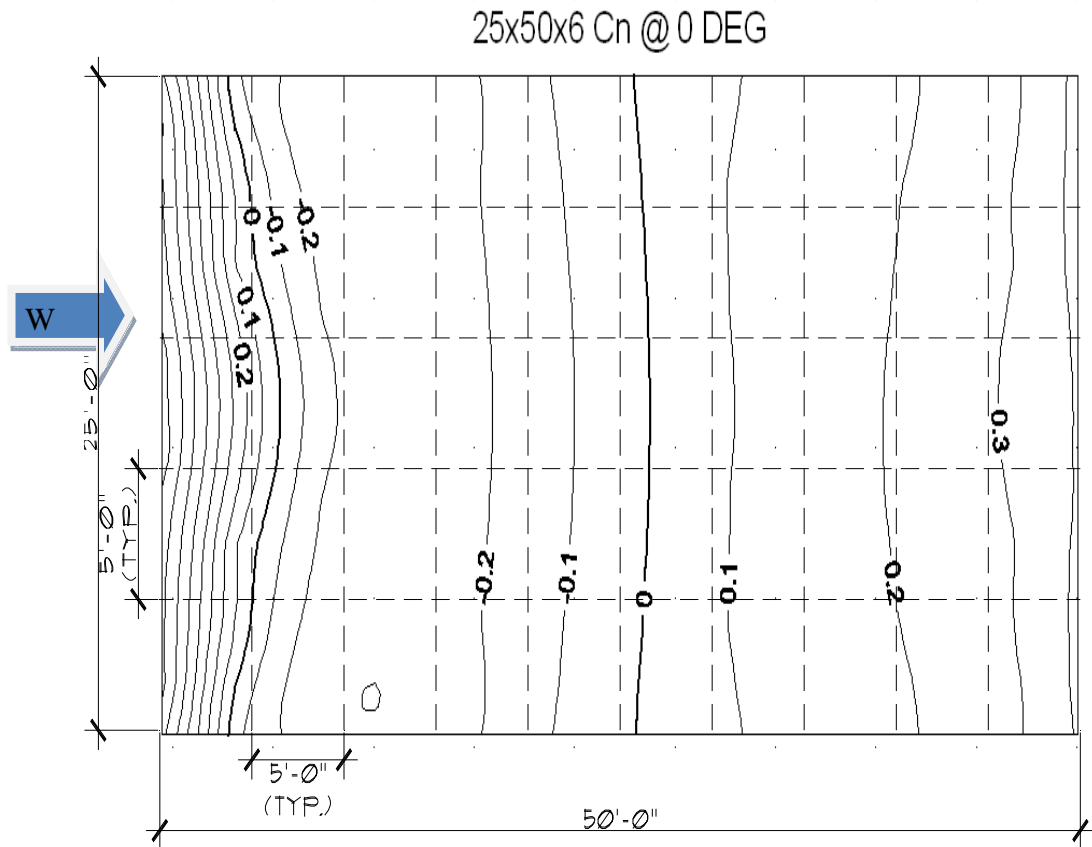


**Figure 7.42: Model #21, contour plan for Cn values for wind at 0 degrees.**

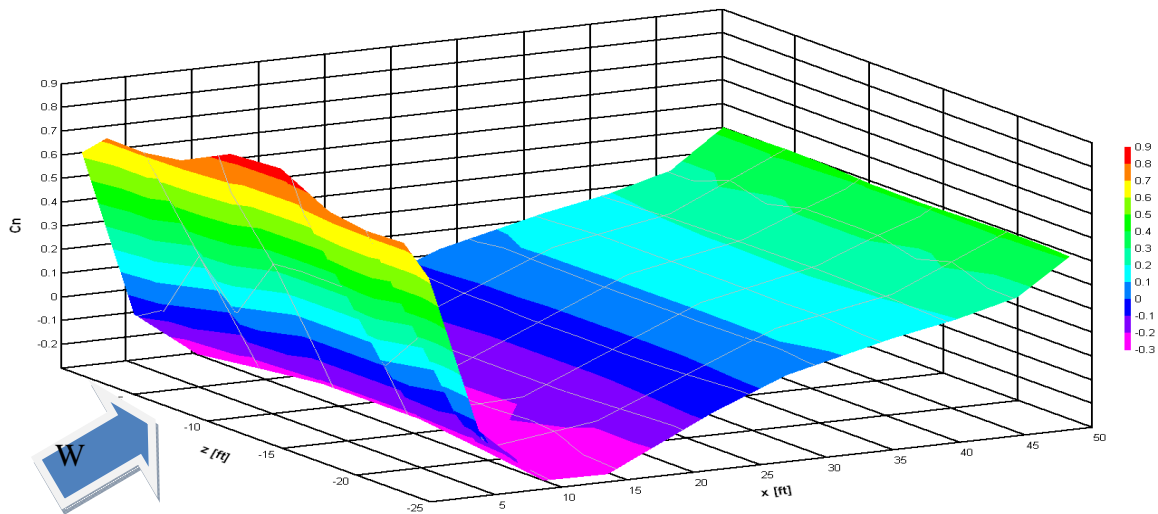


**Figure 7.43: Model #21, 3D contour plot for  $C_n$  values for wind at 0 degrees.**

$C_n$  values for contour plan for Model #23, with dimensions of 7.6 m (25 ft) x 15.2 m (50 ft) x 1.82 m (6 ft) with the wind at 0 degrees are shown in Figure 7.44. Figure 7.45 shows a 3D view of the  $C_n$  values of the canopy. Values range from a maximum of +0.9 to a minimum of -0.3. (Identical values as in Model #15).

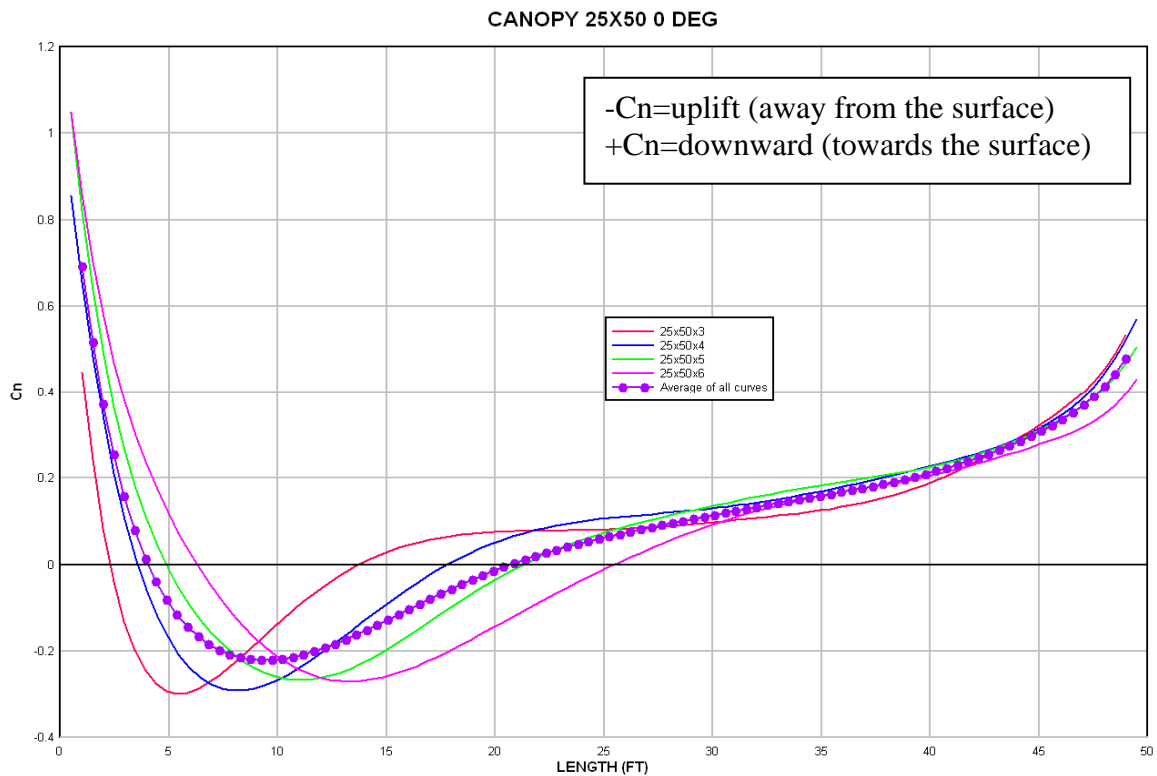


**Figure 7.44: Model #23, contour plan for Cn values for wind at 0 degrees.**  
25x50x6 0 deg



**Figure 7.45: Model #23, 3D contour plot for Cn values for wind at 0 degrees.**

Figure 7.46 shows four Cn graphs, one for each parapet height on the 7.6 m (25 ft) x 15.2 m (50 ft) open canopy. The 2D graph data was obtained from the middle of the canopy width in the principal axis direction. All parapet height exhibits on the graph a segment of positive Cn values (downward) for the first L/5 of the horizontal length. After the downward segment, an uplift pressure length of L/3 occurs, followed by a downward for the rest of the horizontal distance. The results computed for the canopy with 50 ft in the transverse direction are almost identical to those computed for the 40 ft canopy.



**Figure 7.46: 2D graph of Cn values for all 7.6 m (25 ft) x 15.2 m (50 ft) @ 0 degrees.**

## 7.7 Open canopy models, 7.6 m (25 ft) x15.2 m (50 ft) at 30 degrees

Cn values for contour plan for Model #18, with dimensions of 7.6 m (25 ft) x 15.2 m (50 ft) x 0.91 m (3 ft) with the wind at 30 degrees are shown in Figure 7.47. Figure 7.48 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +1.0 to a minimum of -0.6. The values are similar compared to the configuration Model #10 (+1.0 and -0.7 respectively).

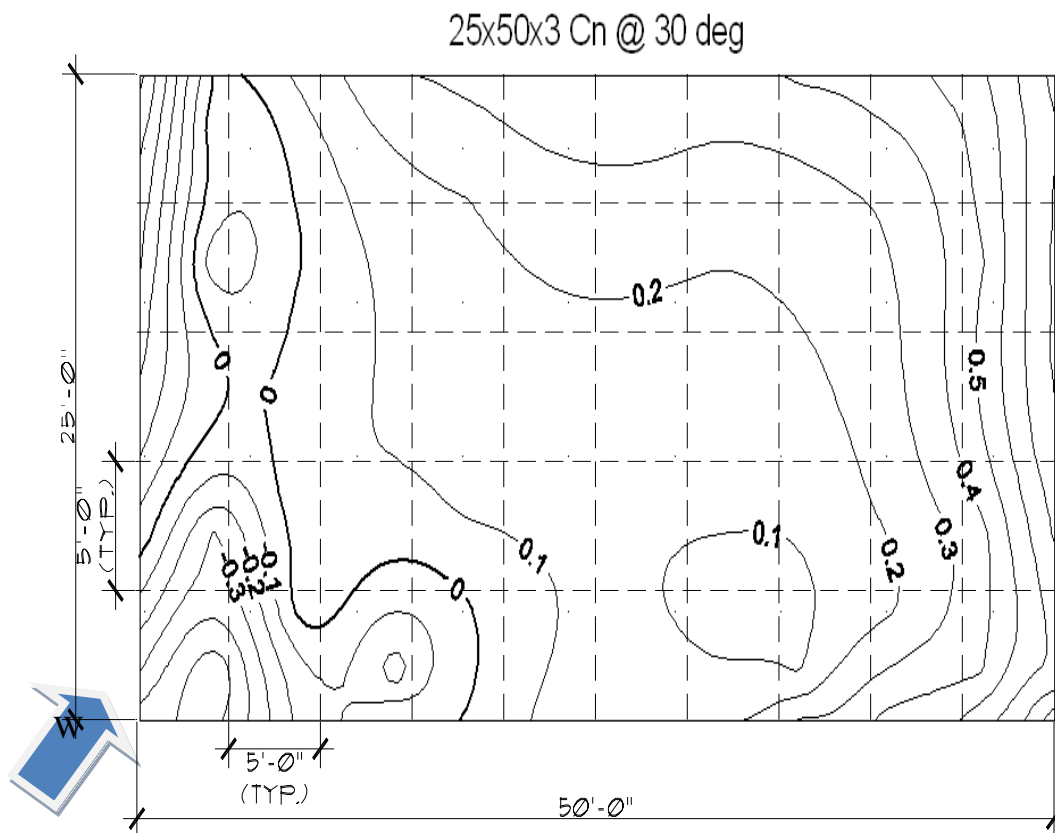
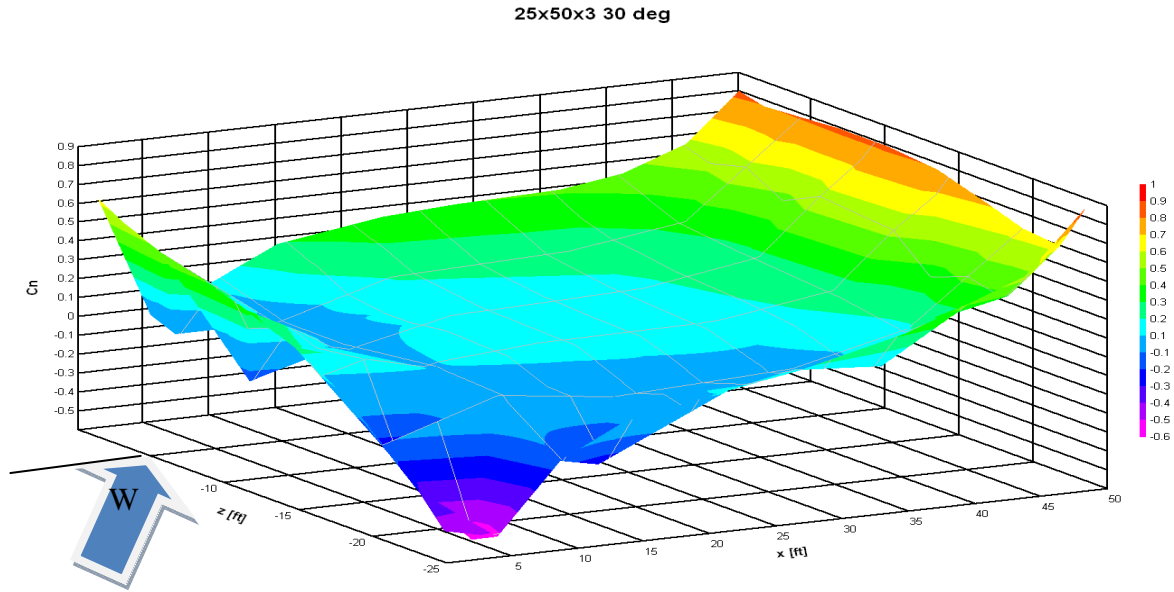
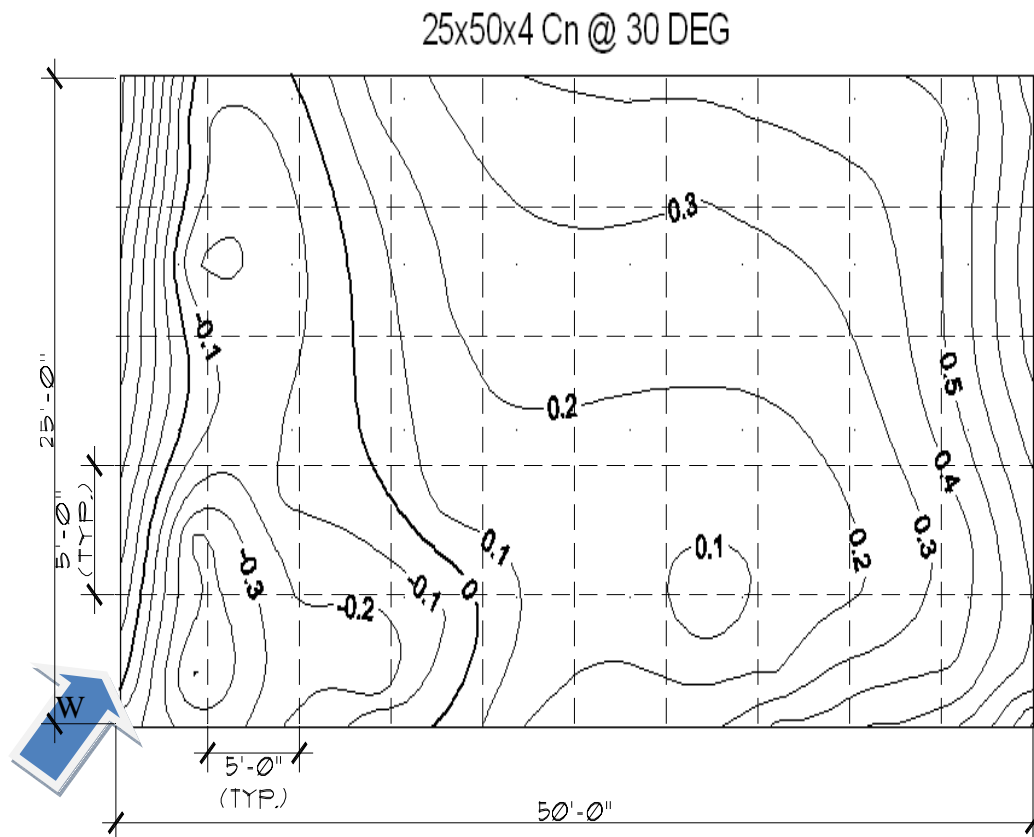


Figure 7.47: Model #18, contour plan for Cn values for wind at 30 degrees.

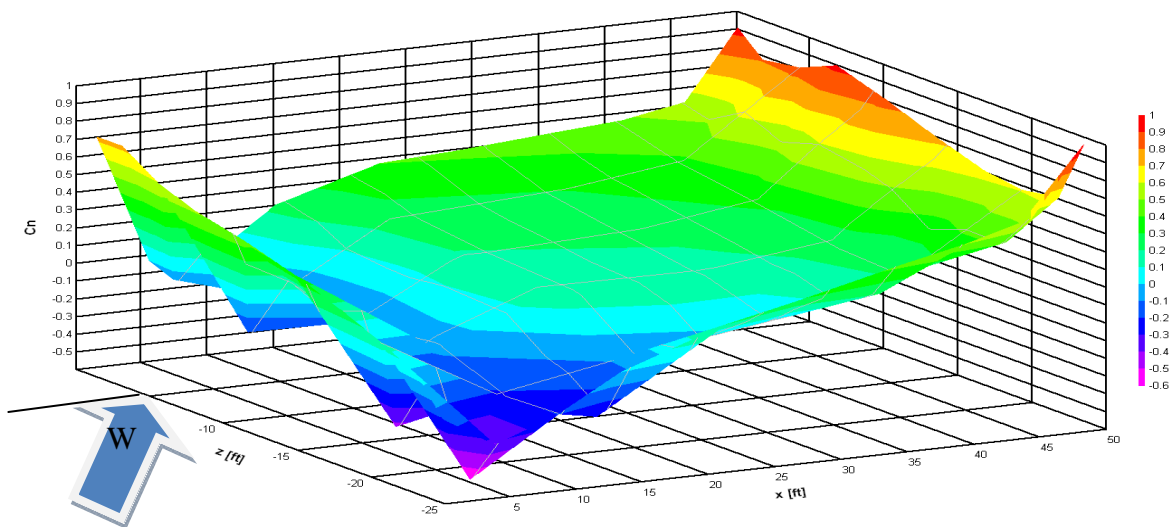


**Figure 7.48: Model #18, 3D contour plot for  $C_n$  values for wind at 30 degrees.**

$C_n$  values for contour plan for Model #20, with dimensions of 7.6 m (25 ft) x 15.2 m (50 ft) x 1.22 m (4 ft) with the wind at 30 degrees are shown in Figure 7.49. Figure 7.50 shows a 3D view of the  $C_n$  values of the canopy. Values range from a maximum of +1.0 to a minimum of -0.6. The values are identical compared to the square configuration Model #12 (+1.0 and -0.6 respectively).

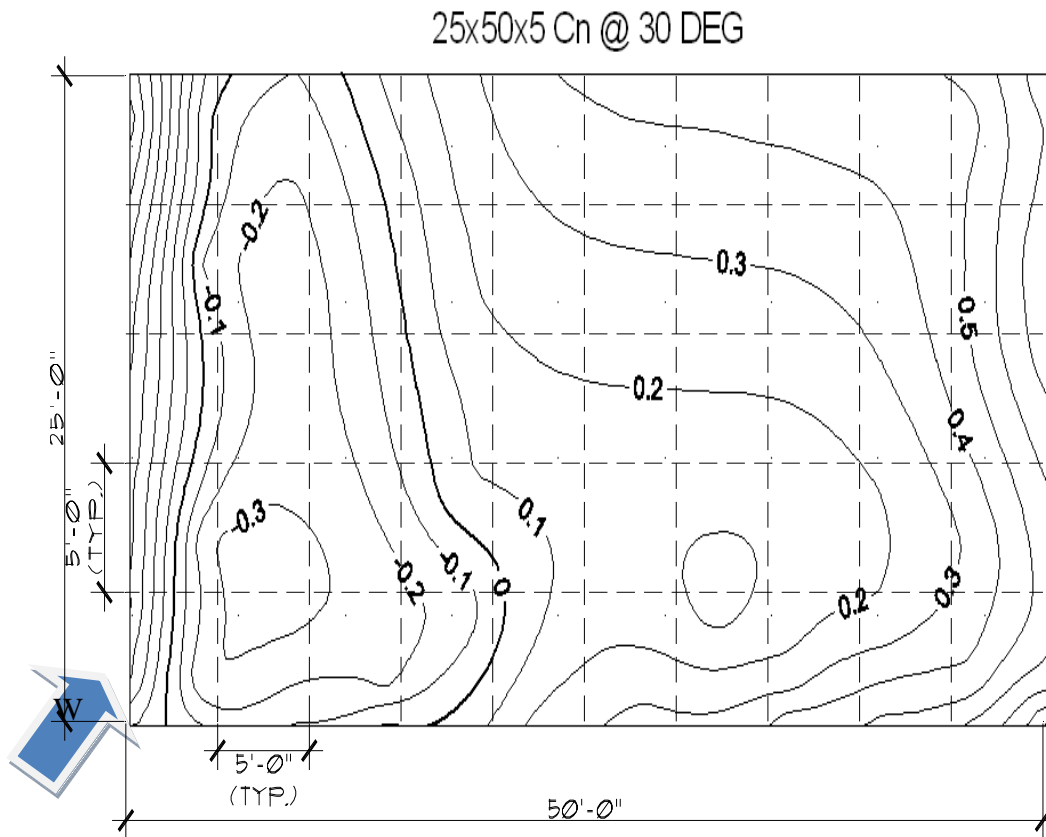


**Figure 7.49: Model #20, contour plan for Cn values for wind at 30 degrees.**  
25x50x4 30 deg

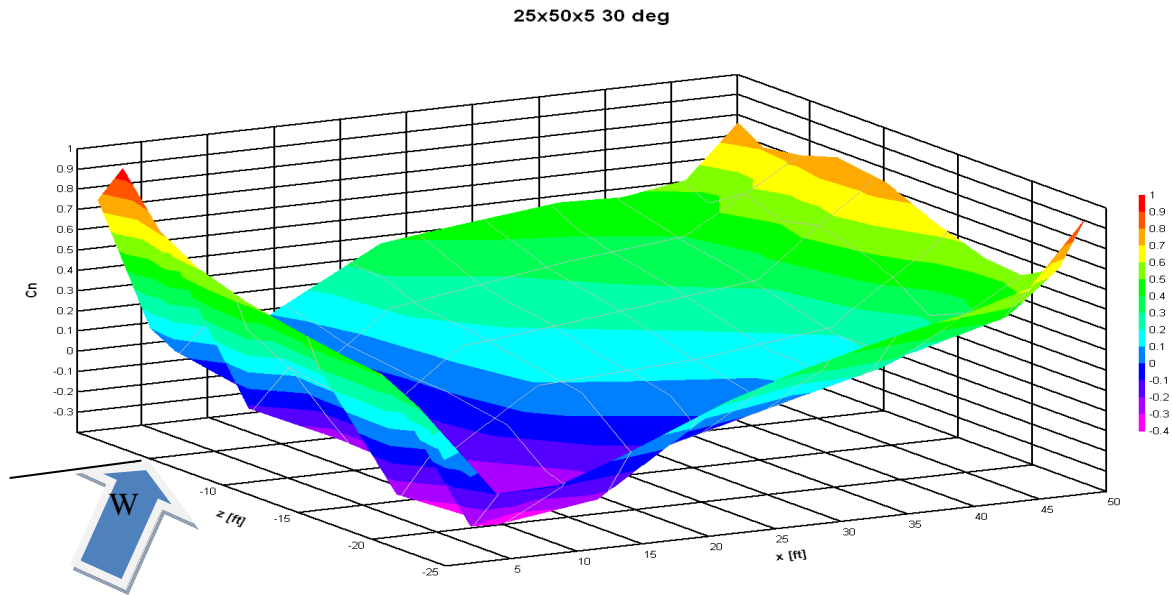


**Figure 7.50: Model #20, 3D contour plot for Cn values for wind at 30 degrees.**

Cn values for contour plan for Model #22, with dimensions of 7.6 m (25 ft) x 15.2 m (50 ft) x 1.52 m (5 ft) with the wind at 30 degrees are shown in Figure 7.51. Figure 7.52 shows a 3D view of the Cn values of the canopy. Values range from a maximum of +1.0 to a minimum of -0.4. The values are very similar compared to the configuration Model #14 (+0.9 and -0.3 respectively).



**Figure 7.51: Model #22, contour plan for Cn values for wind at 30 degrees.**



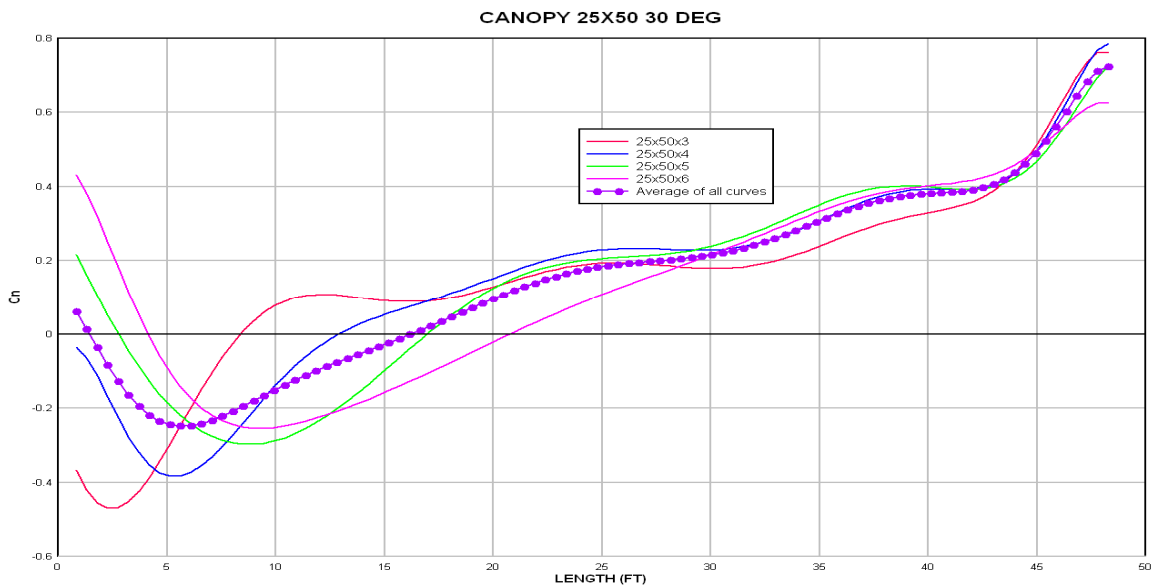
**Figure 7.52: Model #22, 3D contour plot for  $C_n$  values for wind at 30 degrees.**

$C_n$  values for contour plan for Model #24, with dimensions of 7.6 m (25 ft) x 15.2 m (50 ft) x 1.82 m (6 ft) with the wind at 30 degrees are shown in Figure 7.53. Figure 7.54 shows a 3D view of the  $C_n$  values of the canopy. Values range from a maximum of +0.9 to a minimum of -0.3. The values on the configuration Model #14 are +1.0 and -0.3 respectively.



Figure 7.55 shows four  $C_n$  graphs, one for each parapet height on the 7.6 m (25 ft) x 15.2 m (50 ft) open canopy. The 2D graph data was obtained at the center of the canopy width in the angle of the applied wind, in this case, 30 degrees from the principal axis direction. All parapet height exhibits on the graph a segment of positive and negative  $C_n$  values. Depending on the parapet height, downward and uplift is reflected. This is shown on the first  $L/5$ , except the parapet height of 0.91 m (3 ft) and 1.22 m (4 ft), which were all negative  $C_n$  values. After the downward and uplift segment, an uplift pressure length of  $L/3$  occurs followed by a downward for the rest of the horizontal distance.

- $C_n$ =uplift (away from the surface)  
+ $C_n$ =downward (towards the surface)



**Figure 7.55: Graph of  $C_n$  values for all 7.6 m (25 ft) x 15.2 m (50 ft) @ 30 degrees.**

## 7.8 Conclusions

Three canopy configurations were investigated in this chapter using CFD simulations, in which wind incidence angle, parapet height, and transverse direction were changed to understand pressure coefficient distributions. High changes are associated with parapet height, as expected,

as shown in Figures 7.10, 7.19, 7.28, 7.37, 7.46 and 7.55. The influence of wind direction is also important in terms of values and pressure distributions, to visualize that it is necessary to compare pairs of Figures, such as 7.10 and 7.19; 7.28 and 7.37; and 7.46 and 7.55. Finally, the relative dimension of the canopy in plan has been investigated for a square configuration and two rectangular ones. To visualize changes, one should compare Figures 7.10, 7.28 and 7.46; and Figures 7.19, 7.37 and 7.55.

A warning should be made that differences between configurations used to compute pressure values will lead the differences in structural response; however this is not a linear effect and at this point in the thesis it is not clear how the detected differences will affect the response of the structure. This the subject of Chapter 8.

## **CHAPTER 8. CASE STUDIES OF CANOPIES AND COMPARISONS WITH FIELD EVIDENCE**

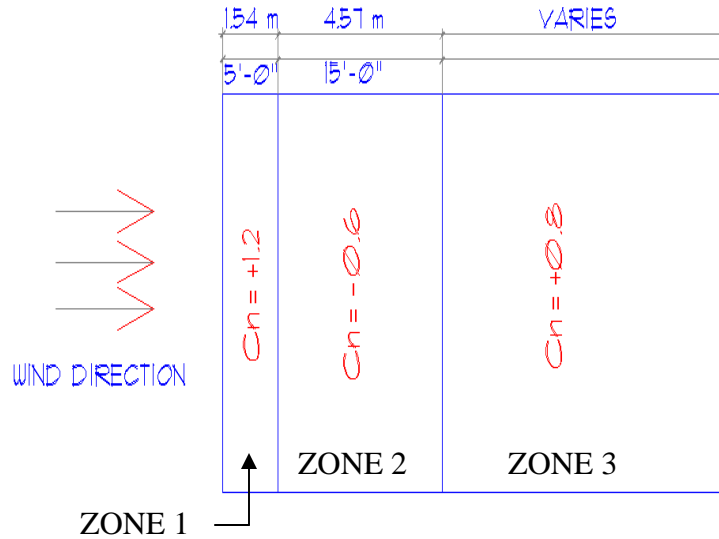
### **8.1 Structural Analysis**

From the point of view of this research, the structural analysis of open canopies that suffered the effect of wind forces is very important for several reasons. First, a structural analysis based on the previous pressure coefficients derived in this research work is extremely useful in order to validate the proposed  $C_n$  wind coefficients on this type of structure. Second the definitions of pressures due to wind make engineering sense in the context of structural analysis and behavior. From the engineering point of view, the relevance of the pressure coefficients depends on the performance of the structures affected by wind. For example, details of pressure coefficients in given zones of a structure would not be relevant if the structural response is not sensitive to those changes. Thus, we estimate wind speed data in order to make structural analysis and improve structural behavior. The importance of wind pressures in the collapse of canopies can be highlighted with reference to the multiple cases of failure identified following hurricanes Katrina and Rita in 2005 (NIST, 2006). From reconnaissance missions to affected areas by Hurricanes Katrina and Rita in 2005, four open canopies have been selected to perform a structural analysis in this chapter. In the cases selected, wind velocity data was not available and could only be estimated based on information available for neighboring areas. Current wind velocity for the comparison of the selected case studies was estimated based on ASCE 7-05, Figure 6-1A, page 34 that uses a 3 second gust speed. Wind speeds in Case 1, 2 and 4 have been estimated to be about 130 mph. Wind speed in case 3 was estimated to be about 110 mph. The wind velocities were used to obtain the wind pressures, and those pressures were used for the

structural analysis. This is a sequential methodology, in which the evaluation of wind pressures is uncoupled for the structural response.

In order to propose and develop a methodology to analyze and calculate the structural response of an open canopy, first we apply the wind pressure obtained from the  $C_n$  results on Chapter 7. The structural effects of the applied wind pressures were analyzed considering two main wind directions, at 0 and 30 degrees from the main horizontal axis. This procedure, in which a detailed distribution of  $C_p$  values, based on contour levels is used, is called Method #1 and is similar to the methodology followed by Portela (2004). A second procedure, in which the values from Method #1 are simplified, so that three zones of uniform pressure are identified with their maximum values. Again, wind directions at 0 and 30 degrees were investigated. This procedure was called Method #2. There is a third level of modeling attempted, which is based on the same methodology employed by the ASCE 7-05, Section 6.5.13. The  $C_n$  values used on the structural analysis for this method are the extreme values found in Chapter 7. This procedure was called Method #3.  $C_n$  values vary from +1.2 to -0.6 and to +0.8 on this proposed method. Typical proposed  $C_n$  values for Method #3 are summarized in Figure 8.1.

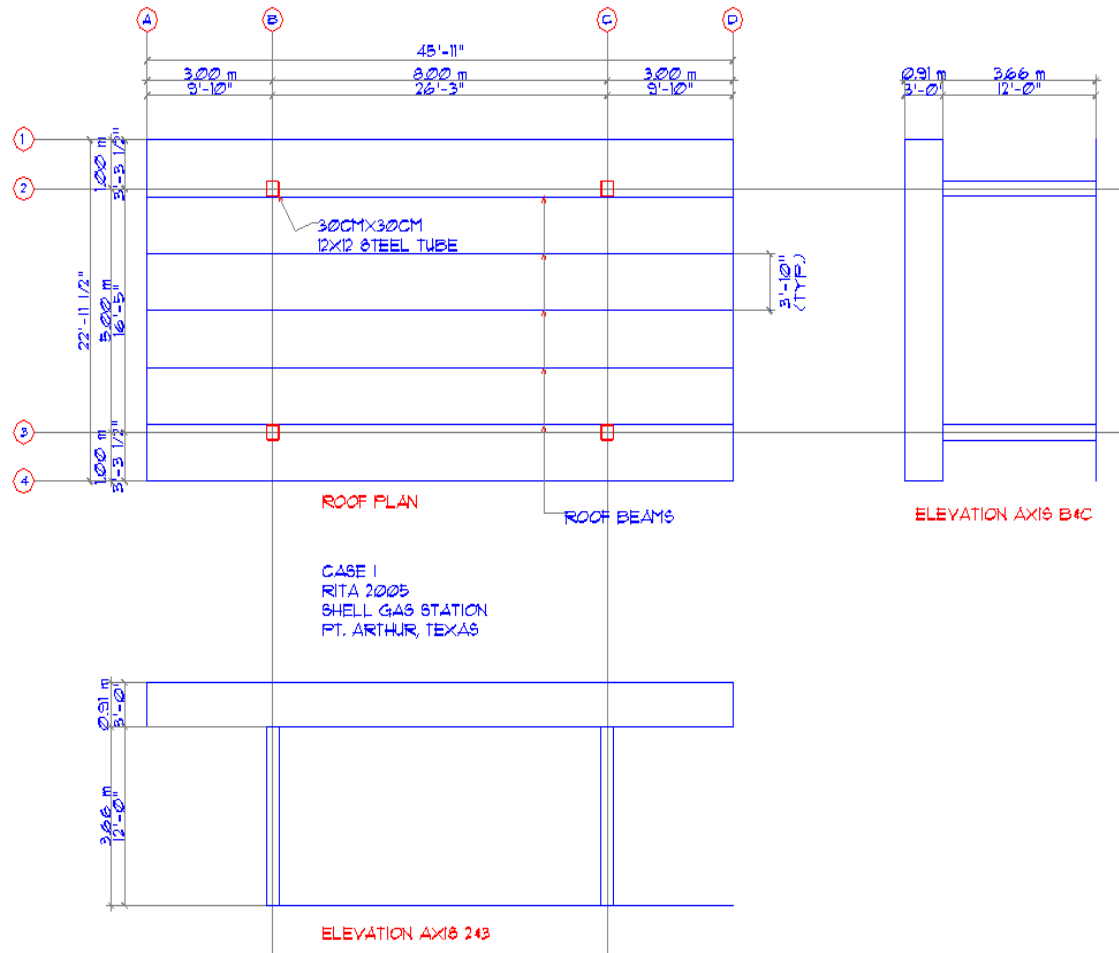
With the wind pressures obtained with all three methods, a structural analysis of the open canopy was performed. The importance of doing structural analysis on case studies this is a firsthand opportunity of analyzing a structure that has actually collapsed and see if that could be anticipated (and prevented) with the use of the proposed  $C_n$  values.



**Figure 8.1: Proposed  $C_n$  values to be applied for Method #3 to the structural analysis model.**

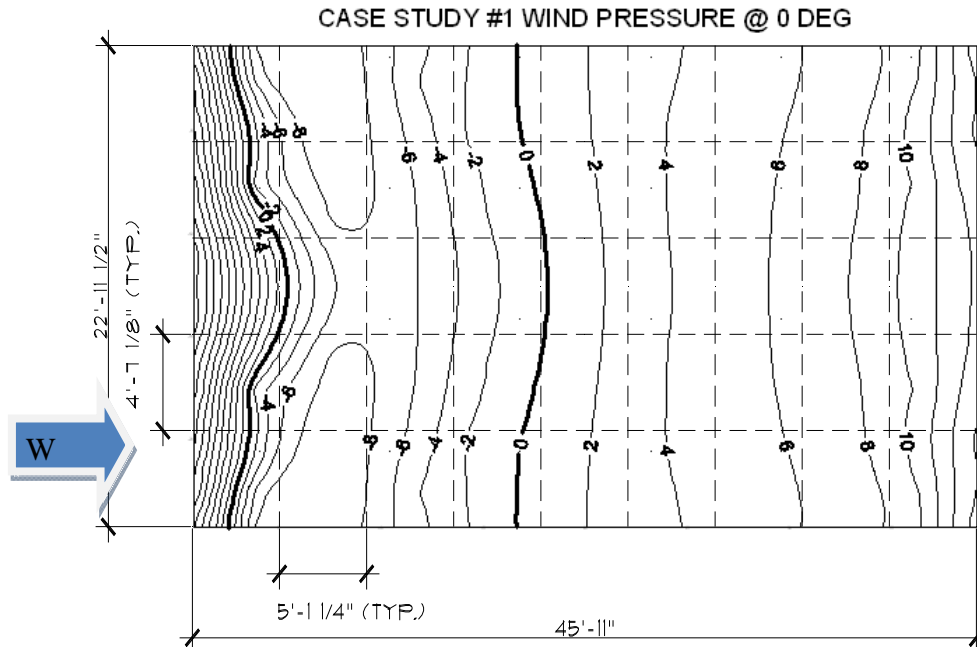
## 8.2 Case study #1, Shell Gas Station in Pt. Arthur, Texas, Method #1

The case study number 1 is a Shell gas station located in Pt. Arthur, Texas. A roof plan and elevations are illustrated in Figure 8.2. The dimensions were taken from measurements in the field. The floor plan dimensions are approximately 14 m (46 ft) in length by 7 m (23 ft) in width and 4.3 m (14 ft) in height. It has a system of four steel columns of 0.3 m (1 ft) x 0.3 m (1 ft) with wide flange steel beams and channels for the gravity system of the structure. Smallest thickness of steel columns and beams were used for the analysis. A steel deck was used for the roof cladding. A parapet of 0.91 m (3 ft) is on the roof perimeter. There is no available data of the foundation size and depth.

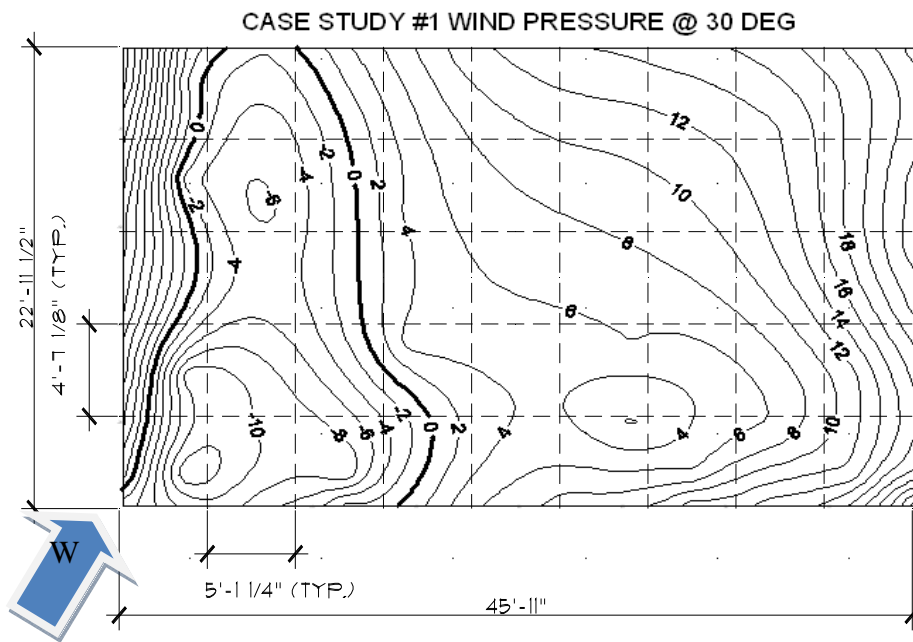


**Figure 8.2: Canopy floor plan and elevations for Case study #1.**

On Chapter 7,  $C_n$  values were obtained for the open canopy geometry for Case study #1. Those  $C_n$  values were converted to wind pressures in pounds per square foot (psf) for two wind cases, one wind direction at 0 and another one at 30 degrees, as shown on Figures 8.3 and 8.4, respectively the negative values indicate uplift pressure and positive values indicate downward pressures.



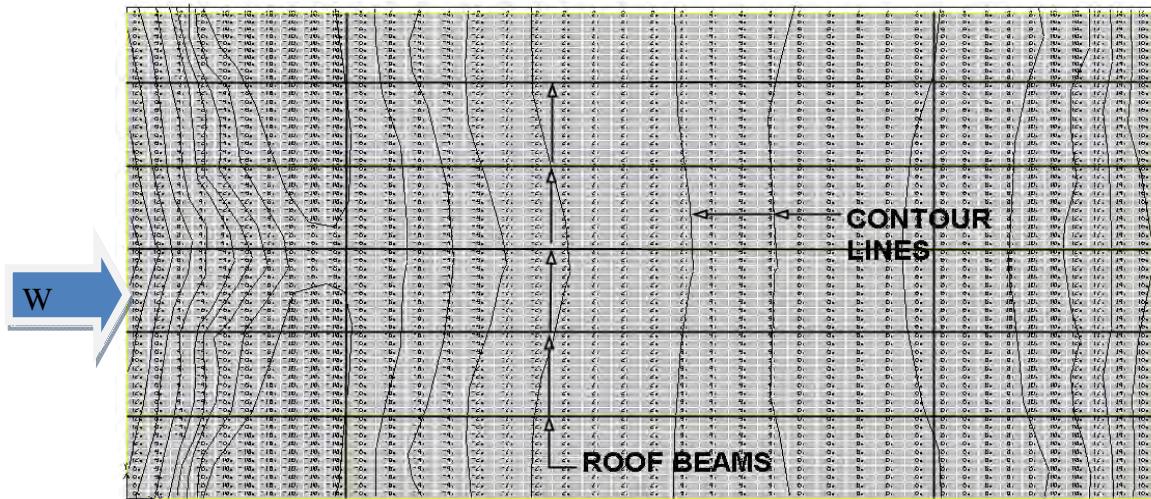
**Figure 8.3: Contours for wind pressures for open canopy case #1 at 0 degrees.**



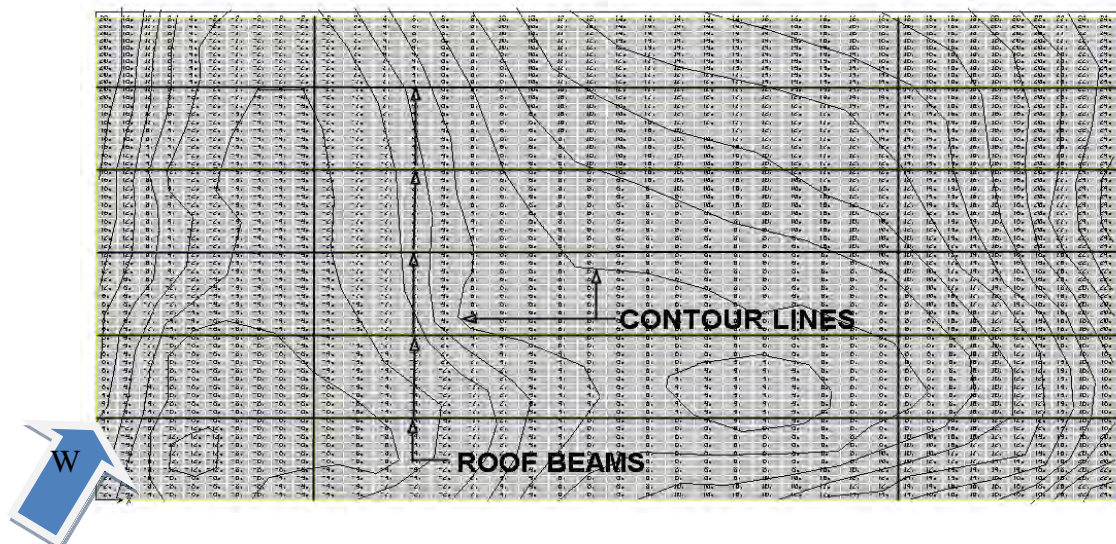
**Figure 8.4: Contours for wind pressures for open canopy case #1 at 30 degrees.**

The resultant wind pressures for each wind case were applied to a computational model using the structural analysis software ETABS (CSI 2009). Using a refined mesh for the roof

cladding, it was possible to input all the wind pressure values on the roof surface as shown on Figures 8.5 and 8.6. The darker lines indicate the contours for wind pressures from the previous figures.



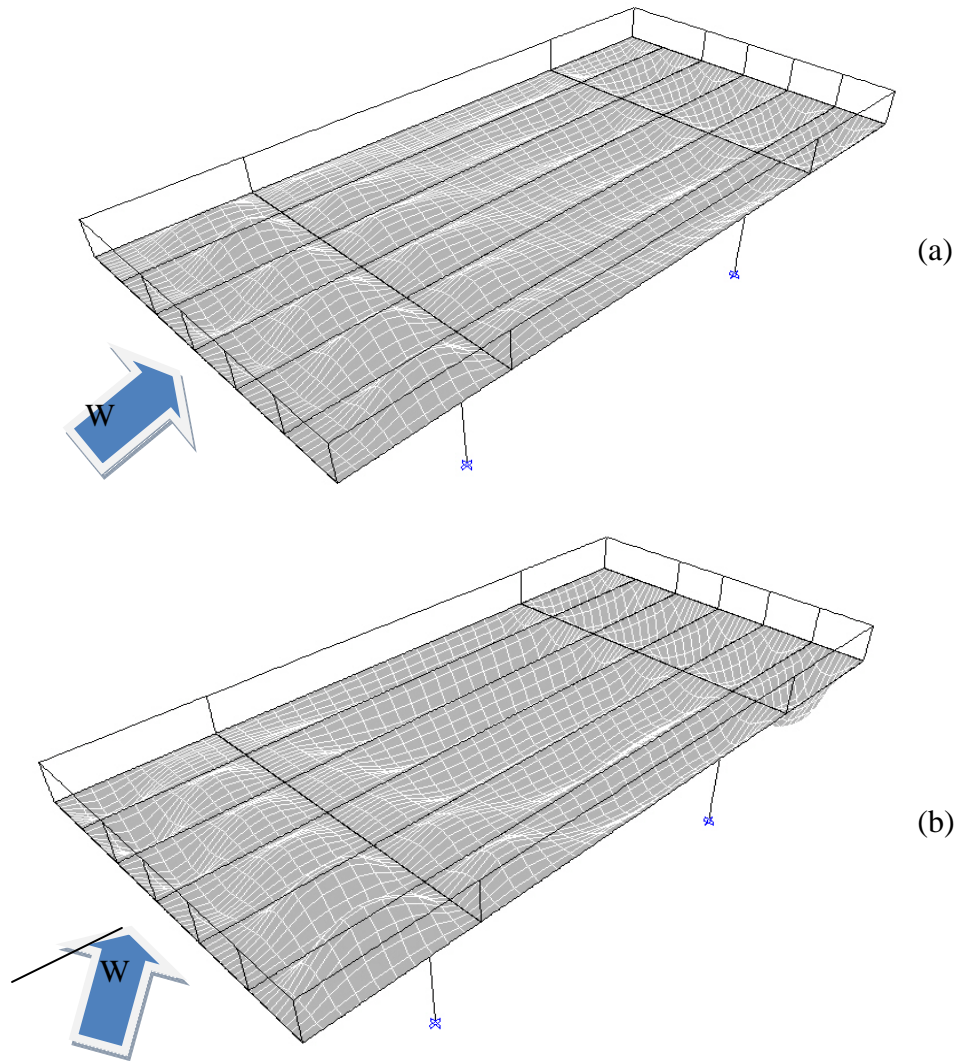
**Figure 8.5: Computer model with wind pressures applied to roof surface at 0 degrees.**



**Figure 8.6: Computer model with wind pressures applied to roof surface at 30 degrees.**

The results of the applied wind load conditions at 0 and 30 degrees were obtained with the use of ETABS, and Figure 8.7a and 8.7b show the deflected roof surface due to winds at 0

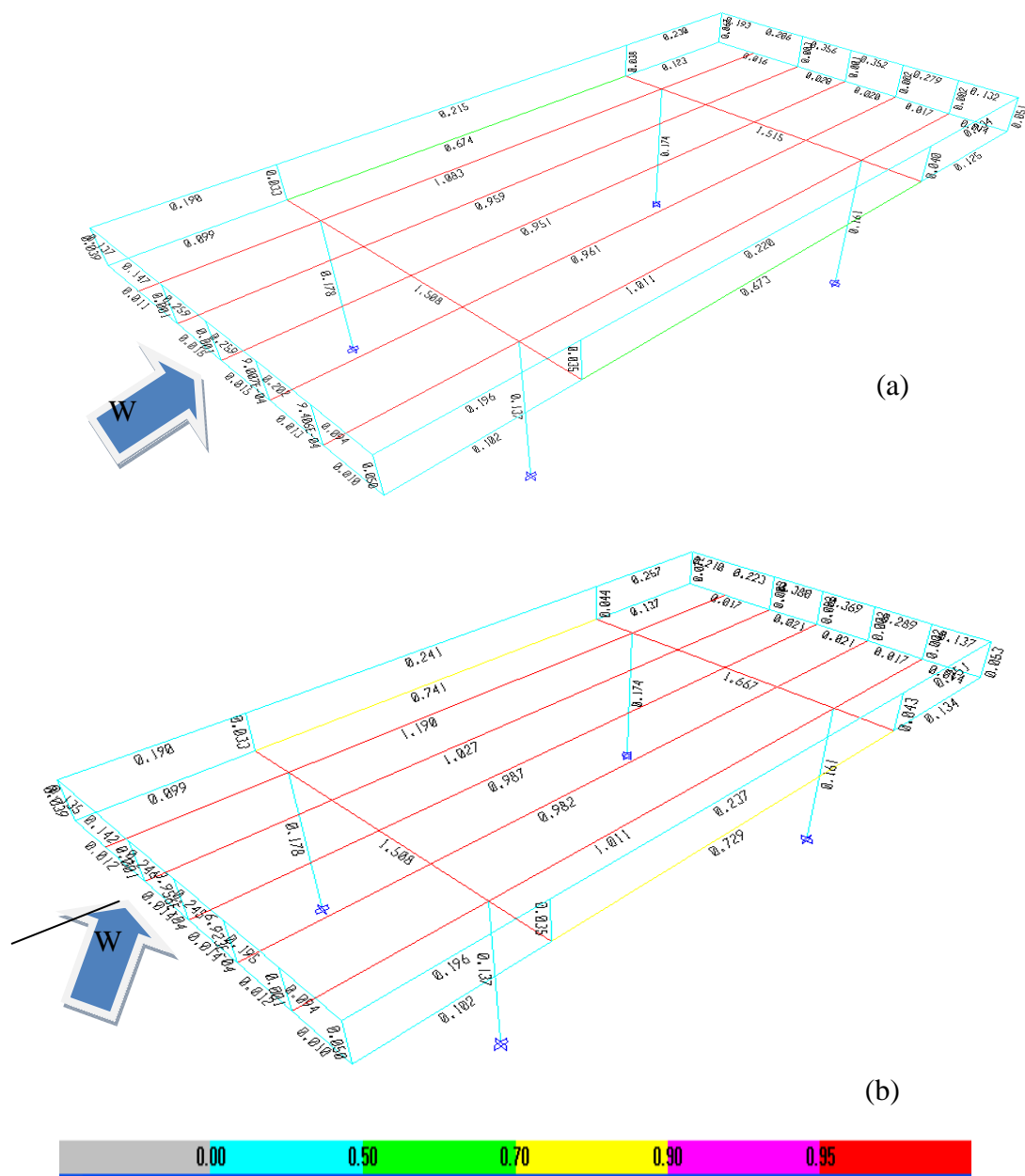
and 30 degrees, respectively. The deflected surfaces show differences between the applied wind pressures for the wind directions tested in this study.



**Figure 8.7: ETABS model for Case #1 showing the refined roof mesh and deflected shape of the wind loading at (a) 0 degrees (b) 30 degrees.**

Figures 8.8a and 8.8b show the steel stress ratios based on ASD 1989 steel code (AISC, 1989) for the wind at 0 and 30 degrees, respectively. A stress ratio is the relation between a given stress state in a structural member and its capacity using plasticity considerations. For this particular canopy, the results in Figures 8.8a and 8.8b show that in both cases there are beam

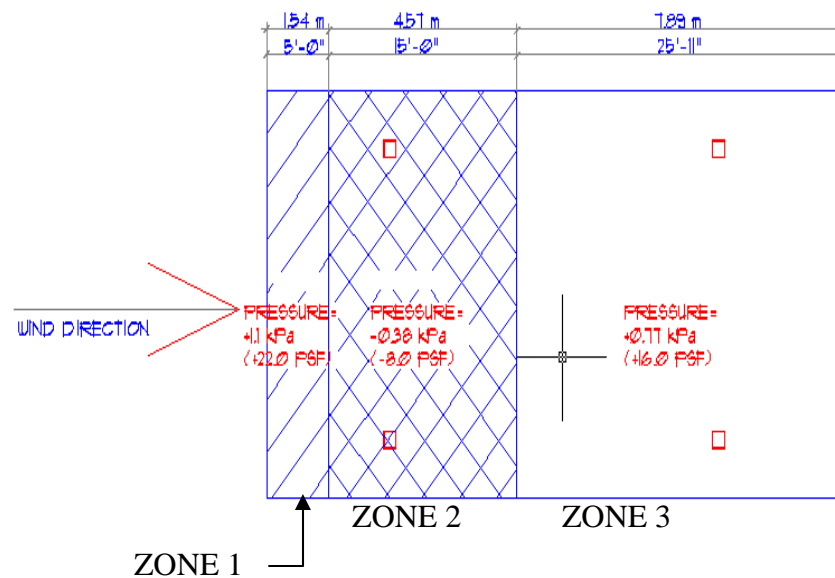
members with stress factors larger than 1. Wind direction at 30 degrees, produce higher stress ratios than wind at 0 degrees. For example, the transverse beam on the leeward side of the roof has a stress ratio of 1.515 for wind blowing at 0 degree, whereas for 30 degrees the ratio is 1.667, that is, a 9.1 % difference. The long beams on the roof are also under conditions of severe stress, showing values of 1.01 for 0° and 1.19 for 30°. The available evidence for this canopy, which can be seen in Figures 8.18 to 8.22, indicate that the columns were still standing after the structure failed and did not show signs of damage. The transverse beams, on the other hand, had clearly failed, and this is in agreement with the computations carried out with Method #1 (also with Methods #2 and #3, as will be seen next). The columns do not seem to be severely compromised by the wind conditions considered. Method #1 has some disadvantages because the wind load input for Method #1 is very complicated. A large number of elements need to be included in the model to represent the contours and this has to be done by hand. Input has to be done one by one, on each of the individual mesh elements, with the consequence that a total of 2400 roof shell elements were assigned wind loading pressures, one at a time. In addition, a considerable amount of computational time is required to solve the model. The wind loading computed and input, represents the calculated wind pressures on the open canopy, for a specific parapet height. The need for a simpler design method is obvious and a second method, Method #2 is proposed and investigated in the following section.



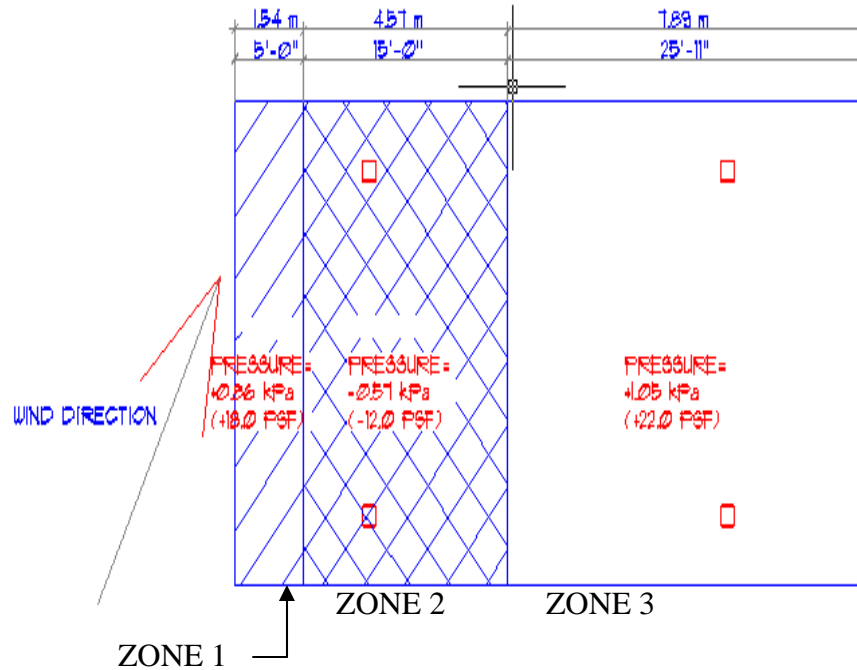
**Figure 8.8: Structural steel results from ETABS for Case #1, showing members stress ratios using Method #1 for (a) wind at 0 degrees (b) wind at 30 degrees.**

### 8.3 Case study #1, Shell Gas Station in Pt. Arthur, Texas, Method #2

Case study #1 was investigated using a simplified loading procedure, named Method #2. It was proposed that instead of input all the contours on the roof surface, the new approach will be to use the maximum wind pressure on the downward and uplift segments, from the extreme wind pressures that were used in Method #1. This method simplifies the wind load input on a structural model. Following recommendations in the literature for identifying the most critical conditions, two wind directions were investigated, the wind at 0 degrees and 30 degrees. The input wind pressures to be used on the structural model are shown on Figures 8.9 for 0° angle of attack and Figure 8.10 for the 30° condition.

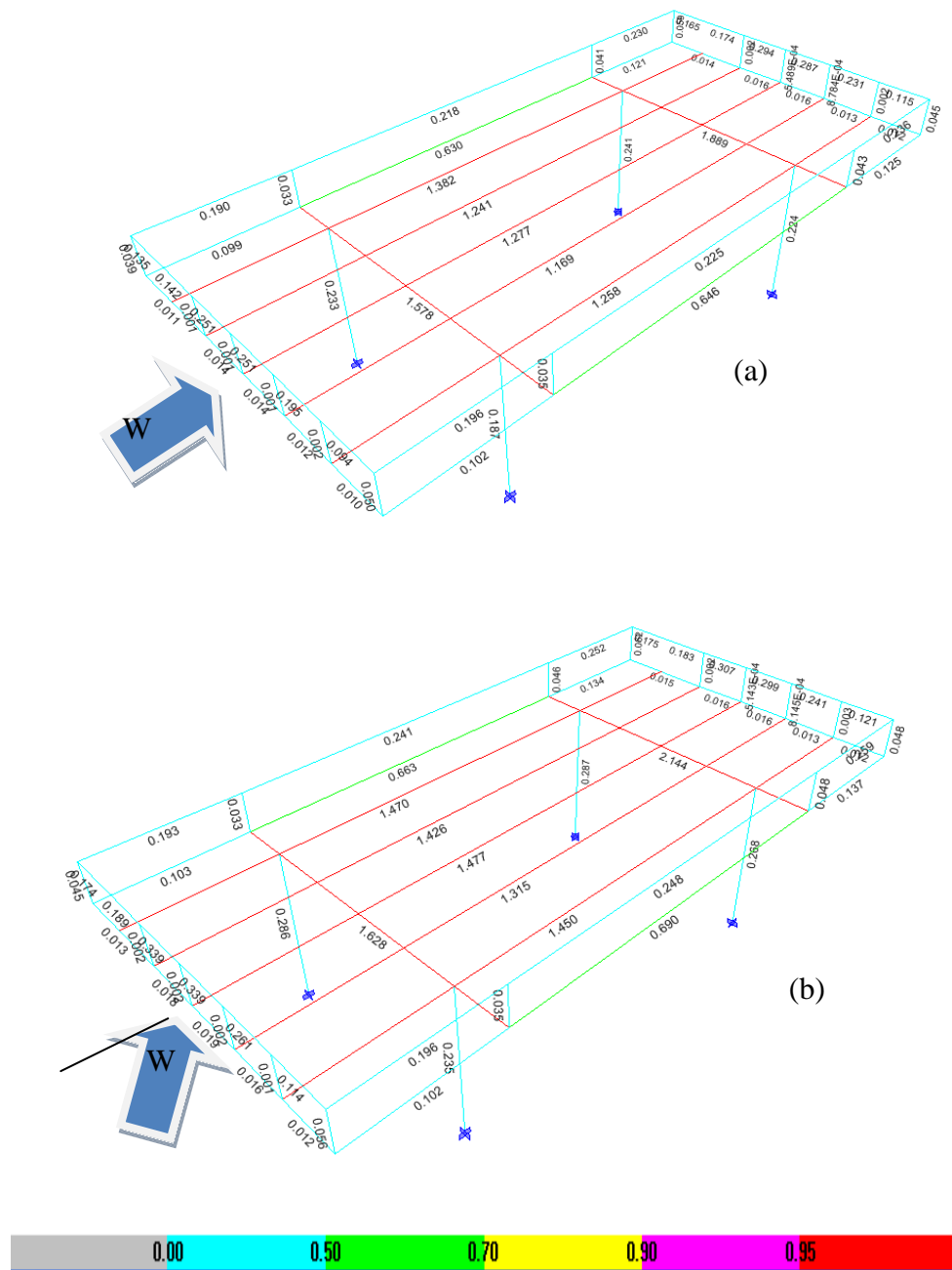


**Figure 8.9: Roof canopy wind pressures for case #1, Method 2, wind at 0 degrees, + means downward pressure, - means uplift pressure.**



**Figure 8.10: Roof canopy wind pressures for case #1, Method 2, wind at 30 degrees, + means downward pressure, - means uplift pressure.**

The wind pressures from both wind load conditions were input on the ETABS program for Case #1. The stress ratios are shown on Figure 8.12a and 8.12b, using wind direction at 0 and 30 degrees, respectively. Higher stress ratios were obtained using Method #2 in comparison with Method #1, and in this sense the method is conservative (similar conclusions were obtained for other cases investigated). For example, the transverse beam on the leeward side receives a stress ratio of 1.515 for wind blowing at 0 degree and Method #1, whereas using Method #2 the ratio becomes 1.889, which is 19.8% higher. For wind at 30 degrees the same beam changes from 1.667 to 2.144 (22.2% higher). Significant changes are detected in the long beams in this method, with higher values of 1.38 for 0° and 1.47 for 30°. The proposed second method used to solve Case #1, is simpler and easier to use, but it needs to include two separate load conditions for wind each time which is not what designers are used to.



**Figure 8.11: Structural steel results from ETABS showing members stress ratios using Method #2 for (a) wind at 0 degrees (b) wind at 30 degrees.**

## **8.4 Case study #1, Shell Gas Station in Pt. Arthur, Texas, Method #3**

For the current case study, calculations following Method #3 were based on the guidelines provided by ASCE 7-05. From the current results, the extreme  $C_n$  values from Chapter 7 were selected. The calculated wind pressures and the resulting  $C_n$  values used are shown in Figure 8.12. The mathematical procedure was performed with Maple, version 11 (Maplesoft, 2008), a symbolic manipulator program. The results of Figure 8.12 show values that vary from +33.1 psf (+1.61 kPa) (downward pressure, zone 1), as well as -16.5 psf (-0.79 kPa) (uplift pressure, zone 2) to +22 psf (1.05 kPa) (downward pressure, zone 3).

Parapet wind pressure values used for structural analysis are indicated in the computer output shown in Figure 8.13. As previously described on Chapter 6,  $C_n$  values from the wind tunnel and CFD simulation generally are in agreement. Wind pressure values were applied to the structural model on the parapets. Wind pressures of 43.88 psf (2.1 kPa) in the windward direction and 31.3 psf (1.5 kPa) in the leeward direction were applied to the structural model. Wind direction was applied on both horizontal axis and wind pressures on the parapets are applied at each wind direction. The proposed Method #3 combines the extreme values of the wind loading of 0 and 30 degrees on the roof surface. It includes different parapet heights calculated on Chapter 7.

**#CASE #1 ASCE 7-05 section 6.5.13 Design on wind loads on open buildings page 29**

$V := 130$ ;  $I_{factor} := 1.0$ ;  $K_z := 0.85$ ;  $K_{zt} := 1.0$ ;  $K_d := 0.85$ ;

#  $K_z$  table 6-3 page 79

#  $K_{zt}$  section 6.5.7.2

#  $K_d$  wind directionality factor table 6-4 page 80

$q := .00256 \cdot K_z \cdot K_{zt} \cdot K_d \cdot V^3 \cdot I_{factor}$  # 6.5.10 (6-15) velocity pressure

$q := 31.25824000$

# Gust effect factor rigid structures

$g_g := 3.4$  # constant 6.5.8.1

$g_v := 3.4$ ;

$z_{bar} := 12$  # equivalent height in structure

$c := 0.2$  # turbulence intensity factor in 6-5 from Table 6-2

$l := 500$  # integral length scale factor from Table 6-2

$e_{bar} := \frac{1}{5}$  # integral length scale power law exponent in 6-7 from Table 6-2

$B := 46$  # horizontal dimension measured normal to wind direction

$h := 12$  # mean roof height of a building

$I_{zbar} := \text{evalf}\left(c \cdot \left(\frac{33}{z_{bar}}\right)^{\frac{1}{6}}\right)$  # 6.5.8.1(6-5)

$I_{zbar} := 0.2367293530$

$L_{zbar} := \text{evalf}\left(l \cdot \left(\frac{z_{bar}}{33}\right)^{e_{bar}}\right)$  # 6.5.8.1(6-7)

$L_{zbar} := 408.4166753$

$Q := \sqrt{\frac{1}{1 + 0.63 \cdot \left(\frac{(B+h)}{L_{zbar}}\right)^{0.63}}}$  # 6.5.8.1(6-6)

$Q := 0.9189381371$

$G := 0.925 \cdot \left(\frac{(1 + 1.7 \cdot g_g \cdot I_{zbar} \cdot Q)}{1 + 1.7 \cdot g_v \cdot I_{zbar}}\right)$  # 6.5.8.1(6-4)

Wind pressure in psf. H=horizontal distance on the roof surface.

$G := 0.8816786140$

$Cn1 := 1.2$  # PROPOSED CNVALUES

$Cn2 := -0.6$ ;

$Cn3 := 0.8$ ;

$q1 := q \cdot G \cdot Cn1$ ;  $h1 := 6$ ;

$q2 := q \cdot G \cdot Cn2$ ;  $h2 := 18$ ;

$q3 := q \cdot G \cdot Cn3$ ;  $h3 := B - h1 - h2$ ;

Proposed Cn values by the author to be used for Method 3, simplified design.

$q1 := 33.07166606$   
 $h1 := 6$

$q2 := -16.53583303$   
 $h2 := 18$

$q3 := 22.04777738$   
 $h3 := 22$

**Figure 8.12: Maple layout of Wind pressure calculations for Method #3, using proposed Cn values and ASCE 7-05 procedure, Case# 1.**

### #Parapet pressure calculation ASCE 7 – 05, section 6.15.2.4 page 28

$V := 130$  ;  $I_{factor} := 1.0$  ;  $K_z := 0.85$  ;  $K_{zt} := 1.0$  ;  $K_d := 0.85$  ;

#  $K_z$  table 6-3 page 79

#  $K_{zt}$  section 6.5.7.2

#  $K_d$  wind directionality factor table 6-4 page 80

$q := .00256 \cdot K_z \cdot K_{zt} \cdot K_d \cdot V^2 \cdot I_{factor}$ ; # 6.5.10 (6-15) velocity pressure

$q := 31.25824000$

# Combined net pressure coefficient

#  $GC_{pn1} = 1.5$  for windward parapet

#  $GC_{pn2} = -1.0$  for leeward parapet

$GC_{pn1} := 1.5$ ;

$GC_{pn1} := 1.5$

$GC_{pn2} := -1.0$ ;

$GC_{pn2} := -1.0$

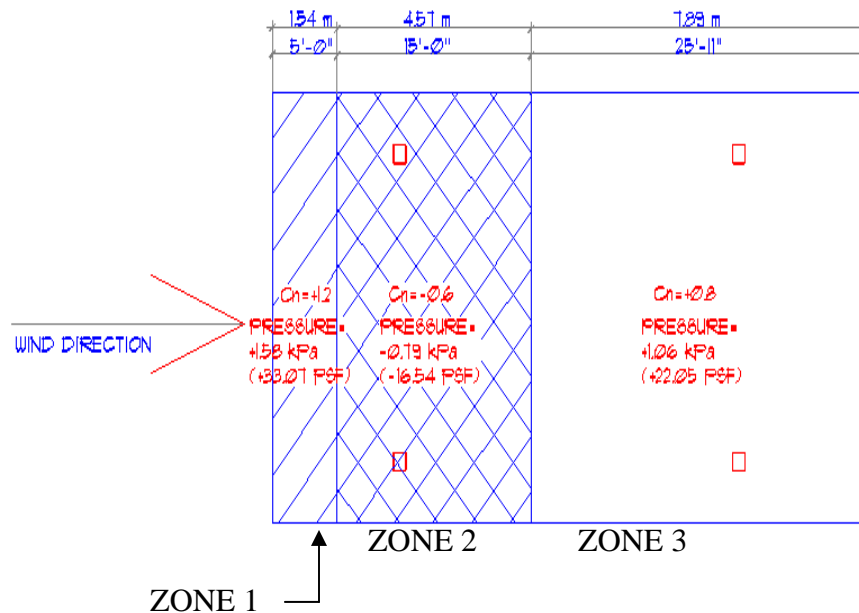
$q1 := q \cdot GC_{pn1}$ ; # formula 6-20

$q1 := 46.88736000$

$q2 := q \cdot GC_{pn2}$ ;

$q2 := -31.25824000$

**Figure 8.13: Maple output of Wind pressure results on parapets using ASCE 7-05  $C_n$  values.**

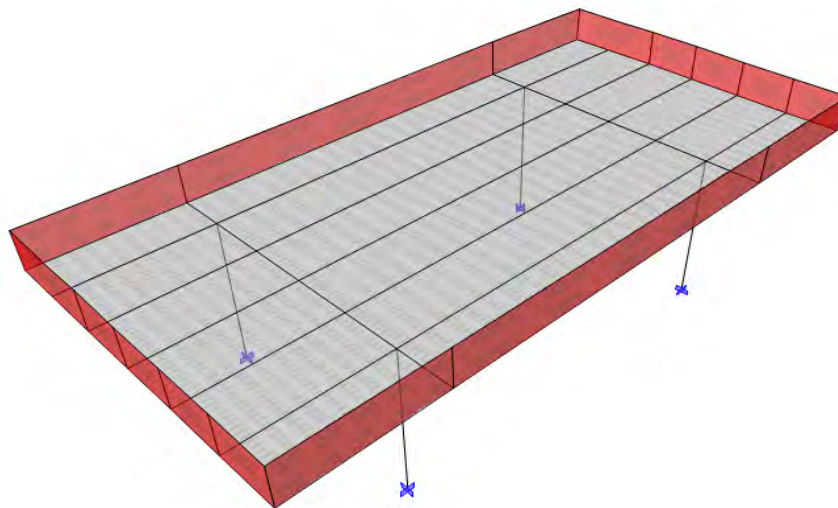


**Figure 8.14: Roof canopy pressure  $C_n$  and wind pressures for case #1, + means downward pressure, - means uplift pressure.**

**Table 8.1: Table of wind pressures for Case #1, Method #3 to be used in structural analysis.**

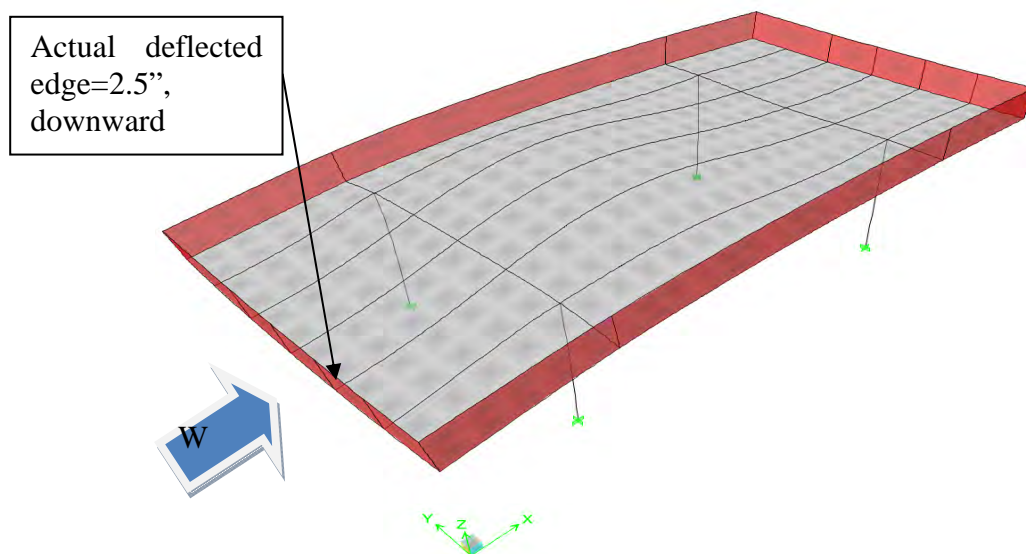
| Case Study # | 1    | Zone Length |        |       |       |      |
|--------------|------|-------------|--------|-------|-------|------|
| Zone #       | Cn   | Pressure    | psf    | kPa   | h(ft) | h(m) |
| 1            | 1.2  | q1=         | 33.04  | 1.58  | 5     | 1.5  |
| 2            | -0.6 | q2=         | -16.54 | -0.79 | 15    | 4.6  |
| 3            | 0.8  | q3=         | 22.05  | 1.06  | 26    | 7.9  |

The wind pressures obtained on Table 8.1 using the ASCE 7-05 method with the proposed Cn values, or Method #3, were imposed on the structural model for the analysis on Case #1. The 3D model used for the structural analysis is shown on Figure 8.15. The plan geometry, height, column and beam sizes of the open canopy were included in the structural model. Wind pressures from Table 8.1 were included and applied to the model. Structural plans for each case were not available, beam spacing and structural layout was approximated from field observations and photographs.

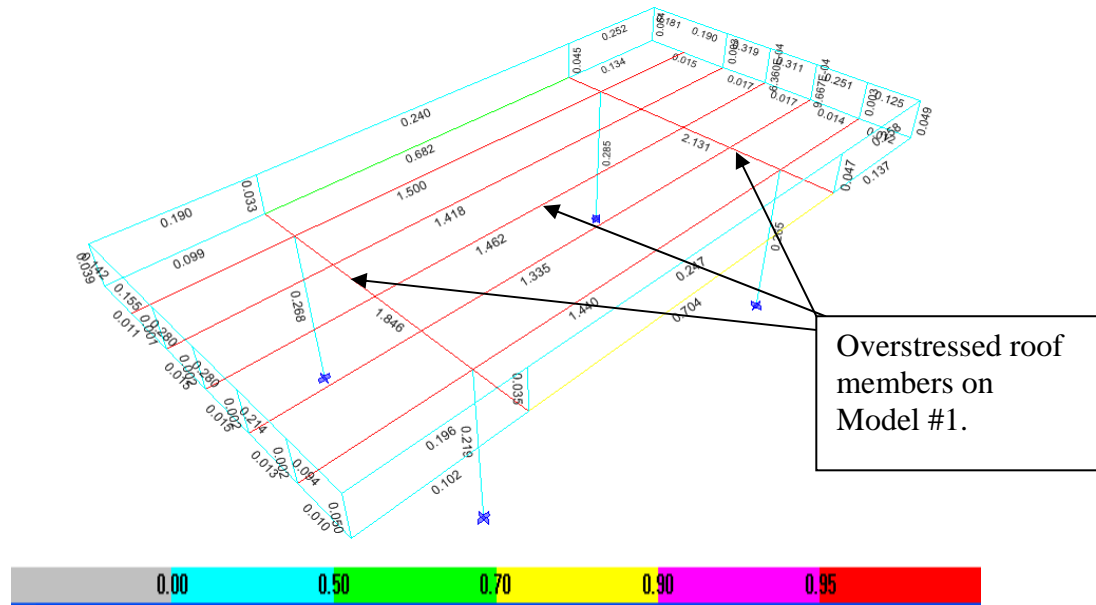


**Figure 8.15: Structural model for Case #1 on ETABS.**

Figure 8.16 show the deflected shape of the open canopy for Case #1 using Method #3. The actual deflection on the roof edge is about 2.5" in the downward direction. Results from the stress analysis are shown on Figure 8.17. Notice that only one wind case is considered here, since the method combines the extreme values of  $0^\circ$  and  $30^\circ$  incident winds. The differences between using Method #2 and #3 are small. For the leeward beam considered previously, Method #2 leads to 1.889 (stress ratio at 0 degrees) and to 2.144 (stress ratio at 30 degrees), whereas Method #3 (which combines both orientations) yields 2.131. The long beams on the roof reach values of 1.5. Notice that Method #3 does not represent a lower bound in terms of stress ratios, but is very close to that.



**Figure 8.16: Structural model on ETABS showing deflected shape due to wind pressures.**



**Figure 8.17: Structural model on ETABS showing overstress ratios on roof steel members.**

Stress ratios on this structural model show overstressed ratios on the majority of the roof beams. The graphical result shown is an envelope of all possible load combinations generated by wind load cases, gravity loads and wind loads. The overstressed members in this case study coincide with the failed steel elements shown in Figures 8.18 to 8.22. The failed elements coincides with the analysis using a wind velocity as specified by the code, using the proposed  $C_n$  values as suggested. All predictions need to be compared with evidence collected during site visits, which are available in the form of photographs. If the  $C_n$  values proposed in this work had been applied to carry out the original design, then, the main wind force resisting system (MWFRS) should have been able to withstand the imposed wind pressures and to perform satisfactory during hurricane Rita.

Therefore, Method #3 is simpler to use. It covers the extreme conditions from the wind direction at 0 degrees and 30 degrees, and it includes the influence of different parapets heights.

Because of the simplicity of data entry of the load in Method #3 as compared with Methods 1 and 2, Method #3 will be used for the rest of the cases investigated in this chapter.



**Figure 8.18: Case #1 collapsed roof layout showing some of the roof beams damages. Permanent deformation is clearly shown on some of the roof members.**



**Figure 8.19: Case #1 partial collapsed roof from below. Bottom of roof cladding has been taken away due to high wind pressures.**



**Figure 8.20: Case #1 roof cladding and beam structural layout. Photograph showing permanent deformation on roof members.**



**Figure 8.21: Case #1 transversal view of the deformed and collapsed roof.**

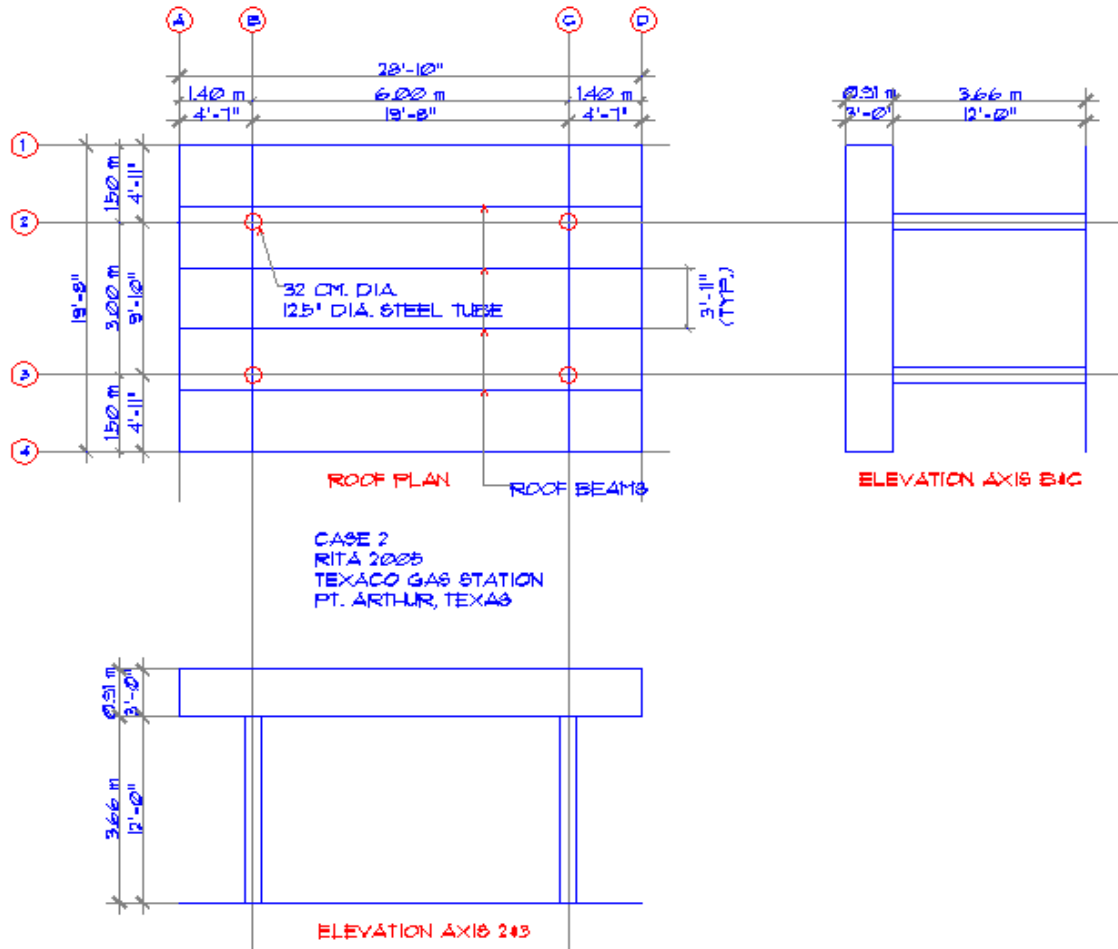


**Figure 8.22: Case #1, close up photograph of roof beams showing buckling and extreme corrosion damage.**

## 8.5 Case study #2, Texaco Gas Station, Port Arthur, Texas

The case study number 2 is a Texaco gas station located in 39th Ave., Pt. Arthur, Texas. Figure 8.23 shows the roof plan and elevations for Case #2. Measurements were taken directly from the field. The floor plan dimensions are 6.1 m (20 ft) in width by 8.8 m (29 ft) in length and 4.3 m (14 ft) in height. The open canopy has a system of four steel columns of 0.3 m (1 ft) diameter with steel beams and channels for the gravity system of the structure. Smallest thickness of steel columns and beams were used for the analysis. A steel deck was used for the roof cladding. A parapet of 0.91 m (3 ft) is on the roof perimeter. There is no data available for the foundation size and depth.

For the current case study, calculations were based on Method #3, which uses the guidelines provided by ASCE 7-05. From the current results, the  $C_n$  values from Chapter 7 were selected. Refer to Figure 8.24 for calculated wind pressure results and  $C_n$  values used. As in other cases, the calculations were done with Maple, and show values that vary from +33.1 psf (+1.61 kPa) (downward pressure, zone 1), -16.5 psf (-0.79 kPa) (uplift pressure, zone 2) to +22 psf (1.05 kPa) (downward pressure, zone 3).



**Figure 8.23: Canopy floor plan and elevations for Case study #2.**

Parapet wind pressure values used for structural analysis are shown on Figure 8.13. Wind pressures of 43.88 psf (2.1 kPa) on the windward direction and 31.3 psf (1.5 kPa) on the leeward direction were applied to the structural model. Wind direction was applied on both horizontal axis and wind pressures on the parapets are applied at each wind direction.

The obtained wind pressures obtained and shown on Table 8.2 using the ASCE 7-05 method and the proposed  $C_n$  values from this thesis were imposed on the structural model for the analysis on Case #2. Refer to Figure 8.26 for the 3D model used for the structural analysis of the

structure on Case #2. Open canopy plan geometry, height, column and beam sizes, were included in the structural model. Wind pressures from Table 8.2 were included and applied to the model.

Stress ratios computed for the structural model show stress ratios values larger than one on all steel columns (Figure 8.26). The graphical result shown in this figure is an envelope of all possible load combinations generated by wind load cases, gravity loads and wind loads, and evaluated using the steel design code (AISC 1989). Unlike what was obtained in our Case study #1, in which the roof beams reached failure, the columns are the weak elements in Case study #2, whereas all roof elements are in safe state. The overstressed members in this case study are the same as the failed steel elements shown on Figure 8.28 to 8.32. The columns used on this open canopy are very slender, and just by looking at the dimensions, only six inches in diameter. Therefore, it does not surprise the author, that they failed so drastically under imposed wind pressures. Plastic hinges are observed in all four columns, and because of the advanced state of deterioration it seems that failure occur for loads (and wind velocities) lower than those considered in the analysis.

**#CASE #2 ASCE 7-05 section 6.5.13 Design on wind loads on open buildings page 29**

$V := 130$ ;  $I_{factor} := 1.0$ ;  $K_z := 0.85$ ;  $K_{zt} := 1.0$ ;  $K_d := 0.85$ ;

#  $K_z$  table 6-3 page 79

#  $K_{zt}$  section 6.5.7.2

#  $K_d$  wind directionality factor table 6-4 page 80

$q := .00256 \cdot K_z \cdot K_{zt} \cdot K_d \cdot V^2 \cdot I_{factor}$  # 6.5.10 (6-15) velocity pressure

$q := 31.25824000$

# Gust effect factor rigid structures

$g_q := 3.4$  # constant 6.5.8.1

$g_v := 3.4$ ;

$z_{bar} := 12$  # equivalent height in structure

$c := 0.2$  # turbulence intensity factor in 6-5 from Table 6-2

$l := 500$  # integral length scale factor from Table 6-2

$e_{bar} := \frac{1}{5}$  # integral length scale power law exponent in 6-7 from Table 6-2

$B := 29$  # horizontal dimension measured normal to wind direction

$h := 12$  # mean roof height of a building

$I_{zbar} := \text{evalf}\left(c \cdot \left(\frac{33}{z_{bar}}\right)^{\frac{1}{6}}\right)$  # 6.5.8.1 (6-5)

$I_{zbar} := 0.2367293530$

$L_{zbar} := \text{evalf}\left(1 \cdot \left(\frac{z_{bar}}{33}\right)^{e_{bar}}\right)$  # 6.5.8.1 (6-7)

$L_{zbar} := 408.4166753$

$Q := \sqrt{\frac{1}{1 + 0.63 \cdot \left(\frac{(B+h)}{L_{zbar}}\right)^{0.63}}}$  # 6.5.8.1 (6-6)

$Q := 0.9332976941$

$G := 0.925 \cdot \left(\frac{(1 + 1.7 \cdot g_q \cdot I_{zbar} \cdot Q)}{1 + 1.7 \cdot g_v \cdot I_{zbar}}\right)$  # 6.5.8.1 (6-4)

$G := 0.8893527027$

$C_{n1} := 1.2$  # PROPOSED CNVALUES

$C_{n2} := -0.6$ ;

$C_{n3} := 0.8$ ;

$q1 := q \cdot G \cdot C_{n1}$ ;  $h1 := 5$ ;

$q2 := q \cdot G \cdot C_{n2}$ ;  $h2 := 15$ ;

$q3 := q \cdot G \cdot C_{n3}$ ;  $h3 := B - h1 - h2$ ;

Wind pressure in psf. H=horizontal distance on the roof surface.

Proposed Cn values by the author to be used for Method 3, simplified design.

$q1 := 33.35952028$

$h1 := 5$

$q2 := -16.67976014$

$h2 := 15$

$q3 := 22.23968018$

$h3 := 9$

**Figure 8.24: Maple output of wind pressure calculations using proposed Cn values with ASCE 7-05 procedure, Case# 2.**

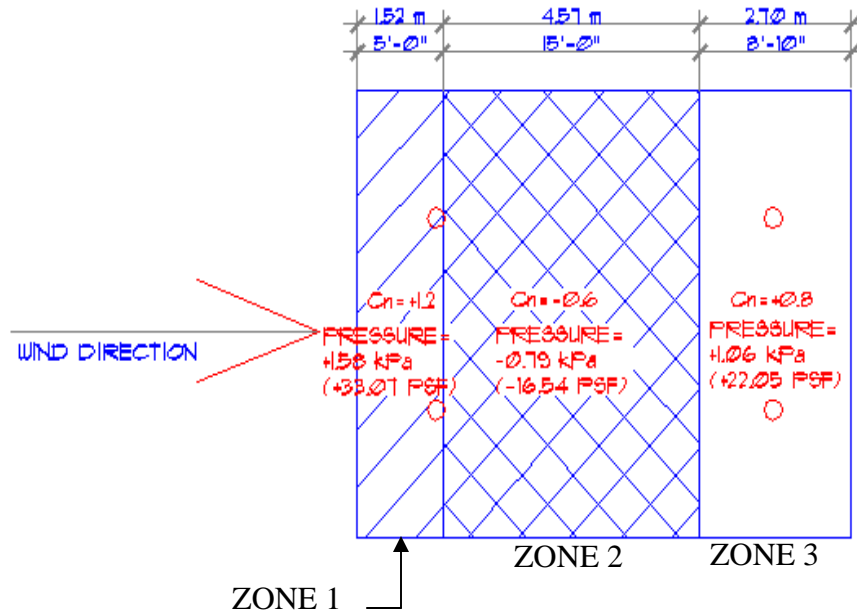
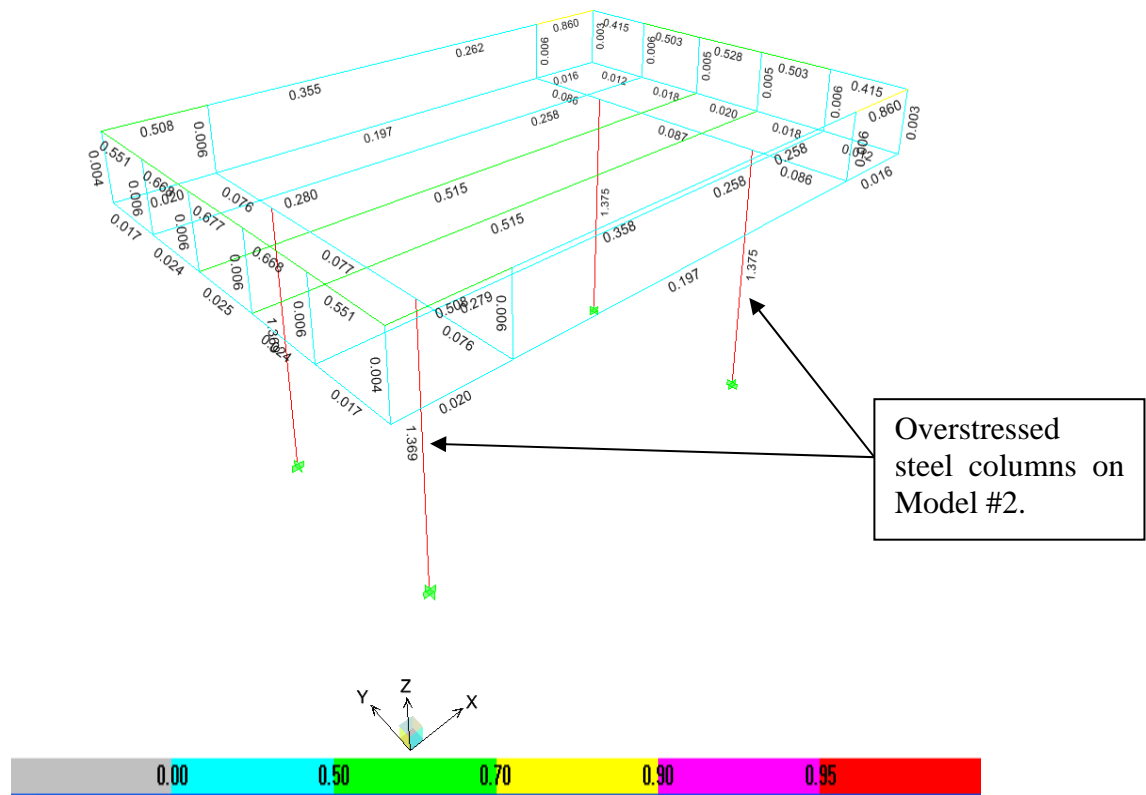


Figure 8.25: Roof canopy pressure  $C_n$  and wind pressures for case #2, + means downward pressure, - means uplift pressure.

Table 8.2: Table of wind pressures for Case #2 to be used in structural analysis.

| Case Study # | 2     | Zone Length |        |       |       |      |
|--------------|-------|-------------|--------|-------|-------|------|
| Zone #       | $C_n$ | Pressure    | psf    | kPa   | h(ft) | h(m) |
| 1            | 1.2   | $q_1 =$     | 33.04  | 1.58  | 5     | 1.5  |
| 2            | -0.6  | $q_2 =$     | -16.54 | -0.79 | 15    | 4.6  |
| 3            | 0.8   | $q_3 =$     | 22.05  | 1.06  | 9     | 2.7  |



**Figure 8.26: Structural model on ETABS showing overstress ratios on steel columns.**



**Figure 8.27: Case #2, open canopy structure completely collapsed.**



**Figure 8.28: Case #2, photograph showing a buckled round steel column.**



**Figure 8.29: Case #2, closer photograph of the round steel column base.**



**Figure 8.30: Case #2, photograph showing the open canopy structure on top of the gas pumps.**

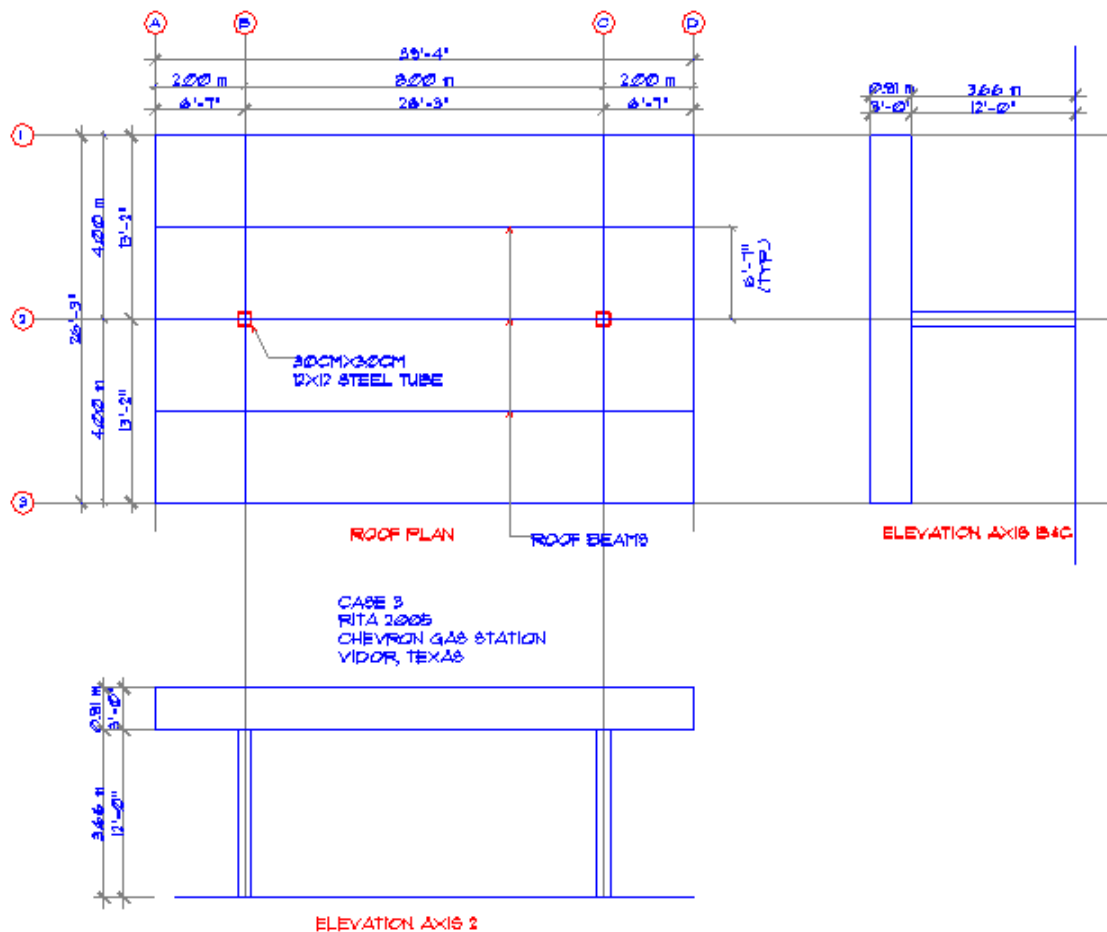


**Figure 8.31: Case #2, photograph showing the roof beam layout.**

## **8.6 Case study #3, Chevron Gas station, Vidor, Texas**

The case study number 3 is a Chevron gas station located in Road 105 and HW 10, Vidor, Texas. A roof plan and elevations are illustrated in Figure 8.32 and were taken from measurements in the field. The floor plan dimensions are approximately 8.0 m (26 ft) in width by

12.2 m (40 ft) in length and 4.3 m (14 ft) in height. It has a system of two steel columns of 0.3 m (1 ft) x 0.3 m (1 ft) with steel beams and channels for the gravity system of the structure. Smallest thickness of steel columns and beams were used for the analysis. A steel deck was used for the roof cladding. A parapet of 0.91 m (3 ft) is on the roof perimeter. There is no data available concerning the foundation size and depth.



**Figure 8.32: Canopy floor plan and elevations for Case study #3.**

For the current case study, calculations are based on Method #3, which follows the guidelines provided by ASCE 7-05. The calculated wind pressure results and used  $C_n$  values are

shown in Figure 8.33. The values vary from +23.8 psf (+1.14 kPa) (downward pressure, zone 1), -11.9 psf (-.57 kPa) (uplift pressure, zone 2) to +15.8 psf (0.76 kPa) (downward pressure, zone 3).

Parapet wind pressure values used for structural analysis are shown in Figure 8.34. Wind pressure values were applied to the structural model on the parapets. Wind pressures of 33.6 psf (1.6 kPa) on the windward direction and 22.4 psf (1.1 kPa) on the leeward direction are applied to the structural model. Wind direction is applied on both horizontal axis and wind pressures on the parapets are applied at each wind direction.

On Figure 8.36 the structural analysis of the open canopy does not indicate overstress ratios on the roof members. This specific case was characterized by extensive corrosion at the base of the columns, associated with the drainage system of the roof. Thus, the steel columns were modeled to represent the loss of material due to corrosion. Because the actual extent of corrosion was difficult to evaluate, an initial value of the columns thickness was used as 1/4" (6.4 mm), and the value was parametrically reduced to almost half of the wall thickness. For the case of 1/8" (3.2 mm), the stress ratios computed are plotted in Figure 8.36. Both columns have values higher than one, whereas the roof members reach half of their capacity at most.

### # CASE 3 ASCE 7 — 05 section 6.5.13 Design on wind loads on open buildings page 29

$V := 110$  : Ifactor := 1.0 :  $K_z := 0.85$  :  $K_{zt} := 1.0$  :  $K_d := 0.85$  :

#  $K_z$  table 6-3 page 79

#  $K_{zt}$  section 6.5.7.2

#  $K_d$  wind directionality factor table 6-4 page 80

$q := .00256 \cdot K_z \cdot K_{zt} \cdot K_d \cdot V^2 \cdot \text{Ifactor}$  # 6.5.10 (6-15) velocity pressure

$q := 22.38016000$

# Gust effect factor rigid structures

$g_q := 3.4$  # constant 6.5.8.1

$g_v := 3.4$  :

Wind speed mph at case  
#3 location

$z_{bar} := 12$  # equivalent height in structure

$c := 0.2$  # turbulence intensity factor in 6-5 from Table 6-2

$l := 500$  # integral length scale factor from Table 6-2

$e_{bar} := \frac{1}{5}$  # integral length scale power law exponent in 6-7 from Table 6-2

$B := 39.5$  # horizontal dimension measured normal to wind direction

$h := 12$  # mean roof height of a building

$I_{zbar} := \text{evalf}\left(c \cdot \left(\frac{33}{z_{bar}}\right)^{\frac{1}{6}}\right)$  # 6.5.8.1(6-5)

$I_{zbar} := 0.2367293530$

$L_{zbar} := \text{evalf}\left(l \cdot \left(\frac{z_{bar}}{33}\right)^{e_{bar}}\right)$  # 6.5.8.1(6-7)

$L_{zbar} := 408.4166753$

$Q := \sqrt{\frac{1}{1 + 0.63 \cdot \left(\frac{(B+h)}{L_{zbar}}\right)^{0.63}}}$  # 6.5.8.1(6-6)

$Q := 0.9241384547$

$G := 0.925 \cdot \left(\frac{(1 + 1.7 \cdot g_q \cdot I_{zbar} \cdot Q)}{1 + 1.7 \cdot g_v \cdot I_{zbar}}\right)$  # 6.5.8.1(6-4)

$G := 0.8844577873$

$C_{n1} := 1.2$  # PROPOSED CNVALUES

$C_{n2} := -0.6$  :

$C_{n3} := 0.8$  :

$q1 := q \cdot G \cdot C_{n1}$ ;  $h1 := 5$ ;

Wind pressure in  
psf. H=horizontal  
distance on the roof  
surface.

$q2 := q \cdot G \cdot C_{n2}$ ;  $h2 := 15$ ;

$q3 := q \cdot G \cdot C_{n3}$ ;  $h3 := B - h1 - h2$ ;

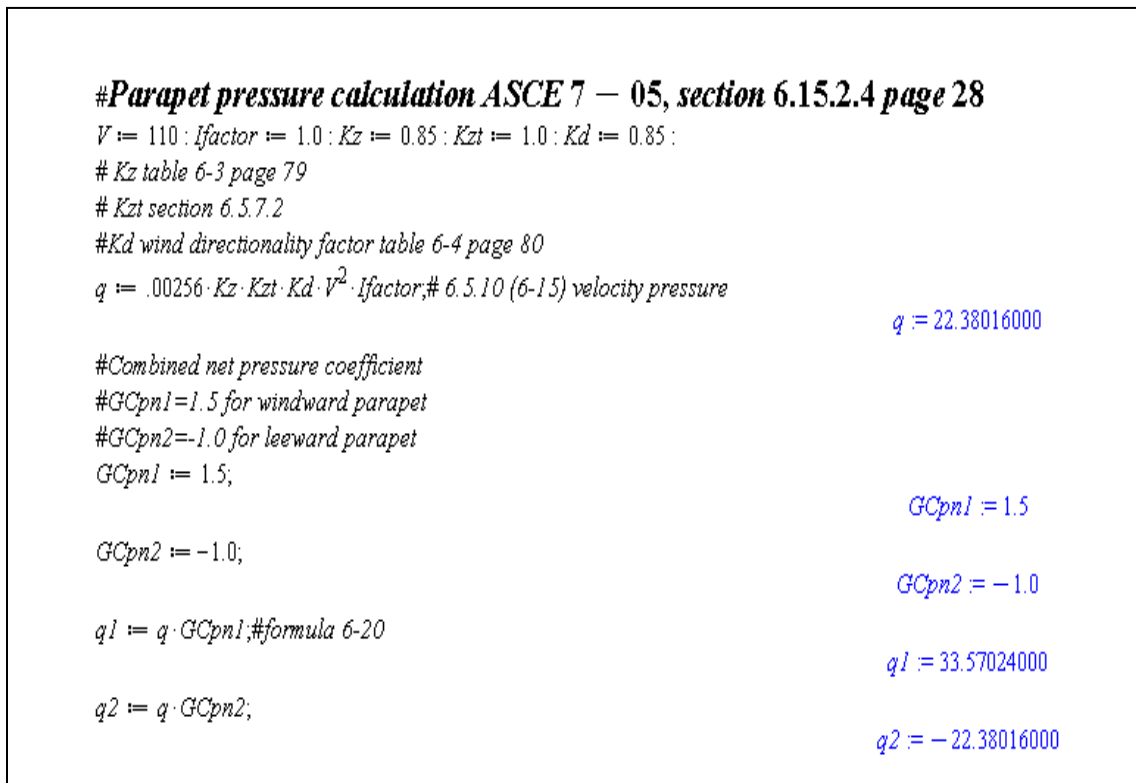
Proposed Cn values by  
the author to be used for  
Method 3, simplified  
design.

$q1 := 23.75316815$   
 $h1 := 5$

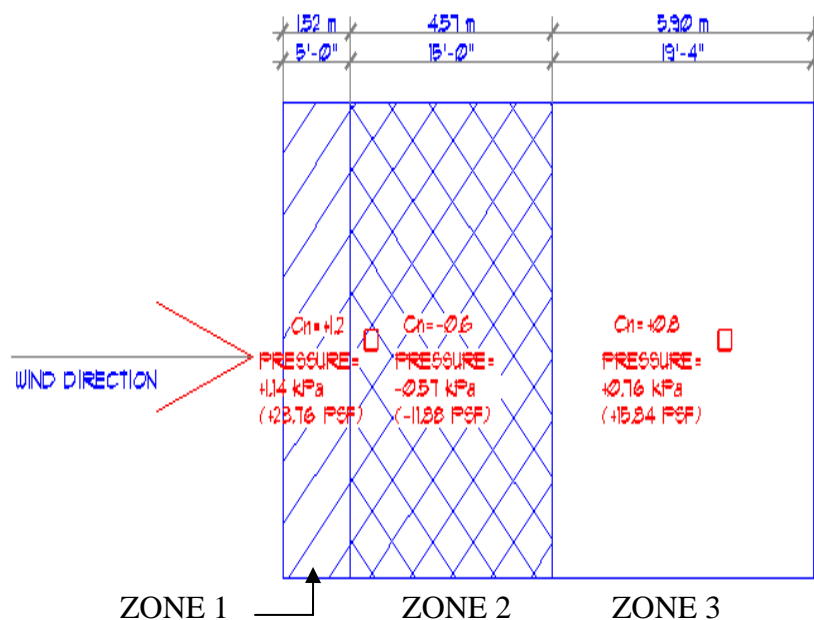
$q2 := -11.87658407$   
 $h2 := 15$

$q3 := 15.83544543$   
 $h3 := 19.5$

Figure 8.33: Maple output of wind pressure calculations using proposed Cn values with ASCE 7-05 procedure, Case# 3.



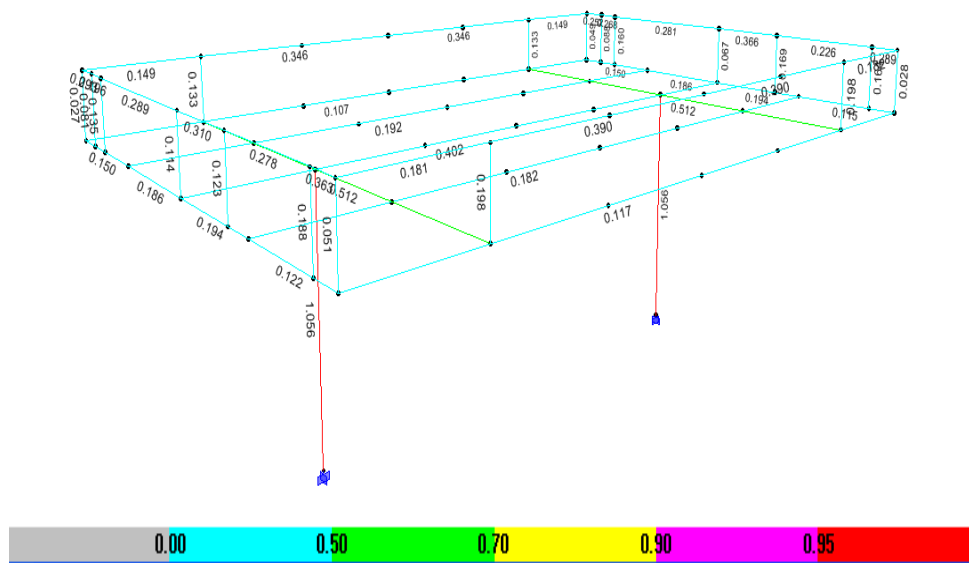
**Figure 8.34: Maple output of wind pressure results on parapets using ASCE 7-05 Cn values**



**Figure 8.35: Roof canopy pressure Cn and wind pressures for case #3, + means downward pressure, - means uplift pressure.**

**Table 8.3: Table of wind pressures for Case #3 to be used in structural analysis.**

| Case Study # | 3    | Zone Length |        |       |       |      |
|--------------|------|-------------|--------|-------|-------|------|
| Zone #       | Cn   | Pressure    | psf    | kPa   | h(ft) | h(m) |
| 1            | 1.2  | q1=         | 23.75  | 1.14  | 5     | 1.5  |
| 2            | -0.6 | q2=         | -11.87 | -0.57 | 15    | 4.6  |
| 3            | 0.8  | q3=         | 15.83  | 0.76  | 19.5  | 5.9  |



**Figure 8.36: Structural model on ETABS showing overstress ratios on roof steel members for Case #3.**

Figures 8.38 to 8.40 show the corrosion damage at the base of the steel columns, which is believed to have triggered the collapse of the structure. The steel columns failed below their stress capacity and even the complete structural layout remained intact after the failure of the steel columns. It is remarkable that the whole superstructure, the MWFRS and the secondary members, remained intact after the structure became upside down after the failure. Without the

corrosion damage on the existing structure, only the roof secondary beams would have suffered damage for the applied wind pressures.



**Figure 8.37: Case #3, open canopy photograph of the inverted steel structure after the steel columns failed due to corrosion and wind pressures.**



**Figure 8.38: Case #3, photograph of the existing corrosion condition at the column base.**

The deterioration of the column at its base illustrated in Figure 8.38 indicated that the drainage system of the roof, which conducts water through the columns, did not function properly and produced severe metal corrosion. This case calls for improvements in the drainage system (to avoid such drastic corrosion) and an effective maintenance schedule. Although just

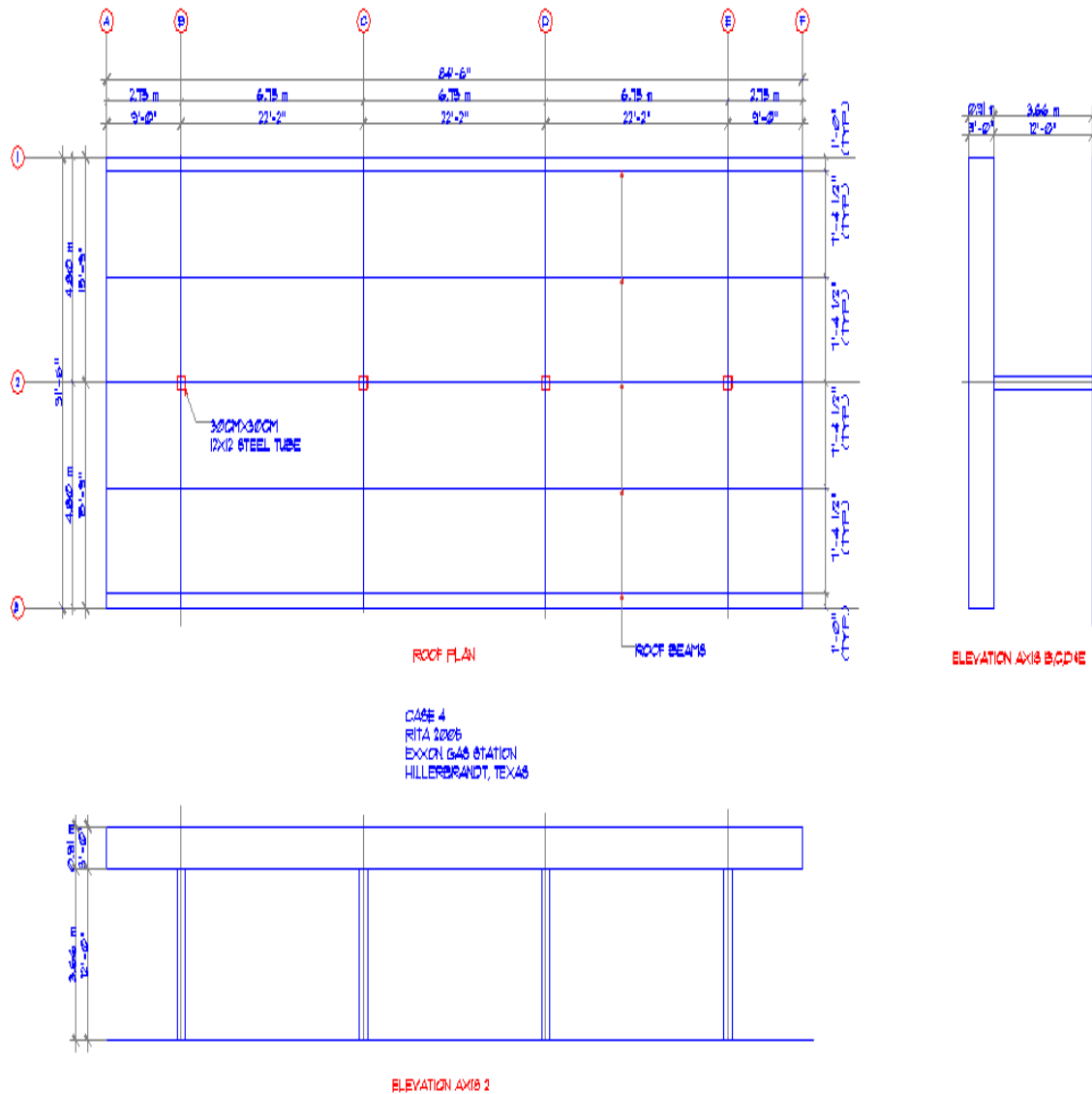
one case involving corrosion was analyzed in this chapter, this pattern of deterioration was observed in many cases during site visits.



**Figure 8.39: Case #3, additional photograph of the inverted steel structure.**

## **8.7 Case study #4, Exxon Gas station, Hillerbrandt, Texas**

The case study number 4 is an Exxon gas station located in Road 365, Hillerbrandt, Texas. A roof plan and elevations as illustrated in Figure 8.40 and were taken from field measurements. The floor plan dimensions are approximately 9.8 m (32 ft) x 25.0 m (85 ft) x 4.3 m (14 ft). It has a system of four steel columns of 0.3 m (1 ft) x 0.3 m (1 ft) with steel beams and channels for the gravity system of the structure. Smallest thickness of steel columns and beams were used for the analysis. A steel deck was used for roof cladding. A parapet of 0.91 m (3 ft) is on the roof perimeter. There is no data available for the foundation size and depth.



**Figure 8.40: Canopy floor plan and elevations for Case study #4.**

For the current case study, calculations are based on the guidelines provided by ASCE 7-05, Method #3. Figure 8.41 shows the calculated wind pressures from the proposed Cn, with values that varied from +33.1 psf (+1.61 kPa) (downward pressure, zone 1), -16.5 psf (-.79 kPa) (uplift pressure, zone 2) to +22 psf (1.05 kPa) (downward pressure, zone 3).

Parapet wind pressure values used for structural analysis are shown on Figure 8.13. Wind pressure values were applied to the structural model on the parapets. Wind pressures of 43.88 psf (2.1 kPa) on the windward direction and 31.3 psf (1.5 kPa) on the leeward direction are applied to the structural model. Wind direction is applied on both horizontal axis and wind pressures on the parapets are applied at each wind direction.

Stress ratios larger than one on this structural model show overstressed ratios on all steel columns. The values displayed in Figure 8.43 is an envelope of all possible load combinations generated by wind load cases, gravity loads and wind loads, using steel design code (AISC 1989). As reference values, the maximum service of the column axial load is 17.1 kips and the maximum service moment in the strong axis is 119 k-ft. The maximum ultimate column axial load is 28.2 kips and the maximum ultimate moment in the strong axis is 154.5 k-ft.

In this particular case is very interesting to identify from the photographs the buckling of the column flanges. The maximum capacity of the flanges before the local flange goes into plastic deformation is 138.9 kips. From the computations, it seems that plastic hinge in one of the local flanges has initially formed, causing the rotation and failure of all steel columns. The actual computational model results using Method #3 produce results that are comparable to the failure mechanism and loads that the actual structure suffered from wind forces. The overstressed members in this case coincide with the failed steel column elements shown on Figure 8.45 to 8.48.

**#CASE #4 ASCE 7-05 section 6.5.13 Design on wind loads on open buildings page 29**

$V := 130$ ;  $I_{factor} := 1.0$ ;  $K_z := 0.85$ ;  $K_{zt} := 1.0$ ;  $K_d := 0.85$ ;

#  $K_z$  table 6-3 page 79

#  $K_{zt}$  section 6.5.7.2

#  $K_d$  wind directionality factor table 6-4 page 80

$q := .00256 \cdot K_z \cdot K_{zt} \cdot K_d \cdot V^2 \cdot I_{factor}$  # 6.5.10 (6-15) velocity pressure

$q := 31.25824000$

# Gust effect factor rigid structures

$g_q := 3.4$  # constant 6.5.8.1

$g_v := 3.4$ ;

Wind speed mph at case #4 location

$z_{bar} := 12$  # equivalent height in structure

$c := 0.2$  # turbulence intensity factor in 6-5 from Table 6-2

$l := 500$  # integral length scale factor from Table 6-2

$e_{bar} := \frac{1}{5}$  # integral length scale power law exponent in 6-7 from Table 6-2

$B := 84.5$  # horizontal dimension measured normal to wind direction

$h := 12$  # mean roof height of a building

$I_{zbar} := \text{evalf}\left(c \cdot \left(\frac{33}{z_{bar}}\right)^{\frac{1}{6}}\right)$  # 6.5.8.1(6-5)

$I_{zbar} := 0.2367293530$

$L_{zbar} := \text{evalf}\left(l \cdot \left(\frac{z_{bar}}{33}\right)^{e_{bar}}\right)$  # 6.5.8.1(6-7)

$L_{zbar} := 408.4166753$

$Q := \sqrt{\frac{1}{1 + 0.63 \cdot \left(\frac{(B+h)}{L_{zbar}}\right)^{0.63}}}$  # 6.5.8.1(6-6)

$Q := 0.8930486990$

$G := 0.925 \cdot \left(\frac{(1 + 1.7 \cdot g_q \cdot I_{zbar} \cdot Q)}{1 + 1.7 \cdot g_v \cdot I_{zbar}}\right)$  # 6.5.8.1(6-4)

$G := 0.8678426830$

$Cn1 := 1.2$  # PROPOSED CN VALUES

$Cn2 := -0.6$ ;

$Cn3 := 0.8$ ;

$q1 := q \cdot G \cdot Cn1$ ;  $h1 := 5$ ;

$q2 := q \cdot G \cdot Cn2$ ;  $h2 := 15$ ;

$q3 := q \cdot G \cdot Cn3$ ;  $h3 := B - h1 - h2$ ;

Wind pressure in psf. H=horizontal distance on the roof surface.

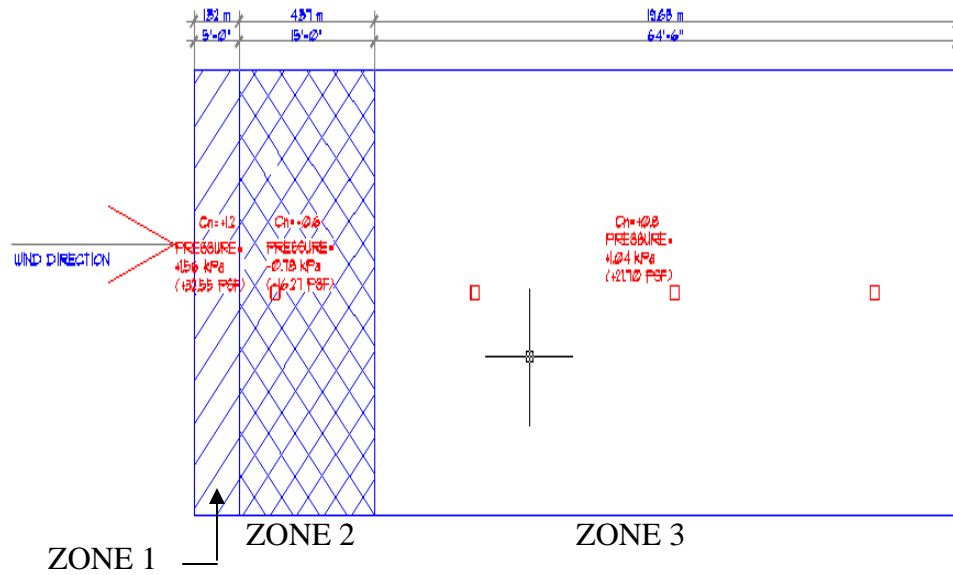
Proposed Cn values by the author to be used for Method 3, simplified design.

$q1 := 32.55268184$   
 $h1 := 5$

$q2 := -16.27634092$   
 $h2 := 15$

$q3 := 21.70178790$   
 $h3 := 64.5$

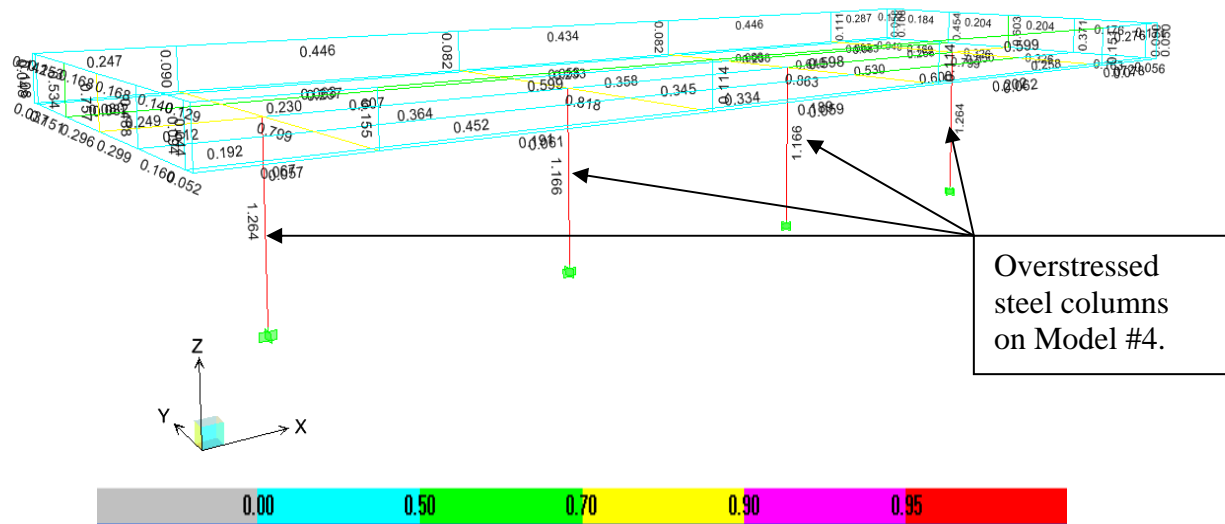
**Figure 8.41: Maple output of wind pressure calculations using proposed Cn values with ASCE 7-05 procedure, Case# 4.**



**Figure 8.42: Roof canopy pressure  $C_n$  and wind pressures for case #4, + means downward pressure, - means uplift pressure.**

**Table 8.4: Table of wind pressures for Case #4 to be used in structural analysis.**

| Case Study # | 4     | Zone Length |        |       |       |      |
|--------------|-------|-------------|--------|-------|-------|------|
| Zone #       | $C_n$ | Pressure    | psf    | kPa   | h(ft) | h(m) |
| 1            | 1.2   | $q_1 =$     | 32.53  | 1.56  | 5     | 1.5  |
| 2            | -0.6  | $q_2 =$     | -16.27 | -0.78 | 15    | 4.6  |
| 3            | 0.8   | $q_3 =$     | 21.7   | 1.04  | 64.5  | 19.7 |



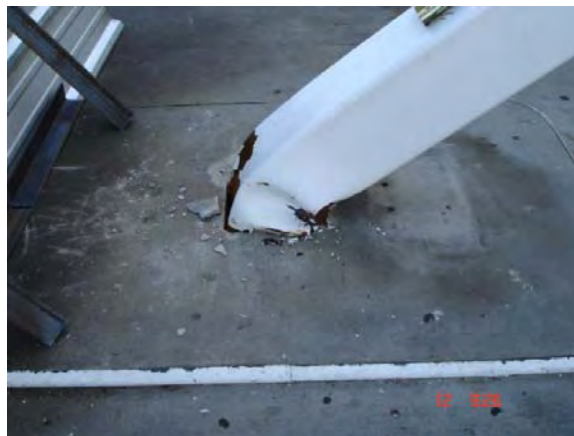
**Figure 8.43: Structural model generated with ETABS showing overstress ratios on all steel columns.**



**Figure 8.44: Case #4, photograph showing closer detail of the buckled steel columns.**



**Figure 8.45: Case #4, photograph showing the collapsed steel columns.**



**Figure 8.46: Case #4, additional photograph of buckled steel columns. The photograph shows the internal column flange.**



**Figure 8.47: Case #4, upper roof beam layout on the collapsed open canopy roof.**

## 8.8 Conclusions drawn from the Case Studies

The  $C_n$  values obtained from wind tunnel experiments and CFD simulation were obtained on Chapter 6. CFD simulations could provide  $C_n$  values for different geometric configurations and were investigated in Chapter 7. In Chapter 8, three methods of performing structural analysis have been investigated. Method #1, with the actual wind pressures, requires large computer time and effort. Method #2, a simplified version of Method #1, investigates the wind effects at 0 and 30 degrees but using bands of pressures rather than detailed local pressures. The suggested Method #3, using the ASCE 7-05 procedure with the modified  $C_n$  factors, combines the wind pressure coefficients in both directions. It has been possible to predict overstressed members and possible failures on open canopy structures using all three methods. In all cases calculated with wind pressures based on Method #3, the suggested  $C_n$  values could confirm the expected overstressed members that lead to the catastrophic collapse of the presented cases.

With the use of the suggested Method #3, the safe design of open canopy structures could be made. The design is conservative only by 18 percent, making the simplified Method #3, a very easy and accessible design method for the structural engineering community.

## **CHAPTER 9. CONCLUSIONS, ORIGINAL RESULTS, FUTURE RESEARCH AND RECOMMENDATIONS**

### **9.1 Conclusions**

There is conclusive evidence that a large number of open canopy structures suffered catastrophic damage due to hurricanes in the US during the last decade. This calls for an urgent reassessment of current design methods and provisions used for this class of structures. The most recent version of the ASCE 7 includes considerations for open structures; however, the recommendations do not account for cases of open structures with parapets. For the specific geometries used in this investigation, the calculated  $C_n$  values for an open structure without parapets using ASCE 7 are negative on the complete surface and range from -0.8 to -0.3 for the Main Wind Force Resisting System (MWFRS) (ASCE 6.5.13, Table 3.5.7).

Based on the evidence presented in this thesis, the following conclusions can be made:

- 1) Current ASCE-7 wind provisions are not adequate to estimate wind pressures on open canopies with parapets. The  $C_n$  results from wind tunnel and computational simulations show that there is a strong influence of the parapets on the pressure coefficients of a canopy structure; this influence was observed in both, wind tunnel testing and Computational Fluid Dynamics (CFD) simulations, and good agreement has been obtained between the two independent methodologies. For the cases studied, using numerical and wind tunnel simulations, the  $C_n$  values obtained in this investigation have a clear pattern of downward pressure, followed by suction, followed by downward pressure along the center line in the direction of the wind. The specific values varies from +1.2 (zone 1), -0.6 (zone 2) and +0.8 (zone 3). The mean roof pressures vary from pressure downward to upward

pressure, depending on the specific locations considered in the roof along the long axis. These values significantly differ from the recommended code values, which predict that only suction or only downward pressure acts in the roof. Further, the resultant forces caused by these  $C_n$  values are 40% higher than the recommended values suggested by the current ASCE 7-05. We conclude that ASCE recommendations for this type of structures are not sufficient to carry out safe designs of canopy structures with parapets. It is suggested that current code recommendations for the MWFRS should be modified for open structures and should include new coefficients for open structures with parapets, such as those presented in this research.

- 2) Parapets have a significant influence on the resultant wind pressures. Wind pressures were first compared on open canopies without parapets with current available data.  $C_n$  results derived from wind tunnel and computational simulations show that there is a strong influence of the parapets on the pressure coefficients of a canopy structure. This influence was observed in both, wind tunnel testing and CFD simulations and good agreement has been obtained between the two independent methodologies. CFD simulations of different plan geometries with different parapet heights verify such conclusion.
- 3) CFD simulations are an adequate tool for the investigation of open canopies with parapets. With the use of the data from the wind tunnel of UNNE, for an open canopy small scale model, a CFD simulation was developed. The  $C_n$  values obtained from the CFD simulation are in good agreement with the wind tunnel model results. Because of the similarity of results and the comparative cost of the

two methods, the present studies conclude that CFD is an adequate tool to perform further parametric studies and investigation of failure cases.

- 4) The present sequential methodology (in which pressures are first determined on a rigid model and subsequently applied to a deformable structure) can represent the failure of real structures occurred during hurricanes. Four cases of real structural collapses were investigated in Chapter 8, showing that the sequential analysis was capable of detecting the specific members that failed in the field.

## **9.2 Original Results**

The investigation reported in this thesis has provided original results in various topics. This is the first time that studies for an open canopy with parapets have been reported in the open literature. Open canopies are frequently used in the construction of civil engineering facilities, either as components of larger structures or as self-supported structures, so that the present results should be taken as a basis for establishing new recommendations in this field. An example of canopies used as self supported structures may be found in most gas stations throughout the nation, in which the roof covers gas pumps and vehicles.

In order to investigate the open canopy with parapets, a wind tunnel study was conducted on a boundary layer wind tunnel at the UNNE. Detailed  $C_p$  and mean pressure  $C_n$  values were obtained from those tests. In the case investigated, wind was applied at 0 and 30 degrees from the principal axis, because they lead to the most severe wind pressure coefficients. No other wind tunnel studies were available for canopies with parapets, so that the present set of results may serve as reference values for research conducted by other investigators.

Concurrent with the wind tunnel tests, a CFD investigation was made. Similar wind conditions, in comparison to the wind tunnel tests, were used in the computational study, considering the same wind directions. Similar  $C_p$  and  $C_n$  results were obtained from the two methods, thus confirming and validating the CFD results. This was an important result for the rest of the research, since parametric studies were further conducted using CFD rather than wind tunnel testing. Thus, the obtained CFD results provided the opportunity to investigate additional cases of this type of structures, including additional geometries and configurations. Three planar geometries were investigated, each of the geometries with four different parapet heights. Each parapet height was considered with two different wind directions. The total number of models obtained was twenty-four. With this investigation, graphical output, in 2D and 3D were obtained and extreme  $C_n$  values were used for investigating actual open canopy collapses.

Four case studies of collapsed open canopy structures were investigated using structural analysis software. With the obtained  $C_n$  values from the wind tunnel investigation and the CFD models, wind pressures from the suggested  $C_n$  values were obtained. The method used for obtaining the wind pressures, was that suggested by the ASCE 7-05. We used the  $C_n$  values obtained on this investigation for the final wind pressure definition. With the obtained wind pressures, structural analysis was performed using structural engineering software. For all the cases studied, the overstressed members coincide with the actual photos of the collapsed structure. This closes the research circle stating from experimental evidence which is used to build computational models that are in turn validated with field observations. The consistency of the research is given by two independent sets of evidence (experimental measures and field observations) used to validate a complete new methodology.

### 9.3 Future Research

Because this is the first time an open canopy with parapets is investigated, further research is considered necessary. We have investigated in this research the Main Wind Force Resisting System. Under severe winds, the collapse of open canopies cause a major disaster in terms of providing gas at the time of an emergency and damaging important facilities. That is why further research is necessary and suggested:

- 1) Investigation of the structure's cladding is very important for this type of construction. Depending on the structural layout, the cladding is a key point, where the structural integrity of the whole structure could be compromised, if not designed correctly.  $C_p$  coefficients for the roof cladding need to be included in future design codes.
- 2) CFD investigation of this type of structures using different simulation programs is suggested in order to compare various ways of representing what is essentially a boundary layer problem. Further calibration of wind energy and dissipation in order to replicate wind turbulence has been obtained in recent investigations and adopting those findings on further research is suggested.
- 3) Further investigation and development of structural details and standard construction practices is recommended. During this research we noticed that there is no uniformity in the construction practices for this type of structures. It is suggested to involve the gas industries to obtain structural drawings and the standards they are using, to help them improving current design practices for this type of structures. This would be necessary in order to investigate if the supplied structural plans need revision and further retrofit of existing gas stations is necessary.

- 4) Open canopies with parapets, depending on the structural design, have several collapse mechanisms. Some of those mechanisms have been identified after hurricanes, including lack of maintenance of the structure, lack of structural redundancy, lack of uniformity of structural details in terms of steel connections and foundation design. Further investigation of those collapse mechanisms would be beneficial in terms of evaluating existing open canopies with parapets and developing retrofitting techniques whenever necessary.

## **9.4 Final Recommendations**

Open canopy structures with parapets are not a trivial type of structure. When a catastrophic event, like a hurricane happens, a large number of those structures collapsed as previously illustrated. At present, it is necessary that those types of structures should be designed, with higher safety margins to withstand the applied forces of nature. Open canopy structures should be designed as an essential facility, due to their importance in terms of human risks and security at the time of an emergency.

Providing structural redundancy and adequate maintenance are some of the suggested recommendations for the design and construction of those types of structures. Redundancy is necessary for a robust structural frame system in order to resist the imposed wind forces. With the frame system, a foundation is necessary in order to counter act the effects of uplift and overturning imposed by the structure. That is why with proper code recommendations and further research, adequate and safe design of those structures will be able to be achieved.

Structural maintenance of open canopies is very important, requiring periodic inspection and repair as necessary. Corrosion of steel and deterioration on exposed surfaces, such as those

illustrated by photographs in Chapter 8, are clear indications of poor maintenance. Future inspections of existing open canopy structures could indicate the need for retrofit and reinforcement of such structures.

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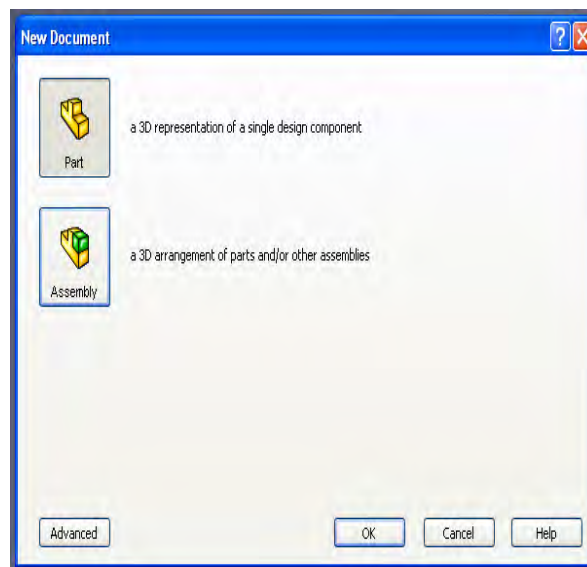
Research Center, University of Toronto Institute for Aerospace Studies, Toronto, Canada.

## APPENDIX A

### EFD.Lab modeling procedure

Several steps are required in order to create an EFD.Lab project, although the specific steps can differ from one project to another. For the cases investigated in this particular research, a step by step description of the procedure for creating an open canopy is as follows;

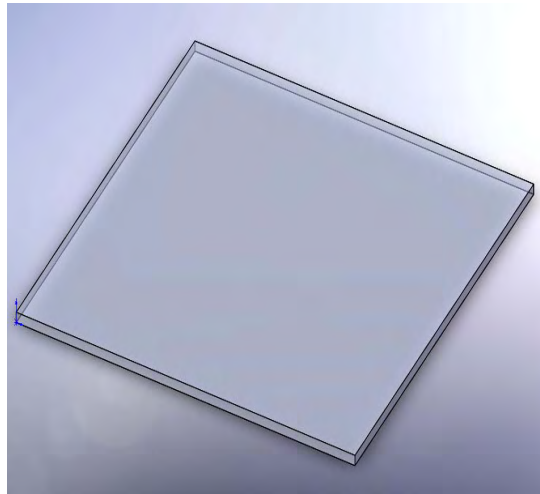
- 1) File, New, New document-At this step, it is possible to create a 3D component of an arrangement of components previously created. First we create the roof component as shown in Figure A. .



**Figure A.1: New document option on EFD.Lab**

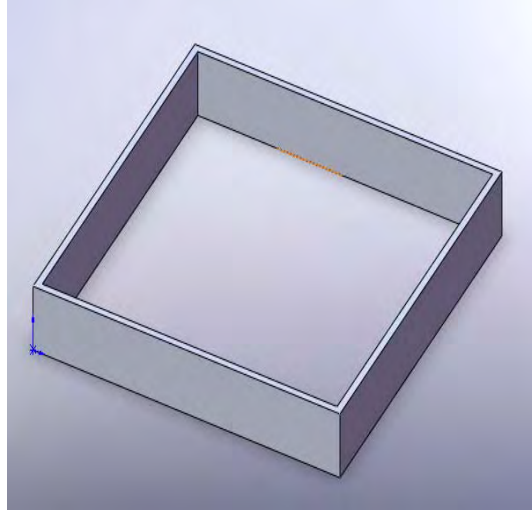
- 2) Using the sketch feature, we define an object to be used as the roof of our open canopy. In this particular case, a flat roof surface of 7.6 m (25 ft) x 7.6m (25ft) with 0.15 m (6 in) thickness is defined as shown in Figure A.2. EFD.Lab has built in a program named

Solidworks (SolidWorks 2009). It is used in the program for creating 3D objects that can have properties, like dimensions, thickness and materials, in order to create parts that can be assembled later for the final CFD analysis.



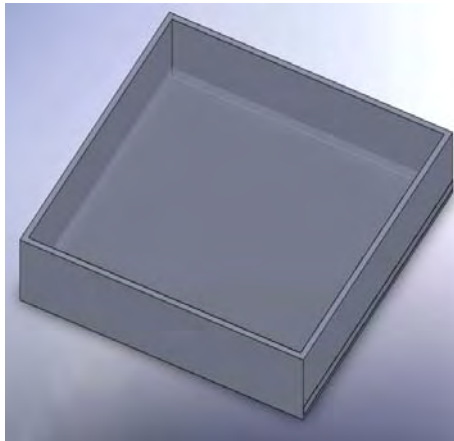
**Figure A.2: Roof surface created using EFD.Lab**

- 3) A new component, created with SolidWorks, to be used as the parapet of the open canopy is modeled with the same dimensions of the roof surface. The thickness of the parapet walls is 6 inches. Refer to Figure A.3.



**Figure A.3: Parapet element created using EFD.Lab**

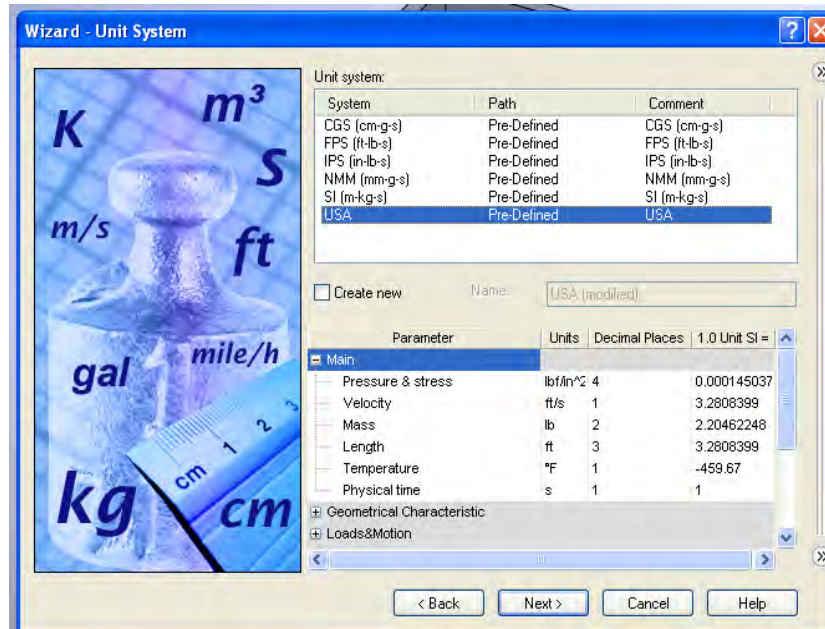
- 4) A third component, called an assembly was created using the previously created single parts, that combines the roof surface and the parapet created on the previous steps. This final component to be used in the analysis is shown in Figure A.4.



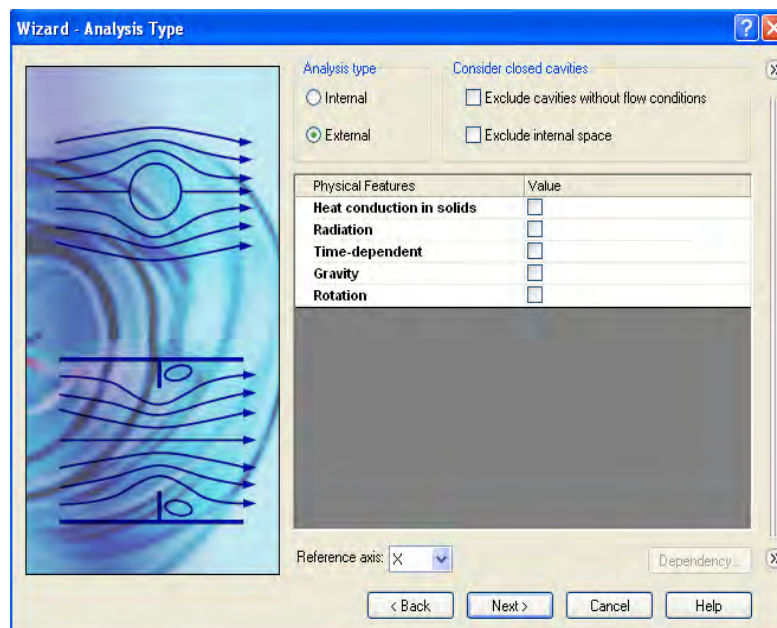
**Figure A.4: Canopy model combining roof element and parapet element using EFD.Lab**

- 5) Using the wizard project configuration, the user should specify the Units to used in the analysis, the analysis type (choosing between internal and external flow), default fluid to be used (in this case air), wall conditions, and initial and ambient conditions. With

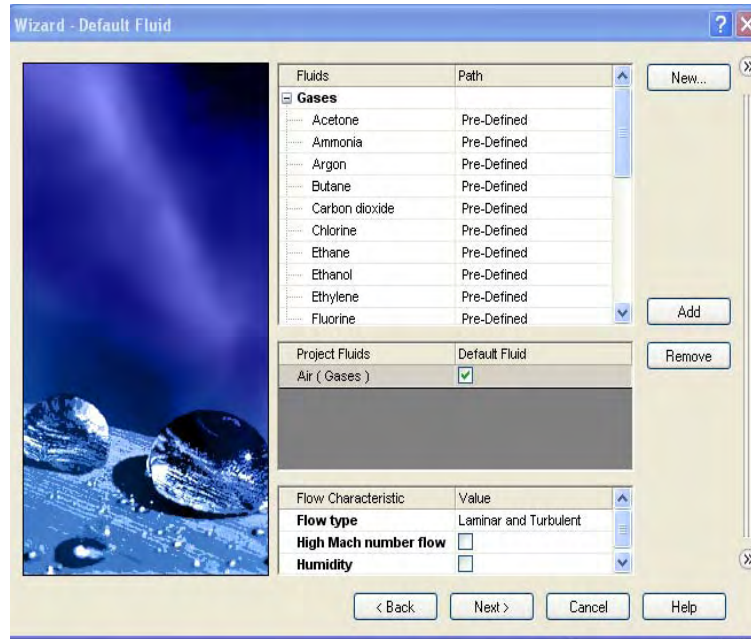
reference to Figure A.5 through Figure A.8, the parameters for pressure, temperature, wind velocity and turbulence parameters can be included.



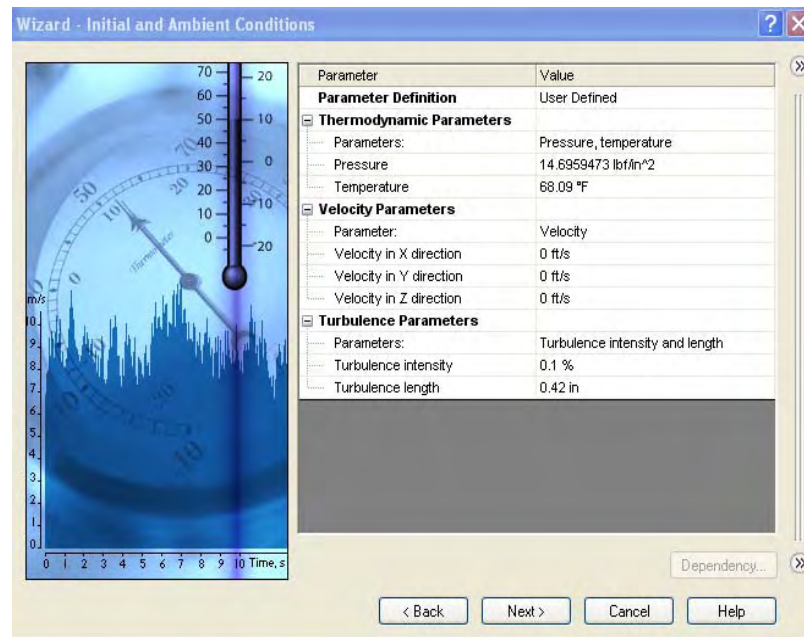
**Figure A.5: EFD.Lab unit selection wizard screen.**



**Figure A.6: EFD.Lab screen for input of internal or external analysis.**



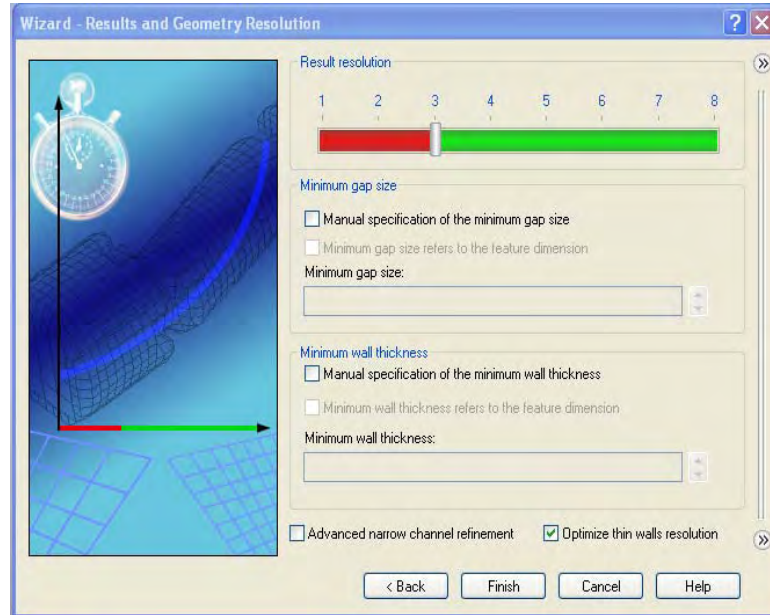
**Figure A.7: Fluid type and analysis type on EFD.Lab wizard.**



**Figure A.8: Initial and ambient conditions in EFD.Lab**

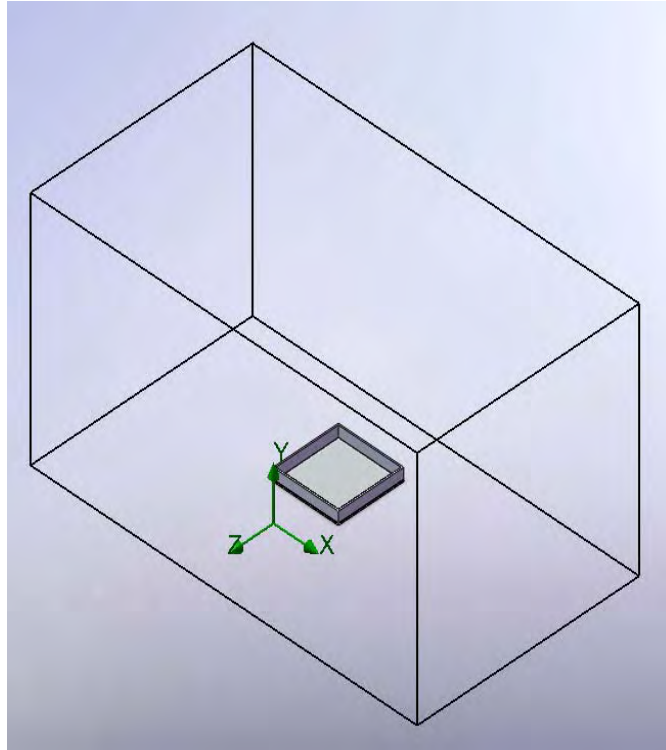
- 6) Results and geometry resolution is the default option that the program uses to mesh the domain. In this option, it is possible to specify the mesh density to be used in the

numerical solution. The program default allows for a quick response of a coarse solution. Further meshing and finer resolutions are changed in the final model for this investigation. Refer to Figure A.9.



**Figure A.9: Default meshing based on the geometry on EFD.Lab.**

- 7) An automatic domain region is created by the program. The program default can be modified to obtain a different sized region. For the current research, the modified simple boundary dimensions in the x, y and z directions respectively, 45.7 m (150 ft), 24.4 m (80 ft) and 25.9 m (85 ft). Refer to Figure A.10.



**Figure A.10: Model and computational domain in EFD.Lab.**

- 8) At this stage, the model is ready to be analyzed. The required time to solve the mathematical model depends on the size of the domain and the meshing on the cells.

## APPENDIX B

Wind tunnel data: Transducer voltage data obtained at the wind tunnel at UNNE for the open canopy with parapets scale model.

Facultad de Ingeniería – UNNE  
Túnel de viento "J. P. Gorecki"



**PRESIONES MEDIAS EXTERIORES EN TECHO DE MODELO 1 CON PARAPETOS**

Fecha 18/02/09 Hora de inicio 9:12  
 $\alpha = 30^\circ$  Hora de finalización 1

|    | Voltaje medio | Temp | Presión atm. | $q_{ref}$ |
|----|---------------|------|--------------|-----------|
| 0  |               | 28°  | 759          |           |
| 1  | 3.440         |      |              |           |
| 2  | 3.424         |      |              |           |
| 3  | 3.390         |      |              |           |
| 4  | 3.369         |      |              |           |
| 5  | 3.349         |      |              |           |
| 6  | 3.414         |      |              |           |
| 7  | 3.399         |      |              |           |
| 8  | 3.369         |      |              |           |
| 9  | 3.389         |      |              |           |
| 10 | 3.356         |      |              |           |
| 11 | 3.333         |      |              |           |
| 12 | 3.401         |      |              |           |
| 13 | 3.382         |      |              |           |
| 14 | 3.433         |      |              |           |
| 15 | 3.384         |      |              |           |
| 16 | 3.364         |      |              |           |

$\alpha = 30^\circ$

|    | Voltaje medio | Temp | Presión atm. | $q_{ref}$ |
|----|---------------|------|--------------|-----------|
| 0  |               |      | 759          |           |
| 1  | 3.390         |      |              |           |
| 2  | 3.321         |      |              |           |
| 3  | 3.323         |      |              |           |
| 4  | 3.324         |      |              |           |
| 5  | 3.329         |      |              |           |
| 6  | 3.333         |      |              |           |
| 7  | 3.326         |      |              |           |
| 8  | 3.322         |      |              |           |
| 9  | 3.321         |      |              |           |
| 10 | 3.326         |      |              |           |
| 11 | 3.326         |      |              |           |
| 12 | 3.327         |      |              |           |
| 13 | 3.320         |      |              |           |
| 14 | 3.332         |      |              |           |
| 15 | 3.352         |      |              |           |
| 16 | 3.325         |      |              |           |

**PRESIONES MEDIAS EXTERIORES EN TECHO DE MODELO 1 CON PARAPETOS**

Fecha

Hora de inicio

$\alpha = 210^\circ$

Hora de finalización 10:59

|    | Voltaje medio | Temp | Presión atm. | Qref |
|----|---------------|------|--------------|------|
| 0  |               |      |              |      |
| 1  | 3.552         |      |              |      |
| 2  | 3.535         |      |              |      |
| 3  | 3.518         |      |              |      |
| 4  | 3.523         |      |              |      |
| 5  | 3.534         |      |              |      |
| 6  | 3.530         |      |              |      |
| 7  | 3.504         |      |              |      |
| 8  | 3.484         |      |              |      |
| 9  | 3.489         |      |              |      |
| 10 | 3.459         |      |              |      |
| 11 | 3.460         |      |              |      |
| 12 | 3.463         |      |              |      |
| 13 | 3.448         |      |              |      |
| 14 | 3.417         |      |              |      |
| 15 | 3.422         |      |              |      |
| 16 | 3.415         |      |              |      |



$\alpha = 150^\circ$

|    | Voltaje medio | Temp | Presión atm. | Qref |
|----|---------------|------|--------------|------|
| 0  |               | 3.00 |              |      |
| 1  | 3.459         |      |              |      |
| 2  | 3.427         |      |              |      |
| 3  | 3.423         |      |              |      |
| 4  | 3.479         |      |              |      |
| 5  | 3.530         |      |              |      |
| 6  | 3.431         |      |              |      |
| 7  | 3.402         |      |              |      |
| 8  | 3.452         |      |              |      |
| 9  | 3.383         |      |              |      |
| 10 | 3.380         |      |              |      |
| 11 | 3.461         |      |              |      |
| 12 | 3.358         |      |              |      |
| 13 | 3.332         |      |              |      |
| 14 | 3.326         |      |              |      |
| 15 | 3.370         |      |              |      |
| 16 | 3.429         |      |              |      |



Fecha: 18/2/09

Temp. inicial: 28°  
Temp. final: 31°

**CALIBRACION DE TRANSDUCTOR**

| PRESION [Pa] | Lecturas [voltios] |       |
|--------------|--------------------|-------|
|              | Inicio             | Fin   |
| 0            | 3.459              | 3.394 |
| 20           | 3.499              | 3.450 |
| 40           | 3.538              | 3.488 |
| 60           | 3.577              | 3.527 |
| -10          | 3.440              | 3.392 |
| -20          | 3.421              | 3.372 |
| -30          | 3.401              | 3.353 |
| -40          | 3.381              | 3.333 |
| -50          | 3.361              | 3.315 |
|              |                    |       |
|              |                    |       |
|              |                    |       |
|              |                    |       |
|              |                    |       |
|              |                    |       |
|              |                    |       |
|              |                    |       |
|              |                    |       |
|              |                    |       |

9:00

11:03

Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 112       | 115       | 113       | 117       | 112       |
| 109       | 109       | 112       | 105       | 116       |
| 108       | 112       | 115       | 110       | 114       |
| 114       | 119       | 112       | 112       | 115       |
| 112       | 117       | 116       | 110       | 111       |
| 113       | 114       | 112       | 111       | 113       |
| 110       | 118       | 115       | 114       | 115       |
| 109       | 113       | 115       | 116       | 112       |
| 108       | 112       | 116       | 115       | 115       |
| 113       | 113       | 117       | 110       | 115       |
| 116       | 118       | 119       | 113       | 117       |
| 113       | 122       | 116       | 112       | 118       |
| 110       | 121       | 117       | 118       | 115       |

- 114 -  
- 113 -  
- 115 -  
- 117 -  
- 114 -  
- 109 -  
- 110 -  
- 104 -  
- 107 -  
- 112 -  
- 115 -  
- 112 -  
- 107 -

PRESIONES MEDIAS INTERIORES SOBRE PARAPETOS DE MODELO I

Fecha 3/25/04

Hora de inicio 10:15

$\alpha = 210^\circ$

Hora de finalización

Temp:

$\alpha = 30^\circ$

Pres. atm.: 760.5 mmHg

|    |       |
|----|-------|
| 1  | 3.641 |
| 2  | 3.653 |
| 3  | 3.633 |
| 4  | 3.638 |
| 5  | 3.636 |
| 6  | 3.632 |
| 7  | 3.606 |
| 8  | 3.663 |
| 9  | 3.651 |
| 10 | 3.617 |
| 11 | 3.673 |
| 12 | 3.662 |
| 13 | 3.632 |
| 14 | 3.656 |
| 15 | 3.650 |
| 16 | 3.605 |
| 17 | 3.592 |
| 18 | 3.584 |
| 19 | 3.533 |
| 20 | 3.534 |

1



2

|       |
|-------|
| 3.409 |
| 3.407 |
| 3.403 |
| 3.397 |
| 3.390 |
| 3.405 |
| 3.403 |
| 3.398 |
| 3.408 |
| 3.408 |
| 3.405 |
| 3.415 |
| 3.413 |
| 3.416 |
| 3.414 |
| 3.415 |
| 3.410 |
| 3.411 |
| 3.405 |
| 3.403 |



3

T = 30°C

Facultad de Ingeniería - UNNE  
Túnel de viento "J. P. Gorecki"

$\alpha = 330^\circ$

|    |       |
|----|-------|
| 1  | 3.422 |
| 2  | 3.446 |
| 3  | 3.438 |
| 4  | 3.444 |
| 5  | 3.503 |
| 6  | 3.511 |
| 7  | 3.500 |
| 8  | 3.515 |
| 9  | 3.526 |
| 10 | 3.516 |
| 11 | 3.504 |
| 12 | 3.506 |
| 13 | 3.486 |
| 14 | 3.489 |
| 15 | 3.481 |
| 16 | 3.455 |
| 17 | 3.456 |
| 18 | 3.417 |
| 19 | 3.504 |
| 20 | 3.500 |

$T = 31^\circ C$

Facultad de Ingeniería - UNNE  
Túnel de viento "J. P. Gorecki"

Fecha: 3/25/24

Temp. inicial:  $26^\circ C$   
Temp. final:  $32^\circ C$

CALIBRACIÓN DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.506              | 3.511 |
| 20              | 3.546              | 3.551 |
| 40              | 3.586              | 3.591 |
| 60              | 3.626              | 3.631 |
| -10             | 3.486              | 3.491 |
| -20             | 3.466              | 3.472 |
| -30             | 3.447              | 3.452 |
| -40             | 3.426              | 3.432 |
| -50             | 3.406              | 3.411 |
|                 | 10.15              | 12.25 |

Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 117       | 110       | 105       | 114       |           |
| 115       | 111       | 109       | 116       |           |
| 111       | 109       | 107       | 113       |           |
| 112       | 109       | 112       | 111       |           |
| 107       | 108       | 114       | 110       |           |
| 108       | 110       | 113       | 108       |           |
| 107       | 111       | 110       | 107       |           |
| 108       | 112       | 108       | 110       |           |
| 111       | 114       | 115       | 112       |           |
| 110       | 115       | 108       | 111       |           |
| 112       | 114       | 112       | 111       |           |
| 114       | 112       | 114       | 114       |           |
|           | 112       | 107       | 113       |           |

PRESIONES MEDIAS INTERIORES SOBRE PARAPETOS DE MODELO 1

Fecha 3/23/09

Hora de inicio 16:20

$\alpha = 0^\circ$

Hora de finalización 18:15

Temp:

$\alpha = 180^\circ$  Pres. atm.:  $757\frac{1}{2}$  mmHg

|    |       |
|----|-------|
| 1  | 3.431 |
| 2  | 3.431 |
| 3  | 3.427 |
| 4  | 3.434 |
| 5  | 3.436 |
| 6  | 3.437 |
| 7  | 3.439 |
| 8  | 3.438 |
| 9  | 3.445 |
| 10 | 3.443 |
| 11 | 3.443 |
| 12 | 3.440 |
| 13 | 3.436 |
| 14 | 3.436 |
| 15 | 3.430 |
| 16 | 3.425 |
| 17 | 3.415 |
| 18 | 3.409 |
| 19 | 3.426 |
| 20 | 3.416 |

1

2

|       |
|-------|
| 3.556 |
| 3.546 |
| 3.573 |
| 3.573 |
| 3.563 |
| 3.543 |
| 3.541 |
| 3.567 |
| 3.547 |
| 3.517 |
| 3.583 |
| 3.570 |
| 3.525 |
| 3.556 |
| 3.535 |
| 3.515 |
| 3.491 |
| 3.475 |
| 3.455 |
| 3.445 |

1

3

$T_2 = 34^\circ C$

$\alpha = 150^\circ$

|    |       |
|----|-------|
| 1  | 3.655 |
| 2  | 3.670 |
| 3  | 3.588 |
| 4  | 3.596 |
| 5  | 3.540 |
| 6  | 3.531 |
| 7  | 3.491 |
| 8  | 3.572 |
| 9  | 3.558 |
| 10 | 3.524 |
| 11 | 3.601 |
| 12 | 3.581 |
| 13 | 3.543 |
| 14 | 3.571 |
| 15 | 3.554 |
| 16 | 3.533 |
| 17 | 3.482 |
| 18 | 3.466 |
| 19 | 3.451 |
| 20 | 3.434 |

4

$T = 32.5^\circ C$

Temp. ambiente  $34^\circ C$   
Pres. atm.:  $757\frac{1}{2}$  mmHg

Fecha: 3/23/09

Temp. inicial: 36°C  
Temp. final: 32°C

CALIBRACION DE TRANSDUCTOR

| PRESION [Pa] | Lecturas [voltios] |       |
|--------------|--------------------|-------|
|              | Inicio             | Fin   |
| 0            | 3.516              | 3.513 |
| 20           | 3.555              | 3.553 |
| 40           | 3.595              | 3.593 |
| 60           | 3.635              | 3.632 |
| -10          | 3.496              | 3.493 |
| -20          | 3.475              | 3.473 |
| -30          | 3.455              | 3.454 |
| -40          | 3.436              | 3.433 |
| -50          | 3.416              | 3.413 |

16:10

Presión dinámica de referencia

725

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 109       | 109       | 108       | 110       |           |
| 109       | 110       | 109       | 108       |           |
| 107       | 108       | 108       | 111       |           |
| 104       | 110       | 109       | 110       |           |
| 98        | 113       | 102       | 107       |           |
| 103       | 112       | 106       | 112       |           |
| 104       | 111       | 105       | 114       |           |
| 106       | 111       | 105       | 108       |           |
| 107       | 106       | 110       | 112       |           |
| 104       | 105       | 109       | 111       |           |
| 103       | 108       | 108       | 111       |           |
| 105       | 104       | 103       | 108       |           |
| 105       | 102       | 102       | 113       |           |

PRESIONES MEDIAS EXTERIORES SOBRE PARAPETOS DE MODELO 1

Fecha 19/03/09

Hora de inicio 18:10

Hora de finalización 19:13

Temp: 31°

Pres. atm.: 761,5

|    | $\alpha = 30^\circ$ |     | $\alpha = 330^\circ$ |     |
|----|---------------------|-----|----------------------|-----|
| 0  |                     |     |                      |     |
| 1  | 3.695               | ← 1 | 3.690                | ←   |
| 2  | 3.713               | ←   | 3.702                | ←   |
| 3  | 3.705               | ←   | 3.693                | ←   |
| 4  | 3.724               | ←   | 3.703                | ←   |
| 5  | 3.688               | ←   | 3.639                | ←   |
| 6  | 3.731               | ←   | 3.670                | ←   |
| 7  | 3.725               | ←   | 3.666                | ←   |
| 8  | 3.696               | ←   | 3.604                | ←   |
| 9  | 3.747               | ←   | 3.624                | ←   |
| 10 | 3.753               | ←   | 3.627                | ←   |
| 11 | 3.563               | ←   | 3.368                | ←   |
| 12 | 3.578               | ←   | 3.370                | ←   |
| 13 | 3.577               | ←   | 3.368                | ←   |
| 14 | 3.569               | ←   | 3.360                | ←   |
| 15 | 3.592               | ←   | 3.360                | ←   |
| 16 | 3.581               | ←   | 3.356                | ←   |
| 17 | 3.578               | ←   | 3.392                | ←   |
| 18 | 3.582               | ←   | 3.393                | ←   |
| 19 | 3.586               | ←   | 3.417                | ←   |
| 20 | 3.587               | ← 2 | 3.416                | ← 3 |

761,5 mm Hg

Fecha: 19/03/09

Temp. inicial: 31°  
Temp. final: 28°

#### CALIBRACION DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.513              | 3.512 |
| 20              | 3.554              | 3.552 |
| 40              | 3.593              | 3.592 |
| 60              | 3.633              | 3.632 |
| -10             | 3.494              | 3.492 |
| -20             | 3.474              | 3.472 |
| -30             | 3.454              | 3.452 |
| -40             | 3.434              | 3.433 |
| -50             | 3.415              | 3.413 |
|                 | 18:00              | 19:20 |

#### Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 108       | 116       | 111       |           |           |
| 108       | 111       | 111       |           |           |
| 111       | 105       | 116       |           |           |
| 115       | 109       | 110       |           |           |
| 116       | 110       | 114       |           |           |
| 110       | 108       | 116       |           |           |
| 108       | 105       | 113       |           |           |
| 105       | 107       | 114       |           |           |
| 102       | 110       | 115       |           |           |
| 107       | 111       | 113       |           |           |
| 105       | 109       | 113       |           |           |
| 104       | 109       | 108       |           |           |
| 108       | 113       | 112       |           |           |

#### PRESIONES MEDIAS EXTERIORES SOBRE PARAPETOS DE MODELO 1

Fecha 19/03/09

Hora de inicio : 9:10

Hora de finalización

Temp: 23°

Pres. atm.: 763,5

|    | $\alpha = 0^\circ$ |    | $\alpha = 180^\circ$ |
|----|--------------------|----|----------------------|
| 0  |                    |    |                      |
| 1  | 3.743              | 1  | 3.449                |
| 2  | 3.760              | 2  | 3.450                |
| 3  | 3.748              | 3  | 3.455                |
| 4  | 3.751              | 4  | 3.448                |
| 5  | 3.699              | 5  | 3.445                |
| 6  | 3.745              | 6  | 3.444                |
| 7  | 3.730              | 7  | 3.445                |
| 8  | 3.678              | 8  | 3.440                |
| 9  | 3.722              | 9  | 3.438                |
| 10 | 3.723              | 10 | 3.431                |
| 11 | 3.303              | 11 | 3.485                |
| 12 | 3.295              | 12 | 3.487                |
| 13 | 3.307              | 13 | 3.481                |
| 14 | 3.365              | 14 | 3.494                |
| 15 | 3.350              | 15 | 3.499                |
| 16 | 3.351              | 16 | 3.493                |
| 17 | 3.466              | 17 | 3.500                |
| 18 | 3.455              | 18 | 3.505                |
| 19 | 3.486              | 19 | 3.499                |
| 20 | 3.482              | 20 | 3.499                |

|    | $\alpha = 150^\circ$ | $\alpha = 210^\circ$ |
|----|----------------------|----------------------|
| 0  |                      |                      |
| 1  | 3.407                | 3.435                |
| 2  | 3.403                | 3.435                |
| 3  | 3.401                | 3.444                |
| 4  | 3.399                | 3.444                |
| 5  | 3.383                | 3.461                |
| 6  | 3.377                | 3.462                |
| 7  | 3.369                | 3.456                |
| 8  | 3.381                | 3.457                |
| 9  | 3.371                | 3.457                |
| 10 | 3.372                | 3.452                |
| 11 | 3.515                | 3.456                |
| 12 | 3.526                | 3.459                |
| 13 | 3.526                | 3.456                |
| 14 | 3.543                | 3.454                |
| 15 | 3.562                | 3.456                |
| 16 | 3.649                | 3.452                |
| 17 | 3.576                | 3.445                |
| 18 | 3.591                | 3.441                |
| 19 | 3.580                | 3.437                |
| 20 | 3.588                | 3.437                |

764 mm Hg

#### ANÁLISIS DE ERROR EN LECTURAS DE PRESIÓN

TOMA N°: 16  $\alpha = 210^\circ$

Inicio Fin

|    | Voltaje medio | Temp | Presión atm. | qref |
|----|---------------|------|--------------|------|
| 1  | 3.451         |      |              |      |
| 2  | 3.451         |      |              |      |
| 3  | 3.449         |      |              |      |
| 4  | 3.451         |      |              |      |
| 5  | 3.447         |      |              |      |
| 6  | 3.450         |      |              |      |
| 7  | 3.453         |      |              |      |
| 8  | 3.448         |      |              |      |
| 9  | 3.449         |      |              |      |
| 10 | 3.446         |      |              |      |

TOMA N°: 2  $\alpha = 0^\circ$

Inicio Fin

|    | Voltaje medio | Temp | Presión atm. | qref |
|----|---------------|------|--------------|------|
| 1  | 3.759         |      |              |      |
| 2  | 3.749         |      |              |      |
| 3  | 3.764         |      |              |      |
| 4  | 3.790         |      |              |      |
| 5  | 3.759         |      |              |      |
| 6  | 3.761         |      |              |      |
| 7  | 3.780         |      |              |      |
| 8  | 3.772         |      |              |      |
| 9  | 3.770         |      |              |      |
| 10 | 3.758         |      |              |      |

Fecha: 19/03/09

Temp. inicial: 23°  
Temp. final: 22°

### CALIBRACION DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.507              | 3.507 |
| 20              | 3.548              | 3.547 |
| 40              | 3.588              | 3.587 |
| 60              | 3.627              | 3.627 |
| -10             | 3.488              | 3.487 |
| -20             | 3.468              | 3.467 |
| -30             | 3.448              | 3.447 |
| -40             | 3.428              | 3.427 |
| -50             | 3.408              | 3.407 |
| 8:50            |                    | 11:15 |

### Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 106       | 112       | 110       | 109       | 112       |
| 108       | 108       | 107       | 111       | 107       |
| 111       | 111       | 109       | 112       | 111       |
| 112       | 109       | 113       | 116       | 110       |
| 108       | 112       | 115       | 113       | 107       |
| 111       | 116       | 115       | 114       | 105       |
| 114       | 115       | 112       | 116       | 111       |
| 117       | 119       | 117       | 112       | 108       |
| 115       | 114       | 112       | 110       | 109       |
| 108       | 113       | 108       | 109       | 109       |
| 110       | 116       | 109       | 107       | 108       |
| 112       | 115       | 116       | 106       | 105       |
| 116       | 113       | 111       | 109       | 105       |

### PRESIONES MEDIAS INTERIORES MODELO I SIN PARAPETOS

Fecha: 05/03/09

Hora de inicio: 10:17

Hora de finalización: 11:25

Temp: 26°

Pres. atm.: 762

|    | $\alpha = 30^\circ$ |   | $\alpha = 30^\circ$ |   |
|----|---------------------|---|---------------------|---|
| 0  |                     |   |                     |   |
| 1  | 3.503               | 1 | 3.427               | 2 |
| 2  | 3.419               |   | 3.417               |   |
| 3  | 3.419               |   | 3.414               |   |
| 4  | 3.426               |   | 3.389               |   |
| 5  | 3.451               |   | 3.385               |   |
| 6  | 3.514               |   | 3.476               |   |
| 7  | 3.515               |   | 3.477               |   |
| 8  | 3.500               |   | 3.483               |   |
| 9  | 3.475               |   | 3.493               |   |
| 10 | 3.521               |   | 3.506               |   |
| 11 | 3.518               |   | 3.510               |   |
| 12 | 3.480               |   | 3.497               |   |
| 13 | 3.518               |   | 3.505               |   |
| 14 | 3.498               |   | 3.503               |   |
| 15 | 3.514               |   | 3.514               |   |
| 16 | 3.517               |   | 3.506               |   |

|    | $\alpha = 210^\circ$ |   |
|----|----------------------|---|
| 0  |                      |   |
| 1  | 3.503                | 3 |
| 2  | 3.504                |   |
| 3  | 3.502                |   |
| 4  | 3.502                |   |
| 5  | 3.500                |   |
| 6  | 3.504                |   |
| 7  | 3.507                |   |
| 8  | 3.507                |   |
| 9  | 3.506                |   |
| 10 | 3.507                |   |
| 11 | 3.506                |   |
| 12 | 3.503                |   |
| 13 | 3.509                |   |
| 14 | 3.511                |   |
| 15 | 3.511                |   |
| 16 | 3.519                | 4 |

Fecha: 05/03/09

Temp. inicial: 26°  
Temp. final: 27°

### CALIBRACION DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.506              | 3.507 |
| 20              | 3.546              | 3.547 |
| 40              | 3.586              | 3.587 |
| 60              | 3.626              | 3.626 |
| -10             | 3.486              | 3.487 |
| -20             | 3.466              | 3.467 |
| -30             | 3.447              | 3.447 |
| -40             | 3.426              | 3.427 |
| -50             | 3.406              | 3.408 |
|                 | 10808              | 11226 |

### Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 106       | 108       | 106       | 111       |           |
| 108       | 109       | 105       | 112       |           |
| 113       | 113       | 107       | 110       |           |
| 112       | 112       | 110       | 112       |           |
| 110       | 113       | 109       | 108       |           |
| 108       | 114       | 110       | 112       |           |
| 112       | 111       | 111       | 116       |           |
| 116       | 106       | 113       | 112       |           |
| 112       | 107       | 110       | 115       |           |
| 113       | 111       | 109       | 111       |           |
| 109       | 111       | 108       | 109       |           |
| 106       | 114       | 111       | 108       |           |
| 107       | 109       | 110       | 106       |           |

### PRESIONES MEDIAS INTERIORES MODELO 1 SIN PARAPETOS

Fecha 04/03/09

Hora de inicio 10:41

Hora de finalización 11:50

Temp: 32°

Pres. atm.: 757

| $\alpha = 0^\circ$ |       | $\alpha = 0^\circ$ |   |
|--------------------|-------|--------------------|---|
| 0                  |       | 3.438              | 1 |
| 1                  | 4.508 | 3.409              |   |
| 2                  | 4.506 | 3.428              |   |
| 3                  | 4.5   | 3.408              |   |
| 4                  |       | 3.407              |   |
| 5                  |       | 3.514              |   |
| 6                  |       | 3.506              |   |
| 7                  |       | 3.496              |   |
| 8                  |       | 3.509              |   |
| 9                  |       | 3.517              |   |
| 10                 |       | 3.519              |   |
| 11                 |       | 3.510              |   |
| 12                 |       | 3.518              |   |
| 13                 |       | 3.520              |   |
| 14                 |       | 3.518              |   |
| 15                 |       | 3.515              |   |
| 16                 |       | 3.515              |   |

| $\alpha = 180^\circ$ |       | $\alpha = 150^\circ$ |  |
|----------------------|-------|----------------------|--|
| 0                    |       | 3.456                |  |
| 1                    | 3.497 | 3.480                |  |
| 2                    | 3.498 | 3.495                |  |
| 3                    | 3.500 | 3.498                |  |
| 4                    | 3.501 | 3.498                |  |
| 5                    | 3.493 | 3.488                |  |
| 6                    | 3.506 | 3.455                |  |
| 7                    | 3.506 | 3.482                |  |
| 8                    | 3.511 | 3.511                |  |
| 9                    | 3.506 | 3.465                |  |
| 10                   | 3.513 | 3.507                |  |
| 11                   | 3.519 | 3.516                |  |
| 12                   | 3.507 | 3.446                |  |
| 13                   | 3.515 | 3.496                |  |
| 14                   | 3.506 | 3.452                |  |
| 15                   | 3.517 | 3.519                |  |
| 16                   | 3.519 | 3.513                |  |

Fecha: 04/03/09

Temp. inicial: 32°  
Temp. final: 35°

CALIBRACION DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.510              | 3.516 |
| 20              | 3.551              | 3.556 |
| 40              | 3.590              | 3.596 |
| 60              | 3.630              | 3.636 |
| -10             | 3.490              | 3.497 |
| -20             | 3.471              | 3.477 |
| -30             | 3.451              | 3.457 |
| -40             | 3.431              | 3.437 |
| -50             | 3.411              | 3.416 |
| 10:30           |                    | 12:00 |

Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 102       | 105       | 106       | 107       |           |
| 100       | 103       | 104       | 106       |           |
| 108       | 101       | 108       | 104       |           |
| 105       | 102       | 108       | 107       |           |
| 100       | 109       | 107       | 108       |           |
| 104       | 106       | 104       | 100       |           |
| 103       | 102       | 105       | 100       |           |
| 106       | 99        | 100       | 105       |           |
| 102       | 104       | 103       | 102       |           |
| 108       | 105       | 107       | 108       |           |
| 101       | 107       | 104       | 104       |           |
| 107       | 106       | 102       | 105       |           |
| 103       | 101       | 106       | 103       |           |

PRESIONES MEDIAS INTERIORES EN TECHO DE MODELO I CON PARAPETOS

Fecha: 02/03/09

Hora de inicio: 17:20

Hora de finalización:

Temp: 34°

Pres. atm.: 259

|    |                |   |  |               |   |
|----|----------------|---|--|---------------|---|
|    | $\alpha = 330$ |   |  | $\alpha = 30$ |   |
| 0  |                |   |  |               |   |
| 1  | 3.361          |   |  | 3.460         |   |
| 2  | 3.357          |   |  | 3.249         |   |
| 3  | 3.341          |   |  | 3.185         |   |
| 4  | 3.332          |   |  | 3.239         |   |
| 5  | 3.297          |   |  | 3.301         |   |
| 6  | 3.381          |   |  | 3.481         |   |
| 7  | 3.368          |   |  | 3.478         |   |
| 8  | 3.344          |   |  | 3.342         |   |
| 9  | 3.396          |   |  | 3.411         |   |
| 10 | 3.406          |   |  | 3.501         |   |
| 11 | 3.399          |   |  | 3.483         |   |
| 12 | 3.439          |   |  | 3.483         |   |
| 13 | 3.448          |   |  | 3.377         |   |
| 14 | 3.472          |   |  | 3.496         |   |
| 15 | 3.470          |   |  | 3.400         |   |
| 16 | 3.491          | 2 |  | 3.492         | 3 |

|    |                |   |  |
|----|----------------|---|--|
|    | $\alpha = 210$ |   |  |
| 0  |                |   |  |
| 1  | 3.483          |   |  |
| 2  | 3.483          |   |  |
| 3  | 3.480          |   |  |
| 4  | 3.473          |   |  |
| 5  | 3.465          |   |  |
| 6  | 3.485          |   |  |
| 7  | 3.489          |   |  |
| 8  | 3.484          |   |  |
| 9  | 3.484          |   |  |
| 10 | 3.495          |   |  |
| 11 | 3.487          |   |  |
| 12 | 3.482          |   |  |
| 13 | 3.494          |   |  |
| 14 | 3.482          |   |  |
| 15 | 3.495          |   |  |
| 16 | 3.505          | 4 |  |

Fecha: 02/03/09

Temp. inicial: 34°  
Temp. final: 39°

# CALIBRACION DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.513              | 3.514 |
| 20              | 3.553              | 3.553 |
| 40              | 3.592              | 3.594 |
| 60              | 3.632              | 3.633 |
| -10             | 3.493              | 3.494 |
| -20             | 3.472              | 3.474 |
| -30             | 3.453              | 3.454 |
| -40             | 3.433              | 3.435 |
| -50             | 3.413              | 3.415 |
|                 | 17:15              | 18:42 |

## Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 104       | 106       | 108       | 109       |           |
| 107       | 108       | 109       | 115       |           |
| 103       | 109       | 107       | 113       |           |
| 107       | 105       | 110       | 116       |           |
| 109       | 109       | 111       | 114       |           |
| 106       | 111       | 108       | 114       |           |
| 101       | 109       | 112       | 113       |           |
| 108       | 107       | 113       | 107       |           |
| 108       | 111       | 107       | 105       |           |
| 106       | 109       | 112       | 103       |           |
| 108       | 110       | 112       | 106       |           |
| 105       | 106       | 113       | 105       |           |
| 106       | 108       | 113       | 107       |           |

## PRESIONES MEDIAS INTERIORES EN TECHO DE MODELO 1 CON PARAPETOS

Fecha 2/03/09

Hora de inicio 10:25

Hora de finalización 11:33

Temp: 29°

Pres. atm.: 760,6

| $\alpha = 0^\circ$ |       | $\alpha = 180^\circ$ |       |
|--------------------|-------|----------------------|-------|
| 1                  | 3.278 | 1                    | 3.475 |
| 2                  | 3.280 | 2                    | 3.473 |
| 3                  | 3.290 | 3                    | 3.474 |
| 4                  | 3.298 | 4                    | 3.474 |
| 5                  | 3.311 | 5                    | 3.475 |
| 6                  | 3.350 | 6                    | 3.475 |
| 7                  | 3.324 | 7                    | 3.482 |
| 8                  | 3.312 | 8                    | 3.487 |
| 9                  | 3.440 | 9                    | 3.487 |
| 10                 | 3.377 | 10                   | 3.487 |
| 11                 | 3.370 | 11                   | 3.493 |
| 12                 | 3.476 | 12                   | 3.491 |
| 13                 | 3.466 | 13                   | 3.497 |
| 14                 | 3.496 | 14                   | 3.486 |
| 15                 | 3.472 | 15                   | 3.501 |
| 16                 | 3.464 | 16                   | 3.497 |

| $\alpha = 150^\circ$ |       |  |
|----------------------|-------|--|
| 0                    |       |  |
| 1                    | 3.393 |  |
| 2                    | 3.397 |  |
| 3                    | 3.438 |  |
| 4                    | 3.449 |  |
| 5                    | 3.455 |  |
| 6                    | 3.386 |  |
| 7                    | 3.389 |  |
| 8                    | 3.475 |  |
| 9                    | 3.391 |  |
| 10                   | 3.472 |  |
| 11                   | 3.490 |  |
| 12                   | 3.385 |  |
| 13                   | 3.426 |  |
| 14                   | 3.377 |  |
| 15                   | 3.493 |  |
| 16                   | 3.486 |  |

760,5

Fecha: 02/03/09

Temp. inicial: 29°  
Temp. final: 29°

#### CALIBRACION DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.508              | 3.508 |
| 20              | 3.548              | 3.549 |
| 40              | 3.588              | 3.588 |
| 60              | 3.628              | 3.629 |
| -10             | 3.488              | 3.488 |
| -20             | 3.469              | 3.469 |
| -30             | 3.449              | 3.449 |
| -40             | 3.429              | 3.429 |
| -50             | 3.409              | 3.409 |
|                 | 10:15              | 11:38 |

#### Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 107       | 106       | 107       | 109       |           |
| 105       | 110       | 106       | 106       |           |
| 103       | 105       | 103       | 105       |           |
| 106       | 105       | 108       | 104       |           |
| 105       | 107       | 106       | 102       |           |
| 104       | 107       | 110       | 101       |           |
| 108       | 105       | 108       | 103       |           |
| 105       | 107       | 103       | 111       |           |
| 101       | 108       | 101       | 109       |           |
| 109       | 106       | 106       | 106       |           |
| 108       | 105       | 105       | 104       |           |
| 104       | 105       | 105       | 104       |           |
| 107       | 104       | 108       | 105       |           |

#### PRESIONES MEDIAS EXTERIORES EN TECHO DE MODELO 1 CON PARAPETOS

Fecha 26/02/09

Hora de inicio

Hora de finalización 19:15

Temp: 32

Pres. atm.: 758,5

|    | $\alpha = 150^\circ$ |   | $\alpha = 210^\circ$ |           |
|----|----------------------|---|----------------------|-----------|
| 0  |                      |   |                      |           |
| 1  | 3.575                | 1 | 3.650                | inestable |
| 2  | 3.537                |   | 3.630                |           |
| 3  | 3.538                |   | 3.624                |           |
| 4  | 3.588                |   | 3.620                |           |
| 5  | 3.637                |   | 3.638                |           |
| 6  | 3.646                |   | 3.636                |           |
| 7  | 3.511                |   | 3.598                |           |
| 8  | 3.565                |   | 3.588                |           |
| 9  | 3.498                |   | 3.597                |           |
| 10 | 3.500                |   | 3.562                |           |
| 11 | 3.574                |   | 3.564                |           |
| 12 | 3.476                |   | 3.559                |           |
| 13 | 3.461                |   | 3.556                |           |
| 14 | 3.445                |   | 3.521                |           |
| 15 | 3.480                |   | 3.523                |           |
| 16 | 3.548                | 2 | 3.510                | 3         |

$p_{atm} = 758,5$

Fecha: 26/02/09

Temp. inicial: 32°  
Temp. final: 34°

CALIBRACION DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.540              | 3.511 |
| 20              | 3.550              | 3.551 |
| 40              | 3.590              | 3.590 |
| 60              | 3.630              | 3.631 |
| -10             | 3.490              | 3.491 |
| -20             | 3.471              | 3.471 |
| -30             | 3.451              | 3.452 |
| -40             | 3.431              | 3.432 |
| -50             | 3.411              | 3.412 |
|                 | 18.28              | 19.17 |

Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 110       | 109       | 104       |           |           |
| 107       | 108       | 109       |           |           |
| 112       | 106       | 111       |           |           |
| 107       | 107       | 112       |           |           |
| 109       | 111       | 111       |           |           |
| 105       | 111       | 110       |           |           |
| 108       | 108       | 114       |           |           |
| 109       | 113       | 110       |           |           |
| 107       | 114       | 110       |           |           |
| 111       | 100       | 115       |           |           |
| 113       | 115       | 116       |           |           |
| 117       | 113       | 115       |           |           |
| 114       | 112       | 112       |           |           |

PRESIONES MEDIAS EXTERIORES EN TECHO DE MODELO I CON PARAPETOS

Fecha: 26/02/09

Hora de inicio: 9:40

Hora de finalización: 11:11

Temp: 25°

Pres. atm.: 760,5

| $\alpha = 0$ |       | $\alpha = 180$ |       |
|--------------|-------|----------------|-------|
| 0            |       | 0              |       |
| 1            | 3.428 | 1              | 3.575 |
| 2            | 3.419 | 2              | 3.563 |
| 3            | 3.414 | 3              | 3.590 |
| 4            | 3.411 | 4              | 3.586 |
| 5            | 3.409 | 5              | 3.562 |
| 6            | 3.417 | 6              | 3.538 |
| 7            | 3.409 | 7              | 3.521 |
| 8            | 3.408 | 8              | 3.531 |
| 9            | 3.408 | 9              | 3.508 |
| 10           | 3.402 | 10             | 3.512 |
| 11           | 3.393 | 11             | 3.490 |
| 12           | 3.403 | 12             | 3.479 |
| 13           | 3.399 | 13             | 3.480 |
| 14           | 3.409 | 14             | 3.440 |
| 15           | 3.401 | 15             | 3.465 |
| 16           | 3.394 | 16             | 3.443 |

| $\alpha = 30^\circ$ |       | $\alpha = 330$ |       |
|---------------------|-------|----------------|-------|
| 0                   |       | 0              |       |
| 1                   | 3.398 | 1              | 3.495 |
| 2                   | 3.389 | 2              | 3.478 |
| 3                   | 3.387 | 3              | 3.442 |
| 4                   | 3.389 | 4              | 3.431 |
| 5                   | 3.400 | 5              | 3.405 |
| 6                   | 3.390 | 6              | 3.422 |
| 7                   | 3.382 | 7              | 3.455 |
| 8                   | 3.389 | 8              | 3.450 |
| 9                   | 3.390 | 9              | 3.419 |
| 10                  | 3.384 | 10             | 3.443 |
| 11                  | 3.384 | 11             | 3.417 |
| 12                  | 3.389 | 12             | 3.443 |
| 13                  | 3.390 | 13             | 3.429 |
| 14                  | 3.409 | 14             | 3.490 |
| 15                  | 3.416 | 15             | 3.431 |
| 16                  | 3.429 | 16             | 3.424 |

Fecha: 26/02/09

Temp. inicial: 25°  
Temp. final: 28°

### CALIBRACION DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.505              | 3.507 |
| 20              | 3.545              | 3.546 |
| 40              | 3.585              | 3.586 |
| 60              | 3.625              | 3.627 |
| -10             | 3.486              | 3.487 |
| -20             | 3.466              | 3.467 |
| -30             | 3.446              | 3.447 |
| -40             | 3.427              | 3.427 |
| -50             | 3.407              | 3.407 |

q: 29

### Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 115       | 112       | 117       | 118       | 114       |
| 113       | 111       | 116       | 118       | 110       |
| 114       | 112       | 119       | 115       | 111       |
| 113       | 117       | 118       | 118       | 115       |
| 124       | 116       | 120       | 114       | 116       |
| 115       | 118       | 114       | 121       | 114       |
| 118       | 118       | 118       | 120       | 109       |
| 116       | 116       | 108       | 116       | 112       |
| 121       | 110       | 113       | 113       | 119       |
| 120       | 119       | 120       | 114       | 116       |
| 118       | 114       | 121       | 111       | 112       |
| 120       | 119       | 119       | 110       | 114       |
| 117       | 116       | 115       | 112       | 112       |

### PRESIONES MEDIAS EXTERIORES EN MODELO 1 SIN PARAPETOS

Fecha: 25/02/09

Hora de inicio: 17:12

Hora de finalización: 18:56

Temp: 31

Pres. atm.: 760,5

| $\alpha = 30^\circ$ |       | $\alpha = 30^\circ$ |  |
|---------------------|-------|---------------------|--|
| 0                   |       |                     |  |
| 1                   | 3.509 | 3.445               |  |
| 2                   | 3.476 | 3.441               |  |
| 3                   | 3.477 | 3.444               |  |
| 4                   | 3.463 | 3.430               |  |
| 5                   | 3.458 | 3.424               |  |
| 6                   | 3.505 | 3.484               |  |
| 7                   | 3.512 | 3.486               |  |
| 8                   | 3.507 | 3.489               |  |
| 9                   | 3.494 | 3.493               |  |
| 10                  | 3.516 | 3.501               |  |
| 11                  | 3.514 | 3.504               |  |
| 12                  | 3.487 | 3.497               |  |
| 13                  | 3.516 | 3.505               |  |
| 14                  | 3.492 | 3.497               |  |
| 15                  | 3.517 | 3.507               |  |
| 16                  | 3.515 | 3.506               |  |

| $\alpha = 150^\circ$ |       | $\alpha = 210^\circ$ |  |
|----------------------|-------|----------------------|--|
| 0                    |       |                      |  |
| 1                    | 3.465 | 3.505                |  |
| 2                    | 3.479 | 3.502                |  |
| 3                    | 3.490 | 3.506                |  |
| 4                    | 3.491 | 3.507                |  |
| 5                    | 3.496 | 3.506                |  |
| 6                    | 3.470 | 3.508                |  |
| 7                    | 3.489 | 3.507                |  |
| 8                    | 3.500 | 3.510                |  |
| 9                    | 3.475 | 3.508                |  |
| 10                   | 3.502 | 3.511                |  |
| 11                   | 3.505 | 3.512                |  |
| 12                   | 3.474 | 3.508                |  |
| 13                   | 3.495 | 3.508                |  |
| 14                   | 3.465 | 3.506                |  |
| 15                   | 3.508 | 3.512                |  |
| 16                   | 3.507 | 3.511                |  |

Patm = 760

Fecha: 25/02/09

Temp. inicial: 37°  
Temp. final: 31°

# CALIBRACION DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.509              | 3.510 |
| 20              | 3.548              | 3.550 |
| 40              | 3.588              | 3.590 |
| 60              | 3.628              | 3.630 |
| -10             | 3.488              | 3.490 |
| -20             | 3.468              | 3.470 |
| -30             | 3.449              | 3.451 |
| -40             | 3.429              | 3.431 |
| -50             | 3.409              | 3.410 |
|                 | 16:44              | 18:57 |

## Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 108       | 107       | 110       | 111       | 111       |
| 107       | 105       | 108       | 112       | 108       |
| 112       | 107       | 112       | 114       | 115       |
| 108       | 109       | 112       | 109       | 116       |
| 106       | 105       | 110       | 110       | 114       |
| 107       | 111       | 107       | 117       | 114       |
| 109       | 107       | 106       | 112       | 117       |
| 107       | 112       | 110       | 105       | 115       |
| 110       | 107       | 108       | 106       | 114       |
| 111       | 111       | 107       | 109       | 112       |
| 110       | 104       | 103       | 111       | 119       |
| 111       | 106       | 103       | 112       | 117       |
| 107       | 103       | 105       | 111       | 112       |

## PRESIONES MEDIAS EXTERIORES EN MODELO 1 SIN PARAPETOS

Fecha 25/02/09

Hora de inicio 10:30

~~25~~

Hora de finalización 11:35

Temp: 25°  
27

Pres. atm.: 763 mm Hg  
763

|    | $\alpha=0$ |   | $\alpha=180$ |
|----|------------|---|--------------|
| 0  | 3.452      | 1 | 3.487        |
| 1  | 3.432      |   | 3.487        |
| 2  | 3.444      |   | 3.491        |
| 3  | 3.428      |   | 3.492        |
| 4  | 3.411      |   | 3.496        |
| 5  | 3.492      |   | 3.496        |
| 6  | 3.487      |   | 3.496        |
| 7  | 3.486      |   | 3.500        |
| 8  | 3.495      |   | 3.497        |
| 9  | 3.502      |   | 3.504        |
| 10 | 3.501      |   | 3.503        |
| 11 | 3.498      |   | 3.500        |
| 12 | 3.503      |   | 3.505        |
| 13 | 3.493      |   | 3.499        |
| 14 | 3.505      |   | 3.507        |
| 15 | 3.507      | 2 | 3.506        |
| 16 | 3.507      |   |              |

Fecha: 25/02/09

Temp. inicial: 25°  
Temp. final: 27°

CALIBRACION DE TRANSDUCTOR

| PRESION<br>[Pa] | Lecturas [voltios] |       |
|-----------------|--------------------|-------|
|                 | Inicio             | Fin   |
| 0               | 3.506              | 3.506 |
| 20              | 3.546              | 3.546 |
| 40              | 3.586              | 3.586 |
| 60              | 3.626              | 3.626 |
| -10             | 3.486              | 3.486 |
| -20             | 3.466              | 3.466 |
| -30             | 3.446              | 3.446 |
| -40             | 3.426              | 3.426 |
| -50             | 3.406              | 3.406 |
|                 |                    |       |
|                 |                    |       |
|                 |                    |       |
|                 |                    |       |
|                 |                    |       |
|                 |                    |       |
|                 |                    |       |
|                 |                    |       |
|                 |                    |       |
|                 |                    |       |
|                 |                    |       |

10:28

11:35

Presión dinámica de referencia

| Lectura 1 | Lectura 2 | Lectura 3 | Lectura 4 | Lectura 5 |
|-----------|-----------|-----------|-----------|-----------|
| 116       | 116       | 112       |           |           |
| 114       | 116       | 111       |           |           |
| 111       | 114       | 115       |           |           |
| 117       | 116       | 113       |           |           |
| 114       | 117       | 115       |           |           |
| 110       | 115       | 114       |           |           |
| 112       | 118       | 114       |           |           |
| 111       | 115       | 115       |           |           |
| 114       | 111       | 116       |           |           |
| 113       | 116       | 114       |           |           |
| 119       | 117       | 117       |           |           |
| 117       | 118       | 119       |           |           |
| 116       | 114       | 115       |           |           |

## APPENDIX C

### Open canopy without parapets: spreadsheet calculations.

| Presiones exteriores, interiores y totales en Modelo1 |               |               |                |       |                      |               |               |                |       |
|---|---------------|---------------|----------------|-------|----------------------|---------------|---------------|----------------|-------|
| Sin parapetos   |               |               |                |       |                      |               |               |                |       |
| $\alpha = 0$  |               |               |                |       |                      |               |               |                |       |
| Presiones exteriores                                  |               |               |                |       | Presiones interiores |               |               |                |       |
| Toma  | Potencial [V] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_e$ | Toma                 | Potencial [V] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_i$ |
| 1   | 3.452         | -27           | 114.2          | -0.2  | 1                    | 3.438         | -38           | 103.8          | -0.4  |
| 2   | 3.432         | -37           | 114.3          | -0.3  | 2                    | 3.409         | -52           | 103.8          | -0.5  |
| 3   | 3.444         | -31           | 114.3          | -0.3  | 3                    | 3.428         | -43           | 103.8          | -0.4  |
| 4   | 3.428         | -39           | 114.4          | -0.3  | 4                    | 3.408         | -53           | 103.8          | -0.5  |
| 5   | 3.411         | -47           | 114.5          | -0.4  | 5                    | 3.407         | -53           | 103.8          | -0.5  |
| 6   | 3.492         | -7            | 114.6          | -0.1  | 6                    | 3.514         | 0             | 103.8          | 0.0   |
| 7   | 3.487         | -9            | 114.7          | -0.1  | 7                    | 3.506         | -4            | 103.8          | 0.0   |
| 8   | 3.486         | -10           | 114.8          | -0.1  | 8                    | 3.496         | -9            | 103.8          | -0.1  |
| 9   | 3.495         | -5            | 114.9          | 0.0   | 9                    | 3.509         | -2            | 103.8          | 0.0   |
| 10  | 3.502         | -2            | 115.0          | 0.0   | 10                   | 3.517         | 2             | 103.8          | 0.0   |
| 11  | 3.501         | -2            | 115.1          | 0.0   | 11                   | 3.519         | 3             | 103.8          | 0.0   |
| 12  | 3.498         | -4            | 115.2          | 0.0   | 12                   | 3.51          | -2            | 103.8          | 0.0   |
| 13  | 3.503         | -1            | 115.3          | 0.0   | 13                   | 3.518         | 2             | 103.8          | 0.0   |
| 14  | 3.493         | -6            | 115.4          | -0.1  | 14                   | 3.52          | 3             | 103.8          | 0.0   |
| 15  | 3.505         | 0             | 115.5          | 0.0   | 15                   | 3.518         | 2             | 103.8          | 0.0   |
| 16  | 3.505         | 0             | 115.6          | 0.0   | 16                   | 3.515         | 1             | 103.8          | 0.0   |



|                      |               |               |                |       |  |                      |               |               |                |       |     |
|----------------------|---------------|---------------|----------------|-------|--|----------------------|---------------|---------------|----------------|-------|-----|
| Sin parapetos        |               |               |                |       |  |                      |               |               |                |       |     |
| $\alpha = 330$       |               |               |                |       |  |                      |               |               |                |       |     |
|                      |               |               |                |       |  |                      |               |               |                |       |     |
| Presiones exteriores |               |               |                |       |  | Presiones interiores |               |               |                |       |     |
| Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_e$ |  | Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_i$ | c   |
| 1                    | 3.445         | -32           | 107.2          | -0.3  |  | 1                    | 3.427         | -40           | 110.6          | -0.4  | 0.1 |
| 2                    | 3.441         | -34           | 107.3          | -0.3  |  | 2                    | 3.411         | -48           | 110.5          | -0.4  | 0.1 |
| 3                    | 3.444         | -33           | 107.3          | -0.3  |  | 3                    | 3.414         | -46           | 110.4          | -0.4  | 0.1 |
| 4                    | 3.43          | -40           | 107.3          | -0.4  |  | 4                    | 3.389         | -59           | 110.3          | -0.5  | 0.2 |
| 5                    | 3.424         | -43           | 107.4          | -0.4  |  | 5                    | 3.385         | -61           | 110.2          | -0.6  | 0.2 |
| 6                    | 3.484         | -13           | 107.4          | -0.1  |  | 6                    | 3.476         | -15           | 110.1          | -0.1  | 0.0 |
| 7                    | 3.486         | -12           | 107.4          | -0.1  |  | 7                    | 3.477         | -15           | 110.0          | -0.1  | 0.0 |
| 8                    | 3.489         | -10           | 107.5          | -0.1  |  | 8                    | 3.483         | -12           | 109.9          | -0.1  | 0.0 |
| 9                    | 3.493         | -8            | 107.5          | -0.1  |  | 9                    | 3.493         | -7            | 109.8          | -0.1  | 0.0 |
| 10                   | 3.501         | -4            | 107.6          | 0.0   |  | 10                   | 3.506         | 0             | 109.7          | 0.0   | 0.0 |
| 11                   | 3.504         | -3            | 107.6          | 0.0   |  | 11                   | 3.51          | 2             | 109.6          | 0.0   | 0.0 |
| 12                   | 3.497         | -6            | 107.6          | -0.1  |  | 12                   | 3.497         | -5            | 109.5          | 0.0   | 0.0 |
| 13                   | 3.505         | -2            | 107.7          | 0.0   |  | 13                   | 3.505         | -1            | 109.4          | 0.0   | 0.0 |
| 14                   | 3.497         | -6            | 107.7          | -0.1  |  | 14                   | 3.503         | -2            | 109.3          | 0.0   | 0.0 |
| 15                   | 3.507         | -1            | 107.7          | 0.0   |  | 15                   | 3.514         | 4             | 109.3          | 0.0   | 0.0 |
| 16                   | 3.506         | -2            | 107.8          | 0.0   |  | 16                   | 3.506         | 0             | 109.2          | 0.0   | 0.0 |

|                      |               |               |                |       |  |                      |               |               |                |       |     |
|----------------------|---------------|---------------|----------------|-------|--|----------------------|---------------|---------------|----------------|-------|-----|
| Sin parapetos        |               |               |                |       |  |                      |               |               |                |       |     |
| $\alpha = 210$       |               |               |                |       |  |                      |               |               |                |       |     |
|                      |               |               |                |       |  |                      |               |               |                |       |     |
| Presiones exteriores |               |               |                |       |  | Presiones interiores |               |               |                |       |     |
| Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_e$ |  | Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_i$ | c   |
| 1                    | 3.505         | -2            | 110.7          | 0.0   |  | 1                    | 3.503         | -2            | 109.2          | 0.0   | 0.0 |
| 2                    | 3.502         | -4            | 110.9          | 0.0   |  | 2                    | 3.504         | -1            | 109.3          | 0.0   | 0.0 |
| 3                    | 3.506         | -2            | 111.2          | 0.0   |  | 3                    | 3.502         | -2            | 109.4          | 0.0   | 0.0 |
| 4                    | 3.507         | -1            | 111.4          | 0.0   |  | 4                    | 3.502         | -2            | 109.5          | 0.0   | 0.0 |
| 5                    | 3.506         | -2            | 111.6          | 0.0   |  | 5                    | 3.5           | -3            | 109.6          | 0.0   | 0.0 |
| 6                    | 3.508         | -1            | 111.8          | 0.0   |  | 6                    | 3.504         | -1            | 109.7          | 0.0   | 0.0 |
| 7                    | 3.507         | -1            | 112.1          | 0.0   |  | 7                    | 3.507         | 0             | 109.9          | 0.0   | 0.0 |
| 8                    | 3.51          | 0             | 112.3          | 0.0   |  | 8                    | 3.507         | 0             | 110.0          | 0.0   | 0.0 |
| 9                    | 3.508         | -1            | 112.5          | 0.0   |  | 9                    | 3.506         | 0             | 110.1          | 0.0   | 0.0 |
| 10                   | 3.511         | 1             | 112.8          | 0.0   |  | 10                   | 3.507         | 0             | 110.2          | 0.0   | 0.0 |
| 11                   | 3.512         | 1             | 113.0          | 0.0   |  | 11                   | 3.506         | 0             | 110.3          | 0.0   | 0.0 |
| 12                   | 3.508         | -1            | 113.2          | 0.0   |  | 12                   | 3.503         | -2            | 110.5          | 0.0   | 0.0 |
| 13                   | 3.508         | -1            | 113.5          | 0.0   |  | 13                   | 3.509         | 1             | 110.6          | 0.0   | 0.0 |
| 14                   | 3.506         | -2            | 113.7          | 0.0   |  | 14                   | 3.511         | 2             | 110.7          | 0.0   | 0.0 |
| 15                   | 3.512         | 1             | 113.9          | 0.0   |  | 15                   | 3.511         | 2             | 110.8          | 0.0   | 0.0 |
| 16                   | 3.511         | 1             | 114.2          | 0.0   |  | 16                   | 3.519         | 6             | 110.9          | 0.1   | 0.0 |

Presiones exteriores, interiores y totales en Modelo1

Sin parapetos

$\alpha = 150$

| Presiones exteriores |               |               |                |       | Presiones interiores |               |               |                |       | Presiones netas |  |
|----------------------|---------------|---------------|----------------|-------|----------------------|---------------|---------------|----------------|-------|-----------------|--|
| Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_e$ | Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_i$ | $c$             |  |
| 1                    | 3.465         | -22           | 107.8          | -0.2  | 1                    | 3.456         | -29           | 104.9          | -0.3  | 0.1             |  |
| 2                    | 3.479         | -15           | 108.0          | -0.1  | 2                    | 3.48          | -17           | 104.9          | -0.2  | 0.0             |  |
| 3                    | 3.49          | -10           | 108.2          | -0.1  | 3                    | 3.495         | -9            | 104.9          | -0.1  | 0.0             |  |
| 4                    | 3.491         | -9            | 108.4          | -0.1  | 4                    | 3.498         | -8            | 104.8          | -0.1  | 0.0             |  |
| 5                    | 3.496         | -7            | 108.5          | -0.1  | 5                    | 3.498         | -8            | 104.8          | -0.1  | 0.0             |  |
| 6                    | 3.47          | -20           | 108.7          | -0.2  | 6                    | 3.455         | -29           | 104.8          | -0.3  | 0.1             |  |
| 7                    | 3.489         | -10           | 108.9          | -0.1  | 7                    | 3.482         | -16           | 104.8          | -0.2  | 0.1             |  |
| 8                    | 3.5           | -5            | 109.1          | 0.0   | 8                    | 3.511         | -1            | 104.7          | 0.0   | 0.0             |  |
| 9                    | 3.475         | -17           | 109.3          | -0.2  | 9                    | 3.465         | -24           | 104.7          | -0.2  | 0.1             |  |
| 10                   | 3.502         | -4            | 109.5          | 0.0   | 10                   | 3.507         | -3            | 104.7          | 0.0   | 0.0             |  |
| 11                   | 3.505         | -2            | 109.7          | 0.0   | 11                   | 3.516         | 1             | 104.7          | 0.0   | 0.0             |  |
| 12                   | 3.474         | -18           | 109.9          | -0.2  | 12                   | 3.446         | -34           | 104.6          | -0.3  | 0.2             |  |
| 13                   | 3.495         | -7            | 110.1          | -0.1  | 13                   | 3.496         | -9            | 104.6          | -0.1  | 0.0             |  |
| 14                   | 3.465         | -22           | 110.3          | -0.2  | 14                   | 3.452         | -31           | 104.6          | -0.3  | 0.1             |  |
| 15                   | 3.508         | -1            | 110.5          | 0.0   | 15                   | 3.519         | 3             | 104.6          | 0.0   | 0.0             |  |
| 16                   | 3.507         | -1            | 110.7          | 0.0   | 16                   | 3.513         | 0             | 104.5          | 0.0   | 0.0             |  |

## APPENDIX D

### Open canopy with parapets: spreadsheet calculations.

| Presiones exteriores, interiores y totales en Modelo1 |               |               |                |       |                      |               |               |                |       |
|---|---------------|---------------|----------------|-------|----------------------|---------------|---------------|----------------|-------|
| Sin parapetos   |               |               |                |       |                      |               |               |                |       |
| $\alpha = 0$  |               |               |                |       |                      |               |               |                |       |
| Presiones exteriores                                  |               |               |                |       | Presiones interiores |               |               |                |       |
| Toma  | Potencial [V] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $C_e$ | Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $C_i$ |
| 1   | 3.428         | -39           | 117.6          | -0.3  | 1                    | 3.278         | -116          | 105.5          | -1.1  |
| 2   | 3.419         | -44           | 117.5          | -0.4  | 2                    | 3.28          | -115          | 105.6          | -1.1  |
| 3   | 3.414         | -46           | 117.3          | -0.4  | 3                    | 3.29          | -110          | 105.7          | -1.0  |
| 4   | 3.411         | -48           | 117.2          | -0.4  | 4                    | 3.298         | -106          | 105.7          | -1.0  |
| 5   | 3.409         | -49           | 117.0          | -0.4  | 5                    | 3.311         | -99           | 105.8          | -0.9  |
| 6   | 3.417         | -45           | 116.9          | -0.4  | 6                    | 3.35          | -80           | 105.8          | -0.8  |
| 7   | 3.409         | -49           | 116.8          | -0.4  | 7                    | 3.324         | -93           | 105.9          | -0.9  |
| 8   | 3.408         | -49           | 116.6          | -0.4  | 8                    | 3.312         | -99           | 106.0          | -0.9  |
| 9   | 3.408         | -49           | 116.5          | -0.4  | 9                    | 3.44          | -34           | 106.0          | -0.3  |
| 10  | 3.402         | -52           | 116.3          | -0.5  | 10                   | 3.397         | -56           | 106.1          | -0.5  |
| 11  | 3.393         | -57           | 116.2          | -0.5  | 11                   | 3.37          | -70           | 106.2          | -0.7  |
| 12  | 3.403         | -52           | 116.0          | -0.4  | 12                   | 3.476         | -16           | 106.2          | -0.2  |
| 13  | 3.394         | -56           | 115.9          | -0.5  | 13                   | 3.466         | -21           | 106.3          | -0.2  |
| 14  | 3.409         | -49           | 115.7          | -0.4  | 14                   | 3.496         | -6            | 106.3          | -0.1  |
| 15  | 3.401         | -53           | 115.6          | -0.5  | 15                   | 3.472         | -18           | 106.4          | -0.2  |
| 16  | 3.391         | -58           | 115.5          | -0.5  | 16                   | 3.464         | -22           | 106.5          | -0.2  |

Presiones exteriores, interiores y totales en Modelo1

Sin parapetos

$\alpha = 180$

| Presiones exteriores |               |               |                |       | Presiones interiores |               |               |                |       | Presiones netas |  |
|----------------------|---------------|---------------|----------------|-------|----------------------|---------------|---------------|----------------|-------|-----------------|--|
| Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_e$ | Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_i$ | c               |  |
| 1                    | 3.575         | 35            | 115.5          | 0.3   | 1                    | 3.475         | -17           | 106.5          | -0.2  | 0.5             |  |
| 2                    | 3.563         | 29            | 115.5          | 0.2   | 2                    | 3.473         | -18           | 106.4          | -0.2  | 0.4             |  |
| 3                    | 3.59          | 42            | 115.6          | 0.4   | 3                    | 3.474         | -17           | 106.3          | -0.2  | 0.5             |  |
| 4                    | 3.586         | 40            | 115.7          | 0.3   | 4                    | 3.474         | -17           | 106.3          | -0.2  | 0.5             |  |
| 5                    | 3.562         | 28            | 115.8          | 0.2   | 5                    | 3.475         | -17           | 106.2          | -0.2  | 0.4             |  |
| 6                    | 3.538         | 16            | 115.9          | 0.1   | 6                    | 3.475         | -17           | 106.2          | -0.2  | 0.3             |  |
| 7                    | 3.521         | 7             | 116.0          | 0.1   | 7                    | 3.482         | -13           | 106.1          | -0.1  | 0.2             |  |
| 8                    | 3.531         | 12            | 116.1          | 0.1   | 8                    | 3.487         | -11           | 106.1          | -0.1  | 0.2             |  |
| 9                    | 3.508         | 1             | 116.2          | 0.0   | 9                    | 3.487         | -11           | 106.0          | -0.1  | 0.1             |  |
| 10                   | 3.512         | 3             | 116.2          | 0.0   | 10                   | 3.487         | -11           | 106.0          | -0.1  | 0.1             |  |
| 11                   | 3.49          | -8            | 116.3          | -0.1  | 11                   | 3.493         | -8            | 105.9          | -0.1  | 0.0             |  |
| 12                   | 3.479         | -14           | 116.4          | -0.1  | 12                   | 3.491         | -9            | 105.8          | -0.1  | 0.0             |  |
| 13                   | 3.48          | -13           | 116.5          | -0.1  | 13                   | 3.497         | -6            | 105.8          | -0.1  | -0.1            |  |
| 14                   | 3.44          | -33           | 116.6          | -0.3  | 14                   | 3.486         | -11           | 105.7          | -0.1  | -0.2            |  |
| 15                   | 3.465         | -21           | 116.7          | -0.2  | 15                   | 3.501         | -4            | 105.7          | 0.0   | -0.1            |  |
| 16                   | 3.443         | -32           | 116.8          | -0.3  | 16                   | 3.497         | -6            | 105.6          | -0.1  | -0.2            |  |

Presiones exteriores, interiores y totales en Modelo1

Sin parapetos

$\alpha = 30$

| Presiones exteriores |               |               |                |       | Presiones interiores |               |               |                |       | Presiones netas |  |
|----------------------|---------------|---------------|----------------|-------|----------------------|---------------|---------------|----------------|-------|-----------------|--|
| Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_e$ | Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_i$ | c               |  |
| 1                    | 3.398         | -54           | 116.8          | -0.5  | 1                    | 3.46          | -27           | 108.3          | -0.2  | -0.2            |  |
| 2                    | 3.389         | -59           | 116.7          | -0.5  | 2                    | 3.249         | -133          | 108.4          | -1.2  | 0.7             |  |
| 3                    | 3.387         | -60           | 116.6          | -0.5  | 3                    | 3.185         | -165          | 108.6          | -1.5  | 1.0             |  |
| 4                    | 3.389         | -59           | 116.5          | -0.5  | 4                    | 3.239         | -138          | 108.7          | -1.3  | 0.8             |  |
| 5                    | 3.4           | -53           | 116.4          | -0.5  | 5                    | 3.301         | -107          | 108.9          | -1.0  | 0.5             |  |
| 6                    | 3.39          | -58           | 116.3          | -0.5  | 6                    | 3.481         | -16           | 109.0          | -0.1  | -0.4            |  |
| 7                    | 3.382         | -63           | 116.2          | -0.5  | 7                    | 3.478         | -18           | 109.1          | -0.2  | -0.4            |  |
| 8                    | 3.389         | -59           | 116.1          | -0.5  | 8                    | 3.342         | -86           | 109.3          | -0.8  | 0.3             |  |
| 9                    | 3.39          | -58           | 116.0          | -0.5  | 9                    | 3.411         | -51           | 109.4          | -0.5  | 0.0             |  |
| 10                   | 3.384         | -62           | 115.9          | -0.5  | 10                   | 3.501         | -6            | 109.6          | -0.1  | -0.5            |  |
| 11                   | 3.384         | -62           | 115.8          | -0.5  | 11                   | 3.483         | -15           | 109.7          | -0.1  | -0.4            |  |
| 12                   | 3.389         | -59           | 115.8          | -0.5  | 12                   | 3.377         | -69           | 109.8          | -0.6  | 0.1             |  |
| 13                   | 3.39          | -58           | 115.7          | -0.5  | 13                   | 3.496         | -9            | 110.0          | -0.1  | -0.4            |  |
| 14                   | 3.409         | -49           | 115.6          | -0.4  | 14                   | 3.4           | -57           | 110.1          | -0.5  | 0.1             |  |
| 15                   | 3.416         | -45           | 115.5          | -0.4  | 15                   | 3.492         | -11           | 110.2          | -0.1  | -0.3            |  |
| 16                   | 3.429         | -39           | 115.4          | -0.3  | 16                   | 3.504         | -5            | 110.4          | 0.0   | -0.3            |  |





## APPENDIX E

Parapets: spreadsheet calculations.

| Presiones exteriores, interiores y totales en Modelo1 |               |               |                |       |                      |      |               |               |                |                 |      |
|---|---------------|---------------|----------------|-------|----------------------|------|---------------|---------------|----------------|-----------------|------|
| Sin parapetos   |               |               |                |       |                      |      |               |               |                |                 |      |
| $\alpha = 0$  |               |               |                |       |                      |      |               |               |                |                 |      |
|   |               |               |                |       |                      |      |               |               |                |                 |      |
| Presiones exteriores                                  |               |               |                |       | Presiones interiores |      |               |               |                | Presiones netas |      |
| Toma  | Potencial [V] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_e$ |                      | Toma | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_i$           | $c$  |
| 1   | 3.743         | 118           | 111.4          | 1.1   |                      | 1    | 3.431         | -42           | 104.9          | -0.40           | 1.5  |
| 2   | 3.76          | 126           | 111.5          | 1.1   |                      | 2    | 3.431         | -42           | 105.1          | -0.40           | 1.5  |
| 3   | 3.748         | 120           | 111.6          | 1.1   |                      | 3    | 3.427         | -44           | 105.3          | -0.42           | 1.5  |
| 4   | 3.751         | 122           | 111.7          | 1.1   |                      | 4    | 3.434         | -40           | 105.5          | -0.38           | 1.5  |
| 5   | 3.699         | 96            | 111.8          | 0.9   |                      | 5    | 3.436         | -39           | 105.7          | -0.37           | 1.2  |
| 6   | 3.745         | 119           | 111.9          | 1.1   |                      | 6    | 3.437         | -39           | 105.8          | -0.37           | 1.4  |
| 7   | 3.73          | 111           | 112.0          | 1.0   |                      | 7    | 3.439         | -38           | 106.0          | -0.36           | 1.4  |
| 8   | 3.678         | 85            | 112.1          | 0.8   |                      | 8    | 3.438         | -38           | 106.2          | -0.36           | 1.1  |
| 9   | 3.722         | 107           | 112.2          | 1.0   |                      | 9    | 3.445         | -35           | 106.4          | -0.33           | 1.3  |
| 10  | 3.723         | 108           | 112.3          | 1.0   |                      | 10   | 3.443         | -36           | 106.6          | -0.34           | 1.3  |
| 11  | 3.303         | -102          | 112.4          | -0.9  |                      | 11   | 3.443         | -36           | 106.7          | -0.3            | -0.6 |
| 12  | 3.295         | -106          | 112.5          | -0.9  |                      | 12   | 3.44          | -37           | 106.9          | -0.3            | -0.6 |
| 13  | 3.307         | -100          | 112.6          | -0.9  |                      | 13   | 3.436         | -39           | 107.1          | -0.4            | -0.5 |
| 14  | 3.365         | -71           | 112.7          | -0.6  |                      | 14   | 3.438         | -38           | 107.3          | -0.4            | -0.3 |
| 15  | 3.35          | -79           | 112.8          | -0.7  |                      | 15   | 3.43          | -42           | 107.5          | -0.4            | -0.3 |
| 16  | 3.351         | -78           | 112.9          | -0.7  |                      | 16   | 3.425         | -45           | 107.7          | -0.4            | -0.3 |
| 17  | 3.466         | -21           | 113.0          | -0.2  |                      | 17   | 3.415         | -50           | 107.8          | -0.5            | 0.3  |
| 18  | 3.455         | -26           | 113.1          | -0.2  |                      | 18   | 3.409         | -53           | 108.0          | -0.5            | 0.3  |
| 19  | 3.486         | -11           | 113.2          | -0.1  |                      | 19   | 3.426         | -44           | 108.2          | -0.4            | 0.3  |
| 20  | 3.482         | -13           | 113.3          | -0.1  |                      | 20   | 3.416         | -49           | 108.4          | -0.5            | 0.3  |

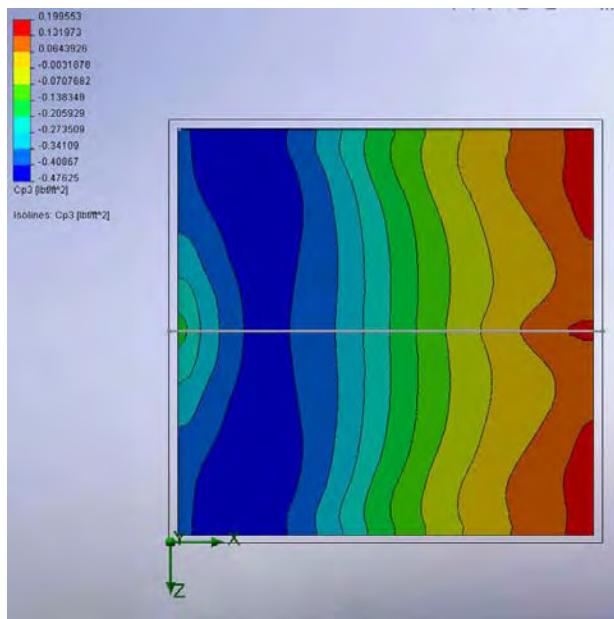




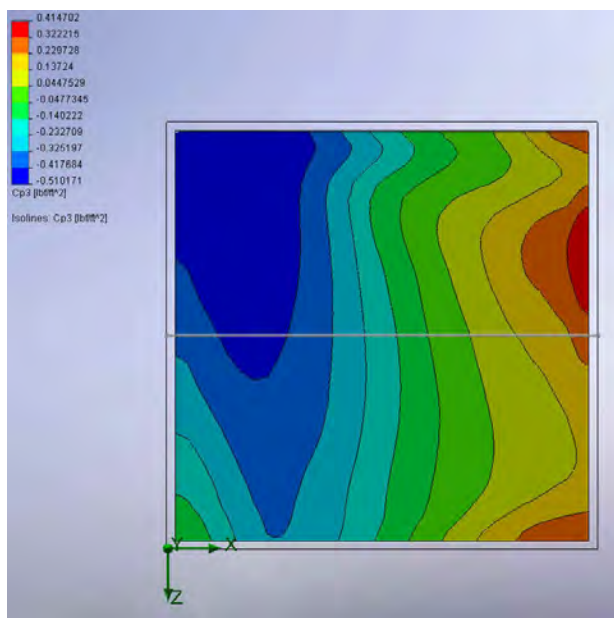
| Presiones exteriores, interiores y totales en Modelo1 |               |               |                |            |                      |               |               |                |       |                 |
|---|---------------|---------------|----------------|------------|----------------------|---------------|---------------|----------------|-------|-----------------|
| Sin parapetos   |               |               |                |            |                      |               |               |                |       |                 |
| $\alpha = 150$  |               |               |                | 104.923077 | 108.384615           | 106.384615    | 110.384615    |                |       |                 |
| Presiones exteriores                                  |               |               |                |            | Presiones interiores |               |               |                |       | Presiones netas |
| Toma  | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_e$      | Toma                 | Potencial [ ] | Pres.Dif.[Pa] | $q_{ref}$ [Pa] | $c_i$ | c               |
| 1   | 3.407         | -50           | 111.8          | -0.4       | 1                    | 3.655         | 71            | 106.4          | 0.7   | -1.1            |
| 2   | 3.403         | -52           | 111.8          | -0.5       | 2                    | 3.67          | 78            | 106.6          | 0.7   | -1.2            |
| 3   | 3.401         | -53           | 111.8          | -0.5       | 3                    | 3.588         | 37            | 106.8          | 0.3   | -0.8            |
| 4   | 3.399         | -54           | 111.7          | -0.5       | 4                    | 3.596         | 41            | 107.0          | 0.4   | -0.9            |
| 5   | 3.383         | -62           | 111.7          | -0.6       | 5                    | 3.54          | 13            | 107.2          | 0.1   | -0.7            |
| 6   | 3.377         | -65           | 111.6          | -0.6       | 6                    | 3.531         | 8             | 107.4          | 0.1   | -0.7            |
| 7   | 3.369         | -69           | 111.6          | -0.6       | 7                    | 3.491         | -12           | 107.6          | -0.1  | -0.5            |
| 8   | 3.381         | -63           | 111.6          | -0.6       | 8                    | 3.572         | 29            | 107.9          | 0.3   | -0.8            |
| 9   | 3.371         | -68           | 111.5          | -0.6       | 9                    | 3.558         | 22            | 108.1          | 0.2   | -0.8            |
| 10  | 3.372         | -68           | 111.5          | -0.6       | 10                   | 3.524         | 5             | 108.3          | 0.0   | -0.7            |
| 11  | 3.515         | 4             | 111.4          | 0.0        | 11                   | 3.601         | 44            | 108.5          | 0.4   | -0.4            |
| 12  | 3.526         | 9             | 111.4          | 0.1        | 12                   | 3.581         | 34            | 108.7          | 0.3   | -0.2            |
| 13  | 3.526         | 9             | 111.4          | 0.1        | 13                   | 3.543         | 14            | 108.9          | 0.1   | 0.0             |
| 14  | 3.543         | 18            | 111.3          | 0.2        | 14                   | 3.571         | 29            | 109.1          | 0.3   | -0.1            |
| 15  | 3.562         | 27            | 111.3          | 0.2        | 15                   | 3.554         | 20            | 109.3          | 0.2   | 0.1             |
| 16  | 3.549         | 21            | 111.2          | 0.2        | 16                   | 3.533         | 9             | 109.5          | 0.1   | 0.1             |
| 17  | 3.576         | 34            | 111.2          | 0.3        | 17                   | 3.482         | -16           | 109.8          | -0.1  | 0.5             |
| 18  | 3.591         | 42            | 111.2          | 0.4        | 18                   | 3.466         | -24           | 110.0          | -0.2  | 0.6             |
| 19  | 3.58          | 36            | 111.1          | 0.3        | 19                   | 3.451         | -32           | 110.2          | -0.3  | 0.6             |
| 20  | 3.588         | 40            | 111.1          | 0.4        | 20                   | 3.434         | -40           | 110.4          | -0.4  | 0.7             |

## APPENDIX F

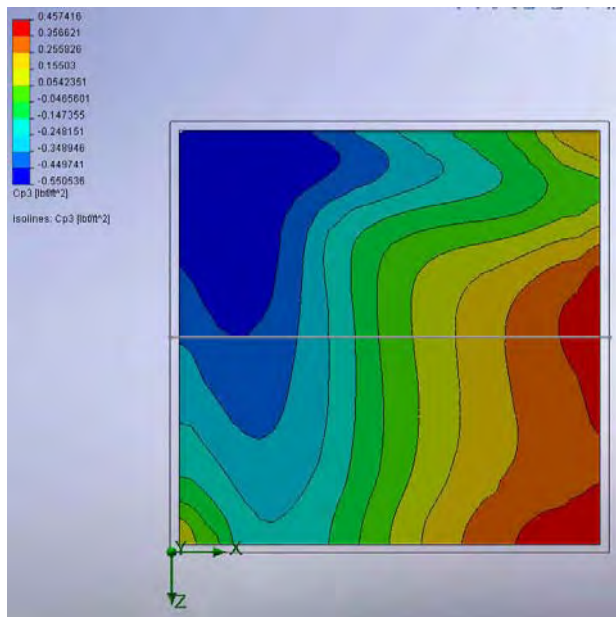
CFD test of different wind angles on open canopy to verify assumption of wind at  $0^\circ$  and  $30^\circ$ .



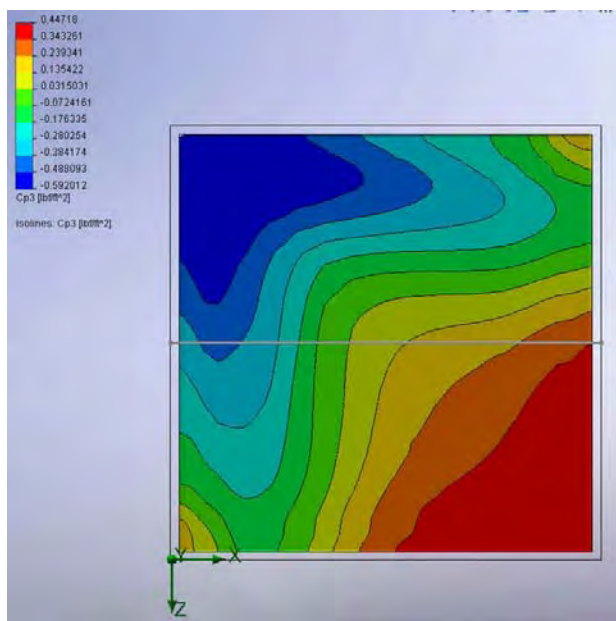
Wind at  $0^\circ$ .



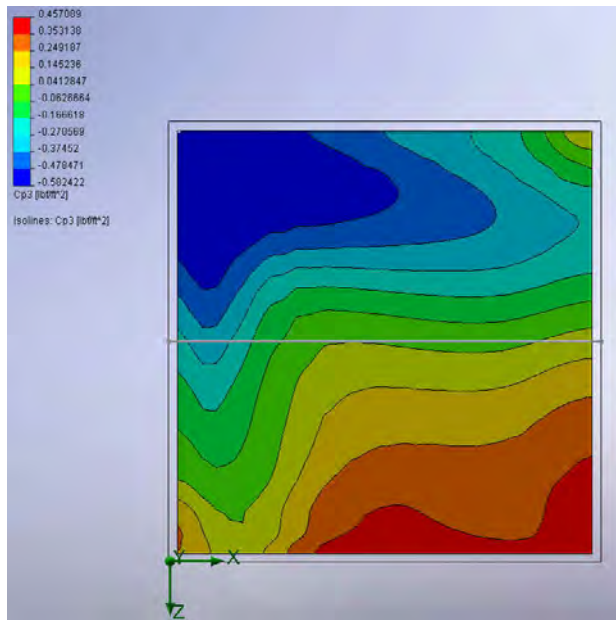
Wind at  $15^\circ$ .



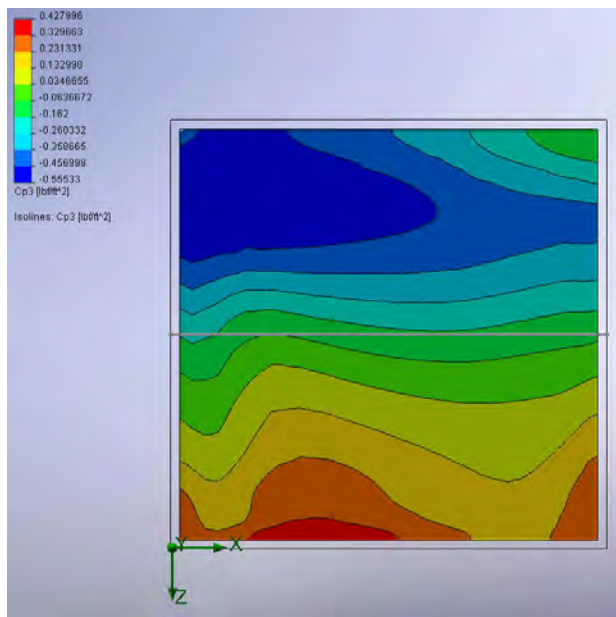
Wind at 30°.



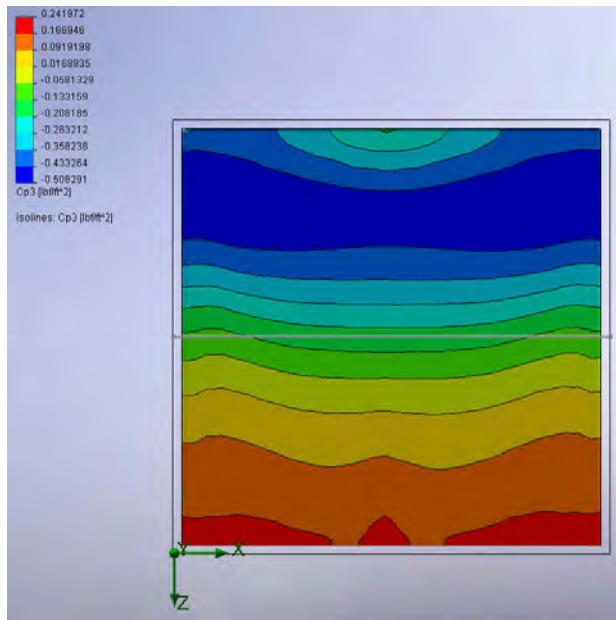
Wind at 45°.



Wind at  $60^\circ$ .



Wind at  $75^\circ$ .



Wind at 90°.

## APPENDIX G

T-test statistical analysis example. Used for determination of sample variation.

| t-Test: Two-Sample Assuming Equal Variances |                |     |             |
|---|----------------|-----|-------------|
|   | CFD values     |     | UNNE values |
| 1   | 0.3867         | 1   | 0.7616      |
| 2   | -0.0752        | 2   | 0.7616      |
| 3   | -0.2290        | 3   | 0.7125      |
| 4   | -0.2367        | 4   | 0.6424      |
| 5   | -0.1391        | 5   | 0.5905      |
| 6   | -0.0254        | 6   | 0.5191      |
| 7   | 0.0709         | 7   | 0.3676      |
| 8   | 0.1355         | 8   | 0.4554      |
| 9   | 0.2192         | 9   | 0.5069      |
| 10  | 0.3489         | 10  | -0.1000     |
| 11  | 0.4938         | 11  | 0.0767      |
| 12  | 0.0230         | 12  | 0.1645      |
| 13  | -0.2308        | 13  | -0.2940     |
| 14  | 0.0840         | 14  | -0.2865     |
| 15  | 0.2286         | 15  | -0.3636     |
| 16  | 0.3610         | 16  | -0.2857     |
| 17  | 0.5852         | 17  | -0.2923     |
| 18  | -0.1559        | 18  | 0.7616      |
| 19  | -0.2048        | 19  | 0.7125      |
| 20  | 0.0193         | 20  | 0.6424      |
| 21  | 0.1365         | 21  | 0.5905      |
| 22  | 0.3430         | 22  | 0.5191      |
| 23  | 0.6572         | 23  | 0.3676      |
| 24  | 0.1803         | 24  | 0.4554      |
| 25  | 0.1881         | 25  | 0.5069      |
| 26  | 0.2812         | 26  | -0.1000     |
| 27  | -0.0961        | 27  | 0.0767      |
| 28  | -0.1968        | 28  | 0.1645      |
| 29  | -0.0011        | 29  | -0.2940     |
| 30  | 0.1084         | 30  | -0.2865     |
| 31  | 0.7633         | 31  | -0.3636     |
| 32  | 0.2665         | 32  | -0.2857     |
| 33  | 0.7561         | 33  | -0.2923     |
| 34  | 0.2667         | 34  | 0.4578      |
| 35  | -0.0975        | 35  | 0.4148      |
| 36  | -0.1978        | 36  | 0.5275      |
| 37  | -0.0021        | 37  | 0.5099      |
| 38  | 0.1071         | 38  | 0.4007      |
| 39  | 0.6543         | 39  | 0.2964      |
| 40  | 0.1801         | 40  | 0.1895      |
| 41  | 0.1875         | 41  | 0.2092      |
| 42  | 0.2816         | 42  | 0.1096      |
| 43  | 0.5841         | 43  | 0.1269      |
| 44  | -0.1571        | 44  | 0.0034      |
| 45  | -0.2064        | 45  | -0.0346     |
| 46  | 0.0187         | 46  | -0.0586     |
| 47  | 0.1356         | 47  | -0.1789     |
| 48  | 0.3443         | 48  | -0.1421     |
| 49  | 0.4855         | 49  | -0.2177     |
| 50  | 0.0194         | 50  | 0.4578      |
| 51  | -0.2313        | 51  | 0.4148      |
| 52  | 0.0837         | 52  | 0.5275      |
| 53  | 0.2267         | 53  | 0.5099      |
| 54  | 0.3621         | 54  | 0.4007      |
| 55  | 0.3567         | 55  | 0.2964      |
| 56  | -0.0821        | 56  | 0.1895      |
| 57  | -0.2336        | 57  | 0.2092      |
| 58  | -0.2397        | 58  | 0.1096      |
| 59  | -0.1392        | 59  | 0.1269      |
| 60  | -0.0263        | 60  | 0.0034      |
| 61  | 0.0709         | 61  | -0.0346     |
| 62  | 0.1341         | 62  | -0.0586     |
| 63  | 0.2158         | 63  | -0.1789     |
| 64  | 0.3437         | 64  | -0.1421     |
| Na  | 64             | Nb  | 64          |
| Ma  | 0.1327         | Mb  | 0.1962      |
| Ssa   | 4.4317         | SSb | 7.2865      |
| Ma-Mb=                                      | -0.0635        |     |             |
| Variance=                                   | 0.0930         |     |             |
| tandard deviation                           | 0.0539         |     |             |
| t=  | <b>-1.1782</b> |     |             |
| df=   | 126            |     |             |

Figure G.1: Excel spreadsheet calculating t-test for a Cn sample. Wind at 0°, Figure 6.10, page 81

**Two-Sample T-Test and CI: C1, C2**

Two-sample T for C1 vs C2

|    | N  | Mean  | StDev | SE Mean |
|----|----|-------|-------|---------|
| C1 | 64 | 0.133 | 0.265 | 0.033   |
| C2 | 64 | 0.196 | 0.340 | 0.043   |

Difference =  $\mu$  (C1) -  $\mu$  (C2)  
 Estimate for difference: -0.0635  
 95% CI for difference: (-0.1702, 0.0432)  
 T-Test of difference = 0 (vs not =): T-Value = -1.18 P-Value = 0.241 DF = 126  
 Both use Pooled StDev = 0.3050

**Figure G-2: Minitab statistical software calculating t-test for a Cn sample. Used for verification.**